ATA Specification 103

Standard for Jet Fuel Quality Control at Airports

Revision 2006.1



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Highlights

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[Section 2-1]	Amended
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[Section 2-4]	Amended
[Section 2-5]	Revised
[Section 2-6]	Amended
[Section 2-7]	Amended
[Section 2-8]	Revised
[Section 2-9]	Amended
[Section 3-3]	Amended
[Section 3-10]	Amended
ATA Form 103-01A [Figure 6-2.1]	Revised
ATA Form 103-01B [Figure 6-2.2]	Revised
ATA Form 103-01C [Figure 6-2.3]	Revised
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ATA Form 103-04C [Figure 6-2.9]	Revised
ATA Form 103-07 [Figure 6-2.14]	Revised
ATA Form 103-08 [Figure 6-2.15]	Revised
[Chapter 4]	Amended
[Annex 1]	Amended

Preface

This document is intended to provide guidance to the user covering the safe storage and distribution of quality jet fuel at airports as currently practiced in the commercial aviation industry. Due to the wide diversity of airport fueling operations, this document is not intended be all-inclusive. Technical information, along with competent judgment, should be considered and followed at all times when overseeing aviation fueling operations. In addition, operators should comply with all applicable rules, regulations, restrictions, ordinances and other laws of federal, state, local and airport entities relating to fuel storage and distribution.

Chapter 1. Introduction

Member airlines of the Air Transport Association of America (ATA) recognize the importance of using quality jet fuel for ensuring the highest degree of flight safety. To achieve this goal, ATA Specification 103, entitled "Standard for Jet Fuel Quality Control at Airports," was developed by member airlines to cover fuel distribution facilities and fuel quality control procedures at airports servicing airline operations.

This standard identifies commonly recognized industry inspection procedures and safety checks of jet fuel storage and distribution facilities at airports that will help minimize introduction of contaminated or unacceptable jet fuel from being delivered to airline aircraft. It is important to note that additional facilities and testing procedures may be required at individual airports based on fuel system complexity and local operating conditions. Alternative procedures and use of non-standard facilities and equipment may also be recognized and determined acceptable for achieving the above safety requirements based on extenuating circumstances.

ATA Specification 103 does not, in itself, impose any performance obligations on any airline, fuel supplier, fuel storage facility, fuel transporter, or any other entity. Its provisions become effective only to the extent they are adopted by an airline and incorporated into its operating manual.

Chapter 2. General

2-1. General

This section covers jet fuel handling issues and procedures that are general in nature and are applicable to all facets of jet fuel handling at airports.

1. Scope

This document contains standards covering airport jet fuel storage facilities, hydrant distribution systems and aircraft refueling equipment to help ensure the safe and dependable flow of quality jet fuel to airline aircraft.

2. Records

All jet fuel quality assurance, airport facility and aircraft refueling equipment maintenance and training records are to be available for inspection and review during normal business hours. All records must be signed, or be adequately identified, by the person performing tasks or the person accepting responsibility that tasks were performed in accordance to this standard.

3. Notification of New or Modified Equipment

Affected airlines are to be notified well in advance whenever new, additional, replacement or modified airport fuel storage, distribution facilities or aircraft refueling equipment is placed into operation. At the airline's option, all airport fueling facilities, into-plane refueling equipment and operator's procedures may be inspected and approved for use prior to servicing airline aircraft.

4. Variance/Waiver

A variance or waiver to the policies and procedures in this document that will not compromise fuel quality, safety or security may be granted. A request for variance or waiver must be made in writing to each affected airline and shall include:

- (a) Requirement from which the variance or waiver is being requested.
- (b) Explanation as to why compliance with airline requirement is not possible or practical.
- (c) Alternate means of compliance to be considered for approval of request.
- (d) Period of time for which variance or waiver is to be effective.

5. Fuel Contamination

If visible contamination of fuel is observed or found, aircraft refueling must be discontinued from that source. Notify all affected aircraft operators if it is anticipated that such contamination might impact operations. Fueling shall not be resumed from the system until the source of fuel contamination is found and removed.

Fuel, suspected of possible contamination, shall be held in quarantine until selected fuel quality, purity or specification tests have determined that it is acceptable for aircraft use. Selected product tests and expected acceptance criteria are to be determined and mutually agreed upon by fueling vendor and all affected customers prior to approving fuel for future use. To the extent the fuel is no longer acceptable for aircraft use, it should be managed and/or disposed consistent with applicable federal, state and local requirements.

6. Defueled Product

Product defueled from an aircraft for purposes other than contamination should be returned to the airline from which it was removed. Defueled product may not be delivered to another airline's aircraft without their approval. Defueling aircraft directly into joint use fueling systems is not authorized unless all system users have unanimously approved a joint use procedure.

7. Inoperative System

If for any reason a fueling system becomes inoperative so as to impair normal refueling operations, all affected airlines must be notified immediately.

8. Training/Qualification

Facility and fueling equipment operators are responsible for ensuring that all personnel under their direction and control are properly trained and qualified for performing tasks assigned to them as specified in this document. Training and qualification records are to be available for review by the airline or its designated representative.

9. Deficiency Reporting

The facility and equipment operator shall establish written procedures for the reporting of any observed deficiencies or safety hazards by its employees to their supervisors.

10. Operations & Maintenance Manuals

All airport fueling vendors having aviation fuel storage facilities and/or aircraft refueling equipment administrative and operational responsibilities should have operations and maintenance (O & M) manuals. These documents are intended to be used by fuel handling vendors and equipment operators to help ensure the safe and dependable flow of quality fuel to aircraft. Guidance on the development of a site-specific O & M manual can be found in the ATA "Airport Fuel Facility Operations and Maintenance Guidance Manual".

2-2. Jet Fuel Specification & Purity Standards

1. Specification Requirements

Jet fuel shall conform to [ASTM D1655], Latest Revision, "Standard Specification for Aviation Turbine Fuels", Jet A or Jet A-1 Kerosene Type.

In all phases of fuel handling, appearance of jet fuel shall be clear and bright (visually free of undissolved water, sediment and suspended matter). The odor of the fuel shall not be nauseating or irritating.

The color of jet fuel generally ranges from water white to light straw or amber. Other colors may be an indication the fuel has been contaminated by other products or unauthorized additives. In such cases, it shall be the facility operator's responsibility to discontinue fuel transfer and/or quarantine product until fuel has been determined acceptable for aircraft use.

2. Upstream Jet Fuel Purity and Specification Parameters

The following jet fuel purity and specification parameters should apply **UPSTREAM** of airport receiving filtration.

Test Property	Maximum Allowable	Test Method
APPEARANCE	CLEAR & BRIGHT	VISUAL Ref. [Section 3-1]
DENSITY (API GRAVITY)	37° TO 51° API (775-840 Kg/m³) Corrected to 60° F (15° C)	[ASTM D1298] Metric measurement shall be used in event of dispute (775-840 Kg/m³).

Table 2-2.1. Jet Fuel Purity and Specification Parameters

NOTE: It is important that the facility operator assigned the task and responsibility to receive jet fuel into airport storage tanks sample inbound deliveries upstream of receiving filtration for potential contamination or excessive water/dirt levels. Inbound jet fuel purity shall permit reasonable receiving filtration system performance and service life. Upstream appearance ratings less than Clear & Bright may indicate excessive contamination levels which could result in shortened receiving filtration life and may increase operational costs. Unacceptable operational and economic issues based on upstream jet fuel purity levels are to be resolved between applicable shipper, facility operator and/or customer.

3. Downstream Jet Fuel Purity and Specification Parameters

The following jet fuel purity and specification limits shall apply **DOWNSTREAM** of the receiving and dispensing filtration as:

- 1. Received into airport storage tanks and dispensed from airport storage facilities which will issue product directly to hydrant systems and to aircraft refueler loading racks
- 2. Dispensed into aircraft

See rejection criteria in reference NOTES or in [Section 2-3] for applicable transportation methods:

Table 2-2.2. Downstream Jet Fuel Purity and Specification Limits

Test Property	Maximum Allowable	Test Method	See Notes
FREE WATER	15 PPM	Ref. [Section 3-3]	
PARTICULATE COLOR*	A, B or G 2 - DRY or A, B or G 3 - WET	[ASTM D2276]	1
PARTICLE ASSESSMENT*	A	VISUAL	2

^{*} Sample sizes are either 1 Gallon or 5 Liter

NOTE 1. A color rating of 3-DRY or greater may indicate a particulate contaminant problem. If a color rating of 3-DRY or greater is observed, proceed as follows:

Perform a subsequent particulate test consisting of two membranes in plastic holder to compare color

difference between top and bottom membranes. If top and bottom membranes have a color rating difference of 2 or less, fuel is to be considered clean and acceptable. If difference is 3 or greater, conduct a gravimetric (weight) analysis. Fuel is unacceptable if gravimetric test ([ASTM D2276]) results exceed 2.0 mg/G or 0.5 mg/L based on test sample size taken.

NOTE 2. An assessment rating of "B" or greater (reference Gammon Technical Products SGTP-3940 "Color and Particle Assessment Rating Guide" or Shell Oil Company "Filter Membrane Evaluation Guide") indicates that solid particles are visible on the test membrane or in the sample container. This observation may be an indication that there is generation of contamination in system or failure of filtration upstream of sample test connection. Particle Assessment is an aid in communicating visual observations of size and distribution of solids as they appear on test membranes or the bottom of sample containers.

2-3. Fuel Receipts Into Airport Storage

1. General

This section covers quality control and safety requirements for receiving jet fuel into airport storage. These requirements can vary depending on method of delivery and facility layout. Receipts of jet fuel at airports are normally made by dedicated or multi-product pipelines, and highway transport trucks. There are some airports receiving product directly from railroad tank cars or marine vessels. It is important that facility operator recognizes that each of these transportation methods has different delivery requirements and that they be addressed in local receiving procedures to ensure fuel quality and safety.

2. Pipeline Deliveries

Prior to delivery, airport facility operator shall receive a certification document from jet fuel supplier or shipping agent certifying product to be delivered to airport meets [ASTM D1655] specification requirements as required in [Section 2-2] with at least the following select property values listed as measured by specified ASTM test methods:

(a)	Visual Appearance in White Bucket	Ref. [Section 3-1]
(b)	Gravity, Corrected to 60° F (15° C)	[ASTM D1298]
(c)	Distillation	[ASTM D86]
	10% Recovered	
	20% Recovered	
	50% Recovered	
	90% Recovered	
	Final Boiling Point	
	Residue	
	Loss	
(d)	Flash Point	[ASTM D56]
(e)	Freezing Point	[ASTM D2386]
(f)	Water Reaction Interface Rating	[ASTM D1094]
(g)	Water Separation (MSEP)	[ASTM D3948]
(h)	Copper Strip Corrosion	[ASTM D130]
(i)	Existent Gum	[ASTM D381]

Shipping document shall also include all delivery information, i.e.: destination; batch number; fuel grade or type; quantity to be shipped.

NOTE	Fuel receipts should not be rejected based on incomplete certification documentation. If the
	certification documentation cannot be delivered prior to delivery, affected airlines must be notified.

Facility operator shall prepare receiving tank(s) and facility items prior to delivery of product, i.e.: gauging, sumping, correct inlet and outlet valve positioning, etc. The facility operator must ensure that the volume available in the receiving tank(s) is greater than the volume of product to be transferred to the tank(s) before the transfer is made.

Fuel receiving process must be monitored constantly by airport facility personnel, and such personnel must be able to communicate with the pipeline company during fuel receipt.

CAUTION: IT IS NOT ACCEPTABLE TO RECEIVE AND DISPENSE FUEL FROM THE SAME TANK SIMULTANEOUSLY

At the beginning, mid-point and near the end of fuel receipts, the facility operator shall conduct the following tests on fuel samples taken downstream of receiving filtration and record the results. This series of tests is to be repeated for each shipper tank or batch. The mid-point test may be omitted on shipments of less than four hours in duration.

- (a) Visual Appearance in White Bucket
- (b) API Gravity, Corrected to 60° F (15° C)
- (c) Color Membrane (Particulate weight to be performed only if necessary per Note 1 of 2-2.3)
- (d) Free Water Detection Test
- (e) Flash Point (Multi-Product Pipeline Deliveries Only)
- NOTE 1. Operators should be aware that pipeline volumes between shipping tanks and sampling points may be from a previous batch. Fuel tests should be timed to coincide with actual arrival of product from shipping tank.
- NOTE 2. If any of the above receiving tests fail reject limits, product in receiving tank(s) must be quarantined and not released for aircraft use pending further investigation and corrective actions as required.
- NOTE 3. Use extreme care and vigilance when performing the Visual Appearance test. Slight traces of water, solids or color may indicate the presence of product mixes or other contaminants that could cause jet fuel to be off-specification and unacceptable for aircraft use. Any unusual results must be investigated.

Fuel is unacceptable and must be rejected if API Gravity, corrected to 60° F (15° C), is not between 37° and 51° API (775-840 Kg/m³) and/or Flash Point is less than 100° F (38° C). Discontinue fuel transfer or receipt and initiate an immediate investigation to determine if there is fuel contamination or a specification problem if there is a change of more than 1° API or 5° F (3° C) in Flash Point from source as shown on shipping document.

If a certification document was not obtained prior to receipt, into airport storage, via a Multi-Product pipeline, the facility operator should immediately conduct the following [ASTM D1655] property tests prior to releasing the tank for aircraft use. Fuel not meeting [ASTM D1655] specification is to be rejected.

Table 2-3.1. ASTM D1655 Property Test

	PROPERTY	SPEC LIMIT	MAX DIFFERENCE
(a)	Visual Appearance in White Bucket	Clear & Bright	
(b)	Gravity, Corrected to 60° F (15° C)	37° to 51° API (775-840 Kg/m³)	1° API
(c)	Distillation		
	10% Recovered	400° F (205° C)	14° F (8° C)
	20% Recovered	Report	14° F (8° C)
	50% Recovered	Report	14° F (8° C)
	90% Recovered	Report	14° F (8° C)
	Final Boiling Point	572° F (300° C)	14° F (8° C)
	Residue	1.5	Spec Limit
	Loss	1.5	Spec Limit
(d)	Flash Point	100° F (38° C)	5° F (3° C)
(e)	Freezing Point		
	Jet A	-40° F (-40° C)	5° F (3° C)
	Jet A-1	-53° F (-47° C)	5° F (3° C)
(f)	Water Reaction: Interface Rating	1b	Spec Limit
(g)	Water Separation (MSEP)	85 min	Spec Limit
(h)	Copper Corrosion Strip	No. 1 max	Spec Limit
(i)	Existent Gum	7 max	Spec Limit

NOTE:	Pipelines are considered "dedicated" only if they do not have inlet connections to any other product from the last tank or point the fuel was completely recertified as jet fuel meeting [ASTM D1655]
	specification. There is a significant increase in exposure to fuel contamination problems when airports
	receive jet fuel by "multi-product" pipelines as compared to "dedicated" pipelines. Additional care must be taken to prevent the deterioration of fuel quality.

While receiving fuel into airport storage, the facility operator is to periodically monitor pressure differential of inlet filtration, tank fill levels and check system for product leaks.

Upon completion of fuel receipt into airport storage, the facility operator should secure receiving tank(s) and facility items, i.e.: gauging, record results of sumping tanks and filters, set inlet and outlet valves for correct positioning, etc. Tank and filter sump results are to be recorded and retained for 12 months.

NOTE:	It is desirable to have one hour settling per vertical foot of product depth.
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3. Transport Truck Deliveries

Airport facility operator shall receive a certification document from jet fuel supplier or shipping agent certifying product to be delivered to airport meets [ASTM D1655] specification requirements required in [Section 2-2] with at least the following select property values listed as measured by specified ASTM test methods (Ref. [Table 2-3.1] "ASTM D1655 Property Test" for Min./Max. Spec. Limits). A single certification document can represent multiple transport truck deliveries:

(a)	Visual Appearance in White Bucket	Ref. [Section 3-1]
(b)	Gravity, Corrected to 60° F (15° C)	[ASTM D1298]
(c)	Distillation	[ASTM D86]
	10% Recovered	
	20% Recovered	
	50% Recovered	
	90% Recovered	
	Final Boiling Point	
	Residue	
	Loss	
(d)	Flash Point	[ASTM D56]
(e)	Freezing Point	[ASTM D2386]
(f)	Water Reaction Interface Rating	[ASTM D1094]
(g)	Water Separation (MSEP)	[ASTM D3948]
(h)	Copper Strip Corrosion	[ASTM D130]
(i)	Existent Gum	[ASTM D381]

Accompanying documentation shall also include all delivery information, i.e.: destination; batch number; fuel grade or type; quantity to be shipped.

NOTE:	Fuel receipts should not be rejected based on incomplete certification documentation. If the
	certification documentation cannot be delivered prior to delivery, affected airlines must be notified.

Facility operator shall prepare receiving tank(s) and facility items prior to delivery of product, i.e., gauging, sumping, correct inlet and outlet valve positioning, etc

At time of delivery and prior to connecting truck discharge hoses, transport truck driver and facility operator are to review and agree that fuel delivery documentation and procedures are in place to ensure satisfactory fuel receipt.

Truck unloading hoses and fittings are to be inspected for deficiencies, pending failures and cleanliness prior to connection to airport facility receiving connections..

Prior to testing and unloading of the transport truck, allow truck to sit for a minimum of 10 minutes with the tank internal valves open.

Facility operator shall conduct the following tests on fuel samples received from each highway transport truck tank compartment and record the results:

- (a) Visual Appearance in White Bucket
- (b) API Gravity, corrected to 60° F (15° C)

1	NOTE:	Use extreme care and vigilance when performing the Visual Appearance test. Slight traces of water,
		solids or color may indicate the presence of product mixes or other contaminants that could cause jet
		fuel to be off-specification and unacceptable for aircraft use. Any unusual results must be investigated

If visible contamination is observed in white bucket, more than one sumping may be required to clear it. If contamination remains after five (5) one gallon individual samples from one tank truck compartment, the load must be rejected. If a load is rejected, affected aircraft operators are to be notified if it is anticipated that such rejections may impact aircraft operations. A representative sample of the rejected product, including supporting documentation, shall be retained in a clean container for a minimum of 30 days for future reference. Documentation shall include copy of Bill of Lading, truck/trailer number and reason for rejection.

Fuel is unacceptable and must be rejected if API Gravity, corrected to 60° F (15° C), is not between 37° and 51° API (775-840 Kg/m3). Discontinue fuel transfer or receipt and initiate an immediate investigation to determine if there is fuel contamination or a specification problem if there is a change of more than 1° API from source as shown on shipping document.

While receiving fuel into airport storage, facility operator is to periodically monitor pressure differential of inlet filtration and check system for product leaks. Fuel receiving process must be monitored constantly by airport facility personnel.

CAUTION: IT IS NOT ACCEPTABLE TO RECEIVE AND DISPENSE FUEL FROM THE SAME TANK SIMULTANEOUSLY

Upon completion of receipt by transport truck, facility operator shall;

- Close inlet valve(s) on receipt tank(s).
- Sump each receiving filter vessel. (Sumping of inbound filter vessels, after each transport truck receipt, will assist facility operator in identifying a particular delivery of product, which may have contained unacceptable levels of water prior to off-loading, but which were undetectable during initial visual receiving checks.).

- Sump tank(s) in which product was received. (A minimum of 1 hour between the end of transport truck receiving and storage tank sumping is recommended to allow any water and/or solids stirred up during tank receiving to settle to the tank sump for removal.).
- Record the results of each sump.

NOTE 1:	It is recognized that many facilities receive fuel via simultaneous off-loading of transport trucks,
	making sumping, of inbound filter vessels, after each individual delivery, unfeasible. Therefore,
	multiple deliveries into a product receiving tank, during a single day, may be considered a single
	receipt, provided the tank does not change from receipt to delivery status during that period.

NOTE 2: It is desirable to have one hour settling per vertical foot of product depth.

Retain fuel receiving records for 12 months.

4. Railroad Tank Car Deliveries

Airport facility operator should follow the "TRANSPORT TRUCK DELIVERIES" section for guidance procedures.

5. Marine Vessel Deliveries

Airport facility operator should follow the "PIPELINE DELIVERIES" section for guidance procedures.

2-4. Fuel Storage Facility Requirements

1. General

Fuel storage facilities which will supply fuel directly into aircraft, refuelers, or hydrant systems must meet the following requirements, unless otherwise indicated. Vendors with facilities that do not meet the requirements of this section shall submit a waiver request to the affected customers under subsection 2-1.4 of this standard. Examples of Waiver/Variance may be found in [Section 5-1] and [Section 5-2].

CAUTION: ZINC GALVANIZED MATERIALS MUST NOT BE USED IN JET FUEL SERVICE. NO COPPER ALLOYS, CADMIUM PLATING OR PLASTIC MATERIALS ARE PERMITTED FOR MAIN FUEL PIPING. THE USE OF COPPER OR COPPER ALLOY MATERIALS FOR OTHER COMPONENTS MUST BE MINIMIZED.

2. Storage Tanks

Storage tanks shall include the following equipment;

- (a) Floating suction with means of verifying proper operation.
- (b) Inlet diffuser
- (c) Gauge hatch with slotted tube
- (d) Access manway (Two are preferred)
- (e) Automatic high liquid level device(s) to prevent tank overfill.
- (f) Suitable secondary containment as applicable.
- (g) A placard, adjacent to tank sump drain(s), indicating the volume of tank drain piping.

Above ground vertical tanks shall also adhere to the following, in addition to 2-4.2. (a) thru (g), above, unless otherwise indicated;

- (a) Fixed roof
- (b) Light color epoxy coated floor and sides up to the top of the first wall panel. Complete internal coating is recommended.
- (c) Cone down bottom to positive center sump with drain. Floor plates shall be arranged to ensure water run-off.
 - Slope of 1 in 20 is recommended.
- (d) Non-Metallic tanks are not acceptable.

Above ground horizontal tanks shall also adhere to the following, in addition to 2-4.2. (a) thru (g), above, unless otherwise indicated;

(a) Carbon steel tanks must have complete internal light colored epoxy coating.

- (b) Sloped bottom to positive sump with drain.
 - Slope of 1 in 20 is recommended.
- (c) Non-Metallic tanks are not acceptable.
- (d) Access manways should be equipped with an internal ladder.

Underground tanks shall also adhere to the following, in addition to 2-4.2. (a) thru (g), above, unless otherwise indicated;

- (a) Carbon steel tanks must have complete internal light colored epoxy coating.
- (b) Access manways should be equipped with an internal ladder.
- (c) Manways and other tank appurtenances must be extended above ground.
- (d) Sloped bottom to positive sump with permanent pump.
 - Slope of 1 in 20 is recommended.

3. Filters

NOTE 3:

Filter/Separators are required for receiving fuel into and dispensing fuel from storage which will supply fuel directly into aircraft, refuelers, or hydrant systems. If only one Filter/Separator is available, it must be installed to perform both fuel receiving and dispensing functions.

CAUTION:	FULL FLOW MONITORS SHOULD NOT BE USED WITH FUELS CONTAINING FUEL SYSTEM ICING INHIBITORS (FSII). THE WATER REMOVAL PERFORMANCE OF FULL FLOW MONITORS MAY BE REDUCED WITH FUEL CONTAINING FSII.
NOTE 1:	Full flow monitors meeting the requirements of [IP 1583] may be used in lieu of Filter/Separators with water defense systems.
NOTE 2:	Additional filtration, such as micronic filters, water coalescers (haypacks) or clay treaters, may be required due to local conditions. If micronic filters are used, they must meet the qualification requirements of [API/IP 1590].

All new vessels and element combinations shall meet [API/IP 1581] latest edition. Existing vessels and element conversions shall meet, by test or similarity, the latest edition of [API/IP 1581] / [API/IP 1582]. For existing vessels, conversion to the latest edition shall occur within 12 months of qualified elements becoming available for a specific vessel. If qualified by similarity, a similarity data sheet must be maintained locally and a data plate reflecting such qualification must be attached to the filter vessel.

An API monogram is not necessary to meet the requirements of this document.

Filter/Separators must be equipped with an automatic water defense system that will stop fuel flow or alert operating personnel when actuated by a high water level.

Float or electronic probe systems must include provisions for an operational test.

All filter vessels must be equipped with:

- (a) Provisions for the elimination of air
- (b) Direct reading differential pressure gauges with an accuracy of +/- 2 PSI
- (c) Manual sump drains Valves with handles spring loaded to the closed position are recommended.
- (d) Upstream and downstream sampling (Millipore) connections, including probes and dust caps or plugs.
- (e) Pressure relief valves
- (f) Placard indicating month and year of last filter change.

Use of automatic water drain valves is not recommended. Existing automatic drain valves should be removed.

4. Physically Segregated Systems

Physically segregated systems are required where more than one grade of fuel is stored to prevent accidental mixing of products.

Isolation valves and/or blind flanges are not acceptable methods of product grade separation.

Connections for receiving and dispensing different grades of fuel must be physically incompatible.

5. Emergency Fuel Shutoff System

An emergency fuel shutoff system is required

Emergency shutoff valves and switches must be clearly marked in accordance with the requirements of [NFPA 407], latest edition, and the area around them must be kept free of obstructions.

6. Fire Extinguishers With Inspection Tags

Fire extinguishers with inspection tags must be positioned in accordance with applicable safety requirements.

7. Fuel Loading and Unloading Hoses

Loading hoses and couplings must meet the following standard when purchased by the end user.

• [API 1529], Grade 2, Type C, latest edition.

Hoses shall be installed within 2 years of the date of manufacture, and have a maximum service life of 10 years from the date of manufacture.

If reusable couplings are installed on loading/unloading hoses the couplings and hose shall meet the requirements of [API 1529] and operator shall abide by the testing requirements of [API/IP 1540], latest edition.

Operators choosing to reattach couplings must undergo training from the hose or coupling manufacturer.

CAUTION:	PRIOR TO THE PERIODIC PRESSURE TESTING OF A FUEL LOADING HOSE, THE
	MAXIMUM ALLOWABLE PRESSURE RATING OF THE ATTACHED VALVES, METERS, OR
	SWIVELS SHOULD BE CHECKED TO PREVENT POSSIBLE INJURY TO THE OPERATOR OR
	DAMAGE TO THE EQUIPMENT. IT MAY BE NECESSARY TO REMOVE THE FUELING HOSE
	PRIOR TO TESTING. FOLLOWING THE SAFETY PRECAUTIONS OUTLINED IN [API/IP 1540]
	IS HIGHLY RECOMMENDED.

Fuel "unloading" hoses shall be compatible with jet fuel and suitable for local conditions. Dust covers or other protective devices must be used to keep out dirt and water.

8. "No Smoking", "Flammable", and Product Identification Signs

"NO SMOKING", "FLAMMABLE", and product identification signs must be prominently displayed.

9. Facility Identification and Color Coding

Fuel storage facilities must be properly identified and color coded in accordance with the standards of [API/IP 1542], latest edition.

10. Metal Underground Tanks and Piping

Metal underground tanks and piping must be cathodically protected.

11. Relaxation Chambers

Relaxation chambers, where installed, shall be equipped with the following:

- (a) Air eliminator
- (b) Protection by pressure relief valve
- (c) Manual sump drain Valves with handles spring loaded to the closed position are recommended.

12. Bulk Air Eliminator

Bulk air eliminators, where installed, shall be equipped with the following:

- (a) Protection by pressure relief valve
- (b) Manual sump drain

13. Refueling Truck Loading Station

A handheld deadman control device is required for all truck loading operations, per NFPA 407. Bottom loading control systems do not negate the need to bond with a separate bonding cable.

Loading station must be equipped with pressure control provisions if necessary to prevent damage to the refueling truck during high level shutdown of truck mounted valves.

All stations must be equipped with static bonding capability.

Bottom-loading nozzles and couplers must be equipped with 60 mesh or finer screens. These screens are not required if an upstream strainer is installed and no hoses are used between the strainer and the nozzle/coupler, i.e., "swing arm" applications.

Fire extinguishers with inspection tags must be positioned in accordance with applicable safety requirements.

Hoses and hose couplings must meet the following standard when purchased by the end user:

• [API 1529], Grade 2, Type C, latest edition.

Hoses shall be installed within 2 years of the date of manufacture, and have a maximum service life of 10 years from the date of manufacture.

Dust covers or other protective devices must be used to prevent debris from accumulating on mating surfaces of fuel loading hose couplers.

14. Product Reclamation Storage Tank

A Reclamation storage tank, where installed, shall include the following equipment:

- Gauge hatch with slotted tube
- Means for access or visual inspection
- Means to prevent overfill
- Suitable secondary containment as applicable
- A placard, adjacent to tank sump drain(s), indicating the volume of tank drain piping.

Tank must be identified as "Jet A Recovery Tank Intended for Aviation Use".

Tank must have sloped bottom to a positive sump with a drain or permanent pump. Sump must be located in the lowest point of the tank. Drain or pump pick-up must be at the lowest point of the sump.

• Slope of 1 in 20 is recommended

All tank appurtenances, access entrances, vents, inspection ports, etc., must be extended above ground, and designed to prevent the ingress of contaminants.

Reclaimed product, except from sump separators, must be filtered prior to returning to off-line storage tank. Filtration must meet the requirements of [Heading 2-4.3] Filters.

2-5. Fuel Facility Checks

1. General

The following checks must be performed on all fuel storage facilities servicing aircraft and at the frequencies specified. Additional tasks or more frequent checks may be required based on local conditions.

Daily checks and inspections should be made at the beginning of each work day including weekends and holidays.

Maintenance requirements specified in this section are generally limited to those items pertaining to fuel quality and safety. Additional programs should be established to ensure mechanical reliability of all facility equipment. Any facility equipment not in daily use must have all daily, monthly, quarterly and annual checks current and recorded before the equipment is returned to service.

2. Facility Check Records

Records, paper or electronic, must be completed by the person performing the tasks, or by the person accepting responsibility for performance of the tasks.

Use of Forms 103.01, A through D, is recommended, but not required. No variance authority is needed to use other forms if they meet or exceed the task and frequency requirements specified in this section. Additional copies of Forms 103.01, A through D, may be reproduced locally (See [Section 6-2]).

The legible signature, initials or employee identification number of the person performing the tasks or the person accepting responsibility for the performance of the tasks is required.

- (a) If initials or employee identification numbers are used, a record of each person's name and initials/identification number must be maintained and available for review.
- (b) Supporting documentation, completed by the person actually performing the tasks and containing their signature, initials or identification number must be available if another person has accepted responsibility for accomplishment of the tasks.

Records must indicate when fueling equipment is not used. Retain records in local files as follows:

Daily, Quarterly, Semi-Annual, and Annual check records – 12 months

Filter Change records – 36 months

Tank Inspection and Cleaning records - Indefinitely

Upon completion of the checks, record results using the following condition codes:

S = Indicates Satisfactory

C = Indicates Comment. Comment required in remarks section. Corrective action must be documented and dated.

N/U = Indicates unit Not Used

N/A = Indicates Task Not Applicable

Sump samples are to be rated according to [Section 3-1].

3. Daily Checks

3.1. General Condition of Tank Yard

- (a) Check the general condition of the yard area for appearance and cleanliness. Report and correct any condition that needs immediate attention, i.e., plugged drainage, weeds, poor housekeeping, etc.
- (b) Evidence of any recent fuel spill, including, but not limited to, staining, strong fuel odors or the presence of fuel in catchment basins, overflow tanks, oil/water separators, or sumps, must be investigated immediately.

3.2. Security, Fire & Safety Deficiencies

- (a) Check tank yard and fuel handling facilities for any security, fire or safety deficiencies or unusual conditions requiring immediate corrective actions.
- (b) Fuel Leaks
 - Check tanks, piping, valves, hoses, meters, filters, and other fuel handling equipment for fuel leaks.
 - Any visible leaks must be immediately reported and repaired.
- (c) Ensure that all gates and access doors are kept locked when area is unattended.
- (d) All broken fences and gates are to be repaired or replaced immediately.
- (e) In unsecured areas, all tank openings, valves, sump drains, fill caps, monitoring ports, loading/unloading hoses, master electrical switches and other accessible fittings must be kept closed and locked at all times when not in use.

3.3. Storage Tank and Product Reclamation Tank Sumps

- (a) Drain fuel, at maximum practical flow, into suitable container (Ref. [Section 3-1]). Sample quantity shall be of sufficient size to ensure displacement of sampling line volume.
- (b) Perform fuel appearance test of sample.
- (c) Record findings of first sample taken, after displacement of sampling line volume, according to [Section 3-1].
- (d) Continue draining until clean, dry fuel is obtained.
- (e) Remove tank from service if unable to obtain clean, dry fuel. Report unusual contamination to aircraft operators if it is anticipated that such contamination may impact aircraft operations.

3.4. Filter Sumps

- (a) Drain fuel, at maximum practical flow, into suitable container (Ref. [Section 3-1]). Vessel must be pressurized, but fuel does not have to be flowing through vessel when sample is taken.
- (b) Perform fuel appearance test of sample.
- (c) Record findings of first sample taken, after displacement of sampling line volume, according to [Section 3-1].
- (d) Remove filter vessel from service if unable to obtain clean, dry fuel. Report unusual contamination to aircraft operators if it is anticipated that such contamination may impact aircraft operations.

3.5. Filter Differential Pressure

Under normal flow conditions, check and record differential pressure across all working filters (Ref. [Section 3-9]).

3.6. Hoses, Swivels, Nozzles & Couplers

- (a) Check condition of all fuel hoses, swivels, nozzles and couplers for wear, damage and leakage.
- (b) Ensure dust covers or other protective devices are available, installed and in good repair.
 - 1) Check hoses for abrasions, cuts, soft spots, carcass separation, worn covers, blisters, exposed reinforcement, cracks, twists and sharp bends that give the appearance of pending failure.
 - 2) Check tightness of all swivel attachment screws and hose couplings.
 - 3) Check condition of nose and poppet seals on nozzles/couplers for cuts, nicks and wear.
- (c) Any item that is defective or is leaking must be replaced or repaired immediately.

3.7. Static Reels, Cables & Clamps

- (a) Check condition of static reels, cables and clamps.
- (b) Any defect that affects continuity <u>must</u> be corrected prior to use.

NOTE: Continuity must be checked after maintenance to static bonding systems.

3.8. Fire Extinguishers

- (a) Verify that fire extinguishers:
 - 1) Are located in their designated place
 - 2) Have unobstructed access and visibility
 - 3) Are tagged to indicate monthly inspections are current
 - 4) Have unbroken safety seals or tamper indicators
 - 5) Have no obvious physical damage, corrosion or leakage
 - 6) When so equipped, the pressure gauge reading or indicator is in the operable range or position
- (b) If any fire extinguisher is missing or does not meet the criteria listed above, it shall be repaired, or removed from service and replaced with a serviceable extinguisher of the same or greater capacity.

4. Monthly Checks

4.1. Filtration & Free Water Test (Millipore)

Perform a membrane color/particle (Millipore) and free water test downstream of all filter/separator vessels (Ref. [Section 3-2] and [Section 3-3]). Record results and attach test membrane to Form 103.08 or equal.

4.2. Bonding Cable Continuity

Perform electrical continuity check on bonding cables and clamps (Ref. [Section 3-10]).

- Resistance must be 25 ohms or less.
- Defective equipment must be repaired prior to fuel transfer.

4.3. Nozzle Screens

- (a) Remove nozzles and examine screens for particles or damage.
 - If particles are found, investigate sources of contamination which could be from inner hose lining, pipe rust, sand, low point sediment, equipment failure, seals, gaskets, etc.
- (b) Screens are to be cleaned if contaminated or replaced if damaged.

4.4. Signs, Labels & Placards

Verify that fueling equipment is clearly marked with the proper type of fuel being dispensed, flammable, no smoking, emergency shutoff and other appropriate information and instructions, signs or labels as required.

4.5. Floating Suctions

Verify satisfactory operation of all tank floating suctions.

4.6. Fire Extinguishers

- (a) Check each fire extinguisher for inspection tag and seal.
- (b) Maintain extinguishers in accordance with the applicable [NFPA 10] guidelines.
- (c) Upon completion of the inspection, update inspection tag.

5. Quarterly Checks

5.1. Emergency Shutoff System

- (a) Operationally check the emergency shutoff system.
 - 1) Coordinate shutoff test with all persons, agents, airlines, fuel suppliers, and other groups having interest in the operation of the system.
 - 2) Each control device must be tested at least once a year.
- (b) Immediately report any operational discrepancies.

5.2. Water Defense System - External Checks

Check operation of water defense systems in accordance with quarterly requirements of [Section 3-12].

5.3. Tank High Level Controls

- (a) Check satisfactory operation of tank high level sensing devices and automatic fuel flow shutoff valves where installed.
- (b) Inoperative controls should be adjusted or repaired immediately or have alternate operating procedures in effect that will provide positive spill prevention while tank is in service.

5.4. Product Reclamation Tank Interior Inspection

On a quarterly basis product reclamation tanks must be visually inspected for cleanliness or pass a microbiological growth test, as recommended, by the affected airlines. Clean as required (Ref. [Section 3-11]).

6. Semi-Annual Check

6.1. Hose Pressure Checks

(a) Loading/unloading hoses fitted with reusable couplings, and being operated under system pressure, must undergo the six-month pressure testing at 225 PSI, per the requirements found in [API/IP 1540].

NOTE:	Pressure testing to 300 PSI is required whenever a new hose attachment or coupling is fitted
CAUTION:	RECOUPLING AND PRESSURE TESTING OF HOSES AND FITTINGS SHOULD ONLY BE
	CARRIED OUT BY PERSONS ADEQUATELY TRAINED IN THEIR PROPER FITTINGS AND
	TESTING, AND APPROVED BY THE OEM.

7. Annual Checks

7.1. Storage Tank Interiors

- (a) Check fuel storage tank interiors for cleanliness and condition of coating.
- (b) Clean as required (Ref. [Section 3-11]).

7.2. Filter Differential Pressure Gauges

Verify proper operation of filter differential pressure gauge(s) in accordance with gauge manufacturers' procedures. Accuracy must be within +/- 2 PSI. Repair or replace as required. (Ref. [Section 3-9])

7.3. Filter Elements

- (a) Change filter elements per [Section 3-13].
- (b) Replace filter elements per criteria found in [Section 3-14].
- (c) All filter vessels must be opened annually to visually check condition of interior for cleanliness, and integrity of elements.

7.4. Filter/Separator Heaters

Where installed, check filter/separator sump and drain line heaters for proper operation per manufacturer specifications before freezing weather.

7.5. Tank Vents

- (a) Where installed, check cleanliness of tank vent screens.
 - Clean, repair or replace vent screens as required.
- (b) Tanks that have pressure/vacuum vents, check satisfactory operation and condition of poppets and inlet screens.
 - Under freezing conditions, additional checks may be required to assure free movement of poppets.

7.6. Cathodic Protection

Where installed, confirm satisfactory operation of cathodic protection systems. This requirement is generally contracted to businesses specializing in this type of service. State or local regulations may require greater frequency of inspection.

7.7. Line Strainers

- (a) If installed, check line strainers for cleanliness and damage.
- (b) Clean or replace screens as required.
- (c) Local conditions may require more frequent check of some strainers, such as those used for truck unloading.

7.8. Water Defense System Inspection & Test

Check operation of water defense systems in accordance with annual requirements of [Section 3-12]

2-6. Hydrant System Checks

1. General

The following checks must be performed on all hydrant fueling systems servicing aircraft and at the frequencies specified. Additional tasks or more frequent checks may be required based on local conditions.

Daily checks and inspections should be made at the beginning of each work day including weekends and holidays.

All personnel engaged in ramp operations must be continuously observant of abnormal conditions that may exist in and around fuel pits. Any fuel leaks, fire/safety hazards, or adverse conditions must be reported immediately.

Aircraft operators shall be notified by contracted fueling agent of any modifications, changes, or construction work to hydrant systems. Hydrant systems must be flushed per [Section 3-15].

Hydrant systems or segments of hydrant systems not in daily use must have all daily, monthly, semi-annual and annual checks current and recorded before the system, or segment is returned to service. Based on fuel test results, flushing may be required by aircraft operator prior to use. Records must indicate when systems are out of service.

2. Hydrant System Check Records

Records, paper or electronic, must be completed by the person performing the tasks, or by the person accepting responsibility for performance of the tasks.

Use of Forms 103.05, A through C, is recommended, but not required. No variance authority is needed to use other forms if they meet or exceed the task and frequency requirements specified in this section. Additional copies of Form 103.05 may be reproduced locally (Ref. [Section 6-2]).

The legible signature, initials or employee identification number of the person performing the tasks or the person accepting responsibility for the performance of the tasks is required.

- (a) If initials or employee identification numbers are used, a record of each person's name and initials/identification number must be maintained and available for review.
- (b) Supporting documentation with the signature, initials or identification number of the person actually performing the tasks must be available if another person has signed the form accepting responsibility for accomplishment of the tasks.

Records must indicate when fueling equipment is not used.

Retain records, including supporting documentation, in local files for 12 months.

Upon completion of the checks, record results using the following condition codes:

S = Indicates Satisfactory

C = Indicates Comment. Comment required in remarks section. Corrective action must be documented and dated.

N/U = Indicates unit Not Used

N/A = Indicates Task Not Applicable

Sump samples are to be rated according to [Section 3-1].

3. Daily Checks

3.1. Hydrant Pit

- (a) Visually check hydrant pits and all components for deficiencies.
- (b) Pits should be clean & free of standing liquid.
- (c) Correct deficiencies in a timely manner.

3.2. Emergency Fuel Shutoff (EFS) Stations

- (a) Verify that all emergency fuel shutoff stations on the ramp have:
 - 1) Clear access
 - 2) A sign or placard identifying EFS location.
- (b) Any deficiencies are to be corrected immediately.

4. Monthly Checks

4.1. Isolation Valve Pits and Control Vaults

Check isolation valve pits for:

- emergency access
- fuel leaks
- standing liquid and debris
- general condition of all components.

Verify proper operation of valves.

Correct any deficiencies found.

4.2. Hydrant Valve Assembly

- (a) Check the general condition of the hydrant pit valve and sense line connectors.
- (b) Verify the satisfactory operation of the hydrant pit valve.
- (c) Check for leaks, excessive coupler mating flange wear and loose or missing fasteners.
- (d) Promptly correct any deficiencies.

4.3. Low Point Drains

- (a) Open all low point drains until all water and/or sediment is removed.
- (b) Flush a minimum of two (2) gallons at each low point until clear fuel is obtained to ensure positive removal of all contaminants.
- (c) Replace missing tags or markings to pit lids or low point drain valves as required to ensure proper identification.

4.4. Emergency Fuel Shutoff

- (a) Verify the satisfactory operation of the emergency shutoff system, by actuating one or more of the control devices for each zone.
- (b) Coordinate the shutoff test with all persons, fueling agents, fuel suppliers, and any other group having an interest in the operation of the facility.
- (c) Each control device must be tested at least once a year.
- (d) Immediately repair any discrepancies.

NOTE: If for any reason the emergency shutoff system cannot be repaired immediately, system operator must have an approved alternate plan in effect for continuing system use until discrepancies are corrected.

5. Quarterly Checks

5.1. High Point Vents

- (a) Bleed all high point vents to ensure the removal of all entrapped air.
- (b) Continue to bleed air until clear fuel is present.
- (c) It is necessary to bleed high point vents more frequently if pipeline was drained or modified, allowing air entry into system.
- (d) Replace missing tags or markings to pit lids or high point vent valves as required to ensure proper identification.

5.2. Surge Absorbers

- (a) Where installed, check the general condition and operating pressure setting of each unit.
- (b) Recharge as required.

5.3. Leak Detection and Piping Isolation Systems

- (a) Where installed, check the satisfactory operation of pipeline leak detection systems and pipeline monitoring wells.
- (b) Monitoring devices and fuel flow shutoff valves are to be tested.
- (c) Immediately report and repair any deficiencies.

NOTE: In critical areas, i.e., baggage rooms, basements, etc., this task may be more frequent based on local needs and exposure.

6. Annual Checks

6.1. Cathodic Protection

Where installed, confirm satisfactory operation of cathodic protection systems. This requirement is generally contracted to businesses specializing in this type of service. State or local regulations may require greater frequency of inspection.

2-7. Aircraft Fueling Equipment Requirements

CAUTION:	ZINC GALVANIZED MATERIALS MUST NOT BE USED IN JET FUEL SERVICE. NO COPPER
	ALLOYS, CADMIUM PLATING OR PLASTIC MATERIALS ARE PERMITTED FOR MAIN
	FUEL PIPING. THE USE OF COPPER OR COPPER ALLOY MATERIALS FOR OTHER
	COMPONENTS MUST BE MINIMIZED.

1. General

All aircraft fueling equipment, including refueling trucks, hydrant vehicles, hydrant carts and fueling cabinets, <u>must</u> comply with the requirements in this section.

Fueling equipment shall be free of leaks.

Tires, wheels, wheel studs/nuts and axle studs/nuts must be maintained in good condition at all times.

Electrical equipment, including lights, light lenses and wiring, must be maintained in a safe and operational condition.

Windows must be clean and free of cracks and crazing.

2. Filter/Separator or Full Flow Fuel Monitor

All aircraft fueling equipment must have a Filter/Separator or a Full Flow Fuel Monitor.

- (a) All new vessels and element combinations shall meet [API/IP 1581] latest edition. Existing vessels and element conversions shall meet, by test or similarity, the latest edition of [API/IP 1581] / [API/IP 1582]. For existing vessels, conversion to the latest edition shall occur within 12 months of qualified elements becoming available for a specific vessel. If qualified by similarity, a similarity data sheet must be maintained locally and a data plate reflecting such qualification must be attached to the filter vessel.
 - 1) Filter/Separators must be equipped with an automatic water defense system that will cause fueling to stop when activated by excessive water.
 - a) Water defense systems must include provisions for an operational test.
- (b) Full flow fuel monitors must meet the requirements of [IP 1583].
 - 1) Full flow fuel monitors must be equipped with a pressure limiting device that will prevent excessive differential pressure from damaging elements in the event of complete blockage.

CAUTION:	FULL FLOW MONITORS SHOULD NOT BE USED WITH FUELS CONTAINING FUEL
	SYSTEM ICING INHIBITORS (FSII). THE WATER REMOVAL PERFORMANCE OF FULL
	FLOW MONITORS MAY BE REDUCED WITH FUEL CONTAINING FSII.

- (c) All filtration vessels must include:
 - 1) Air elimination provisions.
 - 2) Direct reading differential pressure gauges with an accuracy of +/- 2 PSI.

- 3) Manual sump drains Valves with handles spring loaded to the closed position are recommended.
- 4) Upstream and downstream membrane sampling connections, including probes and dust covers.
- 5) Pressure relief valve or other device that will prevent over-pressurization due to thermal expansion of fuel. Include a means for accommodating relieved fuel.

NOTE: An API monogram is not necessary to meet the requirements of this document.

3. Pressure Controls

All aircraft fueling equipment must have separate and independent primary and secondary pressure control devices.

- (a) Primary pressure control is intended to protect the aircraft under conditions of constant flow and also from pressure surge caused during aircraft valve closure.
- (b) Secondary pressure control is intended to protect the aircraft in the event of primary pressure control failure.

CAUTION: FUELING PRESSURE CONTROL SYSTEMS SHALL NEVER ALLOW THE ACTUAL FUEL PRESSURE, MEASURED AT THE FUEL NOZZLE, TO EXCEED THE PRESSURE INDICATED BY THE OPERATOR'S GAUGE.

- (c) Fuel pressure control systems may utilize the following:
 - 1) Pressure controlling hydrant pit valves.
 - 2) Pressure controlling hydrant pit couplers.
 - 3) In-line pressure control valves.
 - 4) Hose End Pressure Control Valves (HEPCV).
 - 5) Pressure switches that will cause rapid shutoff of fuel flow in the event of high fueling pressure.
- (d) Primary pressure control devices must limit fueling pressure, at the fuel nozzle, to 40 psig or less under conditions of constant flow.
- (e) Secondary pressure control devices must limit fueling pressure, at the fuel nozzle, to 50 psig or less under conditions of constant flow.

4. Deadman Control System

All aircraft fueling equipment must have a handheld deadman control device. The deadman control system must completely stop fuel flow within 5 percent of the fuel flow rate at the time of release.

EXAMPLE: If actual fuel flow rate at the time of deadman control release is 500 gpm, total overrun must not exceed 25 gallons.

5. Emergency Fuel Shutoff System

Hydrant vehicles, hydrant carts and fueling cabinets must be equipped with an emergency fuel shutoff system in addition to a deadman control.

- (a) Each unit must have an emergency fuel shutoff control accessible from the ground.
- (b) Units equipped with a lift or platform must have an emergency fuel shutoff control accessible from the lift or platform, in addition to one accessible from the ground.
- (c) The system should stop the fuel flow by automatically closing the hydrant pit valve upon activation.

Refueling trucks must be equipped with an emergency fuel shutoff control accessible from each side of the truck.

- (a) Units equipped with a lift or platform must have an emergency fuel shutoff control accessible from the lift or platform, in addition to one accessible from the ground.
- (b) The emergency fuel shutoff system should also close the tank outlet valve(s).

Each emergency fuel shutoff control must completely stop fuel flow within a maximum of 5 percent overrun.

EXAMPLE: If actual fuel flow rate at the time of emergency fuel shutoff activation is 500 gpm, total overrun must not exceed 25 gallons.

6. Fire Extinguishers

Hydrant vehicles, hydrant carts and fueling cabinets must be equipped with a minimum of one 20-B:C rated fire extinguisher, securely mounted and readily accessible.

Refueling trucks must be equipped with a minimum of two 20- B:C rated fire extinguishers, securely mounted on opposite sides of the truck and readily accessible.

Seals must be intact.

Current inspection, testing and recharging records must be attached.

7. Safety Interlock System

All mobile fueling equipment must have a safety interlock system which will prevent the equipment from being moved when;

- (a) Couplers and/or nozzles are not in their stowed position.
- (b) The pumping system is activated on tank trucks.
- (c) Lift platforms are in the extended position.

The interlock system may stop the engine on motorized equipment, but should also apply the vehicle brakes.

Refueling trucks with bottom loading provisions shall incorporate a brake interlock system that will prevent the vehicle from being moved until the bottom loading coupler has been disconnected from the vehicle.

Interlock systems shall be equipped with an override device i.e., push-button, spring loaded toggle switch, lever device, etc. Regardless of the type and location, it shall be secured in the normal position, with a breakaway seal. Placards must identify normal and override positions. A light, indicating override activation is recommended and should be prominently located in the vehicle cab.

8. Aircraft Fueling Hoses

Hoses and couplings must meet the following standard when purchased by the end user:

• [API 1529], Grade 2, Type C, latest edition.

Hoses shall be installed within 2 years of the date of manufacture, and have a maximum service life of 10 years from the date of manufacture.

If reusable couplings are installed on hoses, the couplings and hose shall meet the requirements of [API 1529] and operator shall abide by the periodic pressure testing requirements of [API/IP 1540] latest edition..

Operators choosing to reattach couplings must undergo training from the hose or coupling manufacturer.

CAUTION:	PRIOR TO THE PERIODIC PRESSURE TESTING OF AN AIRCRAFT FUELING HOSE, THE
	MAXIMUM ALLOWABLE PRESSURE RATING OF THE ATTACHED VALVES, METERS, OR
	SWIVELS SHOULD BE CHECKED TO PREVENT POSSIBLE INJURY TO THE OPERATOR OR
	DAMAGE TO THE EQUIPMENT. IT MAY BE NECESSARY TO REMOVE THE FUELING HOSE
	PRIOR TO TESTING. FOLLOWING THE SAFETY PRECAUTIONS OUTLINED IN [API/IP 1540]
	IS HIGHLY RECOMMENDED.

NOTE: An API monogram is not necessary to meet the requirements of this document.

9. Manual Isolation Valves

Equipment with multiple aircraft delivery hoses must have a manual isolation valve installed upstream of each delivery hose.

10. Dust Covers

Dust covers or other protective devices must be used to prevent debris from accumulating on mating surfaces of hydrant couplers and aircraft fueling nozzles.

11. Strainers and Swivels

Aircraft fueling nozzles must be equipped with 100 mesh strainers.

Hydrant coupler and aircraft fueling nozzle swivel retention devices must be equipped with at least two levels of redundancy, such as collar lock rings and collar retention screws secured by safety wire.

12. Aircraft Fuel Pressure Gauges

A pressure gauge is required for monitoring aircraft fueling pressures.

Gauges should be located where they will be visible to the equipment operator during aircraft fueling operations.

Gauges shall have a minimum face diameter of 4 inches and must have an accuracy of +/-2% of full scale.

Digital pressure displays shall have a minimum character height of 3/4 inch.

13. Fuel Quantity Measurement Meter

Meters must be capable of maintaining accuracy of 1/10 of one percent (0.1%) and repeatability of 1/20 of one percent (0.05%) at flow rates ranging from 100 gpm to the maximum rated flow of the fueling equipment.

Calibrator/adjustor must be sealed.

14. Electrostatic Bonding System

Electrostatic bonding system must have less than 25 ohms total resistance.

15. Signs, Placards & Labels

The following signs, placards or labels must be placed on the equipment as indicated:

- (a) Product identification on each side and rear
- (b) FLAMMABLE on each side and rear
- (c) NO SMOKING posted prominently in cab of vehicles
- (d) NO SMOKING on at least two sides
- (e) EMERGENCY FUEL SHUTOFF placard adjacent to each emergency fuel shutoff control. Placards must also indicate method of operation (e.g., Push, Pull, Turn, etc.).
- (f) Fire extinguishers located in enclosed compartments shall have their location clearly marked.
- (g) Aircraft fueling pressure and filter differential pressure gauges shall be identified.
- (h) Filter and tank drain valves shall be identified.
- (i) A placard indicating the date (Month and Year) during which the filter elements were last changed shall be placed on the filter housing.
- (j) A placard indicating the date (Month and Year) of the last satisfactory single element test, if applicable, shall be placed on the filter housing.
- (k) A sign or placard indicating proper procedure for engaging the pumping system should be prominently displayed adjacent to pump controls.

16. Additional Requirements for Refueling Trucks

Cargo Tanks must be constructed of stainless steel, aluminum or internally light color epoxy coated carbon steel.

Dome covers must be provided with:

- (a) A forward mounted hinge and latches which will automatically cause the cover to close and latch with forward motion of the vehicle.
- (b) Water-tight, fuel resistant seals and gaskets.

Each tank compartment must be equipped with a water drain located at the lowest point.

• Valves with handles spring loaded to the closed position are recommended.

Tank outlets should be equipped with shutoff valves located inside the tank shell.

Refueling trucks with bottom loading capability must be equipped with a high level shutoff system. The system may activate a shutoff device that is mounted on-board the truck or on the loading station. Provisions for ensuring the satisfactory operation of the system (known as a "Pre-Check") shall be included. The pre-check system should simulate a high level condition in the truck by submerging the sensing device in fuel.

Recirculation connections are recommended. If equipped, recirculation systems shall be arranged so that all fuel is recirculated into product tank, i.e., no fuel is to be recirculated into the pump suction.

(c) The refueler must have a means of ensuring that the internal valve is closed except when bottom loading or fueling.

2-8. Aircraft Fueling Equipment Checks

1. General

The following periodic checks must be performed by qualified individuals, at the specified frequencies, on all aircraft fueling equipment, including fueling cabinets. Additional or more frequent checks may be required due to local conditions.

Maintenance requirements specified in this section are generally limited to those items required for maintaining fuel quality and safety. Additional programs should be established to ensure mechanical reliability of all equipment servicing aircraft.

Daily checks should be made at the beginning of each day, including weekends and holidays, but must be made prior to, or during the first aircraft servicing of the day.

Any fueling equipment not in daily use must have all daily, monthly, quarterly and annual checks current and recorded before the equipment is returned to service.

2. Aircraft Fueling Equipment Check Records

Records, paper or electronic, must be completed by the person performing the tasks, or by the person accepting responsibility for performance of the tasks.

Use of Forms 103.04, A through C, is recommended, but not required. No variance authority is needed to use other forms if they meet or exceed the task and frequency requirements specified in this section. Additional copies of Forms 103.04, A through C, may be reproduced locally (Ref. [Section 6-2]).

The legible signature, initials or employee identification number of the person performing the tasks or the person accepting responsibility for the performance of the tasks is required.

- (a) If initials or employee identification numbers are used, a record of each person's name and initials/identification number must be maintained and available for review.
- (b) Supporting documentation, completed by the person actually performing the tasks and containing their signature, initials or identification number must be available if another person has accepted responsibility for accomplishment of the tasks.

Records must indicate when fueling equipment is not used.

2.1 Records Retention

Retain records in local files as follows:

- Daily, Quarterly, Semi-Annual, and Annual check records 12 months
- Filter Change records 36 months

Upon completion of the checks, record results using the following condition codes:

S = Indicates Satisfactory

C = Indicates Comment. Comment required in remarks section. Corrective action must be documented and dated.

N/U = Indicates unit Not Used

N/A = Indicates Task Not Applicable

Sump samples are to be rated according to [Section 3-1].

3. Daily Checks

3.1. General Condition

- (a) Check the general condition of the fueling vehicle for safety defects, fuel leaks, damage and proper appearance.
- (b) Take appropriate corrective action for noted defects.
- (c) Units with fuel leaks are not to be used to service aircraft.

3.2. Filter Sumps

- (a) Filter vessel must be under pressure, but fuel does not have to be flowing through the vessel when the sample is taken.
- (b) Drain approximately one gallon of fuel into a suitable container. Fuel flowing from sump drain valve should be at maximum practicable flow to ensure adequate flushing occurs.
- (c) Perform fuel appearance test of filter sumps according to [Section 3-1].
- (d) Record findings of <u>first</u> sample taken according to [Section 3-1].
- (e) Continue to sample until clean, dry fuel is obtained.
- (f) Remove unit from service if unable to obtain clean, dry sample after three samples have been drained. Report unusual contamination to aircraft operators if it is anticipated that such contamination may impact aircraft operations.

3.3. Filter Differential Pressure

- (a) Observe and record differential pressure with fuel flowing through filter under normal maximum flow conditions (Ref. [Section 3-9]).
- (b) Filter differential pressure must be periodically monitored during fueling operation. Remove unit from service if a sudden drop in differential pressure from previous readings is observed or if differential pressure exceeds 15 psi on filter/separators or 25 psi on full flow monitors.

3.4. Deadman Controls

- (a) Perform a functionality check of the deadman control system.
- (b) Remove vehicle from service if deadman control does not function properly.

3.5. Safety Interlocks

- (a) Verify proper operation of safety interlock system.
 - 1) Remove one nozzle from its storage position and attempt to move unit. Unit should not move.

NOTE: Some refueling trucks may move slightly under heavy engine acceleration due to high gear reduction drive trains. Movement should be minimal and must stop immediately upon returning engine to idle.

- 2) Repeat task for each additional nozzle, coupler, lift platform and bottom loading interlock, as applicable.
- 3) Defective interlock systems should be repaired immediately.

3.6. Nozzle Fueling Pressure

- (a) Check and record nozzle delivery fueling pressure.
- (b) Nozzle pressure must be periodically monitored during fueling operation.
- (c) Nozzle Pressure should not exceed 40 psig under conditions of constant flow.
 - 1) Pressure in excess of 40 psig, but less than 50 psig, indicates an out of adjustment or malfunctioning primary pressure control. Investigate and correct as necessary.
 - 2) Pressure fluctuations greater than +/- 10 psi, under conditions of constant flow, may indicate a malfunctioning pressure control system, and should be investigated.
- (d) Immediately remove unit from service if pressure exceeds 50 psig.

3.7. Hoses, Nozzles & Swivels

- (a) Check condition of all fuel hoses, swivels, nozzles and couplers for damage, leakage or excessive wear.
- (b) Ensure dust covers or other protective devices are available, installed and in good repair.
 - 1) Check hoses for abrasions, cuts, soft spots, carcass separation, worn covers, blisters, exposed reinforcement, cracks, twists and sharp bends that give the appearance of pending failure.
 - 2) Check the tightness and safety wiring of all swivel and collar attachment screws and hose couplings.
 - 3) Check condition of nose and poppet seals on nozzles for cuts, nicks and wear.
- (c) Any item which is defective or leaking must be repaired or replaced before being used to service aircraft.

3.8. Static Reels, Cables & Clamps

- (a) Check the condition of static bonding reels, cables, clamps and connections.
- (b) Any defect that affects continuity must be corrected prior to use.

NOTE: Continuity must be checked after maintenance to static bonding systems.

3.9. Lift Platforms

- (a) Check the general condition and verify proper operation of lift platforms.
- (b) Remove unit from service if deficiencies are noted.

3.10. Fire Extinguishers

- (a) Verify that fire extinguishers:
 - 1) Are located in their designated place
 - 2) Are tagged to indicate monthly inspections are current
 - 3) Have unbroken safety seals or tamper indicators
 - 4) Have no obvious physical damage, corrosion or leakage
 - 5) When so equipped, the pressure gauge reading or indicator is in the operable range or position
- (b) If any fire extinguisher is missing or does not meet the criteria listed above, it shall be repaired or removed from service and replaced with a serviceable extinguisher of the same or greater capacity.

3.11. Surge/Waste Tanks

Check and drain, if applicable, atmospheric surge tanks, thermal relief tanks or waste fuel tanks.

3.12. Air Tanks

Drain all moisture from air tanks to prevent damage to air system components and freezing during cold weather.

3.13. Refueling Truck Troughs

- (a) Check truck troughs for water.
- (b) If standing water is present, clean troughs and drains.

CAUTION: IF STANDING WATER IS FOUND IN TRUCK TROUGHS, EXTRA CARE MUST BE USED IN INSPECTING TANK COMPARTMENTS AND FILTER DRAINS FOR WATER.

3.14. Refueling Truck Sumps

- (a) Drain minimum of one gallon of fuel at high flow rate into a suitable container.
- (b) Perform fuel appearance test on a fuel sample from <u>each</u> tank compartment (Ref. [Section 3-1]).
- (c) Record findings of <u>first</u> sample taken according to [Section 3-1].
- (d) Continue to sample until clean, dry fuel is obtained.
- (e) Additional checks are required during and immediately after inclement weather.
- (f) Remove unit from service if unable to obtain clean, dry fuel after three samples have been drained. Report

unusual contamination to aircraft operators if it is anticipated that such contamination may impact aircraft operations.

3.15. Refueling Truck Bottom Loading Pre-check

- (a) Verify proper operation of high level shutoff systems on refueling trucks, which are bottom loaded by operating pre-check controls during filling.
- (b) Trucks should not be bottom loaded with an inoperative high level shutoff system unless alternate procedures are followed.

4. Monthly Checks

4.1. Filtration Test & Free Water Test

Perform a membrane color/particle (Millipore) and free water test downstream of each filter/separator and monitor vessel. (Ref. [Section 3-2] and [Section 3-3])

NOTE: Use of bottom loading connections on tank trucks for recirculation must be avoided in order to prevent erroneous test results.

4.2. Static System Continuity Test

- (a) Perform electrical continuity check of static bonding system.
- (b) Resistance must be 25 ohms or less.
- (c) Defective equipment must be repaired prior to servicing aircraft.

4.3. Nozzle Screens

- (a) Examine <u>each</u> nozzle screen for particles or other solid contaminants.
 - If particles are found, investigate possible sources of contamination (inner hose lining, pipe rust, sand, seals, gaskets, equipment failure, etc.) and take appropriate corrective action.
- (b) Clean screens as necessary
- (c) Verify that screens are 100 mesh.
- (d) Damaged screens are to be replaced.

4.4. Fuel Hoses

- (a) Lay hoses out full-length with system at full operating pressure and check hoses for abrasions, cuts, soft spots, carcass separation, worn covers, blisters, exposed reinforcement, cracks, twists and sharp bends that give the appearance of pending failure.
- (b) Check couplings at both ends for cracks and signs of slippage or leakage.
- (c) Replace any defective hoses prior to further servicing of aircraft.

4.5. Signs, Labels & Placards

- (a) Verify that unit is clearly marked with applicable signs, placards and labels.
 - 1) Product identification on each side and rear
 - 2) "FLAMMABLE" on each side and rear
 - 3) "NO SMOKING" on at least two sides
 - 4) "NO SMOKING" posted prominently in cab of vehicles
 - 5) "EMERGENCY FUEL SHUTOFF" adjacent to each emergency fuel shutoff control.
 - 6) Placards indicating method of Emergency Fuel Shutoff operation, e.g., Push, Pull, Turn, etc.
 - 7) Signs indicating location of fire extinguishers inside enclosed compartments
 - 8) Placards identifying Nozzle Fueling Pressure
 - 9) Placards identifying Filter Differential Pressure
 - 10) Placards identifying Filter and Tank Drain valves.
 - 11) Placard indicating the last date (Month and Year) during which the filter elements were replaced
 - 12) Placard indicating the date (Month and Year) of a satisfactory single element test was performed, if applicable
 - 13) Other information and instructional markings as required by local conditions

4.6. Meter Seals

- (a) Verify that meter calibrators/adjusters are sealed.
- (b) Meters with missing seals may only be used with airline permission and must be calibrated.

4.7. Fire Extinguishers

- (a) Check each fire extinguisher for inspection tag and seal.
- (b) Maintain extinguishers in accordance with the applicable [NFPA 10] guidelines.
- (c) Upon completion of the inspection, update inspection tag.

4.8. Emergency Fuel Shutoff System

- (a) Verify that each emergency fuel shutoff control device will completely stop fuel flow before overrun has exceeded 5 percent of actual flow rate at the time of release.
 - EXAMPLE: If actual flow rate is 400 gpm, fuel flow must completely stop within 20 gallons of emergency shutoff activation.
- (b) Equipment with a defective emergency fuel shutoff system must be removed from service until the system has been repaired.

4.9. Deadman Control System

- (a) Verify that the deadman control system will completely stop fuel flow before overrun has exceeded 5 percent of actual flow rate at the time of release.
 - EXAMPLE: If actual flow rate is 400 gpm, fuel flow must completely stop within 20 gallons of emergency shutoff activation.
- (b) Equipment with a defective deadman control system must be removed from service until the system has been repaired.

4.10. Lift Platforms

- (a) Verify the safe and dependable operation of all lift platforms.
- (b) Thoroughly inspect the lift, including lift and emergency let-down mechanisms, lift interlocks, hydraulic lines, couplings, lighting, wiring, handrails, steps, working surface and signing.
- (c) Any deficiencies must be repaired prior to returning unit to service.

4.11. Refueling Truck Interiors

- (a) Visually inspect tank interior from dome cover openings for water, debris, surfactants, microbial growth and other contamination.
- (b) Check epoxy coated tanks for coating deterioration.
- (c) Clean and repair as necessary.

4.12. Refueling Truck Vents & Dome Covers

- (a) Check tank dome covers, including latches, hinges, seals and gaskets.
- (b) Verify that hinges are forward mounted and will close with forward motion of the vehicle.
- (c) Verify proper operation of tank vents.
- (d) Correct any deficiencies as necessary.

4.13. Refueling Truck Trough Drains

- (a) Manually check trough drains for plugging.
 - Use cable or wire to ensure that there are no obstructions present.
- (b) More frequent checks may be required during inclement weather.

5. Quarterly Checks

5.1. Vehicle Inspection

Perform a thorough overall inspection of the unit to identify components with excessive wear and pending equipment failure.

5.2. Pressure Controls

(a) Check all primary and secondary pressure control equipment. Adjust as necessary.

CAUTION:	NEVER ADJUST PRESSURE CONTROL EQUIPMENT WHILE FUELING AN AIRCRAFT.
NOTE:	All testing of pressure control equipment should be conducted at a test facility or through test connections on tank trucks.

(b) Record primary and secondary fuel pressure settings.

5.3. Water Defense System Check - External Check

Check operation of water defense system in accordance with quarterly requirements of [Section 3-12]

NOTE:	Filter/separators that have been converted to monitors are not required to have the water defense
	systems checked.

5.4. Internal Valve Check

Test to ensure that the internal valve functions properly, i.e., closes except when bottom loading or dispensing fuel.

6. Semi-Annual Checks

6.1. Hose Pressure Check

(a) Refueling hoses fitted with reusable couplings, and being operated under system pressure, must undergo the six-month pressure testing at 225 PSI, per the requirements found in [API/IP 1540].

NOTE:	Pressure testing to 300 PSI is required whenever a new hose attachment or coupling is fitted.
CAUTION:	RECOUPLING AND PRESSURE TESTING OF HOSES AND FITTINGS SHOULD ONLY BE
	CARRIED OUT BY PERSONS ADEQUATELY TRAINED IN THEIR PROPER FITTINGS AND
	TESTING, AND APPROVED BY THE OEM.

7. Annual Checks

7.1. Filter Element Change

- (a) Replace filter/separator filter elements. (Ref. [Section 3-13])
 - Coalescer element service life may be extended to a maximum of two years, provided the criteria in [Section 3-14] are followed.
- (b) Teflon and synthetic separator elements may be reused, provided that they are cleaned and tested in accordance with the element manufacturer procedures.
- (c) Full flow monitor elements are to be replaced annually.
- (d) A visual inspection of all vessel interiors is to be performed on an annual basis regardless of filter element replacement frequency.
 - 1) Verify that the vessel interior is generally clean and free of water, sediment, evidence of microbial growth or other contamination. Clean interior and repair coating as necessary.
 - 2) Verify that all elements are undamaged and secure.

7.2. Fueling Pressure and Differential Pressure Gauges

- (a) Verify that accuracy of gauges used to monitor fuel delivery to aircraft is within +/- 2% of full scale.
- (b) Verify proper operation of filter differential gauge(s) in accordance with gauge manufacturers' procedures. Accuracy must be within +/- 2 PSI. Repair or replace as required. (Ref. [Section 3-9])
- (c) Replace, or repair and calibrate defective gauges.

7.3. Meter Calibration

- (a) Check accuracy of all aircraft fueling equipment meters.
- (b) Adjust meters to an accuracy of $\pm 0.10\%$. Verify repeatability of $\pm 0.05\%$.
 - Meter adjusters/calibrators are to be sealed upon completion of calibration.

7.4. Water Defense System Inspection and Test

Check operation of water defense system in accordance with annual requirements [Section 3-12].

2-9. Refueling Truck Loading

- CAUTION 1: DURING LOADING OF THE REFUELING TRUCK, THE EQUIPMENT MUST NOT BE LEFT UNATTENDED AT ANY TIME.
- CAUTION 2: IT IS NOT ACCEPTABLE TO TRANSFER FUEL INTO A REFUELING TRUCK WHILE IT IS REFUELING AN AIRCRAFT.
- CAUTION 3: IT IS NOT ACCEPTABLE TO TRANSFER FUEL FROM A TRANSPORT TRUCK INTO A REFUELING TRUCK.
- CAUTION 4: REFUELING TRUCKS SHOULD NOT BE LOADED DIRECTLY FROM HYDRANT SYSTEMS, HYDRANT VEHICLES OR CARTS. IF REFUELING TRUCKS ARE FILLED FROM A HYDRANT SYSTEM, ADDITIONAL PRECAUTIONS SHOULD BE TAKEN TO PROTECT AGAINST OVERPRESSURIZATION, STATIC DISCHARGE AND SPILLAGE.

The refueling truck must be bonded to the loading facility piping (unpainted surface) during filling operations. Bottom loading control systems do not negate the need to bond with a separate bonding cable.

If top loading, the loading arm piping shall be bonded to the truck and the loading tubing shall be extended to the bottom of the truck to prevent "splash" loading.

If bottom loading, the loading operation shall be started and the pre-check operated immediately to ensure proper operation of high level shutoff system.

WARNING: IF THE PRE-CHECK SYSTEM DOES NOT OPERATE PROPERLY, THE FUEL TRUCK OPERATOR MUST MONITOR THE RISING FUEL LEVEL TO PREVENT OVERFILL.

Chapter 3. Procedures and Tests

3-1. Appearance Tests

These procedures provide a quick method for detecting free water, solids, and other traces of possible contamination. The two most common sampling methods used for evaluating the visual appearance of aviation fuel samples are; the White Bucket Test and the Clear and Bright Test. Another test method, also used, is known as the Closed Circuit Sampler, or VisiJar. Unusual color, in aviation fuel, may indicate mixing with another product. Both clear and opaque containers can be used to observe product color. However, informal tests have shown the clean white porcelain bucket to be best suited for the detection of unusual color such as contamination with low concentrations of dyed fuel or color resulting from crude oil characteristics or refinery processing. Hence, only the white porcelain bucket is to be used for the optimum detection of red dye contaminated fuel.

1. Appearance Test (also known as the "White Bucket Test")

- 1. Fill a suitable container (preferably a white porcelain bucket) with approximately 1 gallon of fuel.
- 2. Let the sample settle for 1 minute to remove air bubbles.
- 3. Place the clean bucket on a level surface and inspect the bottom for water droplets, solid contaminants, hazy/cloudy condition, brown slime and observe the color of the fuel.

CAUTION:	TO DETERMINE THE DIFFERENCE BETWEEN A HAZE CAUSED BY ENTRAINED WATER
	OR AIR BUBBLES, IT MAY BE NECESSARY TO PERFORM A WATER DETECTION TEST
	(Ref. [Section 3-3]).

4. Examine the sample for solids and/or sediment on the bottom of the bucket. Sample clarity can be checked by placing a small, shiny object with definition on the bucket's bottom. If the fuel is dry, the definition can be easily distinguished. The amount of sediment observed can be described by a letter category using the "Color and Particle Assessment Rating Guide", SGTP-3940, available from Gammon Technical Products, Manasquan, NJ.

NOTE 1:	The presence of contamination is much more evident when the sample is taken from a pressurized
	system. Samples removed from a static system may indicate little contamination when significant
	contamination actually can be found under flow or pressurized conditions.

NOTE 2: Ensure the fuel sampling tap is free of loose contaminants.

WARNING: IF A PINK OR REDDISH DISCOLORATION IS OBSERVED, PERFORM A REFEREE WHITE BUCKET TEST FOR POSSIBLE DYE CONTAMINATION.

1.1 Rating of White Bucket Sample

1.1.1 Solids Contaminant and Moisture Indicators

Table 3-1.1. Rating of White Bucket Sample

Solids Contaminant Indicators	Moisture Content Indicators
1. Clean	A. Bright
2. Slight Particulate Matter	B. Hazy
3. Particulate Matter	C. Cloudy
4. Dirty	D. Wet (Free Water)
	E. Surfactants

1.1.2 Rating Definitions

Table 3-1.2. Solids Contaminant Indicators

Rating	Rating Guide
Clean	Refers to lack of particles, silt or sediment, flakes, dye, rust or solids
Slight Particulate Matter	Contains several fine to moderate sized particles
Particulate Matter	A sample in which many small particles may be seen floating or settled on the bottom
Dirty	Discoloration or many particles dispersed in the fuel or settled on the bottom

Table 3-1.3. Moisture Content Indicators

Rating	Rating Guide
Bright	Brightness is a quality independent of the color of the sample and refers to the lack of suspended or free water in the sample. Bright fuel tends to sparkle.
Hazy	A condition resulting from fine droplets of moisture dispersed throughout the sample producing a dull hazy appearance. This can be a temporary condition brought about by a drop in temperature. During the first minute, the fuel can appear hazy due to air bubbles.
Cloudy	The result of extremely fine droplets of water dispersed throughout the sample giving it a milky appearance
Wet	Any form of free water appearing as droplets or bulk water on the bottom of the bucket or clinging to the sides
Surfactants	Slime in the bottom of the bucket or at the fuel/water interface appearing as a dark brown/black layer, scum or lacy material floating in or on the sample

1.2. Action Required if Red Dye is Detected (Referee White Bucket Test)

- 1. The white bucket test shall be used to inspect for red dye contamination. The referee white bucket shall be the white porcelain bucket obtained from Gammon Technical Products, GTP-1746B or GTP-1746C. This test requires collecting approximately a two (2) gallon sample with a fuel depth of 6 inches, +/- 1.0 inch. The inspector (with normal vision or wearing corrective non-tinted lenses) performs a visual examination of the fuel for color under normal daylight conditions, or in any well-lighted room such as an office with fluorescent lighting. If the inspector is not sure of the fuel color, a consensus of the color should be obtained from several individuals.
 - a. Jet fuel with no visible red tint should be judged acceptable for use in aircraft.
 - b. A red tint resulting from refinery processing will usually disappear when exposed to light. Jet fuel is acceptable for use if the red tint disappears in less than 30 minutes after being exposed to light.
- 2. Jet fuel produced by blending a red dyed fuel with a non-dyed fuel may be acceptable, providing the resulting fuel has no visible red tint, and fuel satisfies [ASTM D1655] specifications. Blending is not to be performed at an airport fuel storage facility without prior authorization by affected airline(s).

NOTE: Reference: Boeing Service Letter, dated 22 November 1996, ATA: 2800.00, Aircraft Use of Dyed Fuels.

2. Clear and Bright Test

NOTE: The White Bucket test may be used in lieu of the Clear and Bright test where the Clear and Bright test is specified; however, the Clear and Bright test may NOT be used in lieu of the White Bucket test, where specified.

The clear and bright test is performed by filling a 32 oz. Mason jar ¾ full and then visually examining it for clarity, water, solids and/or sediment. The jar is then closed, swirled and examined for traces of water at the vortex bottom. There should be no suspended or visible free water, and sample should be clear and bright (slight sparkle). Air bubbles may cause hazy appearance immediately after the sample is drawn, but haze clears from the bottom up. Therefore, while small water droplets and air bubbles may appear similar, air bubbles will rise while water droplets will settle upon standing. Finely dispersed droplets in the fuel sample may be caused by temperature change due to cooling and may show up as a cloudy white appearance.

3. Visual Detection

Large quantities of water in fuel can be detected visually. The water quickly separates from the fuel and settles on the bottom of the sample container. Jet fuel varies in color from dark straw to water white and it is possible to mistake an all water sample as fuel. Adding a known quantity of water to fuel, the water will quickly separate and settle to the bottom of the sample. If the sample is all water, the added water will not separate. For additional clarity, a drop of liquid food coloring may be added to the sample. The food coloring will separate from the fuel, settle to the bottom of the sample, and color the water when mixed.

NOTE:	The American Society for Testing and Materials is publishing a more detailed and comprehensive
	standard which will be titled, "Free Water, Particulate and Other Contamination in Aviation Fuels
	(Visual Inspection Procedures)", and should be used to complement this procedure.

3.1 Recoverable Product Intended for Aviation Use

Recoverable product must be handled carefully to prevent contamination. Containers or equipment used for handling recovered product must be clean and dedicated to that purpose (e.g. equipment used for cleaning hydrant pits shall not be used for handling reclaim product). Every effort should be made to remove visible solids and water (waste fuel) prior to moving fuel into a reclaim tank or returning to storage from a reclaim tank.

Fuel eligible for reclaim shall be limited to fuel from the following sources:

- Tank sumps
- Filter sumps
- Low point drains
- High point vents

Samples from:

- Fuel truck receipts
- Membrane filtration tests
- Gravity/density checks

Equipment Maintenance:

- Filter changeout
- Draining tanks for inspection/cleaning/repair

CAUTION: FUEL FROM ANY OTHER SOURCE THAN THOSE LISTED ABOVE SHALL NOT BE RECLAIMED.

3-2. Membrane Color Filtration Test

This test provides a field method for detection of particulate matter in jet fuel. It is particularly useful in monitoring the cleanliness of fuel received and in evaluating the performance of filter vessels. Because the method produces results which are not quantitative, it is not to be used as the basis for rejection of product. However, it does provide an alert signal, which indicates the need for further investigation using a gravimetric test to determine weight per unit volume or a double membrane test.

If both single and double color/particle ratings exceed maximum allowable limits or are in dispute, a matched-weight gravimetric test will govern or fuel will be rejected.

The operator should be cognizant of the fact that a wet membrane may appear darker than a dry membrane. This must be taken into consideration if there is a need for immediate evaluation. Ratings are to be made after the membrane is completely dry using the [ASTM D2276] Rating Guide, Shell Oil Evaluation Guide or Gammon Evaluation Guide (SGTP 3940).

Whether wet or dry, a membrane with visible particles is cause for concern and requires prompt investigation of the condition of filtration equipment.

3-3. Free Water Test

1. Background

Dry fuel is a prime contributor to flight safety. There are numerous ways that water enters fuel systems; the most common are leakage at manholes and tank plumbing, water-laden transport deliveries, condensation of atmospheric moisture in partially filled storage tanks.

Water in fuel is in one of two forms: Dissolved water ("water in solution") or free water.

Dissolved water is best described as a condition similar to humidity in the air. The amount of dissolved water in fuel varies with the fuel temperature. The higher the fuel temperature, the higher the potential concentration of dissolved water. When the temperature of the fuel is reduced due to change in ambient temperature or in flight, especially at high altitudes, dissolved water will condense from the fuel and become free water. Free water is heavier than fuel and will settle to the bottom of tanks and accumulate in low points in fuel systems.

Free water in fuel can cause serious problems, e.g., engine flame-out if a slug of water reaches the engine, and the formation of ice crystals, which may block fuel filters and fuel control units. Therefore, free water is not permitted to accumulate or remain in fuel systems and free water testing must be performed as scheduled. Over the years, various tests have been developed. However, experience has shown that the human senses cannot be substituted.

2. Water Sensitive Paste or Paper

Chemically treated paste or paper may be used to indicate the presence of free water. These materials change color when they contact water. They do not readily react to low concentrations of water, such as a hazy fuel sample.

These pastes and papers are normally applied to gauging sticks and tapes when checking storage tank bottoms for bulk water.

3. Water Detection Kits

All free water test kits mentioned herein, are known to be commercially available. Manufacturer's instructions are to be followed when using these kits. Kits that have exceeded their expiration dates are not to be used.

- (1) **Velcon Hydrokit.** The test consists of adding fuel to a vial containing a pre-measured amount of water-sensitive powder. If water is present, the powder turns pink. Two kits are available: one with chemical powder sensitive to water concentrations greater than 15 ppm; the other kit with sensitivity greater than 30ppm.
- (2) **Metrocator Kit.** The test, which may be used in detecting free water in concentrations from 5 ppm to greater than 60 ppm, consists of adding a pre-measured amount of water-sensitive powder to a sample of fuel in a special bottle. A disc of filter paper is placed in the screw cap of the bottle. After shaking the contents blue spots appear on the test wafer or disk, and when compared to a standard, indicate the amount of water that is present in the fuel.
- (3) **Aqua-Glo Kit.** The test is capable of detecting free water concentrations as low as 1 ppm. The test consists of passing a measured amount of fuel through a chemically coated paper disc. The disc is then compared to a known color standard. The chemical will glow in ultraviolet light proportional to the amount of water in the fuel.
- (4) **Shell Water Detection Kit.** The test consists of drawing fuel through a capsule containing a water-sensitive chemical. If free water is present, the chemical changes color. The color changes become progressively more noticeable with increasing water concentrations until, at approximately 30 ppm, with a 5 ml sample, a strong, obvious green color is obtained.

3-4. Fuel Density Test (API Gravity)

1. Introduction and Purpose

This procedure describes the means for measuring the density of fuel with a hydrometer. A significant change in density may indicate contamination by another product. Hydrometers may be calibrated in one of the following units of density measurement: API Gravity, Relative Density (Specific Gravity) or Density.

2. Description

The scale reading at the intersection of the fuel surface, on a freely floating hydrometer, in addition to the temperature of the fuel at the time of the test, are observed and recorded. The observed readings are then used to correct the API gravity to the standard temperature for the test.

3. Equipment

- 1. ASTM approved thermohydrometers are graduated in degrees API Gravity and degrees F as specified in [ASTM E100].
- 2. ASTM plain form hydrometers are graduated in units of density, relative density (specific gravity) or API gravity. These must conform to requirements outlined in [ASTM E100]. Plain form hydrometers do not contain a built-in thermometer.
- 3. ASTM approved thermometers are graduated in either degrees F or degrees C and specified in [ASTM E1]. Specific thermometers recommended are the ASTM 12F (graduated in degrees F) or the ASTM 12C (graduated in degrees C). Thermometers are not required where thermohydrometers are used.
- 4. Use a clear glass, plastic or metal hydrometer cylinder as shown in [Figure 3-4.1]. Clear glass or clear plastic cylinders are preferred since accurate hydrometer readings can only be obtained with opaque plastic or metal cylinders when the sample level is at the top of the cylinder.

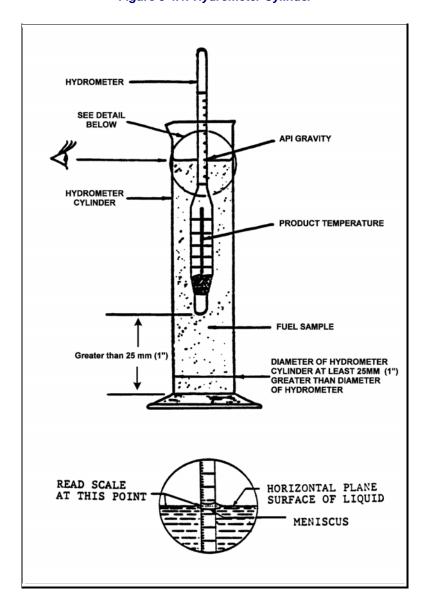


Figure 3-4.1. Hydrometer Cylinder

4. Procedure

- 1. Collect the sample in a clean hydrometer cylinder and place it in a vertical position in a location free from air currents. Allow a minute or two for air bubbles to disappear. Remove any air bubbles that remain on the surface of the sample by touching them with the corner of a clean paper towel.
- 2. When using a thermohydrometer, gently lower it into the sample and, when it has settled, depress it about two scale divisions into the liquid and then release it. Gently spin the hydrometer when releasing it. This will assist in bringing it to rest, floating freely away from the cylinder walls.
- 3. When the thermohydrometer has come to rest and the thermometer is showing a steady reading, read and record the temperature of the sample to the nearest 0.5 C or 1 F. Then read the hydrometer to the nearest scale division and record the value. The correct hydrometer reading is that point on the hydrometer scale at which the principal surface of the liquid cuts the scale (Ref. [Figure 3-4.1]).
- 4. When using a plain form hydrometer, first measure temperature with an approved thermometer. Continuously stir the sample with the thermometer taking care that the mercury is kept fully immersed. As soon as a steady reading is obtained, read and record the temperature of the sample to the nearest 0.5 C or 1 F and then remove the thermometer. To obtain the hydrometer reading, follow the procedure described in paragraphs 2 and 3 above.
- 5. Correct the observed hydrometer reading to the standard temperature of 60 F for API gravity and relative density, or to 15 C for density using the appropriate correction table. (Reference [Figure 3-4.2]).
- 6. Report the corrected gravity measurement.

CORRECTION OF OBSERVED API GRAVITY TO STANDARD TEMPERATURE (1) Enter Table (2) Read down this column to an observed temperature of 78.5 °F Here API GRAVITY AT OBSERVED TEMPERATURE TEMP 45.0 46.0 47.0 49.0 49.5 TEMP 50.0 CORRESPONDING API GRAVITY AT 60 F 75.0 43.6 75.0 75.5 43.6 44.1 45.1 45.6 46.0 46.5 75.5 76.0 44.0 45.0 46.0 43.6 44.5 45.5 46.5 47.0 47.5 47.9 48.4 76.0 76.5 43.5 44.0 45.0 45.5 45.9 46.4 46.9 47.4 47.9 48.4 76.5 43.5 44.0 44.9 45.4 45.9 46.4 46.9 47.4 47.8 48.3 77.0 77.5 78.0 43.4 43.9 45.3 45.8 47.3 47.7 48.2 78.0 44.3 44.8 45.3 45.8 46.7 47.2 47.7 78.5 79.0 43 3 43.8 44.3 45.2 45.7 46.2 46.7 47.2 47.7 48.1 79.0 44.7 79.5 43.2 43.7 44.2 45.2 45.7 47.1 47.6 44.2 80.0 43.7 44.7 45.1 45.6 43.2 46.1 46 6 47 1 47.6 80.0 80.5 43.2 44.1 43.6 44.6 45.1 45.6 46.1 46.5 47.0 47.5 48.0 80.5 81.0 43.1 43.6 44.1 44.6 45.0 45.5 46.0 46.5 47.0 47.5 47.9 81.0 81.5 44.5 45.5 43.1 43.6 44.0 45.0 46.0 46.5 46.9 47.4 47.9 81.5 43.0 43.5 44.0 44.5 45.0 45.4 45.9 46.4 46.9 47.4 47.9 82.0 82.5 43.0 43.9 46.8 47.3 **82**.5 83.0 42.9 43.4 43.9 44.4 44.9 45.3 45.8 46.8 47.3 47.8 83.0 83.5 42.9 43.4 43.9 44.3 44.8 45.3 45.8 46.3 46.7 47.2 47.7 83.5 84.0 42.8 43.3 84.5 42.8 43.3 43.8 44.2 45.2 45.7 47.1 47.6 84.5 47.1 85.0 42.8 43.2 43.7 44.2 47.6 45.2 45.6 46.1 85.0 85.5 44.2 44.6 42.7 43.2 43.7 45.1 45.6 46.1 46.6 47.0 47.5 85.5 44.6 86.0 42.7 43.6 44.1 45.6 47.5 43.2 45.1 46.0 46.5 47.0 86.0 44.1 44.5 46.5 86.5 42.6 43.1 43.6 45.0 45.5 46.0 47.4 86.5 46.9 87.0 42.6 44.0 44.5 87.0 43.I 43.5 46.9 47.4 45.0 45.5 45.9 87.5 42.5 43.0 43.5 45.9 88.0 42.5 43.0 43.5 43.9 44.4 44.9 45.4 45.8 46.3 46.8 47.3 88.0 42.5 42.9 43.4 43.9 44.4 44.8 45.3 45.8 46.8 47.2 88.5 42.4 42.9 43.4 43.8 44.3 44.8 45.3 45.8 47.2 89.0 89.5 42.8 43.3 44.8 45.2 45.7 47.1 89.5 90.0 42.3 42.8 43.3 43.2 44.2 44.7 45.2 45.7 46.1 46.6 47.1 90.0 API GRAVITY 45.0 TO 50.0 EXAMPLE: Hydrometer reading for a jet fuel sample at 78.5°F is 45.5 API. To determine the corrected API Gravity (1) Enter table 5B in the column "API Gravity at Observed Temperature" headed 45.5° API (2) Read down this column to an observed temperature of 78.5°F (3) The corrected API Gravity at 60°F is 43.9°API

Figure 3-4.2. Hydrometer Reading Correction Table

5. Cautions

The hydrometer must float freely to obtain a correct reading. It must not come to rest against the side or bottom of the cylinder during the test.

The thermometer should not be completely removed from the liquid to read temperature. Evaporation of liquid from the thermometer stem and bulb will lower the temperature and cause an incorrect reading.

Hydrometers and thermometers must be inspected periodically to ensure they are not cracked and there is no separation of the mercury column.

3-5. Water Separation Test (MSEP)

This test provides a field method for determining water separation characteristics of jet fuel. Fuel containing little or no surfactant has excellent water separation characteristics. Fuel containing significant amounts of surfactant has very poor water separation characteristics. The better the water separation rating, the more effective system filtration equipment will be in removing free water.

Detailed instructions for performing the test are contained in [ASTM D3948].

The test is particularly useful in monitoring clay treatment vessel performance. Tests are normally performed on samples from the upstream and downstream sides of the vessel and the results compared. Assuming a relatively low rating upstream (e.g., 70), the downstream result should be higher if the clay is active. No improvement or a worsening of the rating on the downstream side of the vessel are indications of spent clay. In a well maintained fuel handling system equipped with a clay treatment vessel, water separation ratings are normally in the range of 90 to 100.

The operator should be cognizant of the fact that this is a very sensitive test. Erroneous results can be obtained if improper sample containers are used. This is especially true with new unlined metal containers.

3-6. Fuel Odor

Acceptable jet fuel produces a distinctive odor which can range from relatively mild or sweet to moderately objectionable. Specification [ASTM D1655] states that the fuel shall not produce an odor which is "nauseating or irritating." Such strong odors can be indicative of problems and requires further investigation.

During all phases of fuel handling, sampling and testing, the operator should be cognizant of the significance of fuel which produces an unfamiliar odor. If fuel producing an unfamiliar, nauseating or irritating odor is detected, further investigation is required. Draw a one gallon sample of the fuel into a suitable sample container (Ref. [Section 3-8]), and notify the affected airline(s).

3-7. Visual Detection of Microorganisms

1. Introduction and Purpose

This method is a visual means for detecting symptoms which indicate the presence of microorganisms in a turbine fuel handling system.

2. Description

The test is performed concurrently with routine drainage of storage tank and filtration equipment sumps and low point drains in system piping. A sample from these locations is drawn into a white bucket and examined visually. Microorganisms produce visual evidence, which can be identified by the operator.

3. Equipment

Equipment consists of a plain white porcelain lined bucket and clear glass petroleum product sample bottles.

3.1. Equipment Preparation

Thoroughly rinse the white bucket and sample bottles in system fuel. Wipe the inside of the bucket dry and free of contaminants with a clean rag or paper towel.

3.2. Cautions

Some contaminants commonly found in turbine fuel handling systems produce visual evidence which may be erroneously interpreted as microorganisms. For this reason, it is essential that the operator does not draw immediate conclusions.

CAUTION: BECAUSE VISUAL EVALUATION IS USED, THE ELEMENTS OF HUMAN PERCEPTION AND JUDGMENT ARE INVOLVED.

The determinations made using this test indicate only the probability that microbial contamination exists. Confirmation must be made by laboratory analysis. (Reference [ASTM D6469] entitled "Standard Guide for Microbial Contamination in Fuels and Fuel Systems")

4. Test Procedure

- (a) Collect sample from storage tank sump, filtration equipment sump, or low point drain in system piping in the white bucket.
- (b) Allow the bucket sample to settle for at least two minutes.
- (c) Tip the bucket from side to side while visually observing sample for any evidence of dark-colored solids, dark-colored water, substances which cling to the side of the bucket, or scummy mucus-like material
- (d) If any suspect materials are found, carefully pour off the fuel contained in the bucket, and collect a sample of the suspect material in a clear glass sample bottle.
- (e) Cap the sample bottle tightly and identify as to date, location, and sampling point.

5. Evaluation of Sample

- (a) Visually examine the contents of the sample bottle in an area providing strong background lighting.
- (b) If suspect portion of the sample is a dark-colored, sludge-like substance, the presence of fungi or microorganisms is a possibility. Further indicators are a matty, lumpy or stringy consistency or a rank moldy odor.
- (c) If visual examination, as described above, indicates the possible presence of fungi or microorganisms, forward the sample to a qualified laboratory for analysis. (Reference [ASTM D6469] entitled "Standard Guide for Microbial Contamination in Fuels and Fuel Systems")

3-8. Sample Containers

The types and preparation of containers most suitable for the handling of aviation fuel samples can be found in [ASTM D4306], entitled "Standard Practice for Sampling Aviation Fuel for Tests Affected by Trace Contamination."

For the purpose of retaining one gallon samples of suspect product as specified in this manual, a one gallon epoxy-coated sample can is recommended. Such cans are commercially available from a number of aviation fueling equipment distributors.

3-9. Filter Vessel Differential Pressure

The purpose of observing differential pressure across a filter vessel is to monitor the changing condition of the elements. Whenever fuel passes through a filter, a drop in pressure should occur. The difference in pressure between the inlet and outlet of the filter vessel is known as differential pressure, and is one of the more apparent indicators of filter element condition. Differential pressure will fluctuate proportionately to flow rate. Fuel must be flowing through a filter vessel in order to have a differential pressure.

The differential pressure reading shall be taken and recorded daily. For accuracy, these checks should be undertaken when the flow rate is steady, and as close as possible to maximum operating flow rate. Tests on individual filter vessels should be carried out at the same flow rate, if possible.

The filter elements must be replaced when a sudden drop in differential pressure occurs under similar flow conditions or the filter vessel differential pressure exceeds the following limits:

- Coalescer elements 15 psi.
- Monitor elements 25 psi.

Verify proper operation of filter differential gauge(s) in accordance with gauge manufacturers' procedures. Accuracy must be within +/- 2 PSI. The differential pressure gauge lines and valves should be checked periodically to ensure they are not plugged or restricted.

Some direct reading gauges have small filters in their inlets. They must be replaced at each filter change out to assure proper operation.

3-10. Bonding Cable Continuity Check

1. General

This section describes the equipment and procedures to check and test bonding cable. Safety of personnel, equipment, and facilities is of primary concern.

2. Bonding Equipment, Checks, and Maintenance

Maintain bonding cables, clips, straps, and adapter plugs in good condition. Check this equipment for frayed wires, broken or damaged clips, and worn or damaged adapter plugs. Make electrical continuity checks monthly between the bonding clamp and the fueling system framework, while unreeling cable slowly. Resistance must be 25 ohms or less.

3-11. Fuel Storage Tank Inspection and Cleaning

CAUTION:	ENTRY OF A FUEL STORAGE TANK REQUIRES SPECIALIZED EQUIPMENT AND
	PERSONNEL TRAINING, AND MUST BE ACCOMPLISHED IN ACCORDANCE WITH LOCAL,
	STATE, AND FEDERAL OCCUPATIONAL SAFETY REGULATIONS. THE VENDOR SHOULD
	NOT ATTEMPT ENTRY OF A STORAGE TANK UNLESS THE REQUIRED EQUIPMENT IS ON
	HAND AND THE REQUIRED TRAINING OF PERSONNEL HAS BEEN ADMINISTERED.

Storage tank floor must be visually inspected at least every 12 months. Check for build-up of sediment, evidence of microbial growth or significant coating failure. If inspection reveals microbial growth or build up of sediment exceeding 1/10 of the area of the tank bottom surface, cleaning must be accomplished.

Jet fuel storage tanks shall be cleaned with water only. High pressure application is recommended.

After cleaning with water, squeegees and lint free mops may be used to dry the tank surfaces. Assure removal of all free water, and allow tank to dry thoroughly.

Inspect internal epoxy coating for evidence of chipping, flaking, or other deterioration. Repair as required.

Maintain a record of tank inspection and cleaning on ATA Form No. 103.07 (or similar). The tank inspection and cleaning dates shall be recorded (month & year) on the tank manhole cover.

3-12. Water Defense Systems

1. General Description

There are primarily two types of water defense systems for filter/separator vessels:

1.1 Float-operated System

This system is actuated by a signal from a float with a sealed air and fluid chamber, which, when properly balanced, will sink in fuel and float in water. The float remains in the downward position, permitting fuel flow, until enough water accumulates in the sump to raise the float. As the float rises, it re-positions a pilot valve or operates an electrical switch, either of which will stop fuel flow. In order to restart fuel flow, the accumulated water must be drained from the sump, allowing the float to return to the downward position.

In some float-operated systems, an additional function is provided whereby, at an intermediate float position, the pilot valve (or an additional electrical switch) is actuated. This causes a drain valve to open and drain the accumulated water. At a higher level, the flow is stopped as previously noted. An automatic water drain valve does not discharge all the water from the sump. Complete water removal must be ensured to prevent microbiological growth.

CAUTION:	USE THE MANUAL WATER DRAIN VALVE FOR DAILY OR MORE FREQUENT DRAINING,
	EVEN WHEN AN AUTOMATIC WATER DRAIN VALVE IS INSTALLED. IT SHOULD BE
	EMPHASIZED THAT THESE AUTOMATIC DRAIN VALVES DO NOT OPERATE UNTIL
	THERE IS A SUBSTANTIAL QUANTITY OF WATER IN THE SUMP OF THE UNIT, AND
	THEREFORE, DOES NOT REPLACE THE REQUIREMENT FOR MANUALLY DRAINING ALL
	WATER FROM THESE SUMPS.

1.2 Electric Probe-Operated Systems

This type of system consists of an electric probe installed in the filter/separator sump, which actuates a relay system to stop fuel flow when enough water accumulates in the sump. In order to restart fuel flow, the accumulated water must be drained from the sump.

In some probe-operated systems, an additional function is provided whereby, at an intermediate water level in the sump, an additional set of contacts, in the probe, are actuated. This causes a drain valve to open and drain the accumulated water. At a higher level, the flow is stopped as previously noted. An automatic water drain valve does not discharge all the water from the sump. Complete water removal must be ensured to prevent microbiological growth.

CAUTION:	USE THE MANUAL WATER DRAIN VALVE FOR DAILY OR MORE FREQUENT DRAINING,
	EVEN WHEN AN AUTOMATIC WATER DRAIN VALVE IS INSTALLED. IT SHOULD BE
	EMPHASIZED THAT THESE AUTOMATIC DRAIN VALVES DO NOT OPERATE UNTIL
	THERE IS A SUBSTANTIAL QUANTITY OF WATER IN THE SUMP OF THE UNIT, AND
	THEREFORE, DO NOT REPLACE THE REQUIREMENT FOR MANUALLY DRAINING ALL
	WATER FROM THESE SUMPS.

2. Check Procedures

Water defense systems must be checked in accordance with the manufacture's procedures, at the frequencies specified in this document. For systems where manufacturer's procedures are not available, guidance may be obtained from [ASTM MNL5].

NOTE:	Checking of water defense systems is not required on filter/separator vessels that have been converted
	to monitors.

- CAUTION 1: NEVER USE PUBLIC OR OTHER POTABLE WATER SUPPLY SYSTEMS FOR WATER DEFENSE SYSTEM TESTS. THE FUEL PRESSURE MAY BE GREATER THAN THE WATER SUPPLY PRESSURE, AND FUEL COULD BE FORCED INTO THE WATER SYSTEM, CONTAMINATING THE SYSTEM.
- CAUTION 2: WHEN INJECTING WATER INTO A FILTER VESSEL DURING TEST OF WATER DEFENSE SYSTEMS, ALWAYS INJECT A MEASURED AMOUNT OF WATER, BEING CAREFUL NOT TO FLOOD THE SEPARATOR STOOLS. THE SAME AMOUNT OF WATER MUST BE RECOVERED AS WAS INITIALLY INJECTED. WHEN TESTING MOBILE EQUIPMENT, NO MORE THAN ONE (1) US GALLON SHOULD BE INJECTED. DUE TO THE VOLUME OF WATER REQUIRED, AS WELL AS FLOW RATES AND PRESSURES ENCOUNTERED, STATIONARY UNITS SHOULD NOT BE TESTED BY INJECTING WATER.
- CAUTION 3: BE SURE THAT THE VENT PORT FROM THE FLOAT-OPERATED PILOT VALVE IS UNOBSTRUCTED. IF THE VENT PORT IS PLUGGED, IMPROPER OPERATION OF THE FUEL FLOW CONTROL VALVE AND/OR AUTOMATIC DRAIN VALVE WILL OCCUR. PLUGS AND/OR VALVES MUST NEVER BE INSTALLED TO STOP LEAKS IN OPERATIONAL SYSTEMS.
- CAUTION 4: MANUAL TESTERS ON SOME FLOAT-OPERATED PILOT VALVES (NON-EXTERNALLY-BALLASTED FLOATS) ONLY CHECK THE OPERATION OF THE CONTROL VALVE, AND NOT THE BUOYANCY OF THE FLOAT.
- CAUTION 5: WATER SHALL NEVER BE INJECTED INTO MOBILE EQUIPMENT WATER DEFENSE SYSTEMS WHILE FUELING AN AIRCRAFT.
- CAUTION 6: FUEL MUST BE RECIRCULATED ON MOBILE FUELING EQUIPMENT AFTER WATER DEFENSE SYSTEM CHECKS.

2.1. Quarterly External Check

- A. Externally check for satisfactory operation of water defense system by mechanically raising the float or float ballast on float-type systems, or injecting water into the probe on probe-type systems. The system(s) may also be checked, by injecting water into the filter sump(s).
- B. Repair any system deficiencies before returning equipment to service.

2.2. Annual System Inspection & Test

- A. Check for satisfactory operation of float-type water defense systems by verifying proper system shutdown when the float is in the raised position, and confirm buoyancy of floats that are not equipped with external ballasts.
- B. Check for satisfactory operation of probe-type water defense systems by injecting water into the probe.
- C. Both system types may also be checked, by injecting water into the filter sump.
- D. Check probe assemblies in probe-type systems for contaminants, which may have accumulated on the outside of the probe. Clean the probe exterior according to manufacturer's instructions. This will ensure that the probe will be sensitive to changes in resistivity and properly signal the presence of water in the filter sump. If the probe is insulated by contaminants, it may not properly activate the water defense system.
- E. Repair any system deficiencies before returning equipment to service.

3-13. Filter Element Change Procedures

A. Stop pumping product to the filter vessel, close the shutoff valves in the inlet and outlet lines, and open vent.

WARNING: IF FILTER VESSEL IS EQUIPPED WITH AN ELECTRICAL HEATER, BE SURE TO TURN OFF HEATER BEFORE OPENING DRAIN.

- B. Open drain valves, allow sufficient time for unit to vent, and drain completely before opening cover.
- C. Open cover and remove old elements.
- D. Wash the interior of the filter vessel with clean jet fuel.
- E. Inspect epoxy-coating for deterioration. Repair as necessary.

WARNING: FILTER/SEPARATOR VESSEL WILL NOT PERFORM CORRECTLY IF POSITION OF COALESCER AND SEPARATOR ELEMENTS IS REVERSED.

F. Install new filter vessel elements in accordance with the manufacturer's recommendations. Disposable polyethylene or cotton gloves should be used when handling the elements. With gloves on, the elements may be handled without disarming them.

CAUTION: FAILURE TO FOLLOW MANUFACTURER'S TORQUE RECOMMENDATIONS MAY RESULT IN DAMAGE TO FILTER ELEMENTS OR COMPROMISE PERFORMANCE.

- G. Install new seal, close cover, and tighten all bolts.
- H. Open shutoff valves and allow the filter vessel to fill slowly, allowing all entrapped air to escape.

WARNING: FILLING VESSEL TOO RAPIDLY MAY CAUSE STATIC DISCHARGE, POSSIBLY RESULTING IN EXPLOSION.

- I. Inspect for leaks. Repair as necessary.
- J. Open inlet and outlet valves to full open position.
- K. Record filter change date on filter change record.
- L. Record month and year of filter change on filter vessel.
- M. Circulate product at normal rate. Check and record the differential pressure.

NOTE: Unusually low or no differential pressure at normal flow may indicate that some of the elements are not properly sealed allowing product to by-pass some of the elements. Further investigation may be required.

3-14. Filter Element Replacement Criteria

1. Element Replacement Criteria and Frequency

1.1. Coalescer Elements

- A. Coalescer elements must be replaced when:
 - 1. Filter Membrane (Millipore ®) Test indicates elements are not performing (Refer to [Section 3-2]).
 - 2. Differential pressure exceeds 15 psi or there is a sudden drop in differential pressure.
 - 3. Continuous sump samples indicate surfactants, microbial or solid contaminants.
 - 4. 12-month service life has expired and approved extension procedure is not accomplished
 - 5. Free water detection test indicates elements are not performing (Ref. [Section 3-3]).
- B. Coalescer Element Time Extension

Although normal coalescer element life is 12 months, the elements' life may be extended, provided all affected airlines approve the coalescer element life extension program. Station, fueling agent, FBO, etc., will submit proposed time extension program in writing. If accepted, airlines will provide written approval, which must be available for review.

1.2. Teflon-coated and Synthetic Separator Elements

Teflon-coated and synthetic separator element life can be extended as long as the elements are cleaned and inspected according to manufacturer's instructions.

1.3. Full Flow Monitor Elements

- (a) Full flow monitor elements must be replaced when:
 - 1. Filter Membrane (Millipore) Test indicates element is not performing (Ref. [Section 3-2]).
 - 2. Differential pressure exceeds 25 psi or there is a sudden drop in differential pressure.
 - 3. Chemical water detection test indicates a positive reading of more than 15 ppm (Ref. [Section 3-3]).
 - 4. 12-month service life has expired.

3-15. Flushing Standards and Specifications

Before a new, modified, repaired or inactive fuel system, or portion thereof, is placed into service, all piping affected by change must be flushed to ensure system cleanliness.

1. Flushing Standards

All flushing procedures shall be pre-approved by the fuel quality assurance (QA) representative(s) of the affected airline(s).

Product used for flushing shall meet [ASTM D1655], latest revision, specs for kerosene Jet A/A-1 type aviation grade turbine fuels.

Desired fuel velocity during flush is 10 feet per second minimum unless a lesser rate is agreed upon by the QA representative(s) of the affected airline(s). Additional temporary pumps and filters may be required to provide minimum flow velocity.

No flushing will be allowed through hydrant pit valves.

Hose assemblies must meet the requirements of [API 1529]. Documented certification of hydrostatic test to at least 300 PSI is required within the last 24 months. Banded hose fittings are prohibited.

Test samples are to be drawn:

- Immediately ahead of filtration on closed loop recirculation systems, or
- Immediately ahead of storage tanks from receiving manifold on recirculation systems returning flushed fuel to tank, or
- Immediately ahead of transport trucks, from the cross-sectional center of the flushing riser, on single line systems.

Sampling connections shall be equipped with probes installed in the flow path.

Where possible, temporary piping connections to form a closed loop piping system should be installed and the system flushed by means of recirculation.

All general service valves and adapters shall be in place throughout the flushing procedure. Remove control valves and metering assemblies prior to initiating flush.

2. Flushing Into Tank Trucks and Portable Tanks

CAUTION: ALL ELECTRICAL AND MOTORIZED EQUIPMENT IN AREA SHOULD BE SHUT DOWN IN CASE OF A MISHAP OR FUEL SPILL. FOR SAFETY, ALL PERSONS NOT INVOLVED IN THE FLUSHING OPERATION MUST BE KEPT A MINIMUM OF 100 FEET AWAY FROM THE TANK TRUCKS AND HYDRANT PITS USED IN THE FLUSHING OPERATION.

- 1. Notify local authorities as necessary.
- 2. Ensure provision of temporary manifolds and a sufficient number of tank trucks and hoses to safely achieve the desired flow rate.
- 3. Tank truck internal valve controls shall be secured in the open position.
- 4. Ensure all seals and gaskets are in place before securing couplings. All quick release type couplings are to be secured by wire, tape or other means to prevent accidental disconnection during flushing.
- 5. Hoses are to be secured in a manner to prevent whipping during flushing.
- 6. Electrically bond tank to system piping.
- 7. Check for leaks and system tightness before reaching flushing velocity.
- 8. Fire extinguisher(s) are to be in place in case of emergency.
- 9. Location of test personnel:
 - One person per tank to monitor fuel level in tank.
 - One person at hose inlet connection to control fuel flow into tank.
 - One person at main pump control station to shut down pumps in emergency.
 - One person at nearest emergency fuel shutoff station to shut down pumps in an emergency.
 - One person must be dedicated to the command of flushing operation.

3. Acceptance Criteria

A minimum of two consecutive acceptable test results is required to ensure piping cleanliness. The system being flushed must be displaced with clean fuel prior to taking the second test. Unusual conditions, if observed, must be acted upon accordingly, i.e., feel, color, odor, etc.

Visual - All fuel samples must be clear and bright.

Solids criteria (one gallon sample) –

- Membrane Color Maximum wet rating of A3, B3 or G3 [ASTM D2276]
- Particle Assessment Maximum rating of "B" [Shell Oil Evaluation Guide or the Gammon Evaluation Guide (SGTP 3940)].

NOTE:	If color rating exceeds the above limits or is in dispute, a gravimetric rating not to exceed 0.5 mg/L
	shall govern.

Free Water - 15 ppm maximum

Water Separation (MSEP) Rating - 85 minimum

4. Final Acceptance

It shall be the responsibility of the airline(s) fuel quality assurance representative(s), or designee(s), to have final decision on system cleanliness and acceptance before aircraft fuel servicing is permitted.

Chapter 4. Terms & Definitions

Additives Compounds used to impart new properties to a product or to improve a property which it

already possesses.

Adsorption A separation method where a component of a mixture is concentrated on a surface e.g.:

Surfactants (surface-active-agents) are separated from jet fuel by adsorption on clay.

API (American

Petroleum Institute) The institute represents and is supported by the petroleum industry. It standardizes the tools

and equipment used by the industry and promotes the advancement of research in the

petroleum field.

API Gravity An arbitrary scale expressing the density of liquid petroleum products, as established by the

American Petroleum Institute (API). The measuring scale is expressed in terms of degrees API from 0 to 100, the higher the API gravity, the lighter the compound. API gravity can be

reported at observed temperature or corrected to 60°F. (See Relative Density.)

Ambient Temperature The air temperature at a specific location

Antioxidant A chemical added to petroleum products to inhibit oxidation.

Appearance Refers to the visual examination of fuel; appearances are, clear, bright, hazy, and cloudy

ASTM (American Society

of Testing and Materials)

A scientific and technical organization formed for the development of standards on

characteristics and performance of materials, products, systems, and services and the

promotion of related knowledge.

Barge A vessel, either motorized or towed, used to carry products in navigable waterways.

Barrel (bbl) A common unit of measurement of liquids in the petroleum industry. It equals 42 US

standard gallons.

Batch A measured amount in which crude oil and refined product shipments are sent through a

pipeline.

Batching Sequence The order in which shipments are sent through a pipeline.

Blending The procedures by which pre-determined quantities of two or more similar products are

homogeneously mixed to upgrade one of the products or to produce an intermediate grade or quality. This term is also used to define the injection of additives, such as corrosion or icing

inhibitors, into fuels.

Breathing The movement of gas (product vapors or air) in and out of the vent lines of storage tanks.

Calibration (1) The graduation of a measuring instrument. (2) The determination of accuracy of

graduation in a measuring instrument.

Cathodic Protection An electrolytic method of protecting a buried pipeline or other metal structure against

corrosion by surrounding it with an electrical field strong enough to overpower the currents seeking to leave the metal to go into the soil. (See Impressed Current and Sacrificial Anode)

Clay Treatment Vessel A vessel containing bulk clay, clay bags or clay canisters used for removing surfactants

(surface-active-agents), by adsorption, from jet fuel.

Clear and Bright Clear is the absence of visible solids, a cloud, a haze, an emulsion, or free water in the

product. Bright is the sparkle of clean, dry product in transmitted light.

Coalescing (1) Drawing together, combining, or uniting to form one body. (2) A method of separating

finely divided or suspended water from a petroleum product by passing the product through

filter media of a filter/separator.

Coalescer Element The first stage cartridge in a filter/separator vessel that removes solid particles and coalesces

free water from jet fuel. It is upstream of the separator cartridge.

Commingling The mixing of two or more products of different ownership, type, or grade.

Conductivity The ability of a given substance to conduct electric current.

Contaminants Substances which may be present in jet fuel that detract from its performance.

Contaminated Product A product in which one or more grades or types of products have been inadvertently mixed;

or a product containing contaminants such as dust, dirt, rust or water.

Copper Strip Corrosion Test A qualitative method of determining the degree of corrosivity of a fuel by suspending or

placing a strip of polished copper in a sample of fuel and comparing the test strip with a

standard.

Correction Factor Due to the natural effects of thermal expansion and contraction of petroleum products in

general, petroleum products must be corrected to a volume at 60°F for inventory and/or quality control purposes. Petroleum products are temperature corrected by using API Tables 5B and 6B for Generalized Products. Table 5B is for Correction of Observed API Gravity to API Gravity at 60°F. Table 6B is for Generalized Products, Correction of volume at 60°F

against API Gravity at 60°F. Reference [ASTM D1250].

Deadman Control A control device which must be hand held by the operator to allow fuel to flow. When

released, fuel flow stops automatically.

Dedicated System A system of pipeline(s), vessel(s), tank(s) and/or truck(s) used solely for the transportation

and storage of one type of product.

Density Specific weight or mass of a substance per unit volume (pounds per cubic foot or gallon, or

kilograms per cubic meter).

Differential Pressure (Delta P)The measured difference in pressure between any two points, generally between inlet and

outlet connections on filtration vessels.

DikeAn embankment or wall erected around a storage tank to contain the product if tank leakage

or rupture should occur.

Direct Reading Differential

Pressure Gauge A pressure gauge which automatically displays the differential pressure between the inlet

and outlet connections of filtration vessels.

Disarming ActionThe process by which elements in filtration systems are rendered incapable of performing

their designed functions.

Dissolved Water Water which is in solution in jet fuel. This water is not free water and cannot be removed by

conventional means.

Effluent Stream of fluid at the outlet of filtration vessels.

Elements A generic term given to different types of decontamination media installed in various types

of filtration vessels.

Emulsion A dispersion of immiscible liquids in a continuous liquid phase.

Entrained Water Entrained water is found in fuels in the form of very small droplets, fog, or mist and it may

or may not be visible. When large quantities of entrained water are present, the fuel will have a hazy or milky appearance. Water usually becomes entrained in the fuel when it is broken up into small droplets and thoroughly mixed with the fuel in equipment such as pumps or meters. Given sufficient time and the proper conditions, entrained water will settle out, however since turbine engine fuel is fairly dense it will hold entrained water in

suspension for long periods of time.

Evaporation The conversion of a liquid into vapor, usually by means of heat.

Filter A device to remove solid particles from fuel.

Filter Membrane (Millipore ®) Test

A standard test in which jet fuel is passed through a small filter membrane. The cleanliness

of the fuel can be determined by measuring the residue or amount of solid contaminates left

on the membrane. (Ref. [ASTM D2276])

Filter/Separator A filtration vessel which removes solids and coalesces free water from jet fuel.

Filter/separators are equipped with two or more types of cartridges, including coalescer

elements (first stage) and separator elements (second stage).

Fixed Base Operator (FBO)Common title for aircraft fueling agents or vendors at airports.

Flash Point The lowest fuel temperature at which the vapor above the fuel will ignite. (Ref. [ASTM

D56] or [ASTM D3828])

Floating Suction Pump suction piping with floatation capability used to draw the cleanest product from an

upper level of the fuel in a jet fuel storage tank.

Free Water Water in fuel other than dissolved water. Free water may be in the form of droplets or haze

suspended in fuel (entrained water or an emulsion) and/or water layered at the bottom of the

container holding the fuel.

Freeze Point The fuel temperature at which the last fuel wax crystals disappear when the fuel is warmed.

(Ref. [ASTM D2386])

Fuel System Icing Inhibitor (FSII) An anti-icing additive for jet turbine engine fuels.

Fungible Interchangeability of like product batches which can be substituted for purposes of shipment

or storage.

Gallon (gal) A unit of measure of volume. A U.S. gallon contains 231 cubic inches or 3.78541 liters.

Gauging To measure the contents of a tank.

Gauging Table

Tank specific measurement tables used to determine the volume of the tank at selected

incremental levels.

Hydrometer A graduated instrument for determining the unit density of liquids. It is usually made of

hollow glass and weighted at one end so as to float upright. The depth to which the

instrument sinks when immersed in a liquid is determined by the density of that liquid. The lighter the liquid, the lower the instrument sinks. Some hydrometers are marked to read in

degrees of API gravity, specific gravity (relative density), or density.

Hydrophilic Attracts water or is water wetting. Has an affinity for water. Opposite of hydrophobic.

Hydrophobic Repels water or is non-water wetting. Resists attracting water. Opposite of hydrophilic.

Immiscible Liquids which are mutually insoluble; Opposite of miscible.

Impressed Current Cathodic protection system utilizing a direct supply of electrical current to develop the

potential difference between the (Energized) anode and the structure being protected. The method involves putting electrical current into the soil so that it flows toward and into the

line or structure.

Influent Stream of fluid at the inlet of filtration vessels.

InnageThe volume of the liquid in a tank or container measured from the bottom of the tank to the

top surface of the liquid.

Interface The common boundary (or surface) of two liquids.

Jet A/A-1 Fuel High-quality kerosene products used primarily as fuel for commercial jet and turboprop

aircraft engines.

Joint Use Fueling Systems A fuel system shared by multiple users.

Liter (L) A metric unit of volume equal to 0.264 US gallons.

Lubricity The ability to lubricate. In fuels, it refers to a value that is measured by the BOCLE test.

(Ref. [ASTM D5001])

Manifold A piping arrangement which permits a stream of liquid or gas to be divided into two or more

streams, or which permits several streams to be collected into one.

Micron A metric unit of linear measurement; One micron is equal to 0.000039 inches and

approximately 25,400 microns equals one inch.

Micronic Filter A filtration vessel or element designed to remove solid particles from aviation fuels.

Miscible Liquids which are mutually soluble; opposite of immiscible.

Monitor Element A cartridge designed to absorb water and remove particulates in aviation fuel. As the

cartridge approaches its maximum capacity, flow is progressively restricted due to the

accumulation of absorbed water and/or retained particulate matter.

NIST National Institute of Standards and Technology.

NFPA National Fire Protection Association.

Must Indicates a mandatory requirement.

Off-Specification Product A product which fails to meet one or more of the physical, chemical, or performance

requirements of the product specification or is otherwise unfit for its intended use.

Ohm A unit of measure of electrical resistance; the higher the value, the greater the resistance to

electrical flow.

Outage (or Ullage)

The volumetric difference between the nominal capacity and the actual contents of a storage

container.

Oxidation A chemical reaction whereby material(s) combine with oxygen to form a new compound.

Particulates Solid contaminants found in jet fuel, i.e., dirt, rust, sand or fibers.

Pipeline Batch The quantity of a product pumped into the pipeline in one continuous operation.

Pipeline Tender A quantity of product offered or designated for pipeline shipment moved in one or more

batches.

Pre-Check System A system used to check the operation of the automatic high level shutoff equipment on tank

trucks for preventing fuel spills.

Pressure Drop See Differential Pressure.

Product Unless indicated otherwise, jet fuel.

PSI Pounds per square inch; a measurement of pressure.

PSIG Pounds per square inch gauge.

Reclaim Fuel Fuel which has been removed from a fueling system (non-aircraft) for purposes of quality

control checks or maintenance, which has been determined acceptable for return to storage.

Relative DensityThe ratio of weight of any volume of fuel to the weight of an equal volume of water at the

same temperature and pressure; sometimes referred to as Specific Gravity. Typically, the

measurement is corrected to 60 degrees Fahrenheit.

RepeatabilityThe allowable difference between two test results on the same sample by the same operator

using the same equipment.

ReproducibilityThe allowable difference between two test results on the same sample by different operators

in different locations.

Sacrificial (Galvanic) Anode Cathodic protection system utilizing a controlled cell, which shifts the corrosion on the

protected structure to the sacrificial anode. Sacrificial anodes are consumed, but are installed so they are easily replaceable. Sacrificial anodes are commonly made of

magnesium, aluminum, zinc, etc.

Sample A representative portion of fuel taken from a batch, tank or system etc., for analysis.

Samples may be taken either manually or automatically.

Screen A filter, sieve, or barrier made of meshed wire or perforated metal intended to remove solid

matter from a flow stream or to segregate particulates by size of solid matter.

Sediment Solid matter that settles to the bottom of a liquid filled container.

Separator Element The second stage cartridge or shroud in a filter/separator vessel that allows passage of jet

fuel but repels free water. It is located downstream of the coalescer cartridge.

Settling Time The elapsed time that a product remains undisturbed between receipt of product into a

storage tank before being discharged.

Shall Indicates a mandatory requirement.

Shelf LifeThe length of time a product, other than jet fuel, may typically be stored from the date of

manufacture without deteriorating.

Should Indicates a recommendation or that which is advised, but not mandatory.

Sludge A deposit consisting of water, dirt, and other sediment.

Soluble Capable of dissolving or passing into solution.

Specification Prescribed limits of control used to maintain uniformity of a product.

Specific Gravity See relative density.

Sump (noun) A chamber or depression installed at the low point of fuel system components; such as

storage tanks, filtration vessels, etc., to facilitate the collection and removal of contaminates.

Sump (verb) The process of removing liquid from a drain for the purpose of quality control.

Sump Fuel Fuel removed from sumps, e.g., storage tanks, filtration vessels, aircraft refuelers, etc., while

performing routine quality control checks and equipment maintenance. Some sump fuel may

be recoverable. See Reclaimed Fuel.

Surfactants An acronym for surface-active-agents, which are chemical substances or detergent-like

compounds frequently found in jet fuels. These chemicals reduce interfacial tension and may disarm the water removing capability of coalescer cartridges in filter/separators. Clay

treatment is the primary means of removing surfactants from jet fuel.

Surge Tanks Small tanks that collect fuel from pressure relief valves and/or air eliminators.

Suspension Dispersion of small particles of a solid, or small droplets of a liquid, in a liquid or gas.

Switch Loading The loading of a high static charge retaining hydrocarbon (diesel, jet fuel, kerosene) into a

tank truck, tank car or other vessel that has previously contained a low flash point hydrocarbon (gasoline Jet B, JP 4) and may contain a flammable mixture of product vapor

and air.

Temperature Degree of heat or cold as measured by a thermometer: Temperature centigrade degrees (C)

is measured on a scale on which water freezes at 0 degrees and boils at 100 degrees.

Temperature Fahrenheit (F) is measured on a scale on which water freezes at 32 degrees and

boils at 212 degrees.

Thermal Stability The quality of a product to resist changes caused by heat such as oxidation or

polymerization. Thermal stability measurements are related to the amount of deposits

formed in the engine fuel system upon heating the fuel in a jet aircraft.

Thermohydrometer A hydrometer used in determining fuel density which has a built in thermometer for

simultaneously measuring fuel temperature.

Thief (Sump) Pump A small pump having a suction line which extends to the low point of a fuel storage tank for

the purpose of drawing off water, which may have accumulated.

Tolerance An allowable variance from a specified limit.

Trace An amount large enough to be detected, but not to be measured.

Turbine Fuel A generic term used for various kerosene-based fuels manufactured for use in jet engines.

Vapor pressureThe measure of the pressure exerted by a product in vapor form on the interior of a

container.

Vendor An agent that provides services or products.

Vent An opening in a tank, container or pipe that permits the flow of air and vapor due to changes

in pressure.

Waste Fuel Fuel that can no longer be used for its intended purpose.

Water Defense System A system or device, which detects excess free water in fuel systems and automatically stops

the flow of fuel or sets off an alarm to prevent downstream contamination.

Water Finding Paste A paste which changes color on contact with water.

Water Slug A large amount of free water.

Working Tank

The fuel storage tank being used to supply fuel to aircraft refuelers, refueling trucks or

hydrant systems.

Chapter 5. Waver/Variance

5-1. Waiver/Variance Request

Figure 5-1.1. Waiver/Variance Request

[VENDOR LETTERHEAD]
[DATE]
[AIRLINE]
RE: REQUEST FOR WAIVER /VARIANCE
Airline Manual SubsectionAirport:
Effective Through:
Dear:
Your airline has adopted as a part of its manual certain provisions relating to inspection tests and safety procedures which are intended to preclude the introduction of contaminated or impure fuel into the airline's aircraft. [Vendor] is unable to comply with the designated subsection of the airline's manual referenced above at the designated airport, for the following reason: [Describe why unable to comply]
However, in order to ensure an equivalent level of fuel purity to that prescribed in the manual, [Vendor] proposes to adopt the following inspection and testing procedures at that airport until the date above when it will be in compliance with the manual:
[Describe substitute procedures]
Accordingly, [Vendor] hereby requests that it be granted a waiver /variance by the airline from the provision of the manual referenced above.
[Signature] [Typed name] [Title]

5-2. Grant of Waiver/Variance

Figure 5-2.1. Grant of Waiver/Variance

[AIRLINE LETTERHEAD]
[DATE] [VENDOR]
RE: GRANT OF WAIVER /VARIANCE Manual Subsection Vendor: Airport: Effective Through:
Dear:
By letter dated you have requested a waiver /variance from the provisions of the subsection of the airline's manual referenced above with respect to operations at the specified airport. In addition, you identified procedures or tests you will substitute for those specified in the manual which you state will ensure that the requisite fuel quality and purity will be achieved. The airline finds these procedures acceptable.
Accordingly, the airline hereby grants a waiver /variance from the manual subsection as described in the reference above, subject to the condition that the substitute procedures or tests will remain in effect during the entire period, as described in your letter.
[Signature] [Typed name] [Title]

Chapter 6. Forms

6-1. List of Forms

NOTE: For the best quality prints, full size forms are available in Acrobat Reader (PDF) format [click here].

Requires Adobe Acrobat Reader (from the ATA Publications CD-ROM or the Web

NOTE: Although use of the forms in this document is recommended, it is not required. No variance authority is needed to use other forms if they meet or exceed the task and frequency requirements specified throughout this document. Additional copies of these forms may be reproduced locally.

Table 6-1.1. List of Forms

ATA FORM NO.	FORM
103.01A	Fuel Facility Checks - Daily & Monthly
103.01B	Fuel Facility Checks (Daily) - Sump Results & Filter Differential Pressure
103.01C	Quarterly Fuel Facility Checks
103.01D	Annual Fuel Facility Checks
103.02	Record of Receipt by Transport Truck
103.03	Record of Receipt by Pipelines
103.04A	Aircraft Fueling Equipment Checks - Daily
103.04B	Aircraft Fueling Equipment Checks - Monthly
103.04C	Aircraft Fueling Equipment Checks - Quarterly & Annual
103.05A	Hydrant System Checks - Daily
103.05B	Hydrant System Checks - Monthly
103.05C	Hydrant System Checks - Quarterly
103.06	Jet Fuel Filter Vessel Record
103.07	Storage Tank Inspection & Cleaning Record
103.08	Fuel Quality Test Record

6-2. Test and Check Forms

To print the forms [click here]

Figure 6-2.1. Form 103.01A - Fuel Facility Checks - Daily & Monthly

DAILY 1. GENERAL CONDITION OF TANK YARD 2. SECURITY, FIRE & SAFETY DEFICIENCIES 3. HOSES, SWIVELS, NOZZIES & COUPLERS 1. STATIC REELS, CABLES AND CLAMPS 5. FIRE EXTINGUISHERS	1		3 4	5	6	7	8					STAT	ION				- 1	FACIL	111											
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Figure 6-2.2. Form 103.01B - Fuel Facility Checks (Daily) - Sump Results & Filter Differential Pressure

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Figure 6-2.3.Form 103.01C - Quarterly Fuel Facility Checks

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1. EMERGENCY SHUTOFF SYS.	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#
DATE & CONDITION CODE										
CHECKED BY:										
2. WATER DEFENSE SYSTEMS										
DATE & CONDITION CODE										
CHECKED BY:										
3. TANK HIGH LEVEL CONTROLS										
DATE & CONDITION CODE										
CHECKED BY:										
4. RECLAIM TANK INSPECTION										
DATE & CONDITION CODE										
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Figure 6-2.4.Form 103.01D - Annual Fuel Facility Checks

DATE & CONDITION CODE CHECKED BY I. FILTER ELEMENTS I. FILTER SEPARATOR HEATERS DATE & CONDITION CODE CHECKED BY I. TANK VENTS DATE & CONDITION CODE CHECKED BY I. CATHODIC PROTECTION DATE & CONDITION CODE CHECKED BY I. CATHODIC PROTECTION DATE & CONDITION CODE CHECKED BY I. CATHODIC PROTECTION DATE & CONDITION CODE CHECKED BY CHECKED BY I. CATHODIC PROTECTION DATE & CONDITION CODE CHECKED BY CHECKED BY I. CATHODIC PROTECTION DATE & CONDITION CODE	NNUAL FUEL FA					STATION		FACILITY		YEAR	
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DATE & CONDITION CODE CHECKED BY A FILTER ELEMENTS DATE & CONDITION CODE CHECKED BY A FILTER SEPARATOR HEATERS DATE & CONDITION CODE CHECKED BY A TANK VENTS DATE & CONDITION CODE CHECKED BY CATHODIC PROTECTION DATE & CONDITION CODE CHECKED BY C	DATE & CONDITION CODE	<u> </u>									
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LEILTER/SEPARATOR HEATERS DATE & CONDITION CODE CHECKED BY TANK VENTS DATE & CONDITION CODE CHECKED BY CATHODIC PROTECTION DATE & CONDITION CODE CHECKED BY CHECKED B	DATE & CONDITION CODE										
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STANK VENTS	DATE & CONDITION CODE										
STANK VENTS	CHECKED BY:										
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CATHODIC PROTECTION DATE & CONDITION CODE CHECKED BY LINE STRAINERS											
DATE & CONDITION CODE CHECKED BY LINE STRAINERS											
CHECKED BY. 7. LINE STRAINERS											
. LINE STRAINERS											
DATE & CONDITION CODE									_		
									_		
CHECKED BY: 3. WATER DEFENSE SYSTEM											
DATE & CONDITION CODE											
CHECKED BY:											

Figure 6-2.5. Form 103.02 - Record of Receipt by Transport Truck

JET FUEL STORAGE FACILITY RECORD OF RECEIPT BY TRANSPORT TRUCK ___DATE___ __FACILITY____ RECEIPT NO. REQUIRED CHECKS 4 1 2 3 PRIOR TO RECEIPT DESIGNATE & SUMP RECEIVING TANK GAGE TANK & RECORD VOLUME SET VALVES FOR RECEIVING CONDITION OF OFF-LOAD HOSE BILL OF LADING/DELIVERY TICKET/NO. CORRECT DESTINATION CORRECT GRADE OF FUEL CORRECT VOLUME TRANSPORT TRUCK CONNECT GROUND CABLE COMPARTMENT SEALS **CLEAR & BRIGHT TEST** API GRAVITY, CORRECTED TO 60 ° F FUEL TEMP, ° F (OBSERVED) DURING RECEIPT DIFFERENTAL PRESS.REC.FILT (PSI) SYSTEM FOR LEAKS AFTER RECEIPT **RE-POSITION VALVES DISCONNECT & STOW HOSE** DISCONNECT GROUND CABLE GAGE TANK & RECORD VOLUME WHITE BUCKET CHECK-TANK SUMP WHITE BUCKET CHECK-FILTER SUMPS SIGNATURE OF PERSON PERFORMING CHECKS ✓- SATISFACTORY X- UNSATISFACTORY (ENTER REMARK): [RETAIN THIS FORM ON FILE FOR 12 MONTHS] ATA FORM NO. 103.02 2/16/2004

Figure 6-2.6. Form 103.03 - Record of Receipt by Pipelines

			LDECEIDT	NO		LDECEIDT	NO	
REQUIRED CHECKS	3		RECEIPT FRONT		END	RECEIPT FRONT		END
PRIOR TO RECEIPT								
DESIGNATE & SUMP RECEIV GAUGE TANK	ING IA	ANK						
SET VALVES FOR RECEIVING	3							
PIPELINE COORDINATION								
CORRECT DESTINATION								
CORRECT GRADE OF FUEL CORRECT VOLUME								
CORRECT VOLOME								
DURING RECEIPT								
APPEARANCE								
API GRAVITY@ 60°F								
OBSERVED API GRAVITY								
OBSERVED FUEL TEMPERA		[⊃] F						
MEMBRANE FILTRATION TES	ST							
WATER TEST SYSTEM LEAKS								
DIFFERENTIAL PRESSURE	NO.					+		
	NO.							
	NO.							
	NO.							
[RECORD PSI]	NO.							
	NO.	\vdash	1			 		
	NO.							
AFTER RECEIPT								
RE-POSITION VALVES			-			-		
GAUGE TANK APPEARANCE CHECK - TANK	(SI IME	<u> </u>	-			-		
APPEARANCE CHECK - FILTI			1			1		
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SIGNATURE OF PERSO PERFORMING CHECK								
. E.a. Olamido Olleon								
S = SATISFACTORY C = COM	IMENT							
REMARKS:								
CLIVIATO.								

Figure 6-2.7. Form 103.04A - Aircraft Fueling Equipment Checks - Daily

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																		COMMENT REQUIRED IN REMARKS SECTION); NIU = NOT USED; NIA = NOT APPLICABLE HT, 3 = PARTICULATE, 4 = DIRTY; MOISTURE CONTENT - A = BRIGHT, B = HAZY, C = CLC											COMMENT REQUIRED IN REMARKS SECTION); INJ = NOT USED; INA = NOT APPLICABLE HT, 3 = PARTICULATE, 4 = DIRTY; MOISTURE CONTENT - A = BRIGHT, B = HAZY, C = CLOUDY, D = WET (FREE WATER), E = SURFACTANTS	

Figure 6-2.8. Form 103.04B - Aircraft Fueling Equipment Checks - Monthly

AIRCRAFT FUELING EQUIPMENT MONTHLY	CONDITION CODE	DATE	CHECKED BY
	CONDITION CODE	DATE	CHECKED BY
FILTRATION & FREE WATER TESTS			
STATIC SYSTEM CONTINUITY TEST (RECORD RESULTS) NOZZLE SCREENS	+		
	+		
FUEL HOSES			
SIGNS, LABELS & PLACARDS METER SEALS	+		
	+		
FIRE EXTINGUISHERS EMERGENCY SHUTDOWN SYSTEM	+	+	
	+		
DEADMAN CONTROL SYSTEM	+		
LIFT PLATFORMS			
REFUELING TRUCK INTERIORS	+		
REFUELING TRUCK VENTS & DOME COVERS	+		
REFUELING TRUCKTROUGH DRAINS			
IARKS:			

Figure 6-2.9.Form 103.04C - Aircraft Fueling Equipment Checks - Quarterly & Annual

UARTERLY, SEMI-ANNUAL & ANNUAL AIRCRA	AFT FUELING EQUIPI	WENT CHECKS ISTATION	EQUIPMENT ID#	YEAR	
UARTERLY CHECKS	1ST QTR	2ND QTR	3RD QTR	4TH QTR	
VEHICLE INSPECTION CONDITION CODE					
CHECKED BY & DATE:					
PRESSURE CONTROLS CONDITION CODE					
RECORD PRIMARY & SECONDARY FUEL PRESSURE SETTING (PSI)					
CHECKED BY & DATE:					
WATER DEFENSE SYSTEM - EXTERNAL CHECK CONDITION CODE					
CHECKED BY & DATE:					
INTERNAL VALVE CHECK CONDITION CODE					
CHECKED BY & DATE:					
EMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQUIRE	ED IF USING REUSEABLE CO	OUPLINGS ON FUEL HOSES)			
PERIODIC HOSE PRESSURE TEST	ID#	D#	ID#	D#	ID#
TEST PRESSURE (PSI)					
CONDITION CODE					
CHECKED BY & DATE:					
NNUAL CHECKS					
FILTER ELEMENT CHANGE CONDITION CODE					
CHECKED BY & DATE:					
FUELING & DIFFERENTIAL PRESSURE GAUGES CONDITION CODE					
CHECKED BY & DATE:					
METER CALIBRATION CODE					
CHECKED BY & DATE:					
WATER DEFENSE SYSTEM INSPECTION & TES: CONDITION CODE					
CHECKED BY & DATE:					
EMARKS:					

Figure 6-2.10. Form 103.05A - Hydrant System Checks - Daily

Statio	n:			Termi	nal:					Date:		
			PERFORM	DAILY	USE A	PPLIC	ABLE CON	DITION C	ODES)			
ONDITION = SATISFA = COMME	CTORY	NTS REQU	JIRED IN REM	ARKS)			NOT IN SE					
PIT	# Pit Leaks & Cleanliness	EFS Stations	PIT#	Pit Leaks & Cleanliness	EFS Stations		PIT#	Pit Leaks & Cleanliness	EFS Stations	PIT#	Pit Leaks & Cleanliness	EFS Stations
IGNATURE	i:	SIGNATL	IRE OF PERSO	ON PERF	ORMIN	G TAS	iks	_				
REMARKS	S:											
		2 MONTHS										Form 103.0

Figure 6-2.11. Form 103.05B - Hydrant System Checks - Monthly

Station	<u>:</u>				Ter	minal: (USE APPI	LICABLI	E COND	TION C	ODES)	 Date:				
SATISFAC	TORY	Pr					N/S = N	NOT IN S	ERVICE	E					
COMMEN	[(COMME	INTS RE	QUIRE	D IN REM	<u>MARI</u> 1	(S)	N/A = N	NOT APP	LICABI	<u>-E</u>	 				
VAULT, F GATE O LOW POI DRAIN	Datio TN	Hydrant Valve Assembly	Low Point Drains	Emergency Fuel Shutoff		VAULT, PIT, GATE OR LOW POINT DRAIN#	io	Hydrant Valve Assembly	Low Point Drains	Emergency Fuel Shutoff	VAULT, PIT, GATE OR LOW POINT DRAIN#	Isolation Valve Pits & Control Vaults	Hydrant Valve Assembly	Low Point Drains	Emergency Fuel Shutoff
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MARKS	:	SIGNAT	TURE O)F PERS	ON P	ERFORMING	TASKS								

Figure 6-2.12.Form 103.05C - Hydrant System Checks - Quarterly

QUARTERLY HYDRANT S	YSIEM	CHECKS	<u> </u>	STATION		TERMINAL		QUARTER	YEAR
1. HIGH POINT VENTS	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#
DATE & CONDITION COD	E								
CHECKED B	Y:								
2. SURGE ABSORBERS	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#
DATE & CONDITION COD	E				_				
CHECKED B	Y:								
3. LEAK DETECTION & PIPING ISOLATION SYSTEMS									
DATE & CONDITION COD	E								
CHECKED B	Υ:								

Figure 6-2.13.Form 103.06 - Jet Fuel Filter Vessel Record

All	RPO	RT	_	_												A	GEI	NC		_											_	_
VE	SSE	LN	IFG													DA	TE	LAS	ΤE	LEN	1EN	IT C	IAH	NGE	:_							_
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	24	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	23 22	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	21	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	20	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	19 18	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
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SS	15 14	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
품	13	0			_	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
Ĭ	12	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
DIFFERENTIAL PRESSURE	11 10	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
逆	9	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
莒	8	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	7 6	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	5	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	4	٥	0	٥	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	3 2	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	1	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
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SIGNATURE																																
																ON F			20													

Figure 6-2.14. Form 103.07 - Storage Tank Inspection & Cleaning Record

STORAGE & PRODUCT RECLAMATION TANK INSPECTION AND CLEANING RECORD* AIRPORT_____FACILITY____ TANK NO. DATE INSPECTED CONDITION ACTION SIGNATURE 1 2 3 4 S R M CL NC MT PLACE AN "X" IN ALL BOXES APPLICABLE TO THE TANK CONDITION CONDITION CODES: 1 - CLEAN 2 - TRACE 3 - MODERATE 4 - HEAVY S - SEDIMENT R - RUST M - MICROBIAL GROWTH PLACE AN "X" IN BOX INDICATING ACTION TAKEN OR "P" (PASS) "F" (FAIL) ON MICROBIAL TEST ACTION CODES: CL - CLEANED NC - NOT CLEANED MT - MICROBIAL TEST REMARKS: * RECORD QUARTERLY VISUAL OR MICROBIAL TEST RESULTS FOR PRODUCT RECLAMATION TANKS ON THIS FORM [RETAIN THIS FORM ON FILE INDEFINITELY] ATA FORM NO. 103.07 11/9/2005

Figure 6-2.15. Form 103.08 - Fuel Quality Test Record

	FACILITY DATE		
SAMPLING POINT	MEMBRANE FILTRATION TEST ASTM D-2276	WATER SEPARATION ASTM D-3948	WATER TEST
BEFORE FILTRATION () PARTICULATE () CLAY () FILTER/SEPARATOR () MONITOR ELEMENT \(\Delta P \ \text{psi} \) UNIT NO	DRY RATING SAMPLE SIZE GAL		PPM
AFTER FILTRATION () PARTICULATE () CLAY () FILTER/SEPARATOR () MONITOR ELEMENT Δ P psi UNIT NO	DRY RATING SAMPLE SIZE GAL		PPM
NOTES:			

Annex 1.

References	
[API 1529]	Aviation Fueling Hose, American Petroleum Institute, (www.api.org), Washington, DC.
[API/IP 1540]	Design, Construction, Operation and Maintenance of Aviation Fuelling Facilities, American Petroleum Institute, (www.api.org), Washington, DC.
[API/IP 1542]	Identification Markings for Dedicated Aviation Fuel, Manufacturing and Distribution Facilities, Airport Storage and Mobile, Fueling Equipment, American Petroleum Institute (www.api.org), Washington, DC.
[API/IP 1581]	Specification and Qualification Procedures for Aviation Jet Fuel Filter/Separators, American Petroleum Institute (www.api.org), Washington, DC.
[API/IP 1582]	Similarity Calculations and Software for Aviation Jet Fuel Filter/Separators, American Petroleum Institute (www.api.org), Washington, DC.
[IP 1583]	Specification and Qualification Procedures for Aviation Fuel Filter Monitors with Absorbent Type Elements, Energy Institute (www.energyinst.org.uk), London, UK.
[API/IP 1590]	Specification and qualification procedures for aviation fuel microfilters, American Petroleum Institute (www.api.org), Washington, DC.
[ASTM D56]	Standard Test Method for Flash Point by Tag Closed Tester, ASTM D56-98a, American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D86]	Standard Test Method for Distillation of Petroleum Products at Atmospheric Pressure, D86-99ae1 American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D130]	Standard Test Method for Detection of Copper Corrosion from Petroleum Products by the Copper Strip Tarnish Test, American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D381]	Standard Test Method for Gum Content in Fuels by Jet Evaporation, D381-99, American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D1094]	Standard Test Method for Water Reaction of Aviation Fuels American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D1250]	Standard Guide for Petroleum Measurement Tables, American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D1298]	Standard Test Method for Density (API Standard 2540, Chapter 9, Section 1) Relative

	Density (Specific Gravity) or API Gravity of Crude Petroleum and Liquid Petroleum Products by Hydrometer Method. American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D1655]	Standard Specification for Aviation Turbine Fuels, ASTM D1655-99a, American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D2276]	Standard Test Method for Particulate Contaminant in Aviation Fuel by Line Sampling, American Society for Testing and Materials American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D2386]	Standard Test Method for Freezing Point of Aviation Fuels, American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D3828]	Standard Test Methods for Flash Point by Small Scale Closed Tester, American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D3948]	Standard Test Method for Determining Water Separation Characteristics of Aviation Turbine Fuels by Portable Separameter, American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D4306]	Standard Practice for Aviation Fuel Sample Containers for Tests Affected by Trace Contamination, American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D5001]	Standard Test Method for Measurement of Lubricity of Aviation Turbine Fuels by the Ball-on-Cylinder Lubricity Evaluator (BOCLE), American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM D6469]	Standard Guide for Microbial Contamination in Fuels and Fuel Systems. American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM MNL5]	Aviation Fuel Quality Control Procedures. Edited by R. Waite. American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM E1]	Standard Specification for ASTM Thermometers, American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[ASTM E100]	Standard Specification for ASTM Hydrometers, American Society for Testing and Materials (www.astm.org), West Conshohocken, Pennsylvania.
[NFPA 10]	Standard for Portable Fire Extinguishers, NFPA, (www.nfpa.org), Quincy, MA
[NFPA 407]	Standard for Aircraft Fuel Servicing, NFPA, (www.nfpa.org), Quincy, MA

FUEL FACILITY CHECKS													STAT	ION					FACIL	_ITY					MON	TH		YEAF	₹		
DAILY	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31
1. GENERAL CONDITION OF TANK YARD																															
2. SECURITY, FIRE & SAFETY DEFICIENCIES																															
3. HOSES, SWIVELS, NOZZLES & COUPLERS																															
4. STATIC REELS, CABLES AND CLAMPS																															
5. FIRE EXTINGUISHERS																															
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	1	1																													
		1																													
		1																													
IDENTIFICATION OF PERSON PERFORMING																															
TASKS OR PERSON ACCEPTING RESPONSIBILITY																															İ
THAT TASKS WERE PERFORMED																															
	-	-																													
MONTHLY		D/	ATE		SIGN	NATUR	RE						HTNC	ILY						D,	ATE		SIGN	NATUF	RE				N CC		í
1. FILTRATION & FREE WATER TEST	-							6. FII	RE EX	TING	JISHE	RS																	CTORY	1	
2. BONDING CABLE CONTINUITY	-	<u> </u>		ļ																						С	= CON	MEN.	Т		
3. NOZZLE SCREENS																										N/U :	= NOT	USE)		
4. SIGNS, LABELS & PLACARDS																										N/A :	= NOT	APPL	ICABL	LE	
5. FLOATING SUCTION																															
NOTE - RECORD SUMP RESULTS & FILTER DIFFERE	NTIAL	PRES	SURE	ON S	EPAR	ATE F	ORM.																								
REMARKS:																															

RETAIN ON FILE FOR 12 MONTHS

ATA FORM 103.01A

FUEL FACILITY CHECKS													STAT	TION					FACI	LITY					MON	TH		YEAF	}		
STORAGE TANK & PRODUCT RECLAM	ATI	ON	SU	MP																											
RECORD RESULTS - SEE NOTE #1	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31
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FILTER SUMP RESULTS																															
RECORD RESULTS - SEE NOTE #1	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31
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FILTED DIFFEDENTIAL DRESSURE																															
FILTER DIFFERENTIAL PRESSURE RECORD DIFFERENTIAL PRESSURE	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31
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NOTE #1 - RATING OF SUMP SAMPLES: SOLIDS - (1) CLEAR (2) SLIGHT (3) PARTICULATE (4) DIRTY; WATER-(A) BRIGHT (B) HAZY (C) CLOUDY (D) WET (E) SURFACTANTS RETAIN ON FILE FOR 12 MONTHS

CHECKED BY: I. TANK HIGH LEVEL CONTROLS DATE & CONDITION CODE CHECKED BY: RECLAIM TANK INSPECTION DATE & CONDITION CODE RECORD INSP. RESULTS ON 103.07 CHECKED BY: SEMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQUIRED IF USING REUSEABLE COUPLINGS ON FUEL HOSES) PERIODIC HOSE PRESS. TEST ID # ID	QUARTERLY FUE	L FACI	ILITY CH	ECKS		STATION		FACILITY		QUARTER	YEAR
CHECKED BY: WATER DEFENSE SYSTEMS DATE & CONDITION CODE CHECKED BY: TANK HIGH LEVEL CONTROLS DATE & CONDITION CODE CHECKED BY: RECLAIM TANK INSPECTION DATE & CONDITION CODE CHECKED BY: CHECKED BY: CHECKED BY: CHECKED BY: DATE & CONDITION CODE DATE & CONDITION CODE CHECKED BY: CHECKED BY: CHECKED BY: DATE & CONDITION CODE CHECKED BY: DATE & CONDITION CODE CHECKED BY: CHECKED BY: DATE & CONDITION CODE CHECKED BY: CHECKED BY: DATE & CONDITION CODE CHECKED BY: CHECKED BY	. EMERGENCY SHUTOFF SYS.	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#
WATER DEFENSE SYSTEMS DATE & CONDITION CODE CHECKED BY: TANK HIGH LEVEL CONTROLS DATE & CONDITION CODE CHECKED BY: RECLAIM TANK INSPECTION DATE & CONDITION CODE CHECKED BY: CHECKED BY: CHECKED BY: CHECKED BY: CHECKED BY: DATE & CONDITION CODE CHECKED BY: CHECKED BY: CHECKED BY: DATE & CONDITION CODE CHECKED BY: DATE & CONDITION CODE											
DATE & CONDITION CODE CHECKED BY: I. TANK HIGH LEVEL CONTROLS DATE & CONDITION CODE CHECKED BY: RECLAIM TANK INSPECTION DATE & CONDITION CODE RECORD INSP. RESULTS ON 103.07 CHECKED BY: SEMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQUIRED IF USING REUSEABLE COUPLINGS ON FUEL HOSES) PERIODIC HOSE PRESS. TEST ID # ID	CHECKED BY:	, <u>=</u>									
CHECKED BY: 3. TANK HIGH LEVEL CONTROLS DATE & CONDITION CODE CHECKED BY: 4. RECLAIM TANK INSPECTION DATE & CONDITION CODE RECORD INSP. RESULTS ON 103.07 CHECKED BY: 1. PERIODIC HOSE PRESS. TEST D# ID# ID# ID# ID# ID# ID# ID# ID# ID# I	2. WATER DEFENSE SYSTEMS										
3. TANK HIGH LEVEL CONTROLS DATE & CONDITION CODE CHECKED BY: 4. RECLAIM TANK INSPECTION DATE & CONDITION CODE RECORD INSP. RESULTS ON 103.07 CHECKED BY: SEMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQUIRED IF USING REUSEABLE COUPLINGS ON FUEL HOSES) 1. PERIODIC HOSE PRESS. TEST ID # ID	DATE & CONDITION CODE	:									
DATE & CONDITION CODE CHECKED BY: 4. RECLAIM TANK INSPECTION DATE & CONDITION CODE RECORD INSP. RESULTS ON 103.07 CHECKED BY: 5. PERIODIC HOSE PRESS. TEST ID#	CHECKED BY:	5									
CHECKED BY: 4. RECLAIM TANK INSPECTION DATE & CONDITION CODE RECORD INSP. RESULTS ON 103.07 CHECKED BY: SEMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQUIRED IF USING REUSEABLE COUPLINGS ON FUEL HOSES) 1. PERIODIC HOSE PRESS. TEST D# D#	3. TANK HIGH LEVEL CONTROLS										
A. RECLAIM TANK INSPECTION DATE & CONDITION CODE RECORD INSP. RESULTS ON 103.07 CHECKED BY: SEMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQUIRED IF USING REUSEABLE COUPLINGS ON FUEL HOSES) 1. PERIODIC HOSE PRESS. TEST D # ID	DATE & CONDITION CODE										
DATE & CONDITION CODE RECORD INSP. RESULTS ON 103.07 CHECKED BY: SEMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQUIRED IF USING REUSEABLE COUPLINGS ON FUEL HOSES) 1. PERIODIC HOSE PRESS. TEST ID #	CHECKED BY:	E									
SEMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQUIRED IF USING REUSEABLE COUPLINGS ON FUEL HOSES) 1. PERIODIC HOSE PRESS. TEST D# ID# ID#	4. RECLAIM TANK INSPECTION										
SEMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQUIRED IF USING REUSEABLE COUPLINGS ON FUEL HOSES) 1. PERIODIC HOSE PRESS. TEST ID # ID #	DATE & CONDITION CODE	<u> </u>									
SEMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQUIRED IF USING REUSEABLE COUPLINGS ON FUEL HOSES) 1. PERIODIC HOSE PRESS. TEST ID # ID #	RECORD INSP. RESULTS ON 103.07	,									
SEMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQUIRED IF USING REUSEABLE COUPLINGS ON FUEL HOSES) 1. PERIODIC HOSE PRESS. TEST ID # ID #											
DATE & CONDITION CODE CHECKED BY:			CILITY C	HECKS (C	ONLY REQU	IRED IF US	ING REUSE	ABLE COU	PLINGS ON	FUEL HOS	ES)
CHECKED BY:	SEMI-ANNUAL FU	EL FA									
CHECKED BY:	SEMI-ANNUAL FU 1. PERIODIC HOSE PRESS. TEST	EL FA									
	SEMI-ANNUAL FU 1. PERIODIC HOSE PRESS. TEST TEST PRESSURE (PSI)	EL FA									
	SEMI-ANNUAL FU 1. PERIODIC HOSE PRESS. TEST TEST PRESSURE (PSI) DATE & CONDITION CODE	EL FA									
	SEMI-ANNUAL FU 1. PERIODIC HOSE PRESS. TEST TEST PRESSURE (PSI) DATE & CONDITION CODE	EL FA									
	SEMI-ANNUAL FU 1. PERIODIC HOSE PRESS. TEST TEST PRESSURE (PSI) DATE & CONDITION CODE CHECKED BY:	EL FA									
	SEMI-ANNUAL FU 1. PERIODIC HOSE PRESS. TEST TEST PRESSURE (PSI) DATE & CONDITION CODE CHECKED BY:	EL FA									
	SEMI-ANNUAL FU 1. PERIODIC HOSE PRESS. TEST TEST PRESSURE (PSI) DATE & CONDITION CODE CHECKED BY:	EL FA									
	SEMI-ANNUAL FU 1. PERIODIC HOSE PRESS. TEST TEST PRESSURE (PSI) DATE & CONDITION CODE CHECKED BY:	EL FA									
	SEMI-ANNUAL FU 1. PERIODIC HOSE PRESS. TEST TEST PRESSURE (PSI) DATE & CONDITION CODE CHECKED BY:	EL FA									
	SEMI-ANNUAL FU 1. PERIODIC HOSE PRESS. TEST TEST PRESSURE (PSI) DATE & CONDITION CODE CHECKED BY:	EL FA									

ANNUAL FUEL FA	CILITY	CHECKS	5		STATION		FACILITY		YEAR	
1. STORAGE TANK INTERIORS	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#
DATE & CONDITION CODE										
CHECKED BY:										
2. DIFFERENTIAL PRESS GAUGES										
DATE & CONDITION CODE										
CHECKED BY:										
3. FILTER ELEMENTS										
DATE & CONDITION CODE										
CHECKED BY:										
4. FILTER/SEPARATOR HEATERS										
DATE & CONDITION CODE										
CHECKED BY:										
5. TANK VENTS										
DATE & CONDITION CODE										
CHECKED BY:										
6. CATHODIC PROTECTION										
DATE & CONDITION CODE										
CHECKED BY:										
7. LINE STRAINERS										
DATE & CONDITION CODE										
CHECKED BY:										
B. WATER DEFENSE SYSTEM										
DATE & CONDITION CODE										
CHECKED BY:										
EMARKS:										

JET FUEL STORAGE FACILITY

RECORD OF RECEIPT BY TRANSPORT TRUCK

AIRPORT	FACILITY			DATE	
			RECEIPT NO.		
REQUIRED CHECKS	1	2	3	4	5
PRIOR TO RECEIPT					
DESIGNATE & SUMP RECEIVING TANK					
GAGE TANK & RECORD VOLUME					
SET VALVES FOR RECEIVING					
CONDITION OF OFF-LOAD HOSE					
BILL OF LADING/DELIVERY TICKET/NO.					
CORRECT DESTINATION					
CORRECT GRADE OF FUEL					
CORRECT VOLUME					
TRANSPORT TRUCK					
CONNECT GROUND CABLE					
CONNECT GROUND CABLE COMPARTMENT SEALS					
CLEAR & BRIGHT TEST					
API GRAVITY, CORRECTED TO 60 ° F					
FUEL TEMP, ° F (OBSERVED)					
DURING RECEIPT					
DIFFERENTAL PRESS.REC.FILT (PSI)					
SYSTEM FOR LEAKS					
0.0.2					
AFTER RECEIPT					
RE-POSITION VALVES					
DISCONNECT & STOW HOSE					
DISCONNECT GROUND CABLE					
GAGE TANK & RECORD VOLUME					
WHITE BUCKET CHECK-TANK SUMP					
WHITE BUCKET CHECK-FILTER SUMPS					
SIGNATURE OF PERSON PERFORMING CHECKS					
✓- SATISFACTORY X- UNSAT	ISFACTORY (ENTER REMAF	 RK):		

[RETAIN THIS FORM ON FILE FOR 12 MONTHS]

JET FUEL STORAGE FACILITY RECORD OF RECEIPT BY PIPELINES

AIRPORT:	FA	CILI	TY:			DATE:		
			RECEIPT	NO		RECEIPT	NO	
REQUIRED CHECKS			FRONT		END		MIDDLE	END
PRIOR TO RECEIPT								
DESIGNATE & SUMP RECEIVI	NG TAN	K						
GAUGE TANK				1				
SET VALVES FOR RECEIVING								
PIPELINE COORDINATION								
CORRECT DESTINATION				4				
CORRECT GRADE OF FUEL				4				
CORRECT VOLUME								
DURING RECEIPT								
APPEARANCE				1 1		<u> </u>		
				+ +				
API GRAVITY@ 60°F								
OBSERVED API GRAVITY								
OBSERVED FUEL TEMPERAT								
MEMBRANE FILTRATION TES	Т							
WATER TEST								
SYSTEM LEAKS								
DIFFERENTIAL PRESSURE	NO.							
	NO.							
	NO.							
	NO.							
[RECORD PSI]	NO.							
	NO.							
	NO.							
	NO.							
AFTER RECEIPT								
RE-POSITION VALVES				1 1				
GAUGE TANK				1 1				
APPEARANCE CHECK - TANK				1 1				
APPEARANCE CHECK - FILTE	R SUMP	S						
SIGNATURE OF PERSO PERFORMING CHECKS								
S = SATISFACTORY C = COM	MENT			-		-		
REMARKS:								

AIRCRAFT FUELING EQUI	PM	EN	IT (CHI	EC	KS					STAT	TION						EQUI	PMEN	IT ID#					мои	тн		YEAF	₹		
DAILY	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31
1. GENERAL CONDITION																															
2. FILTER SUMPS - RECORD RATINGS																															
3. FILTER DIFFERENTIAL PRESSURE - Record PSI																															
4. DEADMAN CONTROLS																															
5. SAFETY INTERLOCKS																															
6. NOZZLE FUELING PRESSURE - Record PSI																															
7. HOSES, NOZZLES & SWIVELS																															
8. STATIC REELS, CABLES & CLAMPS																															
9. LIFT PLATFORMS																															
10. FIRE EXTINGUISHERS																															
11. SURGE/WASTE TANKS																															
12. AIR TANKS																															
13. REFUELING TRUCK TROUGHS																															
14. REFUELING TRUCK SUMPS - RECORD RATINGS																															
15. REFUELING TRUCK BOTTOM LOADING PRE-CHECK																															
IDENDIFICATION OF PERSON PERFORMING TASKS OR PERSON ACCEPTING RESPONSIBILITY THAT TASKS WERE PERFORMED																															
CONDITION CODES: S = SATISFACTORY; C = COMMENT RATING OF SUMP SAMPLES: SOLIDS - 1 = CLEAN, 2 = SI REMARKS:	•																		.OUD	Y, D	= WE	T (FR	EE W	I /ATEF	I	= SU	RFAC	TANT	s		

RETAIN ON FILE FOR 12 MONTHS ATA FORM 103.04A

MONTHLY	CHECKS	STATION	EQUIPMENT ID#	MONTH YEAR
	CONDITION CODE	DATE		CHECKED BY
1. FILTRATION & FREE WATER TESTS				
2. STATIC SYSTEM CONTINUITY TEST (RECORD RESULTS)				
3. NOZZLE SCREENS				
4. FUEL HOSES				
5. SIGNS, LABELS & PLACARDS				
6. METER SEALS				
7. FIRE EXTINGUISHERS				
3. EMERGENCY SHUTDOWN SYSTEM				
9. DEADMAN CONTROL SYSTEM				
10. LIFT PLATFORMS				
1. REFUELING TRUCK INTERIORS				
2. REFUELING TRUCK VENTS & DOME COVERS				
3. REFUELING TRUCK TROUGH DRAINS				

11/9/05

QUARTERLY, SEMI-ANNUAL & ANNUAL AIRC	RAFT FUELING EQUIP	MENT CHECKS STATION	EQUIPMENT ID#	YEAR	
QUARTERLY CHECKS	1ST QTR	2ND QTR	3RD QTR	4TH QTR	
VEHICLE INSPECTION CONDITION CODE					
CHECKED BY & DATE:	<u> </u>				
2. PRESSURE CONTROLS CONDITION CODE					
RECORD PRIMARY & SECONDARY FUEL PRESSURE SETTING (PSI)					
CHECKED BY & DATE:					
3. WATER DEFENSE SYSTEM - EXTERNAL CHECK CONDITION CODE					
CHECKED BY & DATE:					
4. INTERNAL VALVE CHECK CONDITION CODE					
CHECKED BY & DATE:	I.				
SEMI-ANNUAL FUEL FACILITY CHECKS (ONLY REQU	UIRED IF USING REUSEABLE C	OUPLINGS ON FUEL HOSES)			
1. PERIODIC HOSE PRESSURE TEST	ID#	ID#	ID#	ID#	ID#
TEST PRESSURE (PSI)					
CONDITION CODE					
CHECKED BY & DATE:					
ANNUAL CHECKS					
1. FILTER ELEMENT CHANGE CONDITION CODE					
CHECKED BY & DATE:					
2. FUELING & DIFFERENTIAL PRESSURE GAUGES CONDITION CODE					
CHECKED BY & DATE:					
3. METER CALIBRATION CONDITION CODE					
CHECKED BY & DATE:					
4. WATER DEFENSE SYSTEM INSPECTION & TES CONDITION CODE					
CHECKED BY & DATE:					
REMARKS:					

			DAIL	/ HYDI	RAN	T S	YSTEM	1 CHE	ECKS	<u>}</u>				
Station:				Termi	nal:						Date:			
CONDITION COI S = SATISFACTO C = COMMENT (ORY	NTS REQ			USE A	N/S =	ABLE CONI NOT IN SE	RVICE	ODES)					
PIT#	Pit Leaks & Cleanliness	EFS Stations	PIT#	: Leaks & eanliness	EFS Stations		PIT#	Pit Leaks & Cleanliness	EFS Stations		PIT#	Pit Leaks & Cleanliness	EFS Stations	
										-				
SIGNATURE:		SIGNATU	URE OF PER	SON PERF	ORMIN	IG TAS	SKS	-						

Station:				7	Terminal:					Date:				
					(USE APPI	LICABLI	E CONDI	TION C	ODES)					
NDITION COD SATISFACTO COMMENT (0	ORY	NTS RE	QUIRE	D IN REM	ARKS)		IOT IN S							
VAULT, PIT, GATE OR LOW POINT DRAIN #	ਰ હ	Hydrant Valve Assembly	Low Point Drains	Emergency Fuel Shutoff	VAULT, PIT, GATE OR LOW POINT DRAIN #	ਰ હ	Hydrant Valve Assembly	Low Point Drains	Emergency Fuel Shutoff	VAULT, PIT, GATE OR LOW POINT DRAIN #	Isolation Valve Pits & Control Vaults	Hydrant Valve Assembly	Low Point Drains	Emergency Fuel Shutoff
NATURE:		SIGNAT	URE C	F PERSO	N PERFORMING	TASKS								

QUARTERLY HYDRANT SY	STEM C	HECKS		STATION		TERMINAL		QUARTER	YEAR
1. HIGH POINT VENTS	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#
DATE & CONDITION CODE									
CHECKED BY:	i.								
2. SURGE ABSORBERS	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#	ID#
DATE & CONDITION CODE									
CHECKED BY:									
3. LEAK DETECTION & PIPING ISOLATION SYSTEMS									
DATE & CONDITION CODE									
CHECKED BY:									
REMARKS:									

JET FUEL FILTER VESSEL RECORD

AIRPORT	AGENCY
VESSEL MFG.	DATE LAST ELEMENT CHANGE
MODEL NO.	DATE LAST INTERIOR INSP.
FIXED UNIT NO	P / N ELEMENTS INSTALLED
MOBILE UNIT NO	
RATED FLOW (GPM)	DATE WATER DEFENSE CHECK

DIFFERENTIAL PRESSURE RECORD

										DIFFERENTIAL PRESSURE losil	H	5	₹	۳ ا	ES		W	<u>s</u>	<u>-</u> =						
SIGNATURE	ΑY	Ċ		2	2	5	6	7		. 9	10	11	12	13	12	15	16	18 17	19	20	21	22	23	24	25
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	3	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	4	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	2	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	9	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	7	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	٥	0	0	0	0	0
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	6	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	10	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	11	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	12	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	13	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	14	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	٥	0	0	0	0	0
	15	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	16	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	17	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	18	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	19	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
7	20	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
7	21	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
,	22	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
,	23	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	24	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
,	25	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
,	26	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
7	27	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	28	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	٥	0	0	0	0	0
	29	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	30	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	31	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0

MONTH _____ 20 ____

[RETAIN THIS FORM ON FILE FOR 12 MONTHS]

STORAGE & PRODUCT RECLAMATION TANK INSPECTION AND CLEANING RECORD*

AIRPORT_____FACILITY____

INSPECTED			CON	NDITI	ON				ACTIO	N	SIGNATURE
	1	2	3	4	S	R	М	CL	NC	MT	OIOII/TI OILE
	<u> </u>			<u> </u>							
PLACE AN "X"	IN ALL	. BOXI	ES API	PLICA	BLE T	ОТНЕ	ETANI	K CONE	DITION		
CONDITION	COD	ES:	1 – C	CLEA	N	2 –	TRAC	CE	3 – MC	DERAT	ΓE 4 – HEAVY
			S-S	EDIN	IENT	R –	RUS	Т	M - M	ICROBI.	AL GROWTH
PLACE AN "X"	IN BO	X INDI	CATIN	G AC1	ION T	AKEN	OR "I	P" (PAS	SS) "F" (F	AIL) ON	MICROBIAL TEST
											IICROBIAL TEST
7101101100			-					·			
REMARKS:											

[RETAIN THIS FORM ON FILE INDEFINITELY]

ATA FORM NO. 103.07 11/9/2005

THIS FORM

FUEL QUALITY TEST RECORD

AIRPORT	FACILITY
AGENCY	DATE

SAMPLING POINT	MEMBRANE FILTRATION TEST SEPARATION ASTM D-2276 ASTM D-394	
BEFORE FILTRATION () PARTICULATE () CLAY () FILTER/SEPARATOR () MONITOR ELEMENT $\Delta P \underline{\hspace{1cm}} psi$ UNIT NO	MEMBRANE DRY RATING SAMPLE SIZE GAL	PPM
AFTER FILTRATION () PARTICULATE () CLAY () FILTER/SEPARATOR () MONITOR ELEMENT \[\Delta \text{P} \qquad \text{psi} \] UNIT NO	MEMBRANE DRY RATING SAMPLE SIZE GAL	PPM

NOTES:			

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