

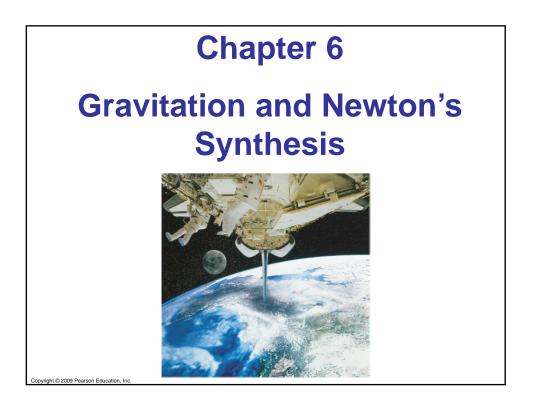
Lecture PowerPoints

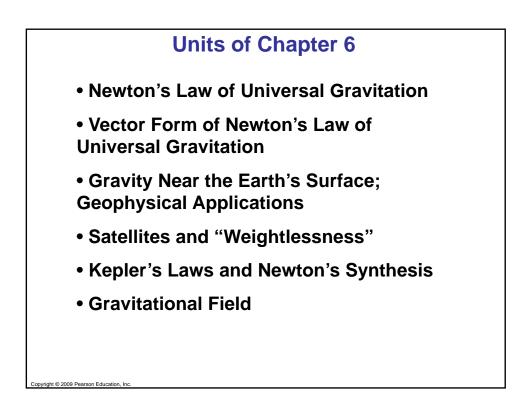
Chapter 6

Physics for Scientists and Engineers, with Modern Physics, 4th edition

Giancoli

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Opening Question

A space station revolves around the Earth as a satellite, 100 km above the Earth's surface. What is the net force on an astronaut at rest inside the space station?

(a) Equal to her weight on Earth.

(b) A little less than her weight on Earth.

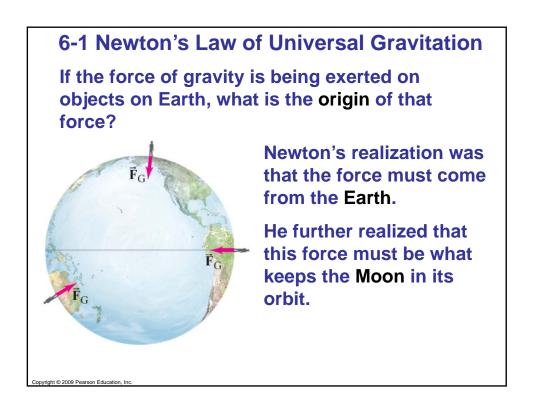
(c) Less than half her weight on Earth.

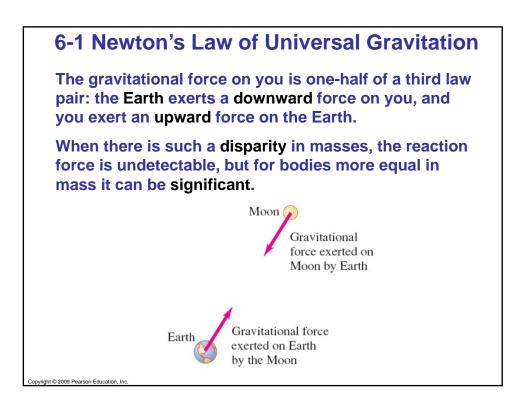
(d) Zero (she is weightless).

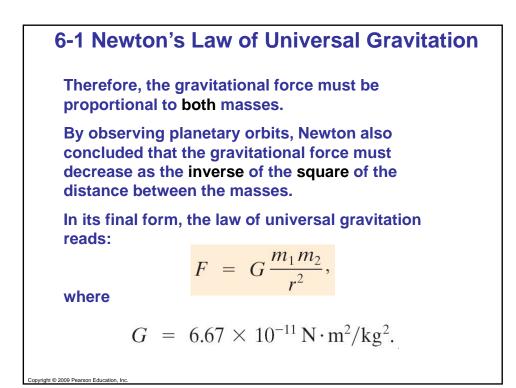
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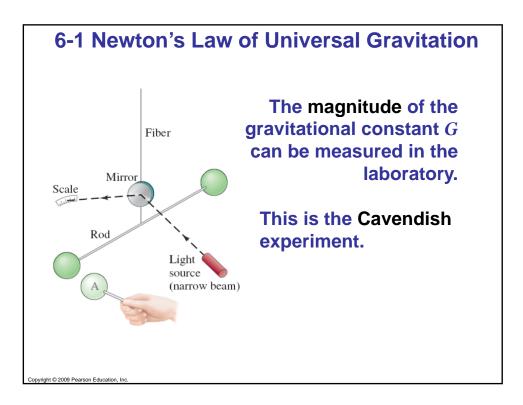
(e) Somewhat larger than her weight on earth.

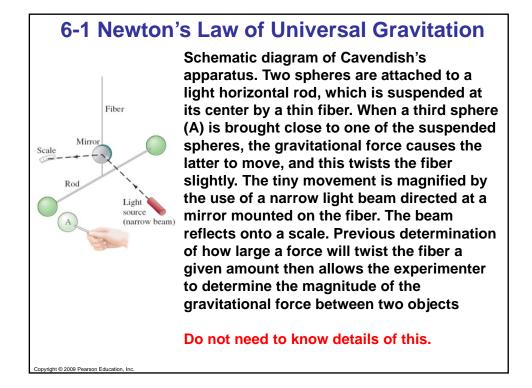
In fact only about 3% less



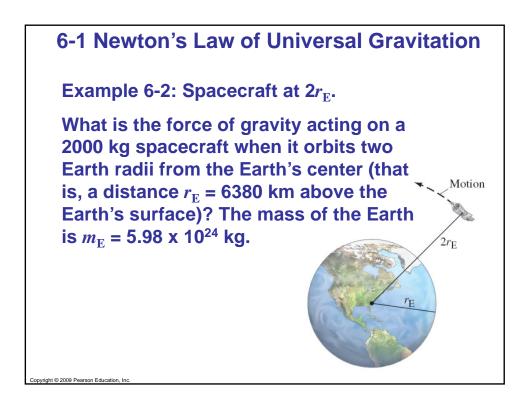




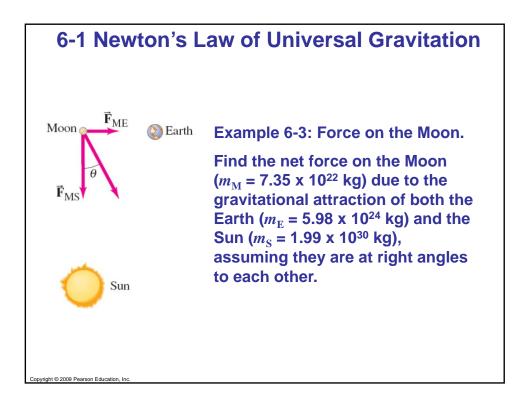


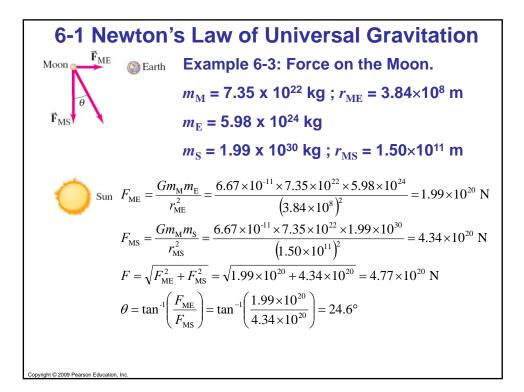


6-1 Newton's Law of Universal Gravitation Example 6-1: Can you attract another person gravitationally? A 50 kg person and a 70 kg person are sitting on a bench about 0.5 m apart. Estimate the magnitude of the gravitational force each exerts on the other. $F = \frac{Gm_1m_2}{r^2}$ $= \frac{6.67 \times 10^{-11} \times 50 \times 70}{0.5^2} = 9.3 \times 10^{-7} \approx 10^{-6} \text{ N}$ This could not be detected without extremely sensitive instruments

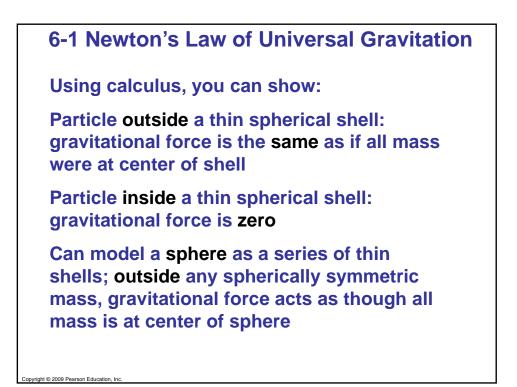


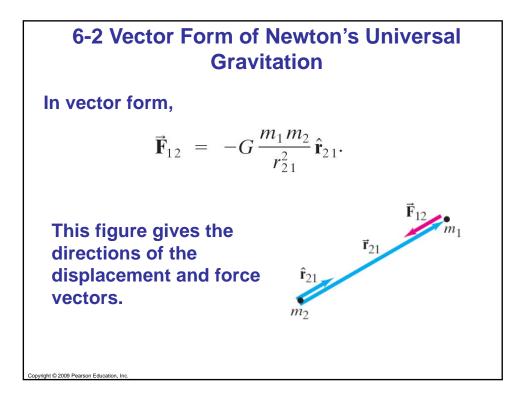
Example 6-2: Spacecraft at $2r_{E}$. What is the force of gravity acting on a 2000 kg spacecraft when it orbits two Earth radii from the Earth's center (that is, a distance $r_{E} = 6380$ km above the Earth's surface)? The mass of the Earth is $m_{E} = 5.98 \times 10^{24}$ kg. Hard Way: $F = \frac{Gm_{E}m_{S}}{r_{S}^{2}}$ $(r_{S} = 2r_{E})$ $= \frac{6.67 \times 10^{-11} \times 5.980 \times 10^{24} \times 2000}{(2 \times 6.380 \times 10^{6})^{2}} = 4900$ N Easy Way: Since $F \propto \frac{1}{r^{2}}$ If $r \rightarrow 2r \implies F \rightarrow \frac{1}{2^{2}} = \frac{1}{4}$ $\therefore F = \frac{1}{4}mg = \frac{2000 \times 9.8}{4} = 4900$ N

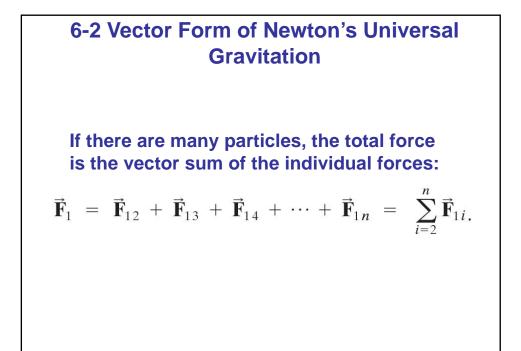




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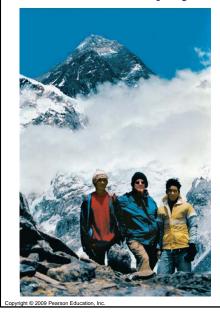


6-3 Gravity Near the Earth's Surface; Geophysical Applications Now we can relate the gravitational constant to the local acceleration of gravity. We know that, on the surface of the Earth: $mg = G \frac{mm_E}{r_E^2}$. Solving for g gives: $g = G \frac{m_E}{r_E^2}$. Now, knowing g and the radius of the Earth, the mass of the Earth can be calculated: $gr_E^2 = 9.80 \times (6.38 \times 10^6)^2$

$$m_{\rm E} = \frac{gr_{\rm E}^2}{G} = \frac{9.80 \times (6.38 \times 10^6)^2}{6.67 \times 10^{-11}} = 5.98 \times 10^{24} \,\rm kg$$

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6-3 Gravity Near the Earth's Surface; Geophysical Applications



Example 6-4: Gravity on Everest.

Estimate the effective value of g on the top of Mt. Everest, 8850 m above sea level. That is, what is the acceleration due to gravity of objects allowed to fall freely at this altitude?

6-3 Gravity Near the Earth's Surface; Geophysical Applications

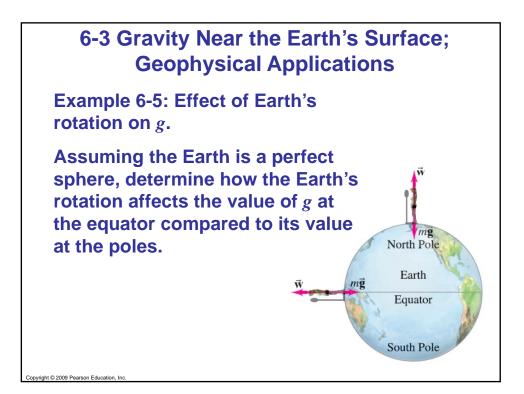
Example 6-4: Gravity on Everest. Mt. Everest: $h = 8850 \text{ m} = 8.85 \text{ x} 10^3 \text{ m}$ above sea level.

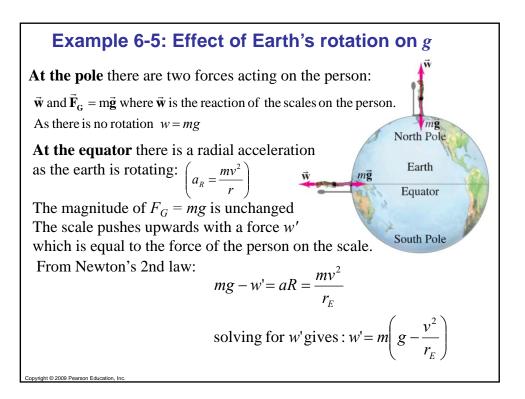
 $r_{\rm E}$ = 6380 km = 6.380 x 10⁶ m

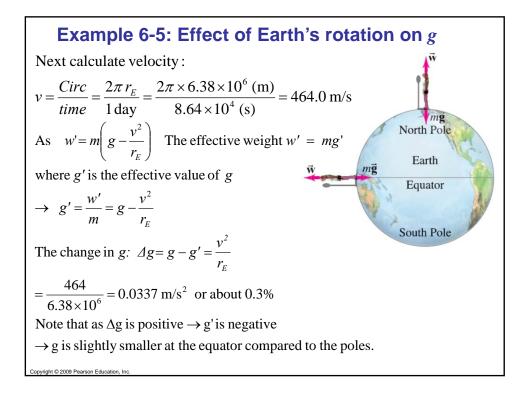
 $r = r_{\rm E} + h = 6.380 \times 10^6 + 8.85 \times 10^3 = 6.389 \times 10^6 \,{\rm m}$

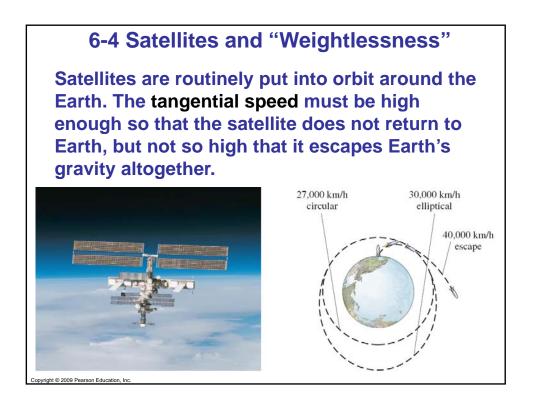
From previous example: $g = \frac{Gm_{\rm E}}{r^2}$ = $\frac{6.67 \times 10^{-11} \times 5.98 \times 10^{24}}{(6.389 \times 10^6)^2} = 9.77 \text{ m/s}^2$

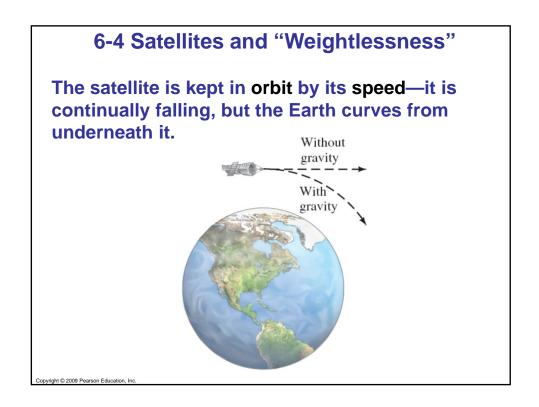
6-3		·	r the Earth's Surface; cal Applications
TABLE 6–1 Acceleration at Various Le			The acceleration due to
Location	Elevation (m)	g (m/s²)	gravity varies over the Earth's surface due to
New York	0	9.803	altitude, local geology,
San Francisco	0	9.800	
Denver	1650	9.796	and the shape of the
Pikes Peak	4300	9.789	Earth, which is not quite
Sydney, Australia	0	9.798	spherical.
Equator	0	9.780	
North Pole (calculated)	0	9.832	











Example 6-6: Geosynchronous satellite. (a) Relation between v and $r [r = r_E + h;$ where h = altitude]For circular orbit : $F_C = \frac{mv^2}{r} = \frac{Gmm_E}{r_2} \rightarrow \therefore v^2 = \frac{Gm_E}{r}$ For geosynchronous orbit : $v = \frac{Circ}{time} = \frac{2\pi r}{t} = \frac{2\pi r}{86400}$ (1 day = 86400 s) Substitute into above equation and solve for r $r = \frac{Gm_E}{v^2} = \frac{6.67 \times 10^{-11} \times 5.98 \times 10^{24} \times 86400^2}{4\pi^2 r^2} = \frac{7.542 \times 10^{22}}{r^2}$ $\therefore r = \sqrt[3]{7.542 \times 10^{22}} = 4.23 \times 10^7 \text{ m} = 42300 \text{ km}$ $\therefore h = r - r_E = 42300 - 6380 = 35920 \text{ km} (\approx 36000 \text{ km})$

Example 6-6: Geosynchronous satellite. (b) the satellite's speed. From previous slide : $v = \sqrt{\frac{Gm_E}{r}} = \sqrt{\frac{6.67 \times 10^{-11} \times 5.98 \times 10^{24}}{4.23 \times 10^7}} = 3070 \text{ m/s}$ (c) Compare to the speed of a satellite orbiting 200 km above Earth's surface. From part (b) we see that $v \propto \sqrt{\frac{1}{r}}$ $v' = v \sqrt{\frac{r}{r'}} = 3070 \sqrt{\frac{42300}{6380 + 200}} = 7780 \text{ m/s}$

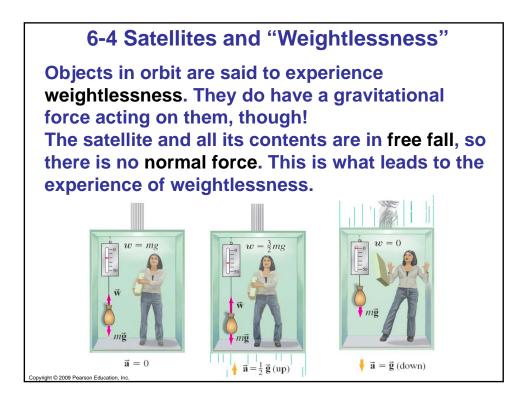
6-4 Satellites and "Weightlessness"

Conceptual Example 6-7: Catching a satellite.

You are an astronaut in the space shuttle pursuing a satellite in need of repair. You find yourself in a circular orbit of the same radius as the satellite, but 30 km behind it. How will you catch up with it?

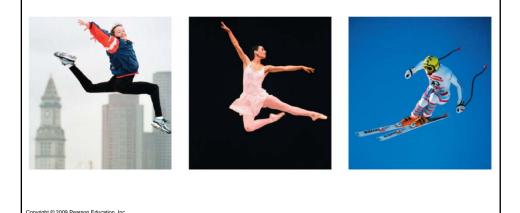
As $r = \frac{Gm_E}{v^2}$ If v increases – r must decrease or if v decreases – r must increase

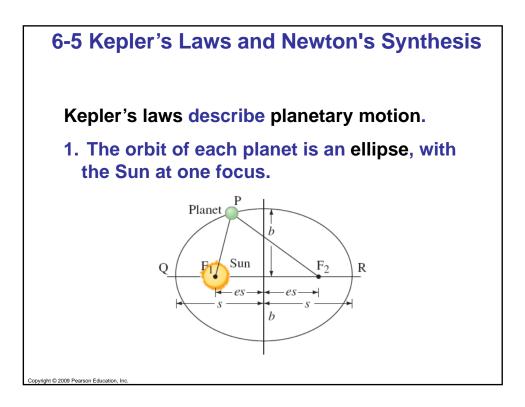
You have to drop into a lower orbit to speed up; when you get ahead of the satellite you need to slow down and get back into the higher orbit.



6-4 Satellites and "Weightlessness"

More properly, this effect is called apparent weightlessness, because the gravitational force still exists. It can be experienced on Earth as well, but only briefly:





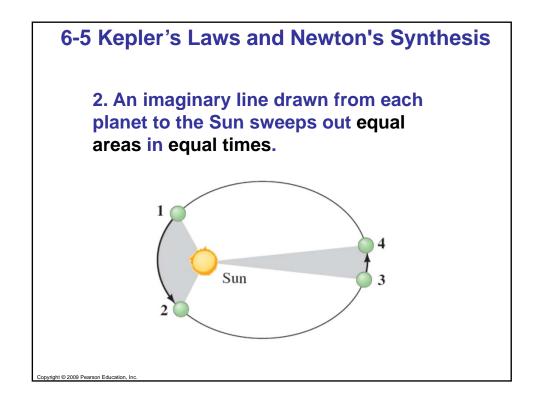


	TABLE 6–2 Planetary Data Applied to Kepler's Third Law			
3. The square of a planet's orbital period	Planet	Mean Distance from Sun, s (10 ⁶ km)	Period, <i>T</i> (Earth yr)	$\frac{s^3/T^2}{\left(10^{24}\frac{\mathrm{km}^3}{\mathrm{yr}^2}\right)}$
is proportional to the	Mercury	57.9	0.241	3.34
cube of its mean	Venus	108.2	0.615	3.35
distance from the Sun.	Earth	149.6	1.0	3.35
	Mars	227.9	1.88	3.35
	Jupiter	778.3	11.86	3.35
	Saturn	1427	29.5	3.34
	Uranus	2870	84.0	3.35
	Neptune	4497	165	3.34
	Pluto	5900	248	3.34

6-5 Kepler's Laws and Newton's Synthesis

Kepler's laws can be derived from Newton's laws. In particular, Kepler's third law follows directly from the law of universal gravitation —equating the gravitational force with the centripetal force shows that, for any two planets (assuming circular orbits, and that the only gravitational influence is the Sun):

$$\left(\frac{T_1}{T_2}\right)^2 = \left(\frac{s_1}{s_2}\right)^3.$$

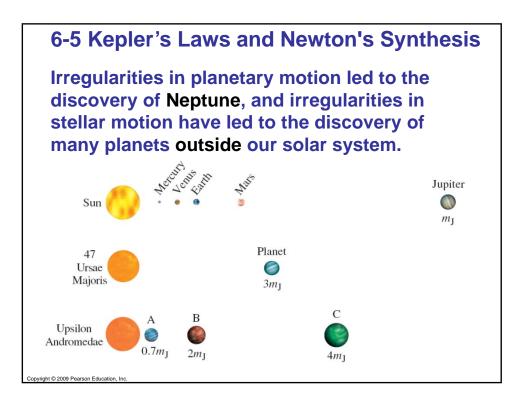
6-5 Kepler's Laws and Newton's Synthesis Example 6-8: Where is Mars? Mars' period (its "year") was first noted by Kepler to be about 687 days (Earth-days), which is (687 d/365 d) = 1.88 yr (Earth years). Determine the mean distance of Mars from the Sun using the Earth as a reference. $(\mathbf{r}_{ES} = 1.50 \times 10^{11} \text{ m})$ $\left(\frac{T_1}{T_2}\right)^2 = \left(\frac{T_1}{T_2}\right)^3$ $\therefore \frac{r_{MS}}{r_{ES}} = \left(\frac{T_M}{T_E}\right)^{\frac{2}{3}} = \left(\frac{1.88}{1}\right)^{\frac{2}{3}} = 1.52$ $\therefore r_{MS} = 1.52 \times 1.50 \times 10^{11} = 2.28 \times 10^{11} \text{ m}$ 6-5 Kepler's Laws and Newton's Synthesis Example 6-9: The Sun's mass determined.

Determine the mass of the Sun given the Earth's distance from the Sun as $r_{\rm ES} = 1.5 \times 10^{11}$ m.

(Note: This an easier approach than Giancoli without having to prove Kepler's 3rd law.)

$$F_{C} = \frac{m_{E}v_{E}^{2}}{r_{ES}} = \frac{Gm_{E}m_{S}}{(r_{ES})^{2}}$$

and $v_{E} = \frac{2\pi r_{ES}}{t} = \frac{2\pi \times 1.5 \times 10^{11}}{365 \times 24 \times 60 \times 60} = 3.00 \times 10^{4} \text{ m/s}$
 $\therefore m_{S} = \frac{r_{ES}v_{E}^{2}}{G} = \frac{1.5 \times 10^{11} \times (3.00 \times 10^{4})^{2}}{6.67 \times 10^{-11}}$
 $= 2.0 \times 10^{30} \text{ kg}$



6-6 Gravitational Field

The gravitational field is the gravitational force per unit mass:

$$\vec{\mathbf{g}} = \frac{\vec{\mathbf{F}}}{m}$$

The gravitational field due to a single mass *M* is given by:

$$\vec{\mathbf{g}} = -\frac{GM}{r^2}\,\hat{\mathbf{r}}\,.$$

