Lecture Note of Topics in Ship Design Automation

Optimum Design

Fall 2015

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Ch. 8 Case Study of Optimal Dimension Design

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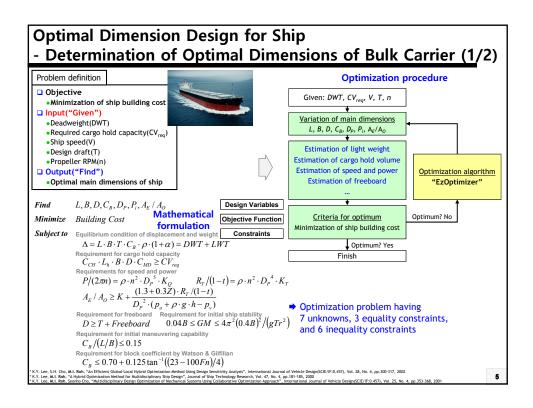
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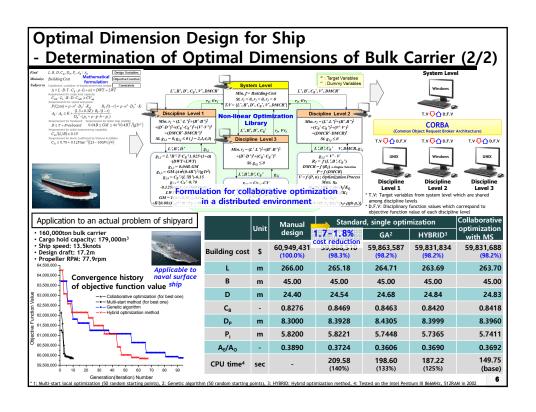
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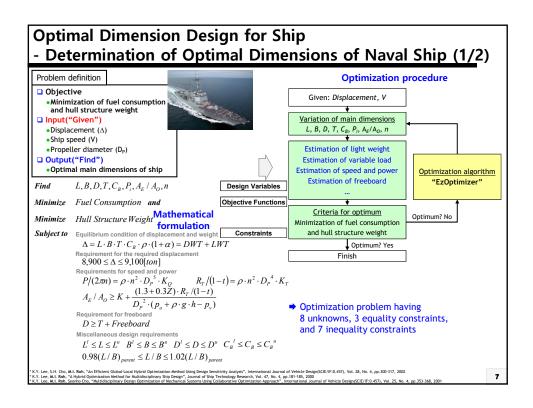
8.1 Overview

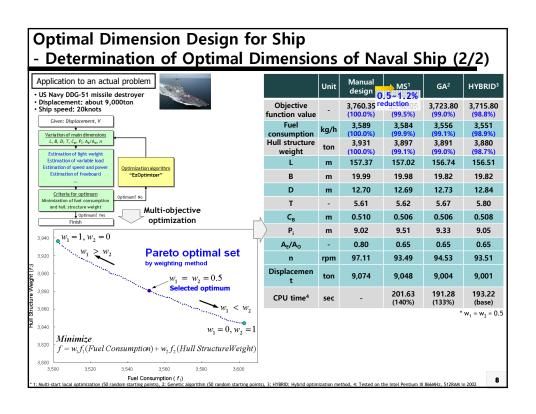
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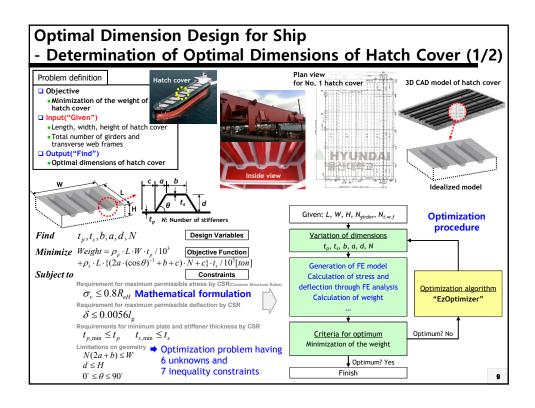
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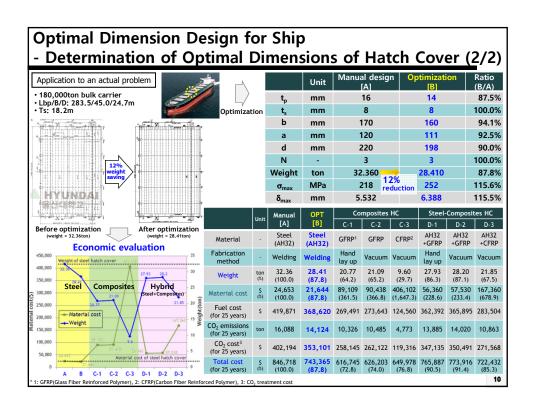


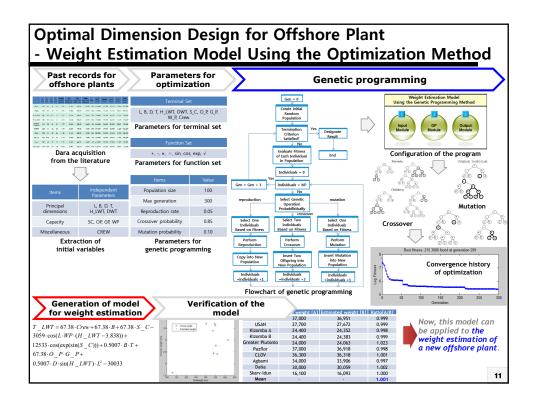












8.2 Determination of Optimal Principal Dimensions of Propeller Generals Mathematical Formulation and Its Solution Example

Generals

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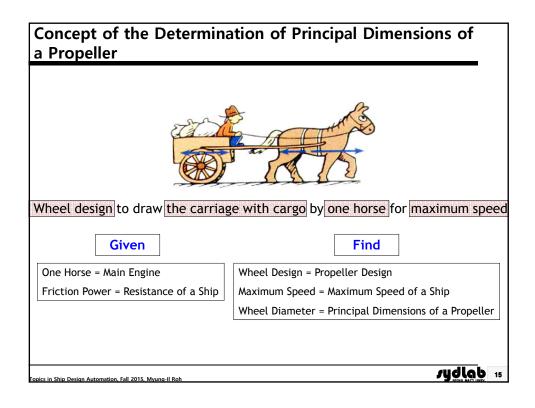
Example of a Propeller

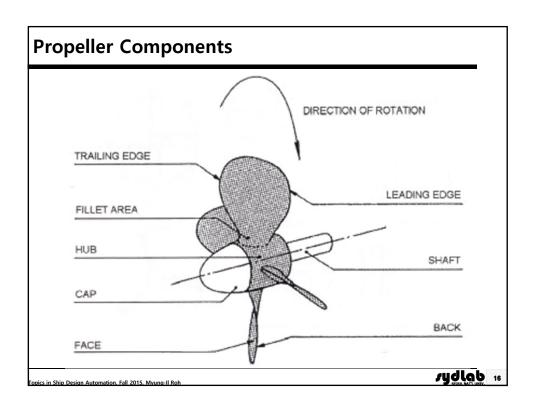


- ☑ Ship: 4,900 TEU Container Ship
- ☑ Owner: NYK, Japan
- ☑ Shipyard: HHI (2007.7.20)
- ☑ Diameter: 8.3 m☑ Weight: 83.3 ton☑ No of Blades: 5

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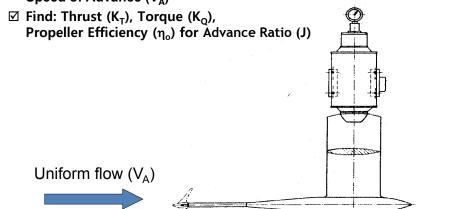
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Propeller Open Water (POW) Test

- ☑ This test is carried out under ideal condition in which the propeller does not get disturbed by the hull.



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Main Non-dimensional Coefficients of Propeller

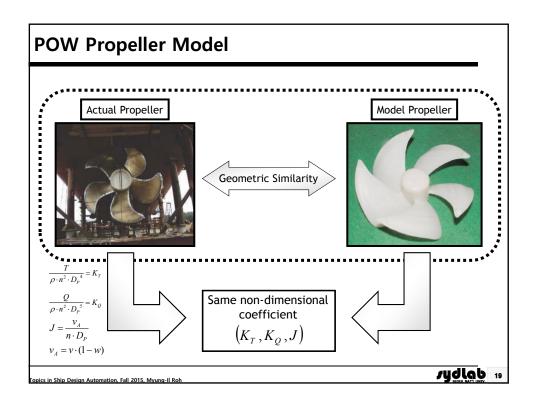
From dimensional analysis:

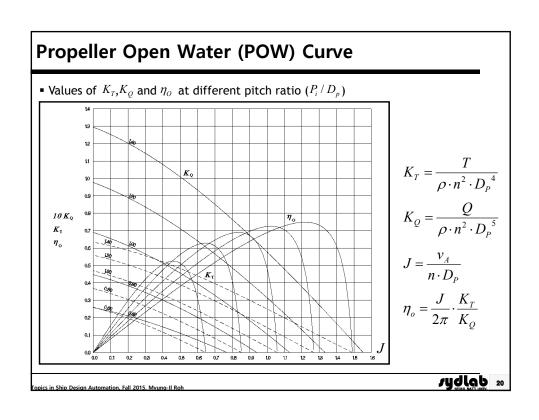
- ① Thrust coefficient: $\frac{T}{\rho \cdot n^2 \cdot {D_P}^4} = K_T$
- ③ Advance ratio: $J = \frac{v_A}{n \cdot D_P}$ $v_A = v \cdot (1 w)$

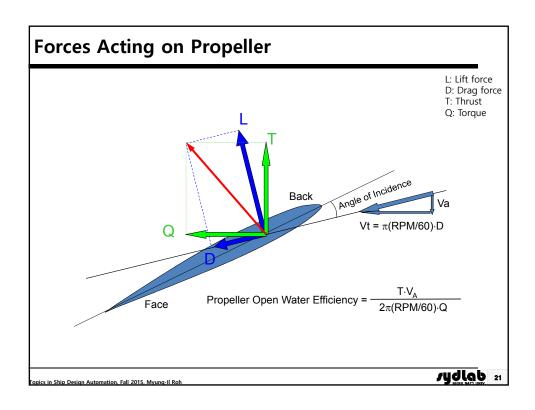
- v: Ship Speed [m/s]
- w: Wake fraction
- T: Thrust of the propeller [kN]
- Q: Torque absorbed by propeller [kN·m]
- n: Number of Revolutions [1/s]
- $D_{\scriptscriptstyle p}$: Propeller Diameter [m]
- P_i : Propeller Pitch [m]
- $V_{\scriptscriptstyle A}$: Speed of Advance [m/s]

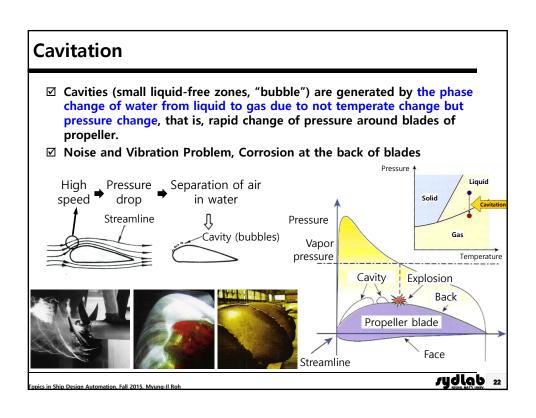
Thrust deduction coefficient: The ratio of the resistance increase due to rotating of a propeller at after body of ship

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Mathematical Formulation and Its Solution

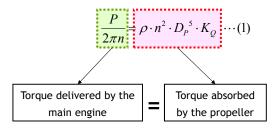
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Governing Equations for the Determination of Principal Dimensions of a Propeller (1/3)

Given	z ; P_{NCR} [kW], n_{MCR} [1/s]; $R_T(v)$ [kN]				
Find	$D_p[m]$, $P_i[m]$, A_E/A_O ; $v[m/s]$				

• Condition 1: The propeller absorbs the torque delivered by main engine.



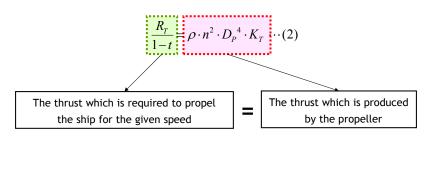
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Governing Equations for the Determination of Principal Dimensions of a Propeller (2/3)

Given	z ; P_{NCR} [kW], n_{MCR} [1/s]; $R_T(v)$ [kN]				
Find	$D_p[m]$, $P_i[m]$, A_E/A_O ; $v[m/s]$				

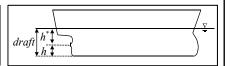
• Condition 2: The propeller should produce the required thrust at a given ship speed.



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Governing Equations for the Determination of Principal Dimensions of a Propeller (3/3)

Given	z ; P_{NCR} [kW], n_{MCR} [1/s]; R_T (v) [kN]
Find	$D_p[m]$, $P_i[m]$, A_E/A_O ; $v[m/s]$



- Condition 3: Required minimum expanded blade area ratio for non-cavitating criterion can be calculated by using one of the two formulas.
 - ① Formula given by Keller

$$A_E / A_O \ge K + \frac{\left(1.3 + 0.3z\right) \cdot T}{{D_P}^2 \cdot \left(p_0 + \rho g h^* - p_v\right)} \\ \text{h: Shaft Immersion Depth [m]} \\ \text{h: Shaft Center Height (height from the baseline) [m]}$$

K: Single Screw = 0.2, Double Screw = 0.1

 P_0 - P_v = 99.047 [kN/m²] at 15°C Sea water

T: Propeller Thrust [kN]

or ② Formula given by Burrill

$$\begin{split} A_{E}/A_{O} &\geq F \cdot (\eta_{0}/(1/J)^{2})/[\{1+4.826(1/J)^{2}\} \cdot (1.067-0.229 \cdot P_{i}/D_{p})] \\ F &= \frac{\eta_{R}}{287.4(10(18+h)^{0.625}} \\ B_{P} &= n \cdot P^{0.5}/v_{A}^{2.5} \\ &\qquad \qquad V_{A} = v \cdot (1-w)[knots] \end{split} \qquad \begin{array}{c} P &= DHP \cdot \eta_{R}[HP] \\ n[rpm] \end{array}$$

Determination of the Propeller Principal Dimensions for Maximum η_0 (1/6)

By Using Optimization Method

Given z; P_{NCR} [kW], n_{MCR} [1/s]; $R_T(v)$ [kN] Find $D_p[m]$, $P_i[m]$, A_E/A_O ; v[m/s]

Condition 1: The propeller absorbs the torque delivered by main engine.

$$\frac{P}{2\pi n} = \rho \cdot n^2 \cdot D_P^5 \cdot K_Q$$

Condition 2: The propeller should produce the required thrust at a given ship's speed.

$$\frac{R_T}{1-t} = \rho \cdot n^2 \cdot D_P^4 \cdot K_T$$

 Condition 3: Required minimum expanded blade area ratio for non-cavitating criterion.

$$A_E / A_O \ge K + \frac{(1.3 + 0.3z) \cdot T}{D_P^2 \cdot (p_0 + \rho g h^* - p_v)}$$

4 Unknowns

2 Equality constraints 1 Inequality constraint

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Nonlinear indeterminate equation

Objective Function: Maximum η_0

Propeller diameter(D_D), pitch(P_i), expanded blade area $ratio(A_E/A_O)$, and ship speed are determined to maximize the objective function by iteration.

Determination of the Propeller Principal Dimensions for Maximum η_0 (2/6) Calculation By Hand

Assume the Expanded Area Ratio (A_E/A_o) .

 A_O : Disc area $(\pi D_P^2/4)$

 $A_{\scriptscriptstyle E}$: Expanded propeller area

Assume that the expanded area ratio of the propeller of the design ship is the same as that of the basis ship.

Assume the ship speed v.

Condition 1:
$$\frac{1}{2\pi n} = \rho \cdot n \cdot D_P \cdot K_Q$$
$$J = \frac{v_A}{R} \implies \frac{nJ}{R} = \frac{1}{R}$$

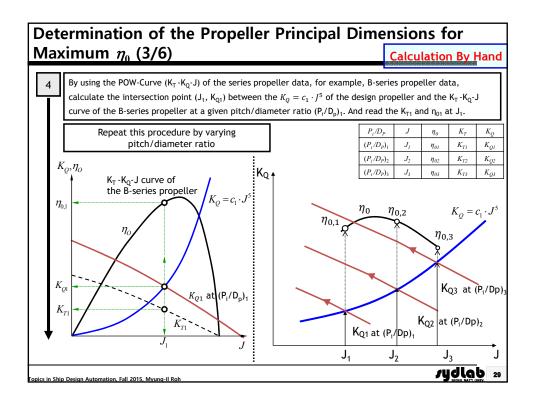
Express the condition 1 as
$$K_Q = C_1 J^5$$
.

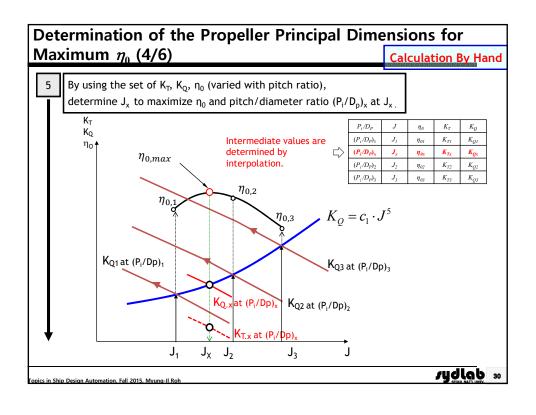
Condition 1: $\frac{P}{2\pi n} = \rho \cdot n^2 \cdot D_P^5 \cdot K_Q$, $K_Q = \frac{P}{2\pi n^3 \rho} \cdot \frac{1}{D_P^5} = \frac{P}{2\pi n^3 \rho} \cdot \left(\frac{nJ}{v_A}\right)^5$

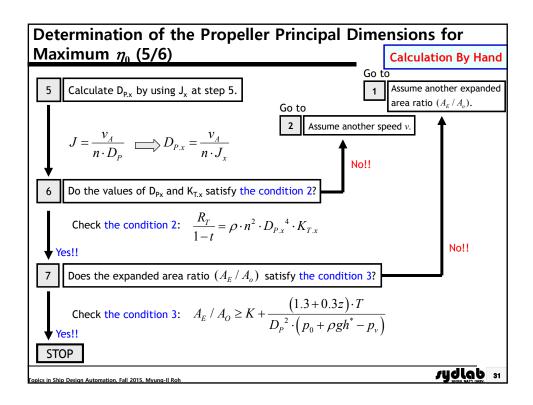
$$J = \frac{v_A}{n \cdot D_P} \implies \frac{nJ}{v_A} = \frac{1}{D_P}$$

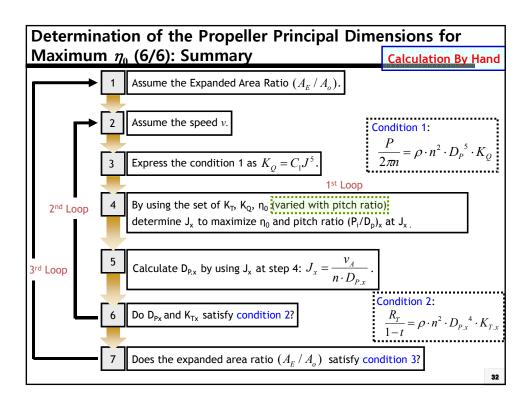
$$= \frac{P \cdot n^2}{2\pi \rho v_A^5} J^5 = C_1 J^5, \quad \left(C_1 = \frac{P \cdot n^2}{2\pi \rho v_A^5}\right)$$

$$K_O = C_1 J^5$$









Determination of the Optimal Principal Dimensions of a Propeller by Using the Lagrange Multiplier (1/5)

Given
$$P, n, A_E / A_O, V$$

Find
$$J_i, P_i / D_p$$

Maximize
$$\eta_O = \frac{J}{2\pi} \cdot \frac{K_T}{K_Q}$$

Because K_T and K_Q are a function of J and P/D_p , the objective is also a function of J and P/D_p .

Subject to
$$\frac{P}{2\pi n} = \rho \cdot n^2 \cdot D_P^{-5} \cdot K_Q$$

: The propeller absorbs the torque delivered by Diesel Engine

Where,
$$J = \frac{V(1-w)}{n \cdot D_P}$$

$$K_T = f(J, P_i/D_P)$$

$$K_O = f(J, P_i/D_P)$$

P: Delivered power to the propeller from the main engine, KW n: Revolution per second, 1/sec D_p: Propeller diameter, m P_i: Propeller pitch, m

A_E/A_O: Expanded area ratio V: Ship speed, m/s η_O: Propeller efficiency (in open water)

→ Optimization problem having two unknown variables and one equality constraint

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Determination of the Optimal Principal Dimensions of a Propeller by Using the Lagrange Multiplier (2/5)

$$\frac{P}{2\pi n} = \rho \cdot n^2 \cdot D_P^{5} \cdot K_Q \quad \cdots \quad \text{(a)} \quad \text{: The propeller absorbs the torque delivered by main engine}$$

The constraint (a) is reformulated as follows:

$$C = \frac{K_Q}{J^5} = \frac{P \cdot n^2}{2\pi\rho \cdot V_A^5}$$

$$G(J, P_i/D_P) = K_Q - C \cdot J^5 = 0 \quad \cdots \quad (a')$$

Propeller efficiency in open water η_0 is as follows.

$$F(J, P_i/D_P) = \eta_O = \frac{J}{2\pi} \cdot \frac{K_T}{K_O} \quad \cdots \quad \text{(b)}$$

The objective F is a function of J and P_i/D_{p_i}

It is to determine the optimal principal dimensions (J and P_i/D_p) to maximize the propeller efficiency in open water satisfying the constraint (a').

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Determination of the Optimal Principal Dimensions of a Propeller by Using the Lagrange Multiplier (3/5)

$$F(J, P_i/D_p) = \eta_0 = \frac{J}{2} \cdot \frac{K_T}{K_T} \cdot \cdots \cdot (b)$$

Introduce the Lagrange multiplier λ to the equation (a') and (b). $F(J,P_i/D_p)=\eta_0=\frac{J}{2\pi}\cdot\frac{K_T}{K_{\varrho}}$ ····· (b)

$$H(J, P_i/D_p, \lambda) = F(J, P_i/D_p) + \lambda G(J, P_i/D_p) \cdots$$
 (c)

Determine the value of the P_i/D_p and λ to maximize the value of the function H.

$$\frac{\partial H}{\partial J} = \frac{1}{2\pi} \left(\frac{K_T}{K_Q} \right) + \frac{J}{2\pi} \frac{\left\{ \left(\frac{\partial K_T}{\partial J} \right) \cdot K_Q - \left(\frac{\partial K_Q}{\partial J} \right) \cdot K_T \right\}}{K_Q^2} + \lambda \left\{ \left(\frac{\partial K_Q}{\partial J} \right) - 5 \cdot C \cdot J^4 \right\} = 0 \cdots (1)$$

$$\frac{\partial H}{\partial (P_i/D_P)} = \frac{J}{2\pi} \frac{\left\{ \left(\frac{\partial K_T}{\partial P_i/D_P} \right) \cdot K_Q - \left(\frac{\partial K_Q}{\partial P_i/D_P} \right) \cdot K_T \right\}}{K_Q^2} + \lambda \left(\frac{\partial K_Q}{\partial P_i/D_P} \right) = 0 \quad \cdots \quad (2)$$

$$\frac{\partial H}{\partial \lambda} = K_Q - C \cdot J^5 = 0 \quad \dots \tag{3}$$

Determination of the Optimal Principal Dimensions of a Propeller by Using the Lagrange Multiplier (4/5)

Eliminate λ in the equation (1), (2), and (3), and rearrange as follows.

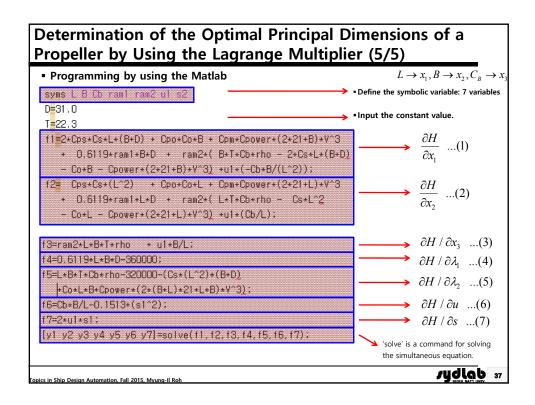
$$\left(\frac{\partial K_{\varrho}}{\partial (P_{i}/D_{p})}\right)\left\{J\cdot\left(\frac{\partial K_{T}}{\partial J}\right)-4K_{T}\right\} + \left(\frac{\partial K_{T}}{\partial (P_{i}/D_{p})}\right)\left\{5K_{\varrho}-J\cdot\left(\frac{\partial K_{\varrho}}{\partial J}\right)\right\} = 0 \quad \cdots \quad (4)$$

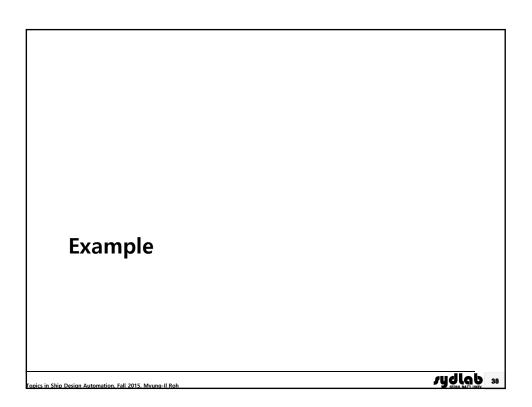
$$K_{\mathcal{Q}} - C \cdot J^5 = 0 \quad \cdots \quad (5)$$

By solving the nonlinear equation (4) and (5), we can determine J and P_i/D_p to maximize the propeller efficiency.

By definition $J = \frac{V(1-w)}{n \cdot D_p}$, if we have J, we can find D_p . Then P_i is obtained from P_i/D_p .

Thus, we can find the propeller diameter (D_n) and pitch (P_i) .



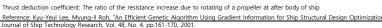


Determination of Optimal Principal Dimensions of a Propeller Problem Definition

- ☑ Problem for determining optimal principal dimensions of a propeller of a 9,000ton missile destroyer (DDG)
 - Objective
 - ullet Maximization of the efficiency of propeller ($\eta_{
 m O}$)
 - Input (Given, Ship owner's requirements)
 - P: Delivered power
 - D_p: Diameter of propeller
 - Data related to resistance: R_T (total resistance), w (wake fraction), t (thrust deduction coefficient*), η_R (relative rotative efficiency)



- P_i: Propeller pitch
- A_E/A_O: Expanded area ratio
- n: Propeller RPS (Revolution Per Second)
- V: Ship speed



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Determination of Optimal Principal Dimensions of a Propeller - Problem Formulation

Find
$$P_i, A_E / A_O, n, V$$

Design Variables

Maximize
$$\eta_O = \frac{J}{2\pi} \cdot \frac{K_T}{K_Q}$$

Objective Function

Maximize
$$\eta_O = \frac{J}{2\pi} \cdot \frac{K_T}{K_Q}$$
Subject to $\frac{P}{2\pi n} = \rho \cdot n^2 \cdot D_P^5 \cdot K_Q$
: The condition

Constraints

$$\frac{R_T}{1} = \rho \cdot n^2 \cdot D_P^4 \cdot K_T$$

 $\frac{2\pi n}{1-t} = \frac{r}{1-t} = \frac{Q}{r}$: The condition that the propeller absorbs the torque delivered by main engine $\frac{R_T}{1-t} = \rho \cdot n^2 \cdot D_P^{-4} \cdot K_T$: The condition that the propeller should produce the required thrust the condition should produce the required thrust at a given ship's speed

$$A_E / A_O \ge K + \frac{(1.3 + 0.3Z) \cdot T_h}{D_P^2 \cdot (p_o + \rho \cdot g \cdot h - p_v)}$$

: The condition about the required minimum expanded area ratio

Where

$$J = \frac{V(1-w)}{n \cdot D_P}, K_T = f(J, P_i / D_P, A_E / A_O, Z),$$

$$K_O = f(J, P_i / D_P, A_E / A_O, Z), T_h = R_T / (1 - t)$$

⇒ Optimization problem having 4 design variables, 2 equality constraints, 40nd 1 inequality constraint

Determination of Optimal Principal Dimensions of a Propeller - Optimization Result

Optimization results according to optimization methods						;	
	Unit	DDG-51	MFD	MS	GA	HYBRID w/o Refine	HYBRID with Refine
Pi	m	8.90	9.02	9.38	9.04	9.06	9.06
A _E /A _O	-	0.80	0.80	0.65	0.80	0.80	0.80
n	rpm	88.8	97.11	94.24	96.86	96.65	96.64
V*	kts	20.00	19.98	20.01	20.01	19.99	20.00
ηο	-	-	0.6439	0.6447	0.6457	0.6463	0.6528
Δ	LT	8,369	9,074	8,907	8,929	9,016	9,001
ВНР	HP	13,601	14,654	14,611	14,487	14,447	14,443
Iteration No	-	-	5	267	89	59	63
CPU Time	sec	-	0.88	38.07	41.92	40.45	41.39

* V*: Cruising Speed

* MFD: Method of feasible directions, MS: Multi-start local optimization method, GA: Genetic algorithm, HYBRID: Global-local hybrid optimization method

* Text system: Pentium 3 866MHz 512MR RAM

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8.3 Determination of Optimal Principal Dimensions of Ship

Generals

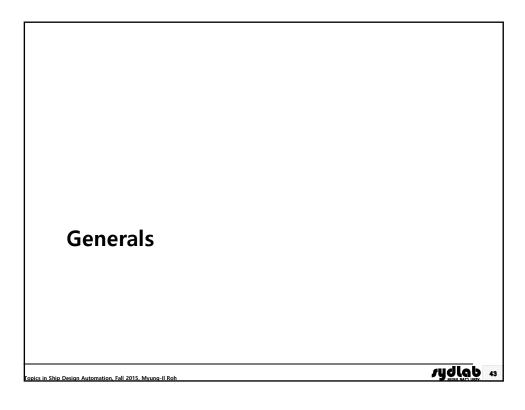
Design Equations

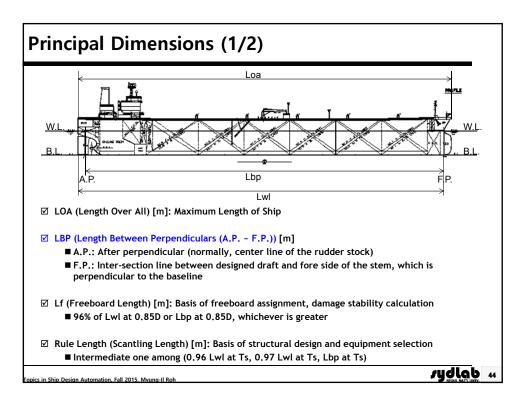
Mathematical Formulation and Its Solution

Example for the Determination of Optimal Principal Dimensions of a Bulk Carrier

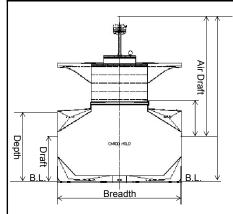
Example for the Determination of Optimal Principal Dimensions of a Naval Ship

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Principal Dimensions (2/2)



- B (Breadth) [m]: Maximum breadth of the ship, measured
 - B_{molded} : excluding shell plate thickness
 - $\mathbf{B}_{\text{extreme}}\text{:}$ including shell plate thickness
- D (Depth) [m]: Distance from the baseline to the deck side line
 - D_{molded} : excluding keel plate thickness
 - D_{extreme} : including keel plate thickness
- Td (Designed Draft) [m]: Main operating draft
- In general, basis of ship's deadweight and speed/power
- Ts (Scantling Draft) [m]: Basis of structural design
- Air Draft [m]: Distance (height above waterline only or including operating draft) restricted by the port facilities, navigating route, etc.
 - Air draft from baseline to the top of the mast
 - Air draft from waterline to the top of the mast
 - Air draft from waterline to the top of hatch cover

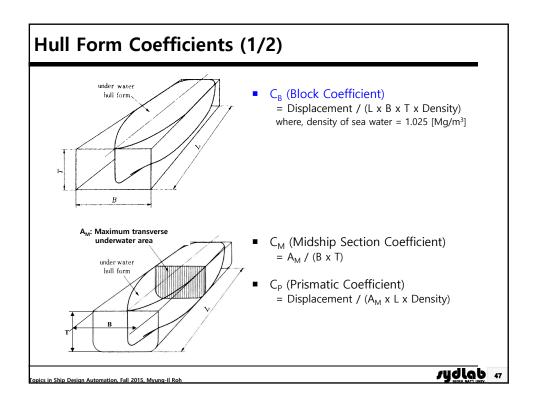
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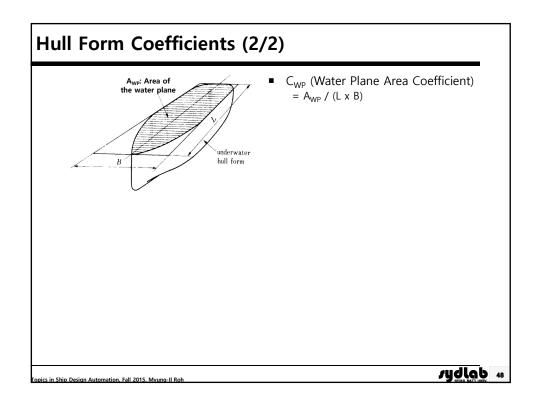
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Weight and COG (Center Of Gravity)

- ☑ Displacement [ton]
 - Weight of water displaced by the ship's submerged part
- Deadweight (DWT) [ton]: Cargo payload + Consumables (F.O., D.O., L.O., F.W., etc.) + DWT Constant
 - = Displacement Lightweight
- ☑ Cargo Payload [ton]: Weight of loaded cargo at the loaded draft
- ☑ DWT Constant [ton]: Operational liquid in the machinery and pipes, provisions for crew, etc.
- ☑ Lightweight (LWT) [ton]: Total of hull steel weight and weight of equipment on board
- - Trim = {Displacement x (LCB LCG)} / (MTC x 100)
- ☑ LCB: Longitudinal Center of Buoyancy
- ☑ LCG: Longitudinal Center of Gravity

F.O.: Fuel Oil, D.O.: Diesel Oil, L.O.: Lubricating Oil, F.W.: Fresh Water ics in Ship Design Automation, Fall 2015, Myung-Il Roh





Speed and Power (1/2)

- ☑ MCR (Maximum Continuous Rating) [PS x rpm]
 - NMCR (Nominal MCR)
 - DMCR (Derated MCR) / SMCR (Selected MCR)
- ✓ NCR (Normal Continuous Rating) [PS x rpm]
- ☑ Trial Power [PS x rpm]: Required power without sea margin at the service speed (BHP)
- ☑ Sea Margin [%]: Power reserve for the influence of storm seas and wind including the effects of fouling and corrosion.
- ☑ Service Speed [knots]: Speed at NCR power with the specific sea margin (e.g., 15%)

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Speed and Power (2/2)

- ☑ DHP: Delivered Horse Power
 - Power actually delivered to the propeller with some power loss in the stern tube bearing and in any shaft tunnel bearings between the stern tube and the site of the torsion-meter
- **☑** EHP: Effective Horse Power
 - Required power to maintain intended speed of the ship
- \square η_D : Quasi-propulsive coefficient = EHP / DHP
- ☑ RPM margin
 - To provide a sufficient torque reserve whenever full power must be attained under unfavorable weather conditions
 - To compensate for the expected future drop in revolutions for constant-power operation

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Tonnage

- ☑ Tonnage: normally, $100 \text{ ft}^3 \text{ (=2.83 m}^3\text{)} = 1 \text{ ton}$
 - Basis of various fee and tax
 - GT (Gross Tonnage): Total sum of the volumes of every enclosed space
 - NT (Net Tonnage): Total sum of the volumes of every cargo space
 - GT and NT should be calculated in accordance with "IMO 1969 Tonnage Measurement Regulation".
 - CGT (Compensated Gross Tonnage)
 - Panama and Suez canal have their own tonnage regulations.

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Unit (1/2)

- ☑ LT (Long Ton, British) = 1.016 [ton], ST (Short Ton, American) = 0.907 [ton], MT (Metric Ton, Standard) = 1.0 [ton]
- ☑ Density

 [ton/m³ or Mg/m³]
 - e.g., density of sea water = 1.025 [ton/m³], density of fresh water = 1.0 [ton/m³], density of steel = 7.8 [ton/m³]
- ☑ 1 [knots] = 1 [NM/hr] = 1.852 [km/hr] = 0.5144 [m/sec]
- ☑ 1 [PS] = 75 [kgf·m/s] = 75×10^{-3} [Mg]·9.81 [m/s²]·[m/s] = 0.73575 [kW] (Pferdestarke, German translation of horsepower)
 - NMCR of B&W6S60MC: 12,240 [kW] = 16,680 [PS]
- ☑ 1 [BHP] = 76 [kgf·m/s] = 76×10^{-3} [Mg]·9.81 [m/s²]·[m/s] = 0.74556 [KW] (British horsepower)

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Unit (2/2)

☑ SG (Specific Gravity) ➡ No dimension

- SG of material = density of material / density of water
- e.g., SG of sea water = 1.025, SG of fresh water = 1.0, SG of steel = 7.8

☑ SF (Stowage Factor) ➡ [ft³/LT]

- e.g., SF = 15 [ft³/LT] **⇒** SG = 2.4 [ton/m³]
- ☑ API (American Petroleum Institute) = (141.5 / SG) 131.5
 - e.g., API 40 **⇒** SG = 0.8251
- \square 1 [barrel] = 0.159 [m³]
 - e.g., 1 [mil. barrels] = 159,000 [m³]

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Basic Functions of a Ship

☑ Going fast on the water

- Hull form: Streamlined shape having small resistance
- Propulsion: Diesel engine, Helical propeller
- The speed of ship is represented with knot(s). 1 knot is a speed which can go 1 nautical mile (1,852 m) in 1 hour.

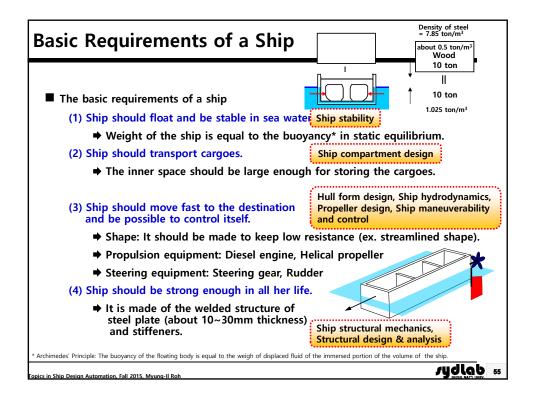
☑ Containing like a strong bowl

- Welded structure of plates (thickness of about 20 ~ 30mm), stiffeners, and brackets
- A VLCC has the lightweight of about 45,000 ton and can carry crude oil of about 300,000 ton.

☑ Navigable safely

- A ship has less motion for being comfortable and safe of passengers and cargo.
- Maneuvering equipment: Rudder

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Criteria for the Size of a Ship

☑ Displacement

- Weight of water displaced by the ship's submerged part
- Equal to total weight of ship
- Used when representing the size of naval ships

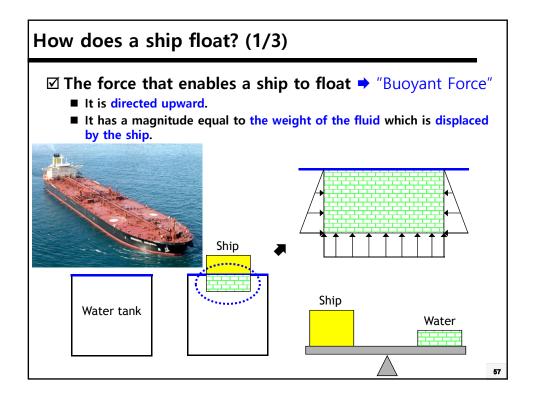
☑ Deadweight

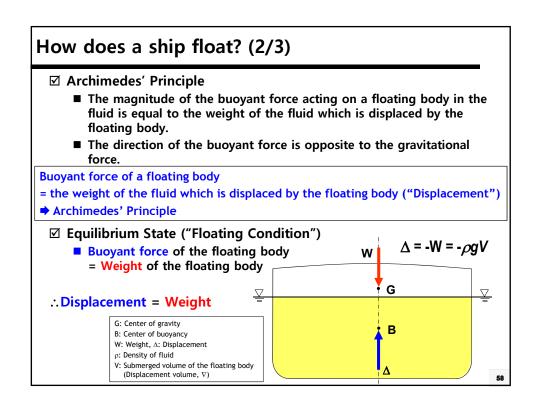
- Total weight of cargo. Actually, Cargo payload + Consumables (F.O., D.O., L.O., F.W., etc.) + DWT Constant
- Used when representing the size of commercial ships (tanker, bulk carrier, ore carrier, etc.)

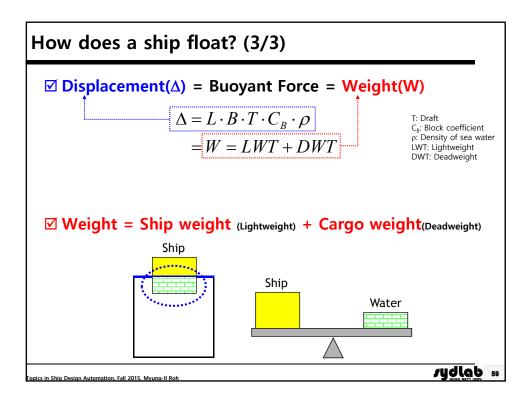
☑ Tonnage

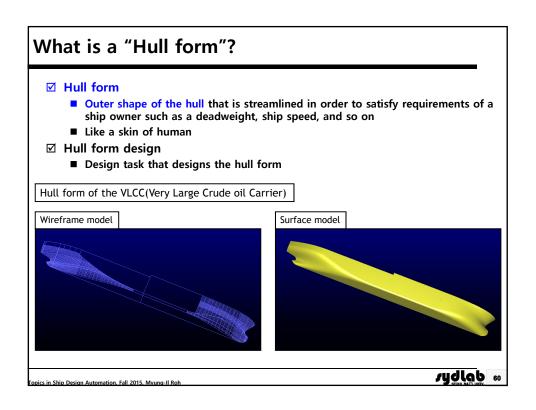
- Total volume of cargo
- Basis for statics, tax, etc.
- Used when representing the size of passenger ships

* F.O.: Fuel Oil, D.O.: Diesel Oil, L.O.: Lubricating Oil, F.W.: Fresh Water Copics in Ship Design Automation, Fall 2015, Myung-II Roh









What is a "Compartment"?

☑ Compartment

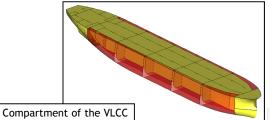
- Space to load cargos in the ship
- It is divided by a bulkhead which is a diaphragm or peritoneum of human.
- ☑ Compartment design (General arrangement design)
 - Compartment modeling + Ship calculation

☑ Compartment modeling

Design task that divides the interior parts of a hull form into a number of compartments

☑ Ship calculation (Naval architecture calculation)

■ Design task that evaluates whether the ship satisfies the required cargo capacity by a ship owner and, at the same time, the international regulations related to stability, such as MARPOL and SOLAS, or not



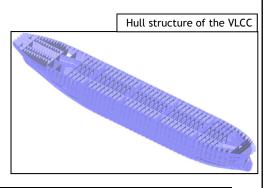
What is a "Hull Structure"?

☑ Hull structure

- Frame of a ship comprising of a number of hull structural parts such as plates, stiffeners, brackets, and so on
- Like a skeleton of human

☑ Hull structural design

■ Design task that determines the specifications of the hull structural parts such as the size, material, and so on



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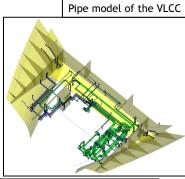
What is a "Outfitting"?

☑ Outfitting

- All equipment and instrument to be required for showing all function of the ship
 - Hull outfitting: Propeller, rudder, anchor/mooring equipment, etc.
 - Machinery outfitting: Equipment, pipes, ducts, etc. in the engine room
 - Accommodation outfitting: Deck house (accommodation), voyage equipment, etc.
 - Electric outfitting: Power, lighting, cables, and so on
- Like internal organs or blood vessels of human

☑ Outfitting design

Design task that determines the types, numbers, and specifications of outfitting



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Design Equations

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(1) Owner's Requirements

Owner's Requirements

☑ Owner's Requirements

- Ship's Type
- Deadweight (DWT)
- Cargo Hold Capacity (V_{CH})
 - Cargo Capacity: Cargo Hold Volume / Containers in Hold & on Deck / Car Deck Area
 - Water Ballast Capacity
- Service Speed (*V_s*)
 - Service Speed at Design Draft with Sea Margin, MCR/NCR Engine Power & RPM
- Dimensional Limitations: Panama canal, Suez canal, Strait of Malacca, St. Lawrence Seaway, Port limitations
- Maximum Draft (T_{max})
- Daily Fuel Oil Consumption (DFOC): Related with ship's economy
- Special Requirements
 - Ice Class, Air Draft, Bow/Stern Thruster, Special Rudder, Twin Skeg
- Delivery Day
 - Delivery day, with ()\$ penalty per delayed day
 - Abt. 21 months from contract
- The Price of a Ship
 - Material & Equipment Cost + Construction Cost + Additional Cost + Margin

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Principal Particulars of a Basis Ship

At early design stage, there are few data available to determine the principal particulars of the design ship. Therefore, initial values of the principal particulars can be estimated from the basis ship (called also as 'parent ship' or 'mother ship'), whose main dimensional ratios and hull form coefficients are similar with the ship being designed.

The principal particulars include main dimensions, hull form coefficients, speed and engine power, DFOC, capacity, cruising range, crew, class, etc.

Example) VLCC (Very Large Crude oil Carrier)



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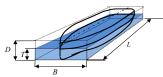
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Principal Dimensions & Hull Form Coefficients

The principal dimensions and hull form coefficients decide many characteristics of a ship, e.g. stability, cargo hold capacity, resistance, propulsion, power requirements, and economic efficiency.

Therefore, the determination of the principal dimensions and hull form coefficients is most important in the ship design.

The length L, breadth B, depth D, immersed depth (draft) T, and block coefficient C_B should be determined first.



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(2) Design Constraints

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Design Constraints

In the ship design, the principal dimensions cannot be determined arbitrarily; rather, they have to satisfy following design constraints:

1) Physical constraint

- Floatability: Hydrostatic equilibrium **→** "Weight Equation"

2) Economical constraints

Owner's requirements Ship's type, Deadweight (DWT) [ton], Cargo hold capacity (V_{CH}) [m^3], \Rightarrow "Volume Equation" Service speed (V_S) [knots], \Rightarrow Daily fuel oil consumption(DFOC)[ton/day] Maximum draft (T_{max}) [m],

Limitations of main dimensions (Canals, Sea way, Strait, Port limitations : e.g. Panama canal, Suez canal, St. Lawrence Seaway, Strait of Malacca, Endurance $[\mathcal{N}/\mathcal{M}^0]$,

3) Regulatory constraints

1) N/M: Nautical Mile 1 N/M = 1.852 km

International Maritime Organization [IMO] regulations, International Convention for the Safety Of Life At Sea [SOLAS], International Convention for the Prevention of Marin Pollution from Ships [MARPOL], International Convention on Load Lines [ICLL], Rules and Regulations of Classification Societies

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(3) Physical Constraints

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Physical Constraint

Physical constraint

- Floatability

For a ship to float in sea water, the total weight of the ship (W) must be equal to the buoyant force (F_R) on the immersed body **→** Hydrostatic equilibrium:

$$F_{\scriptscriptstyle D} \stackrel{!}{=} W$$
 ...(1)

$$F_{B} \stackrel{!}{=} W \dots (1)$$

$$W = LWT + DWT$$

*Lightweight(LWT) reflects the weight of vessel being ready to go to sea without cargo and loads. And lightweight can be composed of:

LWT = Structural weight + Outfit weight + Machinery weight

*Deadweight(DWT) is the weight that a ship can load till the maximum allowable immersion(at the scantling draft(T_s)).

And deadweight can be composed of:

DWT= Payload+ Fuel oil + Diesel oil+ Fresh water +Ballast water + etc.

• Physical constraint: hydrostatic **Physical Constraint** $F_B = W \dots (1)$ W=LWT+DWT ∇ : the immersed volume of the ship. ρ : density of sea water = 1.025 Mg/m³ (L.H.S) What is the buoyant force (F_B) ? According to the Archimedes' principle, the buoyant force on an immersed body has the same magnitude as the weight of the fluid displaced by the body. $F_{B} = g \cdot \rho \cdot V$ Mass In shipbuilding and shipping society, those are called as follows : $\begin{array}{c|c} \hline & \text{Displacement volume} & \nabla \\ \hline & \text{Displacement mass} & \Delta_m \\ \hline & \text{Displacement} & \Delta \end{array}$ Buoyant Force is the weight of the displaced fluid. In shipbuilding and shipping society, buoyant force is called in another word, **displacement** (Δ).

(4) Weight Equation

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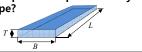
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Block Coefficient (C_B)

V: immersed volume V_{box} : volume of box L: length, B: breadth T: draft

Does a ship or an airplane usually have box shape?



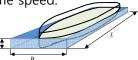






Why does a ship or an airplane has a streamlined shape?

They have a streamlined shape **to minimize the drag force** experienced when they travel, so that the propulsion engine needs a smaller power output to achieve the same speed.



Block coefficient(C_B) is the ratio of the immersed volume to the box bounded by L, B, and T.

$$C_B \equiv \frac{V}{V_{hox}} = \frac{V}{L \cdot B \cdot T}$$

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Shell Appendage Allowance

 $= \frac{V}{L \cdot B \cdot T} \begin{array}{c} V : \text{immer} \\ V_{box} : \text{volu} \\ L : \text{length} \\ T : \text{draft} \end{array}$

V: immersed volume V_{bax} : volume of box L: length, B: breadth T: draft C_B : block coefficient

The immersed volume of the ship can be expressed by block coefficient.

$$V_{molded} = L \cdot B \cdot T \cdot C_B$$

In general, we have to consider the **displacement of shell plating and appendages such as propeller, rudder, shaft, etc.** additionally.

Thus, The total immersed volume of the ship can be expressed as following: $\mathbf{U} = \mathbf{U} \mathbf{D} \mathbf{T} \mathbf{C} \mathbf{C} \mathbf{C} \mathbf{C} \mathbf{C} \mathbf{C}$

 $V_{total} = L \cdot B \cdot T \cdot C_B \cdot (1 + \alpha)$

Where the hull dimensions length L, beam B, and draft T are the **molded** dimensions of the immerged hull to the inside of the shell plating,

thus α is a fraction of the shell appendage allowance which adapts the molded volume to the actual volume by accounting for the volume of the shell plating and appendages (typically about 0.002~0.0025 for large vessels).

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Design Equations - Weight Equation

- Physical constraint: hydrostatic
 - uilibrium $F_B = W$...(1) (R.H.S) W = LWT + DWT(L.H.S) $F_B = \rho \cdot g \cdot L \cdot B \cdot T \cdot C_B \cdot (1+\alpha)$
 - ρ : density of sea water = 1.025 Mg/m³ α : displacement of shell, stern and appendages
 - C_n: block coefficient g : gravitational acceleration

$$\rho \cdot g \cdot L \cdot B \cdot T \cdot C_B \cdot (1 + \alpha) = LWT + DWT...(2)$$

The equation (2) describes the physical constraint to be satisfied in ship design,

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Unit of the Lightweight and Deadweight

• Physical constraint: hydrostatic equilibrium

...(1)

 $\rho \cdot \mathbf{g} \cdot L \cdot B \cdot T \cdot C_B \cdot (1 + \alpha) = LWT + DWT \quad \dots (2)$



What is the unit of the lightweight and deadweight?

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Design Equations

• Physical constraint : hydrostatic equilibrium $F_{\it B} = W$

...(1)

- Mass Equation

In shipping and shipbuilding world, "ton" is used instead of "Mg (mega gram)" for the unit of the lightweight and deadweight in practice.

Actually, however, the weight equation is "mass equation".



$$\rho \cdot L \cdot B \cdot T \cdot C_B \cdot (1 + \alpha) = LWT + DWT \qquad \dots (3)$$

"Mass equation"

where, ρ = 1.025 Mg/m³

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(5) Volume Equation

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Economical Constraints: Required Cargo Hold Capacity → Volume Equation

• Economical constraints

- Owner's requirements (Cargo hold capacity[m3])
- The main dimensions have to satisfy the required cargo hold capacity (V_{CH}).

$$V_{CH} = f(L, B, D)$$

: Volume equation of a ship

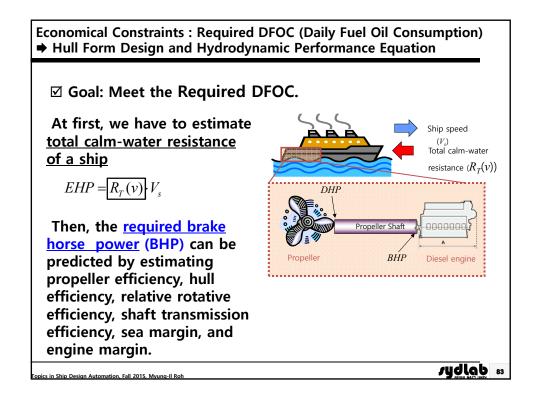
- It is checked whether the depth will allow the <u>required cargo hold</u> <u>capacity</u>.

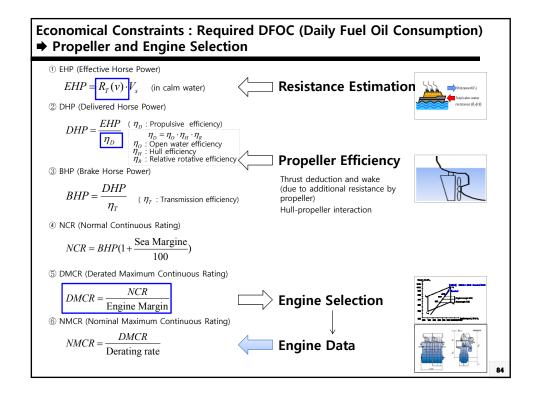
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(6) Service Speed & DFOC (Daily Fuel Oil Consumption)

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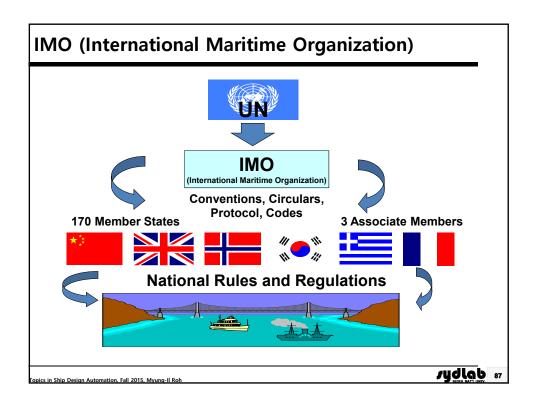


(7) Regulatory Constraints	
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Regulatory Constraints - Rules by Organizations

- - International Maritime Organizations (IMO)
 - International Labor Organizations (ILO)
 - Regional Organizations (EU, ...)
 - Administrations (Flag, Port)
 - **■** Classification Societies
 - International Standard Organizations (ISO)

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IMO Instruments □ Conventions □ SOLAS / MARPOL / ICLL / COLREG / ITC / AFS / BWM □ Protocols □ MARPOL Protocol 1997 / ICLL Protocol 1988 □ Codes □ ISM / LSA / IBC / IMDG / IGC / BCH / BC / GC □ Resolutions □ Assembly / MSC / MEPC □ Circulars □ MSC / MEPC / Sub-committees

Regulatory Constraints - Rules by Classification Societies ☑ 10 Members ■ ABS (American Bureau of Shipping) Council ■ DNV (Det Norske Veritas) ■ LR (Lloyd's Register) Permanent ■ BV (Bureau Veritas) Representative ■ GL (Germanischer Lloyd) General to IMO ■ KR (Korean Register of Shipping) **Policy** ■ RINA (Registro Italiano Navale) Group ■ NK (Nippon Kaiji Kyokai) ■ RRS (Russian Maritime Register of Shipping) ■ CCS (China Classification Society) Working Group **☑** 2 Associate Members ■ CRS (Croatian Register of Shipping) ■ IRS (Indian Register of Shipping) sydlab *

(8) Required Freeboard

Required Freeboard of ICLL 1966

Regulatory constraints

- International Convention on Load Lines (ICLL)1966 Fb Tmld 1

$$D_{Fb} - T \geq Fb_{ICLL}(L, B, D_{mld}, C_B)$$

: Freeboard Equation

✓ Check: Actual freeboard ($D_{Fb} - T$) of a ship should not be less than the freeboard required by the ICLL 1966 <u>regulation</u> (Fb_{ICLL}).

Freeboard (Fb) means the distance between the water surface and the top of the deck at the side (at the deck line). It includes the thickness of freeboard deck plating.

The freeboard is closely related to the draught.
 A 'freeboard calculation' in accordance with the regulation determines whether the desired depth is permissible.

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Mathematical Formulation and Its Solution

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Mathematical Model for Determination of Optimal Principal Dimensions of a Ship Summary ("Conceptual Ship Design Equation")

Find (Design variables) L, B, D, C_B length breadth depth block

Physical constraint

→ Displacement - Weight equilibrium (Weight equation) - Equality constraint

$$\begin{split} L \cdot B \cdot T \cdot C_B \cdot \rho_{sw} \cdot C_\alpha &= DWT_{given} + LWT(L, B, D, C_B) \\ &= DWT_{given} + C_s \cdot L^{1.6}(B+D) + C_o \cdot L \cdot B \\ &+ C_{power} \cdot (L \cdot B \cdot T \cdot C_B)^{2/3} \cdot V^3 \cdot \cdots (2.3) \end{split}$$

Economical constraints (Owner's

Regulatory constraint

- DFOC (Daily Fuel Oil Consumption) : It is related with the resistance and proper

requirements) \rightarrow Required cargo hold capacity (Volume equation) - Equality constraint - Delivery date : It is related with the shipbuilding process. $CC_{req} = C_{CH} \cdot L \cdot B \cdot D \cdot \cdot \cdot (3.1)$

Min. Roll Period .e.g.,

$$T_R \ge 12 \text{ sec}.....(6)$$

→ Freeboard regulation (ICLL 1966) - Inequality constraint $D \ge T + C_{FB} \cdot D \cdots (4)$

Stability regulation (MARPOL, SOLAS, ICLL)

$$GM \ge GM_{\text{Re quired}} \cdots (5)$$

Objective function (Criteria to determine the proper principal dimensions)

 $GZ \geq GZ_{\text{Re}\,quired}$

Building $Cost = C_{PS} \cdot C_s \cdot L^{1.6}(B+D) + C_{PO} \cdot C_o \cdot L \cdot B + C_{PM} \cdot C_{power} \cdot (L \cdot B \cdot T \cdot C_B)^{2/3} \cdot V^3$

4 variables (L, B, D, C_B), 2 equality constraints ((2.3), (3.1)), 3 inequality constraints ((4), (5), (6))

→ Optimization problem

Determination of the Optimal Principal Dimensions of a Ship by Using the Lagrange Multiplier (1/5)

- Given: DWT, CC_{req}, D, T_s, T_d
- Find: L, B, C_R
 - Hydrostatic equilibrium (Weight equation)

$$\begin{split} L \cdot B \cdot T_s \cdot C_B \cdot \rho_{sw} \cdot C_\alpha &= DWT_{given} + LWT(L, B, D, C_B) \\ &= DWT_{given} + \frac{C_s \cdot L^{1.6} \cdot (B+D)}{C_s \cdot L \cdot B} + C_o \cdot L \cdot B + \frac{C_{power} \cdot (L \cdot B \cdot T_d \cdot C_B)^{2/3} \cdot V^3}{C_{power} \cdot (L \cdot B \cdot T_d \cdot C_B)^{2/3} \cdot V^3} \right) \quad \dots \Big(a\Big) \\ &= \sum_{\substack{\text{Simplify } @ \\ O \setminus B \cdot T_d \cdot C_B \geq^{2/3} \text{ is (Volume)}^{2/3} \text{ and means the submerged area of the ship.}} \end{split}$$

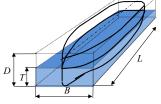
So, we assume that the submerged area of the ship is equal to the submerged area of the rectangular box.

• Required cargo hold capacity (Volume equation)

 $CC_{rea} = C_{CH} \cdot L \cdot B \cdot D \quad \dots(b)$ Recommended range of obesity coefficient

considering maneuverability of a ship

 $\frac{C_B}{(L/B)} < 0.15 \quad ...(c)$



- ▶ Indeterminate Equation: 3 variables (L, B, C_B) , 2 equality constraints ((a), (b))
- It can be formulated as an optimization problem to minimize an objective function.

Determination of the Optimal Principal Dimensions of a Ship by Using the Lagrange Multiplier (2/5)

- Given: DWT, $V_{H.reg}$, D, T_s , T_d
- Find: L, B, C_R
- Minimize: Building Cost

$$f(L,B,C_B) = C_{PS} \cdot C_s' \cdot L^{2.0} \cdot (B+D) + C_{PO} \cdot C_o \cdot L \cdot B + C_{PM} \cdot C_{power}' \cdot (2 \cdot B \cdot T_d + 2 \cdot L \cdot T_d + L \cdot B) \cdot V^3$$

Subject to

...(d)

• Hydrostatic equilibrium (Simplified weight equation)

$$\begin{split} L \cdot B \cdot T_{s} \cdot C_{B} \cdot \rho_{sw} \cdot C_{\alpha} &= DWT_{given} + LWT(L, B, D, C_{B}) \\ &= DWT_{given} + C_{s}' \cdot L^{2.0} \cdot (B + D) + C_{o} \cdot L \cdot B + C_{power}' \cdot (2 \cdot B \cdot T_{d} + 2 \cdot L \cdot T_{d} + L \cdot B) \cdot V^{3} \\ & ... (a') \end{split}$$

$$CC_{req} = C_{CH} \cdot L \cdot B \cdot D \quad \dots (b)$$

$$\frac{C_B}{\left(L/B\right)} < 0.15 \quad ...(c)$$

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Determination of the Optimal Principal Dimensions of a Ship by Using the Lagrange Multiplier (3/5)

• By introducing the Lagrange multipliers λ_1, λ_2, u , formulate the Lagrange function H.

$$H\left(L,B,C_{_B},\lambda_{_1},\lambda_{_2},u,s\right) = f\left(L,B,C_{_B}\right) + \lambda_{_1} \cdot h_{_1}\left(L,B,C_{_B}\right) + \lambda_{_2} \cdot h_{_2}\left(L,B,D\right) + u \cdot g\left(L,B,C_{_B},s\right) \quad ...(e)$$

$$\begin{split} f\left(L,B,C_{B}\right) &= C_{PS} \cdot C_{s}^{\ \prime} \cdot L^{2} \cdot (B+D) + C_{PO} \cdot C_{o} \cdot L \cdot B + C_{PM} \cdot C_{power}^{\ \prime} \cdot \{2 \cdot \left(B+L\right) \cdot T_{d} + L \cdot B\} \cdot V^{3} \\ h_{1}\left(L,B,C_{B}\right) &= L \cdot B \cdot T_{s} \cdot C_{B} \cdot \rho_{sw} \cdot C_{a} - DWT_{given} - C_{s}^{\ \prime} \cdot L^{20} \cdot (B+D) - C_{o} \cdot L \cdot B - C_{power}^{\ \prime} \cdot \{2 \cdot \left(B+L\right) \cdot T_{d} + L \cdot B\} \cdot V^{3} \\ h_{2}\left(L,B,D\right) &= C_{CH} \cdot L \cdot B \cdot D - CC_{req} \\ g\left(L,B,C_{B},s\right) &= \frac{C_{B}}{\left(L/B\right)} - 0.15 + s^{2} \\ L \rightarrow x_{1},B \rightarrow x_{2},C_{B} \rightarrow x_{3} \end{split}$$

$$H(x_1, x_2, x_3, \lambda_1, \lambda_2, u, s)$$

$$\begin{split} &= C_{PS} \cdot C_{s}^{'} \cdot x_{1}^{2} (x_{2} + D) + C_{PO} \cdot C_{o} \cdot x_{1} \cdot x_{2} + C_{PM} \cdot C_{power}^{'} \cdot \{2 \cdot (x_{2} + x_{1}) \cdot T_{d} + x_{1} \cdot x_{2}\} \cdot V^{3} \\ &+ \lambda_{1} \cdot [x_{1} \cdot x_{2} \cdot T_{s} \cdot x_{3} \cdot \rho_{sw} \cdot C_{a} - DWT_{given} - C_{s} \cdot x_{1}^{2} \cdot (x_{2} + D) - C_{o} \cdot x_{1} \cdot x_{2} - C_{power}^{'} \cdot \{2 \cdot (x_{2} + x_{1}) \cdot T_{d} + x_{1} \cdot x_{2}\} \cdot V^{3}] \\ &+ \lambda_{2} \cdot \left(C_{CH} \cdot x_{1} \cdot x_{2} \cdot D - CC_{req}\right) \end{split}$$

$$+u \cdot \{x_3 / (x_1 / x_2) - 0.15 + s^2\}$$
 ...(f)

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Determination of the Optimal Principal Dimensions of a Ship by Using the Lagrange Multiplier (4/5)

$$\begin{split} H\left(x_{1}, x_{2}, x_{3}, \lambda_{1}, \lambda_{2}, u, s\right) &= C_{pS} \cdot C_{s}^{\ \prime} \cdot x_{1}^{\ 2}(x_{2} + D) + C_{pO} \cdot C_{o} \cdot x_{1} \cdot x_{2} + C_{pM} \cdot C_{power}^{\ \prime} \cdot \{2 \cdot \left(x_{2} + x_{1}\right) \cdot T_{d} + x_{1} \cdot x_{2}\} \cdot V^{3} \\ &+ \lambda_{1} \cdot \left[x_{1} \cdot x_{2} \cdot T_{s} \cdot x_{3} \cdot \rho_{sw} \cdot C_{\alpha} - DWT_{given} - C_{s} \cdot x_{1}^{\ 2} \cdot (x_{2} + D) - C_{o} \cdot x_{1} \cdot x_{2} - C_{power}^{\ \prime} \cdot \{2 \cdot \left(x_{2} + x_{1}\right) \cdot T_{d} + x_{1} \cdot x_{2}\} \cdot V^{3}\right] \\ &+ \lambda_{2} \cdot \left(C_{CH} \cdot x_{1} \cdot x_{2} \cdot D - CC_{req}\right) + u \cdot \left\{x_{3} \cdot \left(x_{1} / x_{2}\right) - 0.15 + s^{2}\right\} & \dots(f) \end{split}$$

• To determine the stationary point (x_1, x_2, x_3) of the Lagrange function H (equation (f)), use the Kuhn-Tucker necessary condition: $\nabla H(x_1, x_2, x_3, \lambda_1, \lambda_2, u, s) = 0$.

$$\begin{split} \frac{\partial H}{\partial x_{1}} &= 2C_{PS} \cdot C_{s}^{'} \cdot x_{1} \cdot (x_{2} + D) + C_{PO} \cdot C_{o} \cdot x_{2} + C_{PM} \cdot C_{power}^{'} \cdot \left(2 \cdot T_{d} + x_{2}\right) \cdot V^{3} \\ &+ \lambda_{1} \cdot \left(x_{2} \cdot T_{s} \cdot x_{3} \cdot \rho_{sw} \cdot C_{\alpha} - \left[2 \cdot C_{s} \cdot x_{1} \cdot (x_{2} + D) + C_{o} \cdot x_{2} + C_{power}^{'} \cdot (2 \cdot T_{d} + x_{2}) \cdot V^{3}\right]\right) \\ &+ \lambda_{2} \cdot \left(C_{CH} \cdot x_{2} \cdot D\right) + u \cdot \left(-x_{3} \cdot x_{2} / x_{1}^{2}\right) = 0 \quad \dots(1) \end{split}$$

$$\begin{split} \frac{\partial H}{\partial x_{2}} &= C_{PS} \cdot C_{s}' \cdot x_{1}^{2} + C_{PO} \cdot C_{o} \cdot x_{1} + C_{PM} \cdot C_{power}' \cdot (2 \cdot T_{d} + x_{1}) \cdot V^{3} \\ &+ \lambda_{1} \cdot \left[x_{1} \cdot T_{s} \cdot x_{3} \cdot \rho_{sw} \cdot C_{\alpha} - C_{s}' \cdot x_{1}^{2} - C_{o} \cdot x_{1} - C_{power}' (2 \cdot T_{d} + x_{1}) \cdot V^{3} \right] \\ &+ \lambda_{2} \cdot \left(C_{CH} \cdot x_{1} \cdot D \right) + u \cdot \left(x_{3} / x_{1} \right) = 0 \quad ...(2) \end{split}$$

Determination of the Optimal Principal Dimensions of a Ship by Using the Lagrange Multiplier (5/5)

$$\begin{split} H\left(x_{1}, x_{2}, x_{3}, \lambda_{1}, \lambda_{2}, u, s\right) &= C_{PS} \cdot C_{s}^{'} \cdot x_{1}^{2}(x_{2} + D) + C_{PO} \cdot C_{o} \cdot x_{1} \cdot x_{2} + C_{PM} \cdot C_{power}^{'} \cdot \left\{2 \cdot \left(x_{2} + x_{1}\right) \cdot T_{d} + x_{1} \cdot x_{2}\right\} \cdot V^{3} \\ &+ \lambda_{1} \cdot \left[x_{1} \cdot x_{2} \cdot T_{s} \cdot x_{3} \cdot \rho_{sw} \cdot C_{\alpha} - DWT_{given} - C_{s} \cdot x_{1}^{2} \cdot (x_{2} + D) - C_{o} \cdot x_{1} \cdot x_{2} - C_{power}^{'} \cdot \left\{2 \cdot \left(x_{2} + x_{1}\right) \cdot T_{d} + x_{1} \cdot x_{2}\right\} \cdot V^{3}\right] \\ &+ \lambda_{2} \cdot \left(C_{CH} \cdot x_{1} \cdot x_{2} \cdot D - CC_{req}\right) + u \cdot \left\{x_{3} \cdot \left(x_{1} / x_{2}\right) - 0.15 + s^{2}\right\} \quad(f) \end{split}$$

■ Kuhn-Tucker necessary condition: $\nabla H(x_1, x_2, x_3, \lambda_1, \lambda_2, u, s) = 0$

$$\frac{\partial H}{\partial x_3} = \lambda_1 \cdot x_1 \cdot x_2 \cdot T_s \cdot \rho_{sw} \cdot C_\alpha + u \cdot (x_2 / x_1) = 0 \quad ...(3)$$

$$\frac{\partial H}{\partial x_3} = x_1 \cdot x_2 \cdot T \cdot x_3 \cdot \rho_{sw} \cdot C_\alpha - DWT_{sym} - C_x \cdot x_1^2 \cdot (x_2 + DWT_{sym}) = 0$$

$$\frac{\partial H}{\partial \lambda_1} = x_1 \cdot x_2 \cdot T_s \cdot x_3 \cdot \rho_{sw} \cdot C_{\alpha} - DWT_{given} - C_s \cdot x_1^2 \cdot (x_2 + D) - C_o \cdot x_1 \cdot x_2$$

$$-C_{power}' \cdot \{2 \cdot (x_2 + x_1) \cdot T_d + x_1 \cdot x_2\} \cdot V^3 \cdots (4)$$

$$\frac{\partial H}{\partial \lambda_2} = C_{CH} \cdot x_1 \cdot x_2 \cdot D - CC_{req} = 0 \quad ...(5)$$

$$\frac{\partial H}{\partial u} = x_3 \cdot x_2 / x_1 - 0.15 + s^2 = 0 \quad ...(6)$$

$$\frac{\partial H}{\partial s} = 2 \cdot u \cdot s = 0, \quad \left(u \ge 0 \right) \quad ...(7)$$

 $\nabla H(x_p, x_p, \lambda_p, \lambda_p, \lambda_p, u, s)$: Nonlinear simultaneous equation having the 7 variables ((1)~(7)) and 7 equations

➡ It can be solved by using a numerical method!

yydlab "

Example for the Determination of Optimal Principal Dimensions of a Bulk Carrier

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sydlab »

Determination of Optimal Principal Dimensions of a Bulk Carrier - Problem Definition

- ☑ Criteria for determining optimal principal dimensions (Objective function)
 - Minimization of shipbuilding cost or Minimization of hull structure weight or Minimization of operation cost
- ☑ Given (Ship owner's requirements)
 - Deadweight (DWT)
 - Cargo hold capacity (CC_{req})
 - Maximum draft (T_{max})
 - Ship speed (V)
- ☑ Find (Design variables)
 - Length (L)
 - Breadth (B)
 - Depth (D)
 - Block Coefficient (C_B)

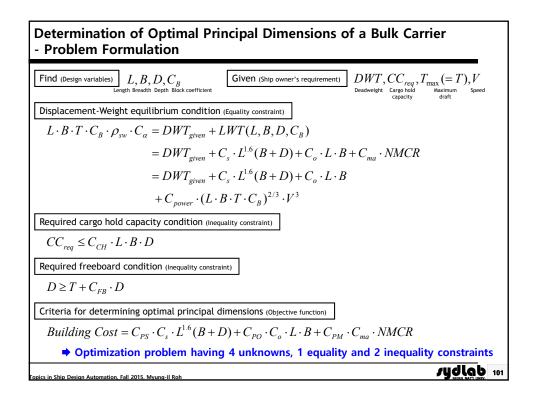


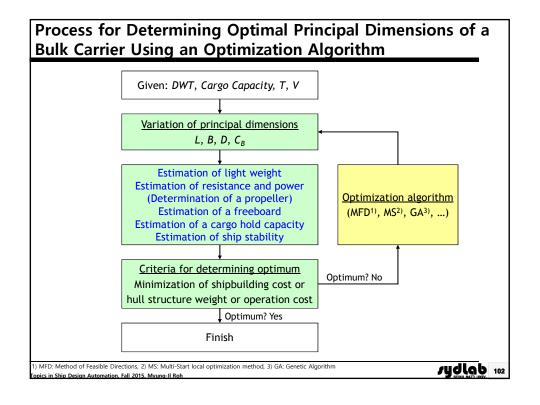
- Constraint about the displacement-weight equilibrium condition
- Constraint about the required cargo hold capacity
- Constraint about the required freeboard condition

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Determination of Optimal Principal Dimensions of a Bulk Carrier - Given Information

Principal particulars of a deadweight 150,000 ton bulk carrier (parent ship) and ship owner's requirements

	Item	Parent Ship	Design Ship	Remark
L _{OA}		abt. 274.00 m	max. 284.00 m	
	L _{BP}	264.00 m		
Principal	B _{mld}	45.00 m	45.00 m	
Dimensions	D _{mld}	23.20 m		
	T _{mld}	16.90 m	17.20 m	
	T _{scant}	16.90 m	17.20 m	
Deadweight		150,960 ton	160,000 ton	at 17.20 m
	Speed			90 % MCR (with 20 % SM)
	TYPE	B&W 5S70MC		
M / E	NMCR	17,450 HP×88.0 RPM		Derating Ratio = 0.9
Ē	DMCR	15,450 HP×77.9 RPM		E.M = 0.9
Ī	NCR	13,910 HP×75.2 RPM		
F	SFOC	126.0 g/HP.H		
O TON/DAY		41.6		Based on NCR
Cruising Range		28,000 N/M	26,000 N/M	7
Midship Section		Single Hull Double Bottom/Hopper /Top Side Wing Tank	Single Hull Double Bottom/Hopper /Top Side Wing Tank	
	Cargo	abt. 169,380 m ³	abt. 179,000 m ³	Including Hatch Coamin
Capacity	Fuel Oil	abt. 3,960 m ³		Total
Capacity	Fuel Oil	abt. 3,850 m ³		Bunker Tank Only
Ī	Ballast	abt. 48,360 m ³		Including F.P and A.P Tan

Determination of Optimal Principal Dimensions of a Bulk Carrier - Optimization Result

Minimization of Shipbuilding Cost

		Unit	MFD ¹⁾	MS ²⁾	GA ³⁾	HYBRID ⁴⁾ w/o Refine	HYBRID ⁴⁾ with Refine
G	DWT	ton	160,000				
l V	Cargo Capacity	m³	179,000				
E	T _{max}	m	17.2				
N	V	knots	13.5				
L		m	265.54	265.18	264.71	264.01	263.69
	B m		45.00	45.00	45.00	45.00	45.00
	D		24.39	24.54	24.68	24.71	24.84
	C _B	-	0.8476	0.8469	0.8463	0.8427	0.8420
	D _P	m	8.3260	8.3928	8.4305	8.4075	8.3999
	P _i		5.8129	5.8221	5.7448	5.7491	5.7365
	A _E /A _O	-	0.3890	0.3724	0.3606	0.3618	0.3690
Е	Building Cost	\$	59,889,135	59,888,510	59,863,587	59,837,336	59,831,834
	Iteration No	-	10	483	96	63	67
	CPU Time ⁵⁾	sec	4.39	209.58	198.60	184.08	187.22

1) MFD: Method of Feasible Directions, 2) MS: Multi-Start local optimization method, 3) GA: Genetic Algorithm 4) HYBRID: Global-local hybrid optimization method, 5) 테스트 시스템: Pentium 3 866Mhz, 512MB RAM

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Example for the Determination of Optimal Principal Dimensions of a Naval Ship

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Determination of Optimal Principal Dimensions of a Naval Ship

- ☑ Problem for determining optimal principal dimensions of a 9,000ton missile destroyer (DDG)
 - Objective
 - Minimization of a power (BHP) or Fuel Consumption (FC) of a main engine (f₁)

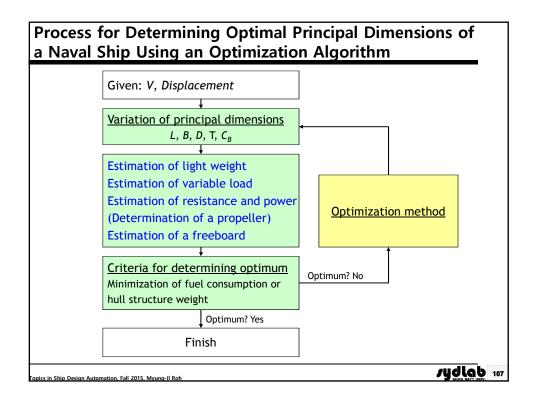
or

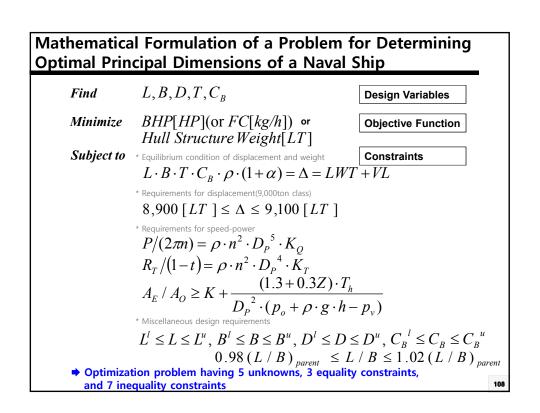
- Minimization of hull structure weight (f₂)
- Input (Given, Ship owner's requirements)
 - Δ: Displacement
 - V: Speed
- Output (Find)
 - L: Length
 - B: Moulded breadth
 - D: Moulded depth
 - T: Draft
 - C_R: Block coefficient



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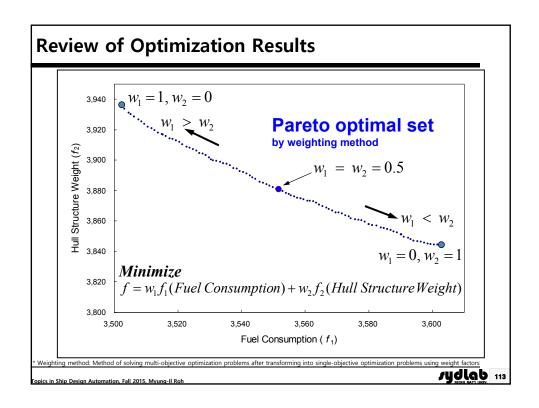


•		Result for ion of F		sumptio	n		
CASE 1: Mi	nimize f	uel consum	ption (f_1)				
	Unit	DDG-51	MFD	MS	GA	HYBRID w/o Refine	HYBRID with Refine
L	m	142.04	157.68	157.64	157.60	157.79	157.89
В	m	17.98	20.11	19.69	19.47	19.60	19.59
D	m	12.80	12.57	12.67	12.79	12.79	12.74
Т	m	6.40	5.47	5.57	5.69	5.68	5.63
C _B	-	0.508	0.520	0.506	0.506	0.508	0.512
P _i	m	8.90	9.02	9.38	9.04	9.06	9.06
A _E /A _O	-	0.80	0.80	0.65	0.80	0.80	0.80
n	rpm	88.8	97.11	94.24	96.86	96.65	96.64
F.C (f ₁)	kg/h	3,391.23	3,532.28	3,526.76	3,510.53	3,505.31	3,504.70
H.S.W	LT	3,132	3955.93	3901.83	3910.41	3942.87	3,935.39
Δ	LT	8,369	9,074	8,907	8,929	9,016	9,001
Iteration No	-	-	6	328	97	61	65
CPU Time	sec	-	3.83	193.56	195.49	189.38	192.02

				7	•		
CASE 2: Mi		null structure				HYBRID	HYBRID
	Unit	DDG-51	MFD	MS	GA	w/o Refine	with Refine
L	m	142.04	157.22	155.92	155.78	155.58	155.56
В	m	17.98	20.09	20.09	20.12	20.10	20.09
D	m	12.80	12.72	12.66	12.63	12.66	12.67
Т	m	6.40	5.64	5.63	5.61	5.65	5.66
C _B	-	0.508	0.510	0.506	0.508	0.508	0.508
P _i	m	8.90	8.98	9.42	9.04	9.46	9.45
A _E /A _O	-	0.80	0.80	0.65	0.80	0.65	0.65
n	rpm	88.8	97.40	94.06	97.29	93.93	93.98
F.C	kg/h	3,391.23	3,713.23	3,622.40	3,618.71	3,603.89	3,602.60
H.S.W (f ₂)	LT	3,132	3,910.29	3,855.48	3,850.56	3,844.43	3,844.24
Δ	LT	8,369	9,097	9,014	9,008	9,004	9,003
Iteration No	-	-	7	364	95	64	68
CPU Time	sec	-	3.91	201.13	192.32	190.98	192.41

•				/linimiza tructure		1	
CASE 3: Mi		* w ₁ = w ₂ =					
	Unit	DDG-51	MFD	MS	GA	HYBRID w/o Refine	HYBRID with Refine
L	m	142.04	157.37	157.02	156.74	156.54	156.51
В	m	17.98	19.99	19.98	19.82	19.85	19.82
D	m	12.80	12.70	12.69	12.73	12.82	12.84
Т	m	6.40	5.61	5.62	5.67	5.77	5.80
C _B	-	0.508	0.510	0.506	0.506	0.508	0.508
P _i	m	8.90	9.02	9.51	9.33	9.50	9.05
A _E /A _O	-	0.80	0.80	0.65	0.65	0.65	0.65
N	rpm	88.8	97.11	93.49	94.53	93.52	93.51
F.C (f ₁)	kg/h	3,391.23	3,589.21	3,583.56	3,556.15	3,551.98	3,551.42
H.S.W (f ₂)	LT	3,132	3,931.49	3,896.54	3,891.45	3,880.74	3,880.18
$w_1f_1 + w_2f_2$	-	3,261.62	3,760.35	3,740.05	3,723.80	3,716.36	3,715.80
Δ	LT	8,369	9,074	9,048	9,004	9,001	9,001
Iteration No	-	-	7	351	93	65	68
CPU Time	sec	-	3.99	201.63	191.28	190.74	193.22

			CASE 1	CASE 2	CASE 3
	Unit	DDG-51	Minimize f ₁ (fuel consumption)	Minimize f ₂ (hull structure weight)	Minimize w ₁ f ₁ +w ₂ f ₂
L	m	142.04	157.89	155.56	156.51
В	m	17.98	19.59	20.09	19.82
D	m	12.80	12.74 12.67		12.84
Т	m	6.40	5.63 5.66		5.80
Св	-	0.508	0.512	0.508	0.508
P _i	m	8.90	9.06	9.45	9.05
A _E /A _O	-	0.80	0.80	0.65	0.65
n	rpm	88.8	96.64	93.98	93.51
F.C	kg/h	3,391.23	3,504.70	3,602.60	3,551.42
H.S.W	LT	3,132	3,935.39	3,844.24	3,880.18
Objective	-	-	3,504.70 3,844.24		3,715.80
Δ	LT	8,369	9,001	9,003	9,001
Iteration No	-	-	65	68	68
CPU Time	sec	-	192.02	192.41	193.22



8.4 Determination of Optimal Principal Dimensions of Hatch Cover

Generals

Mathematical Formulation and Its Solution

Example

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JUDIO 114

Generals Julia peics in Ship Design Automation, Fall 2015, Myung-II Roh

Hatch Cover of a Bulk Carrier as Optimization Target (1/2)

- ☑ Bulk carrier: Dry cargo ship of transporting grains, ores, coals, and so on without cargo packaging
- $\ensuremath{\square}$ Hatch: Opening for loading and off-loading the cargo





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Hatch Cover of a Bulk Carrier as Optimization Target (2/2)

☑ Hatch cover

- Cover plate on the hatch for protecting the cargo
- Having a structure of stiffened plate which consists of a plate and stiffeners
- In general, the cost of hatch cover equipment is accounting for 5~8% of shipbuilding cost.
- In spite of the importance of the hatch cover in the B/C, it has hardly been optimized. Thus, the hatch cover was selected as an optimization target for the lightening of the ship weight in this study.





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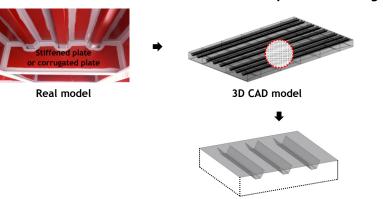
Mathematical Formulation and Its Solution

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Jydlab 118

Idealization of Hatch Cover of a Bulk Carrier

- ☑ The hatch cover has a structure of stiffened plate which consists
 of a plate and stiffeners and looks like a corrugated plate.
- ☑ The hatch cover can be idealized for the effective optimization.
- ☑ Thus, the idealized model will be used as the optimization target.



Idealized model

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rydlab 119

Determination of Optimal Principal Dimensions of a Hatch Cover - Problem Definition

- ☑ Criteria for determining optimal principal dimensions (Objective function)
 - Minimization of the weight of hatch cover

☑ Given

- Length (L), width (W), height (H) of hatch cover
- Total number of girders and transverse web frames
- Load (p_H) on the hatch cover
- The largest span of girders (l_a)
- Materials of the hatch cover

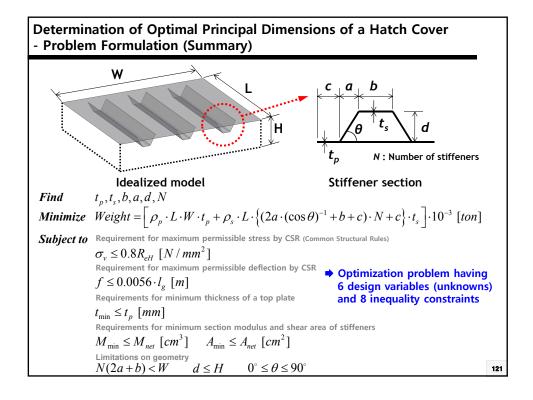
☑ Find (Design variables)

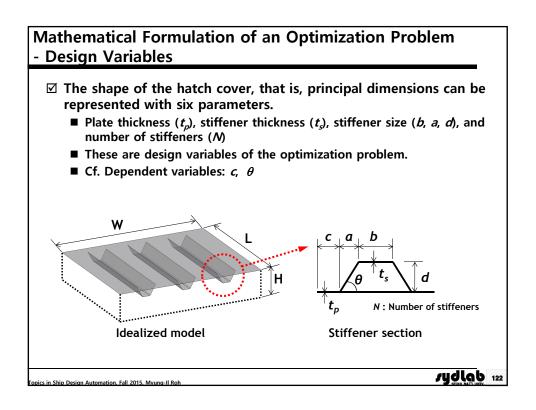
 Plate thickness (tp), stiffener thickness (ts), stiffener size (b, a, d), and number of stiffeners (N)

☑ Constraints

- Constraints about the maximum permissible stress and deflection
- Constraint about the minimum thickness of a top plate
- Constraints about the minimum section modulus and shear area of stiffeners
- Constrains about geometric limitations

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Mathematical Formulation of an Optimization Problem - Constraints (1/6)

☑ Maximum Permissible Stress of the Hatch Cover

$$\sigma_{v} \leq 0.8 R_{eH} [N/mm^{2}]$$

where,

$$\sigma_{\scriptscriptstyle V} = \sqrt{\sigma^2 + 3\tau^2} \ [N \, / \, mm^2] \quad \text{or} \quad \sigma_{\scriptscriptstyle V} = \sqrt{\sigma_{\scriptscriptstyle X}^2 - \sigma_{\scriptscriptstyle X} \cdot \sigma_{\scriptscriptstyle Y} + \sigma_{\scriptscriptstyle Y}^2 + 3\tau^2} \ [N \, / \, mm^2]$$

($\sigma_{\!v}$: equivalent stress, τ : shear stress, $\sigma_{\!x}$ and $\sigma_{\!y}$: normal stress in x- and y- direction)

$$\sigma = \sigma_b + \sigma_n$$

(σ_b : bending stress, σ_n : normal stress)

 R_{eH} : yield strength, given as: 235 [N/mm²] for mild steel, 315 [N/mm²] for AH32, 355 [N/mm²] for AH36

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Mathematical Formulation of an Optimization Problem - Constraints (2/6)

☑ Maximum Permissible Deflection of the Hatch Cover

$$f \le 0.0056 \cdot l_g [m]$$

where,

f: deflection [m] of the hatch cover

 l_g : The largest span [m] of girders in the hatch cover

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Mathematical Formulation of an Optimization Problem - Constraints (3/6)

☑ Minimum Thickness of a Top Plate of the Hatch Cover

$$t_{\min} \le t_p \ [mm]$$

where,

$$t_{\min} = \max(t_1, t_2, t_3)$$
 $t_1 = 16.2 \cdot c_p \cdot c \cdot \sqrt{\frac{p}{R_{eH}}} + t_k [mm]$

$$t_2 = 10 \cdot c + t_k \ [mm]$$
 $t_3 = 6.0 + t_k \ [mm]$

 t_k : corrosion additions (2.0 mm for hatch covers in general, See Table 17.1 in [1]) c_n : coefficient, defined as

$$c_p = 1.5 + 2.5 \cdot \left(\frac{|\sigma|}{R_{eH}} - 0.64 \right) \ge 1.5 \text{ for } p = p_H$$

 \emph{c} : spacing [m] of stiffeners

p: design load [kN/m²]

 p_H : load on the hatch cover [kN/m²] (See Table 17.2 in [1])

 Germanischer Lloyd, 2014. Rules for classification and construction, Rules I. Ship Technology, Part 1. Seagoing Ships, Chapter 1. Hull Structures, Section 17. Cargo Hatchways. Germanischer Hoyd

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Mathematical Formulation of an Optimization Problem - Constraints (4/6)

☑ Minimum Section Modulus of Stiffeners of the Hatch Cover

$$M_{\min} \le M_{net} [cm^3]$$

where,

 M_{net} : net section modulus [cm³]

 M_{\min} : minimum section modulus, defined as

$$M_{net} = \frac{104}{R_{eH}} \cdot c \cdot l^2 \cdot p \ [cm^3]$$

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Mathematical Formulation of an Optimization Problem - Constraints (5/6)

☑ Minimum Shear Area of Stiffeners of the Hatch Cover

$$A_{\min} \le A_{net} [cm^2]$$

where,

 A_{net} : net shear area [cm²]

 A_{\min} : minimum shear area, defined as

$$A_{\min} = \frac{10 \cdot c \cdot l \cdot p}{R_{eH}} [cm^2]$$

l: unsupported span [m] of stiffener

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Mathematical Formulation of an Optimization Problem - Constraints (6/6)

☑ Geometric Limitations Related to the Shape of the Hatch Cover

$$N(2a+b) < W$$
 $d \le H$ $0^{\circ} \le \theta \le 90^{\circ}$

where,

W: width [m] of the hatch cover

D: depth [m] of the hatch cover

 θ angle between the plate and stiffener

→ This optimization problem has total 8 inequality constraints.

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Mathematical Formulation of an Optimization Problem - Objective Function

- ☑ An optimal hatch cover means a hatch cover having minimum weight.
- ☑ Thus, the weight of the hatch cover was selected as the objective function of the optimization problem.

$$\begin{aligned} \textit{Minimize} \quad \textit{Weight} = & \left[\rho_p \cdot L \cdot W \cdot t_p + \rho_s \cdot L \cdot \left\{ (2a \cdot (\cos \theta)^{-1} + b + c) \cdot N + c \right\} \cdot t_s \right] \cdot 10^{-3} \ [\textit{ton}] \end{aligned}$$
 where,

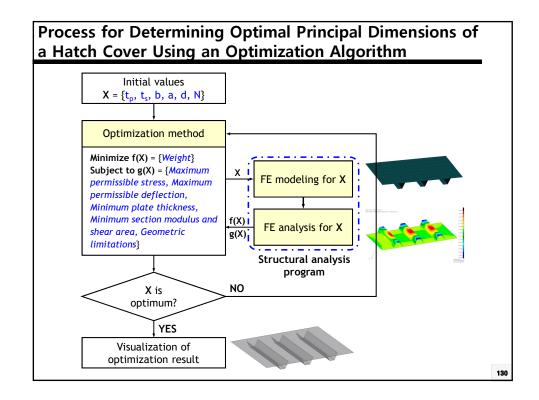
 ρ_{p} and ρ_{s} : specific gravity [ton/m³] of plate and stiffener, respectively

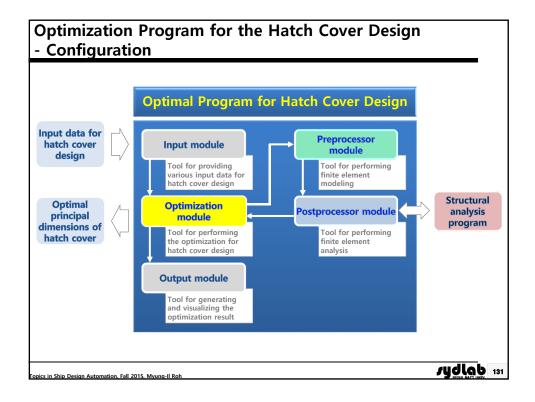
L: length [m] of the hatch cover

 A_{\min} : stiffener thickness [mm]

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Optimization Program for the Hatch Cover Design - Components (1/5)

☑ Input Module

- The input module inputs some data for optimization of the hatch cover from a designer.
- The data includes the size (length, width, and depth) of the hatch cover, materials of plate and stiffeners, and so on.
- In addition, the input module generates initial values for design variables and transfers them to the optimization module.

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Optimization Program for the Hatch Cover Design - Components (2/5)

☑ Optimization Module

- The optimization module includes the multi-start optimization algorithm.
- The module calculates the values of an objective function and constraints are calculated.
- By using the values, the module improves the current values of the design variables.
- At this time, the finite element modeling and analysis for the current values of the design variables should be performed in order to calculate some structural responses such as the stress and deflection of the hatch cover for the values of the design variables.
- Thus, this module is linked with the preprocessor and postprocessor modules, and calls them when needed.

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Optimization Program for the Hatch Cover Design - Components (3/5)

☑ Preprocessor Module

- To calculate the structural responses by using a structural analysis program, a finite element model is required.
- The preprocessor module is used to generate the finite element model for the current values of the design variables.
- That is, the role of the module is the finite element modeling.
- In this module, an input file for the execution of the structural analysis program is generated with the current values of the design variables.
- The input file is transferred to the postprocessor module.

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Optimization Program for the Hatch Cover Design - Components (4/5)

☑ Postprocessor Module

- In the post processor module, the structural analysis program is executed with the input file from the preprocessor module.
- That is, the role of the module is to perform the finite element analysis.
- In this study, the ANSYS which is one of commercial structural analysis programs was used for the structural analysis.
- After performing the finite element analysis with the structural analysis program, the structural responses such as the stress and deflection of the hatch cover can be acquired.
- The values of the structural responses are written in the output file by the structural analysis program.
- The postprocessor module parses the output file by the structural analysis program, and transfers the values of the structural responses to the optimization module.

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Optimization Program for the Hatch Cover Design - Components (5/5)

☑ Output Module

- The output module outputs an optimization result from the optimization module.
- The result includes optimal dimensions (optimal values of the design variables), weight, maximum stress, maximum deflection of the hatch cover, and so on.

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Example

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Hatch Cover Design of a Deadweight 180,000 ton Bulk Carrier - Mathematical Formulation

$$\begin{aligned} \textit{Find} & t_p, t_s, b, a, d, N \\ \textit{Minimize} & \textit{Weight} = \left[\rho_p \cdot L \cdot W \cdot t_p + \rho_s \cdot L \cdot \left\{ (2a \cdot (\cos \theta)^{-1} + b + c) \cdot N + c \right\} \cdot t_s \right] \cdot 10^{-3} \ [ton] \\ & = \left[7.85 \cdot 14.929 \cdot 8.624 \cdot t_p + 7.85 \cdot 14.929 \cdot \left\{ (2a \cdot (\cos \theta)^{-1} + b + c) \cdot N + c \right\} \cdot t_s \right] \cdot 10^{-3} \\ & : \text{weight of top plate and stiffeners} \end{aligned}$$

Subject to

$$\begin{split} &\sigma_{_{\!V}} \leq 0.8 \cdot 315 \left[N \, / \, mm^2 \right] \; : \; \text{maximum permissible stress} \\ &f \leq 0.0056 \cdot 3.138 \left[m \right] \quad : \; \text{maximum permissible deflection} \\ &t_{\min} \leq t_p \left[mm \right] \qquad : \; \text{minimum thickness of a top plate} \\ &M_{\min} \leq M_{\textit{net}} \left[cm^3 \right] \qquad : \; \text{minimum section modulus of stiffeners} \\ &A_{\min} \leq A_{\textit{net}} \left[cm^2 \right] \qquad : \; \text{minimum shear area of stiffeners} \end{split}$$

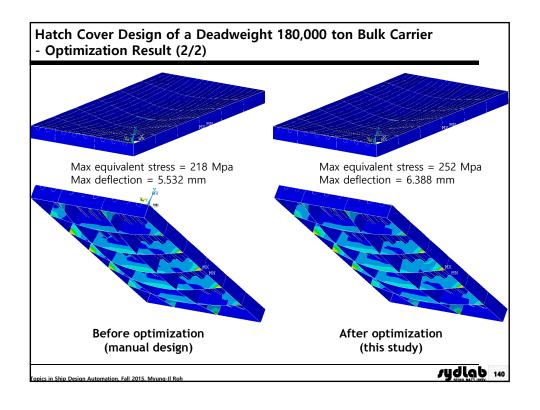
N(2a+b) < W : geometric limitation d < H : geometric limitation $0^\circ < \theta \le 90^\circ$: geometric limitation

→ Optimization problem having 6 design variables and 8 inequality constraints

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ltem	Unit	Manual design	Optimization result
t _p	mm	16	14
t _s	mm	8	8
b	m	0.170	0.160
a	m	0.120	0.111
d	m	0.220	0.198
N	-	8	8
Weight	ton	26.225	23.975
Maximum stress	MPa	218	252
Maximum deflection	mm	5.532	6.388



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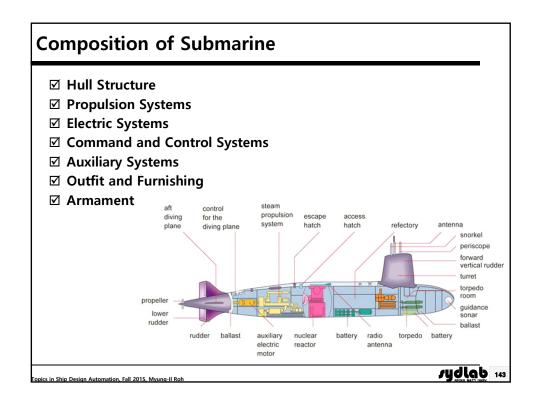
8.5 Determination of Optimal Principal Dimensions of Submarine

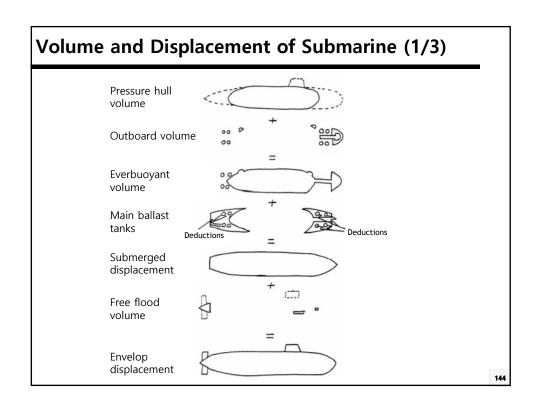
Generals Mathematical Formulation and Its Solution Example

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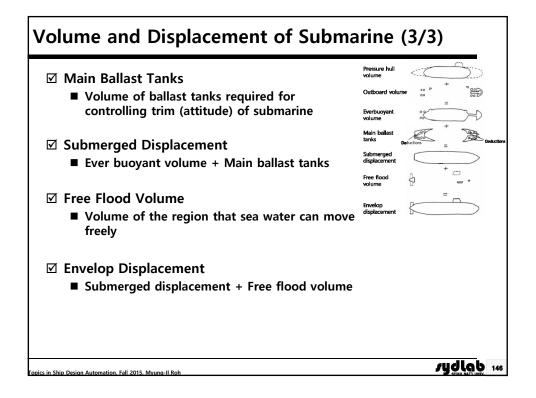
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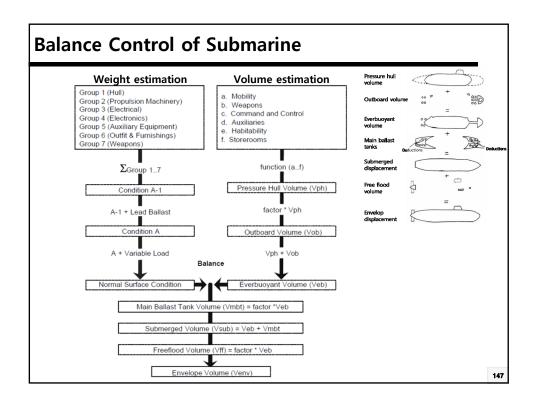
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Volume and Displacement of Submarine (2/3) ✓ Pressure Hull Volume ■ Watertight volume having important parts of submarine ✓ Outboard Volume ■ Volume of weapons and propulsion systems which are installed outside of pressure hull ✓ Everbuoyant Volume ■ Total volume related to buoyancy among volumes of submarine ■ Basis for calculating Normal Surface Condition Weight (NSCW) ■ NSCW = Ever buoyant volume / density of sea water





Weight Estimation of Submarine

- ☑ Composition of Weight (Displacement)
 - Lightweight (LWT) + Variable Load (VL, cargo weight)
 - Most of displacement becomes the lightweight.
- ☑ Weight Estimation Method (SWBS* Group of US Navy)

Group	ltem
100	Hull Structure
200	Propulsion
300	Electric Systems
400	Communication and Control
500	Auxiliary System
600	Outfitting and Furnishing
700	Armament

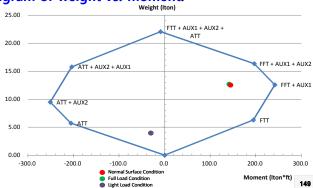
Straubinger, E.K., Curran, V.L., "Fundamentals of Naval Surface Ship Weight Estimating, Naval Engineers Journal, pp.127-143, 1983. SWBS: Ships Work Breakdown Structure

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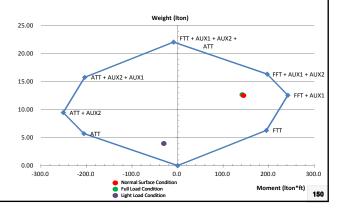
Meaning of Equilibrium Polygon (1/2)

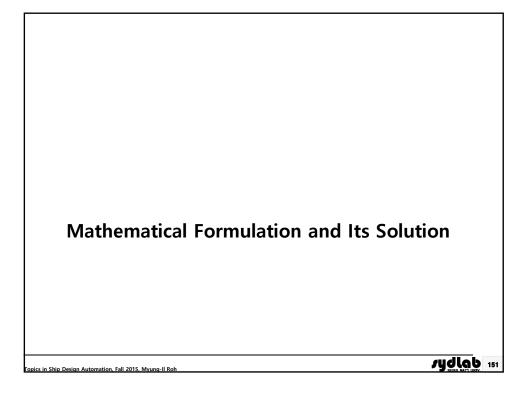
- ☑ The equilibrium polygon is a graphical tool that is used to ensure that the submarine will be able to remain neutrally buoyant and trimmed level while submerged in any operating (loading) condition.
- ☑ In all operating conditions the ship must be able to compensate which is accomplished through the variable ballast tanks.
- ☑ The polygon is a diagram of weight vs. moment.

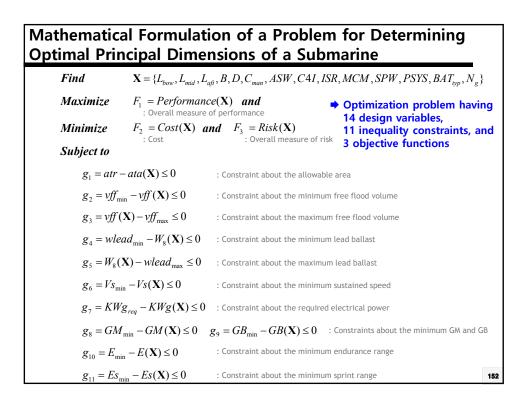


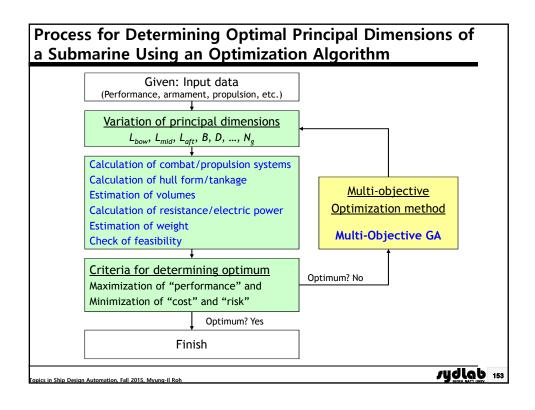
Meaning of Equilibrium Polygon (2/2)

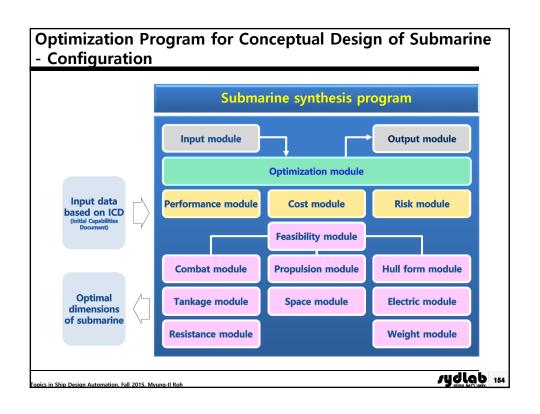
- ☑ The boundaries of the graphic are calculated from the variable tanks.
- ☑ Weights and moments are then calculated based on their compensation for all extreme loading conditions.
- ☑ The ship is adequately able to compensate for each loading conditions if each point lies within the polygon.











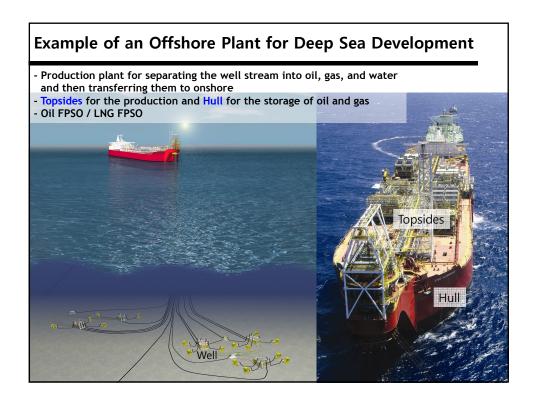
8.6 Generation of Weight Estimation **Model Using the Optimization Method**

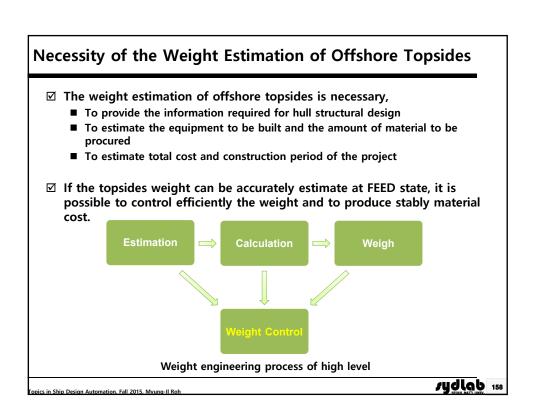
Generals Generation of Weight Estimation Model by Using **Genetic Programming Example**

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Generals JULIAN 156

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Classification of Weight Estimation Methods (1/3)

☑ Volumetric Density Method

- A method of estimating the detailed weight group by the multiplication of space volume and bulk factor (density)
- e.g., detailed weight = space volume * bulk factor

☑ Parametrics

- A method of representing the weight with several parameters, and an essential prerequisite of the following ratiocination
- e.g., hull structural weight = L^{1.6}(B + D)

☑ Ratiocination

- A method of estimating the weight with a ratio from past records and a parametric equation
- e.g., hull structural weight = $C_S \cdot L^{1.6}(B + D)$)

☑ Baseline Method

A method of estimating the weight by using the result of the first one for a series of ships and offshore plants

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Classification of Weight Estimation Methods (2/3)

☑ Midship Extrapolation Method

- A method of estimating the weight by the multiplication of the length and the midship weight per unit length
- e.g., fore body weight = midship weight per unit * fore body length * coeff.

☑ Deck Area Fraction Method

- A method of estimating the weight by the multiplication of the deck area and the deck weight per unit area
- e.g., detailed weight = deck weight per area * deck area * coeff.

☑ Synthesis Method

- A method of estimating by using a delicate synthesis program which was made from the integration all engineering fields (e.g., performance) based on requirements
- Most ideal method but it needs much time and efforts.

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Classification of Weight Estimation Methods (3/3)

☑ Statistical Method

- A method of developing a weight equation from statistical analysis of various past records, and of estimating the weight by using the equation
- ☑ Optimization Method
 ➡ To be presented here
 - A method of developing a weight equation by optimization method such as genetic programming

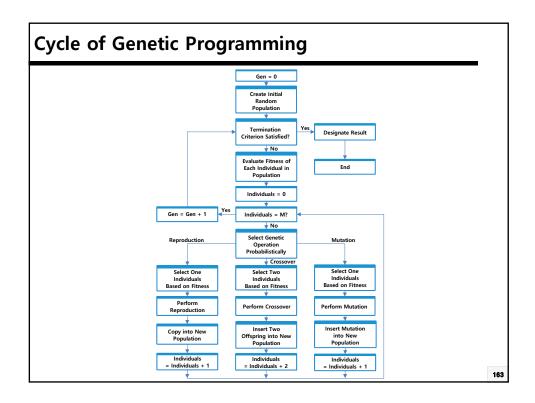
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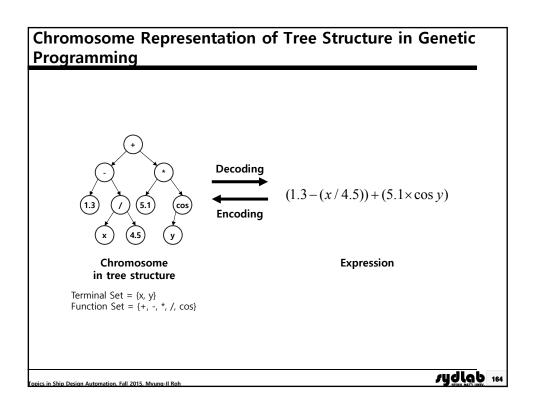
rydlab 161

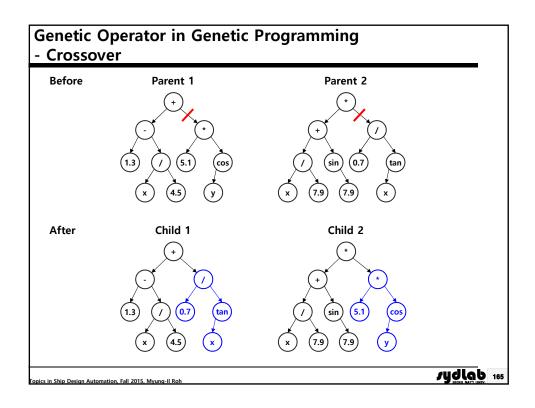
Generation of Weight Estimation Model by Using Genetic Programming

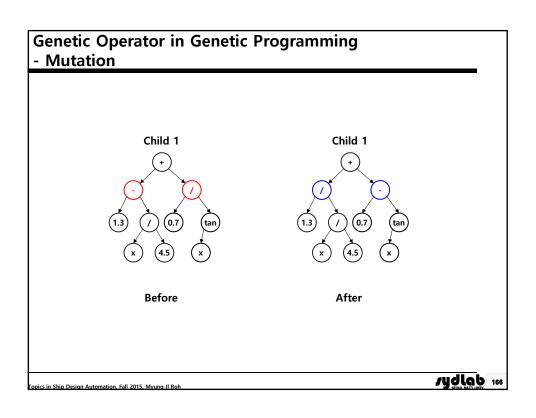
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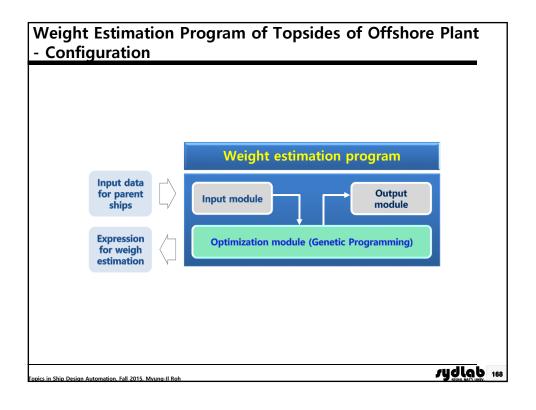


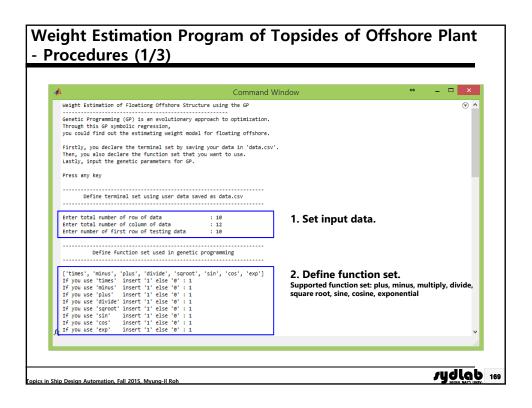


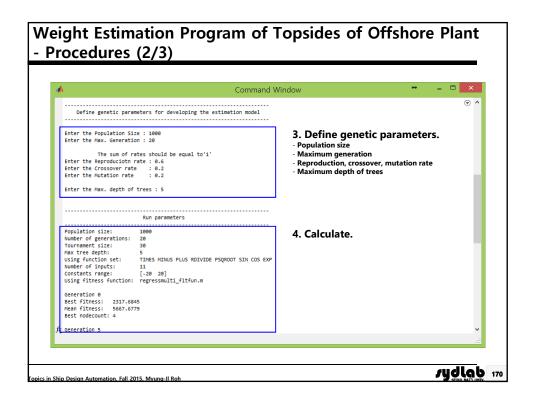


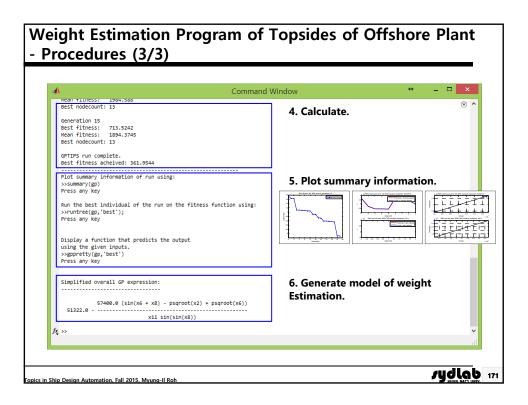


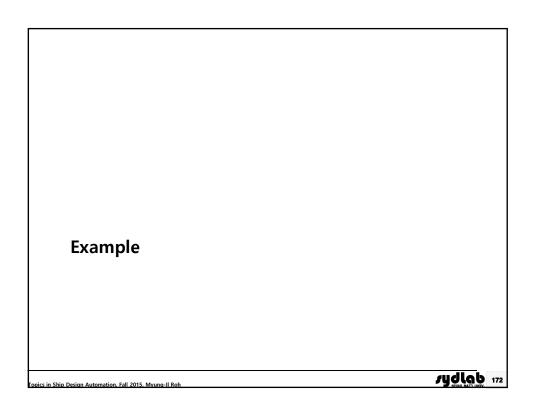
	Genetic algorithms (e.g., Binary-string coding)	Generic Programming
	Binary string of 0 and 1	Function
Expression	String	Tree
	Fixed length	Length variable
Main operator	Crossover	Crossover
Structure	1010110010101011	1.3 / (5.1 cos) x (4.5 y)



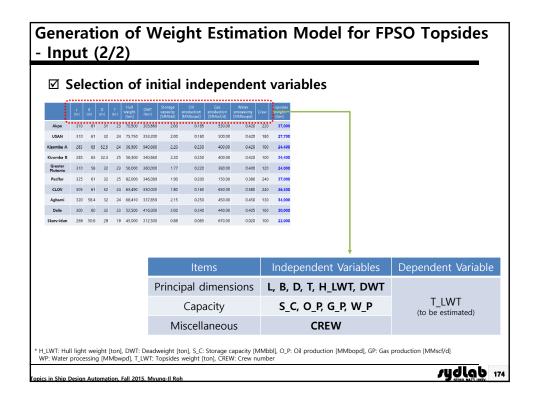








☑ Past records for FPSOs from the literature survey												
	L [m]	B [m]	D [m]	T [m]	Hull weight [ton]	DWT [ton]	Storage capacity [MMbbl]	Oil production [MMbopd]	Gas production [MMscf/d]	Water processing [MMbwpd]	Crew	Topsides weight [ton]
Akpo	310	61	31	23	70,500	303,669	2.00	0.185	530.00	0.420	220	37,00
USAN	310	61	32	24	75,750	353,200	2.00	0.160	500.00	0.420	180	27,70
Kizomba A	285	63	32.3	24	56,300	340,660	2.20	0.250	400.00	0.420	100	24,40
Kizomba B	285	63	32.3	25	56,300	340,660	2.20	0.250	400.00	0.420	100	24,40
Greater Plutonio	310	58	32	23	56,000	360,000	1.77	0.220	380.00	0.400	120	24,00
Pazflor	325	61	32	25	82,000	346,089	1.90	0.200	150.00	0.380	240	37,00
CLOV	305	61	32	24	63,490	350,000	1.80	0.160	650.00	0.380	240	36,30
Agbami	320	58.4	32	24	68,410	337,859	2.15	0.250	450.00	0.450	130	34,00
Dalia	300	60	32	23	52,500	416,000	2.00	0.240	440.00	0.405	160	30,00
Skarv-Idun	269	50.6	29	19	45,000	312,500	0.88	0.085	670.00	0.020	100	22,00



Generation of Weight Estimation Model for FPSO Topsides - Output

☑ Simplified model for the weight estimation

■ The model can be represented as the nonlinear relationship between 11 independent variables and the corresponding coefficients.

$$\begin{split} T_{LWT} &= 67.38 \cdot CREW + 67.38 \cdot B + 67.38 \cdot S _C - \\ 3059 \cdot \cos(L \cdot W _P \cdot (H _LWT - 3.838)) + \\ 12533 \cdot \cos(\exp(\sin(S _C))) + 0.5007 \cdot B \cdot T + \\ 67.38 \cdot O _P \cdot G _P + \\ 0.5007 \cdot D \cdot \sin(H _LWT) \cdot L^2 - 30033 \end{split}$$

* H_LWT: Hull light weight [ton], DWT: Deadweight [ton], S_C: Storage capacity [MMbbl], O_P: Oil production [MMbopd], GP: Gas production [MMscf/d] WP: Water processing [MMbwpd], T_LWT: Topsides weight [ton], CREW: Crew number

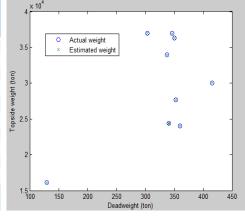
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Generation of Weight Estimation Model for FPSO Topsides - Verification of the Weight Estimation Model

 $T_{LWT} = 67.38 \cdot CREW + 67.38 \cdot B + 67.38 \cdot S - C - 3059 \cdot \cos(L \cdot W - P \cdot (H - LWT - 3.838)) + 12533 \cdot \cos(\exp(\sin(S - C))) + 0.5007 \cdot B \cdot T + 67.38 \cdot O - P \cdot G - P + 0.5007 \cdot D \cdot \sin(H - LWT) \cdot L^2 - 30033$

FPSOs	Actual weight [A]	Estimated weight [B]	Ratio [A/B]
Akpo	37,000	36,951	0.9987
USAN	27,700	27,672	0.9990
Kizomba A	24,400	24,352	0.9980
Kizomba B	24,400	24,383	0.9993
Greater Plutonio	24,000	24,063	1.0226
Pazflor	37,000	36,918	0.9978
CLOV	36,300	36,318	1.0005
Agbami	34,000	33,906	0.9972
Dalia	30,000	30,059	1.0020
Skarv-Idun	16,100	16,093	0.9996
Test	25,000	24,928	0.9971
	Mean		1.0011



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