

Development and Qualification of Space Flight Hardware for Optical Systems

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Optical Fiber Communications Conference

Session on Space Photonics; Disruptive Satellite Laser Communications and Astrophotonics

Meet the Photonics Group of NASA Goddard Over 20 years of space flight hardware development, testing, & integration



Back row L-R: Erich Frese, Joe Thomes, Marc Matyseck
Middle row L-R: Rick Chuska, Eleanya Onuma, Cameron Parvini, Rob Switzer Clairy Reiher
Front row L-R: Hali Jakeman, Melanie Ott, Diana Blair,



Trevon Parker



Alexandros Bontzos



Alejandro Rodriguez

All great things require a great team!

<https://photonics.gsfc.nasa.gov>

Introduction: How to Get to Space

- History of some of our successful remote and earth missions.
- Here is how we accomplished TRL 9 mission success with commercial off the shelf components (COTS).



Incoming Inspection



Integration & Test

**Working with our industry partners to provide full service
From incoming inspection of materials all the way to integration of hardware.
Our missions require rigor in our work and documentation.**

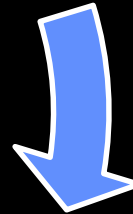
Custom Spaceflight Optical & Optoelectronic Subsystems using Commercial Components



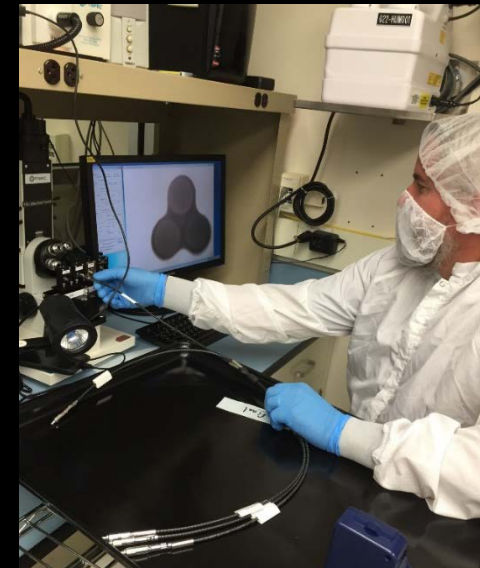
Integration



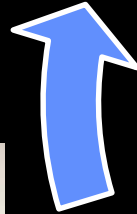
Materials Selection and Inspections



Manufacturing

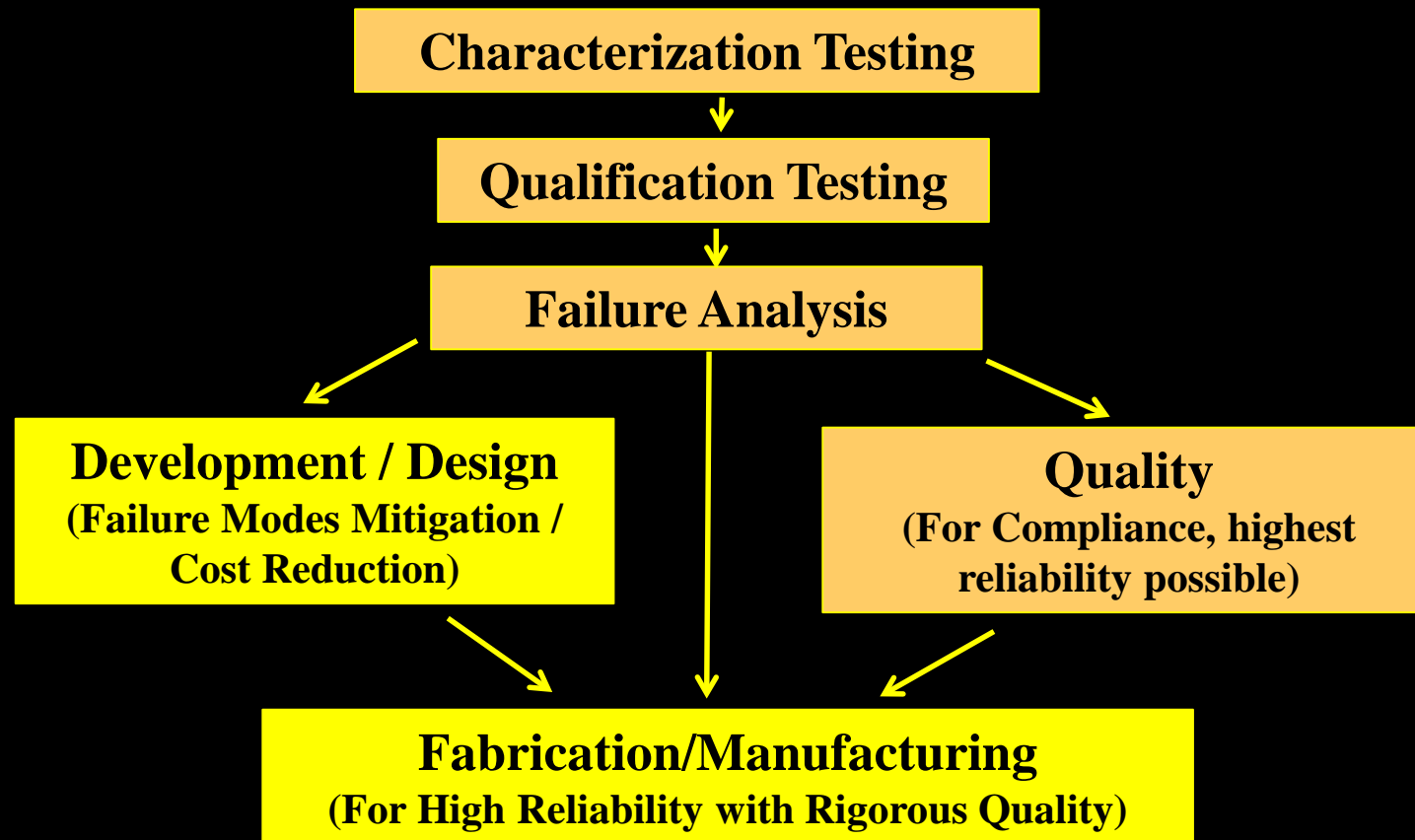


Environmental Testing



One Stop Shopping for
Concept through Delivery

How Do You Develop and Fabricate Hardware?



**Risk mitigation to reduce cost - use space flight component failure mode knowledge;
Design out what you can –through configuration; packaging, materials, processes, screening.**

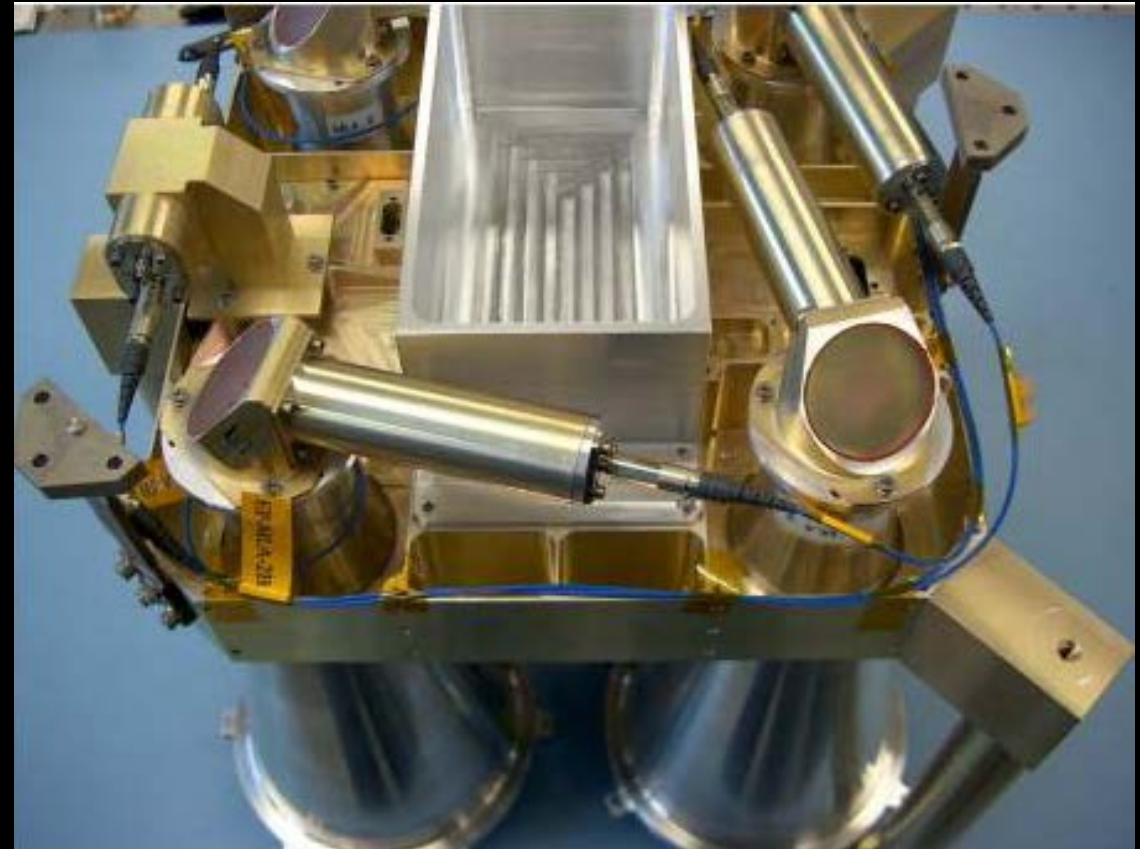
Planetary and Earth Orbiting LIDARS

Mercury



Mercury Laser Altimeter on Mercury Surface, Space Environment, Geochemistry and Ranging (MESSENGER); development 1999-2003, built by NASA Goddard Space Flight Center

Launch 2004, Operation 2011-2015 (travel time 7 years, 4 years usage, decommissioned in 2015)



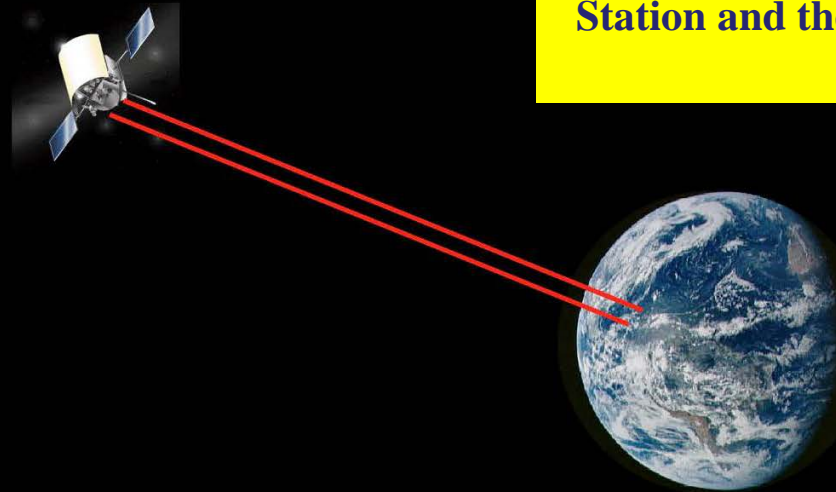
SPIE Vol. 5104

The 24 Million Km Link with the Mercury Laser Altimeter

Jay Steigelman
Dave Skillman
Barry Coyle
John F. Cavanaugh
Jan F. McGarry
Gregory A. Neumann
Xiaoli Sun
Thomas W. Zagwodzki
Dave Smith
Maria Zuber

MOLA Science Team Meeting
Bishop's Lodge, Santa Fe, NM
August 24-25, 2005

Smith, D. E., *et al.*, Two-way laser link over interplanetary distance, *Science*, 311, 5757, 53, Jan. 2006.



On the way to Mercury a link between NASA GSFC Greenbelt Station and the MLA was established - Longest Laser Link in Space Flight @ 24 Million Km.

The success of this experiment led the way for the Laser Ranging investigation on the Lunar Reconnaissance Orbiter.

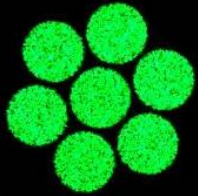
Planetary and Earth Orbiting LIDARS

The Moon

<https://lunar.gsfc.nasa.gov>

Laser Ranging Experiment & Lunar Orbiter Laser Altimeter (LOLA) –Lunar Reconnaissance Orbiter (LRO) Developed 2005-2008; Launch 2009, lifetime requirement 14 months, 3 years desired, actual 8 years and counting.....

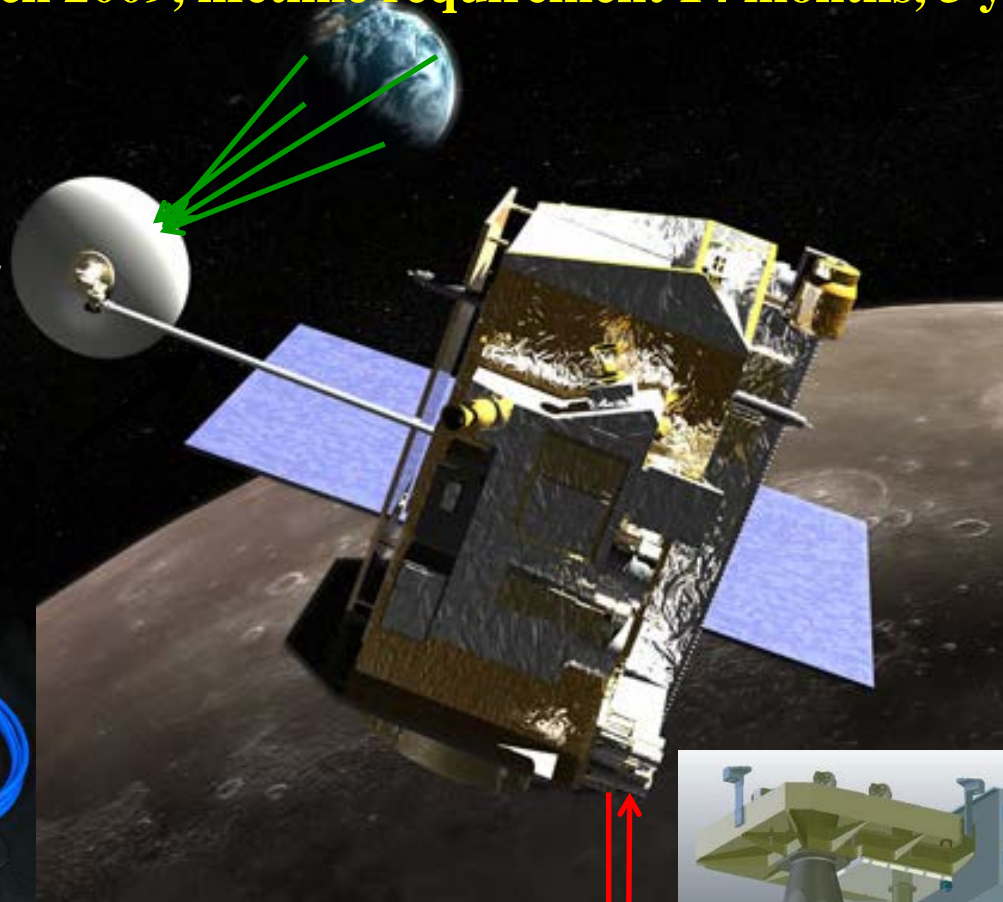
**LASER RANGING @ 532 nm -
 Stations Around the World
 Transmitting to the receiver telescope/
 7 optical fiber array**



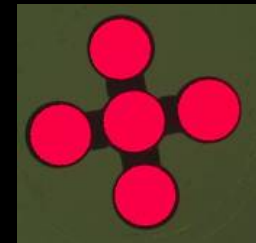
LRO Fiber Optic Laser Ranging Array Flight Assemblies



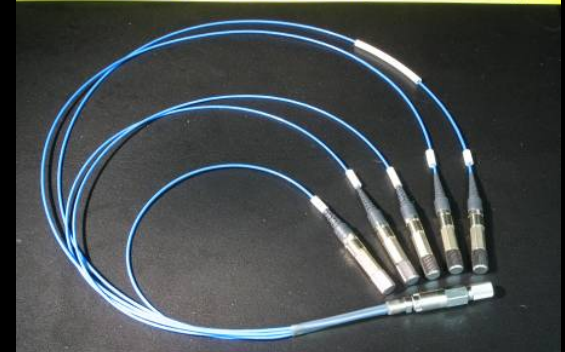
The assemblies traverse two moving gimbals, and a deployable mandrel 10 meters away to LOLA.



Lunar Orbiter Laser Altimeter (LOLA) Measuring moon topography @ 1064 nm with a 5 fiber array

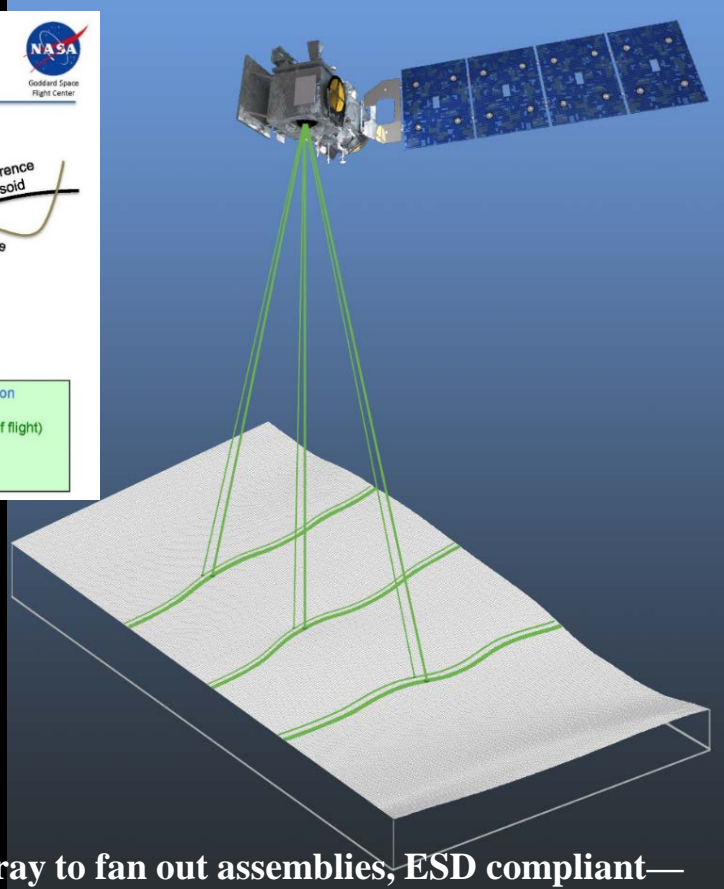
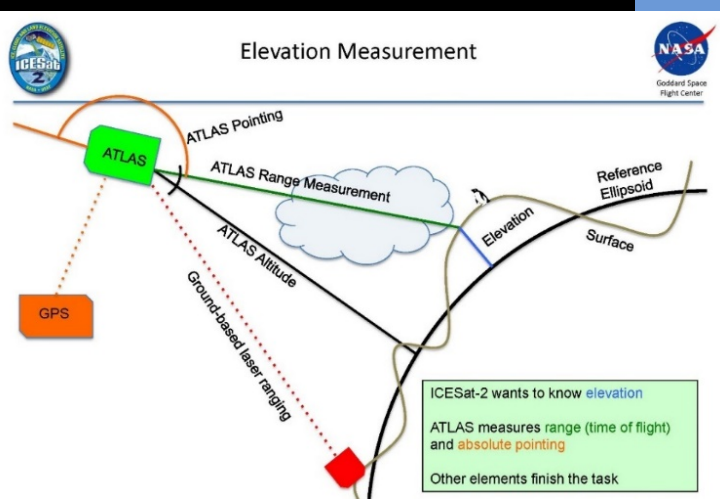


LRO Fiber Optics LOLA Flight Assembly



SPIE Vol. 7095

Ice, Cloud and Land Elevation Satellite (ICESat-2) – (ATLAS) Advanced Topographic Laser Altimeter System (2012 – 2018)
 Launched 2018, currently in operation. Expected lifetime > 3 years – measuring the height of sea ice to within an inch.



ATLAS uses ranging measurements with 532 nm and has a sophisticated real time, calibration system.

25 simplex, 4 bundle/array to fan out assemblies, ESD compliant—
 5 different types of fiber; dual and quad fiber arrays; 52 interconnections.
 Commercial LED - on board calibration system
 Fibertek lasers

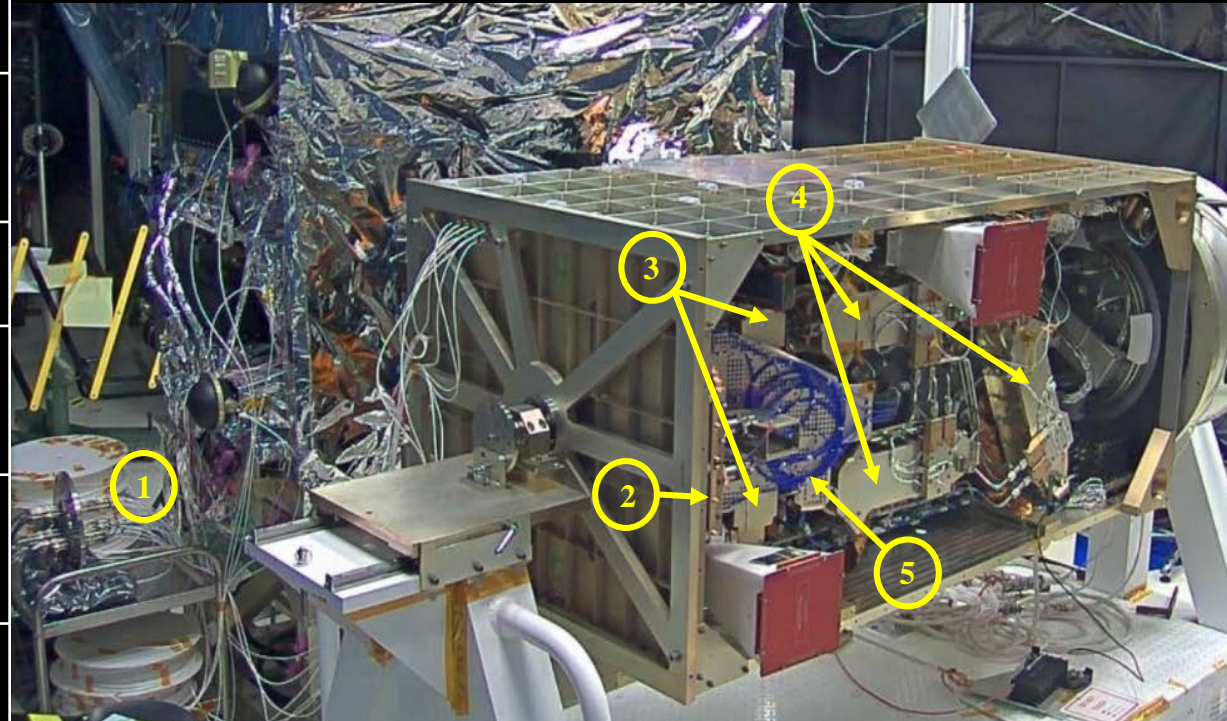
Melanie Ott (fiber system lead) inspecting the final flight configuration for fiber optic system. Transmission requirement of >98% for optical fiber receiver system.

Reference: <http://icesat.gsfc.nasa.gov>

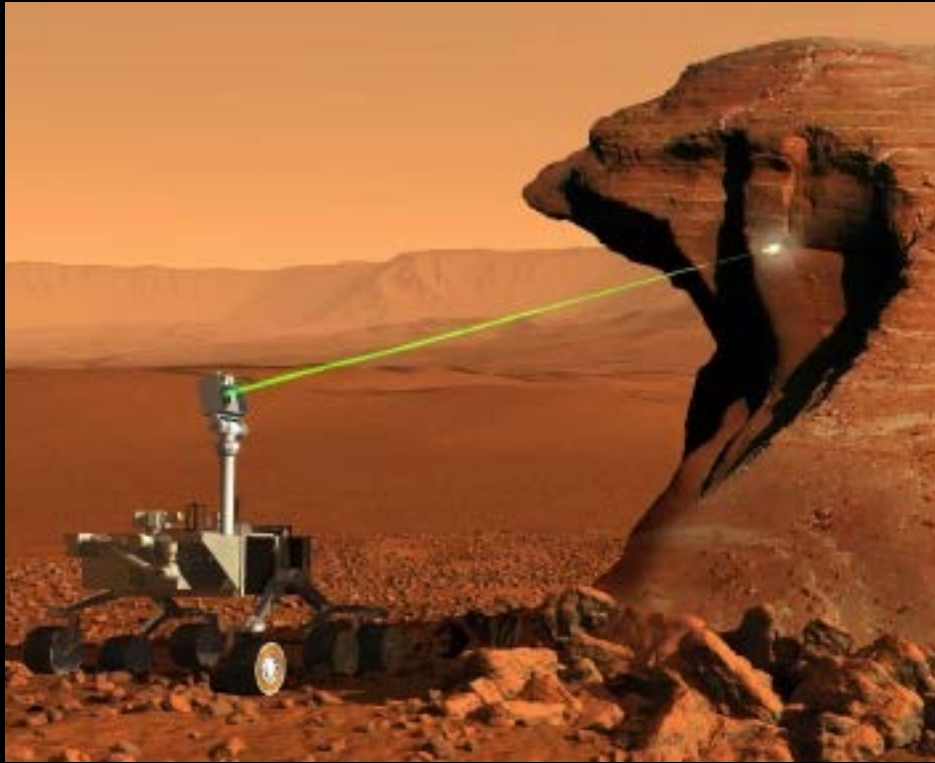
GEDI: Global Ecosystem Dynamics Investigation LIDAR (2016-2018)

Launched Dec 2018, operating currently on International Space Station

#	GEDI Subsystem	Hardware Deliveries
1	Checkout Equipment	Development, fabrication & integration: laser & detector test rack used for qualification of flight instrument, TVAC fiber assemblies down to -120°C.
2	Detector Qualification	Qualification of engineering & flight unit detectors
3	Laser Beam Dithering Unit	Development, fabrication, qualification & integration of engineering and flight units
4	Optical Laser Components	Development, qualification & fabrication of flight laser fiber optic feedthrough. Incoming inspection of laser components.
5	Flight Fiber Optic System	Development, qualification & integration of flight 600/600µm fiber optics transmission >97%; 200/220µm triple fiber arrays for start pulse. Adapter inspections and screening.



Science, Rovers and Communications Mars



**Mars Curiosity Rover; ChemCam Instrument
Launch Nov. 2011,
currently in operation.**



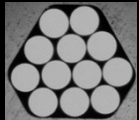
Hali Jakeman inspects the flight Mars2020 assemblies



**Mars 2020 Rover, SuperCam Instrument
Currently in integration and test.**

Development, fabrication, qualification of flight hardware delivery for JPL

SPIE Vol. 10565



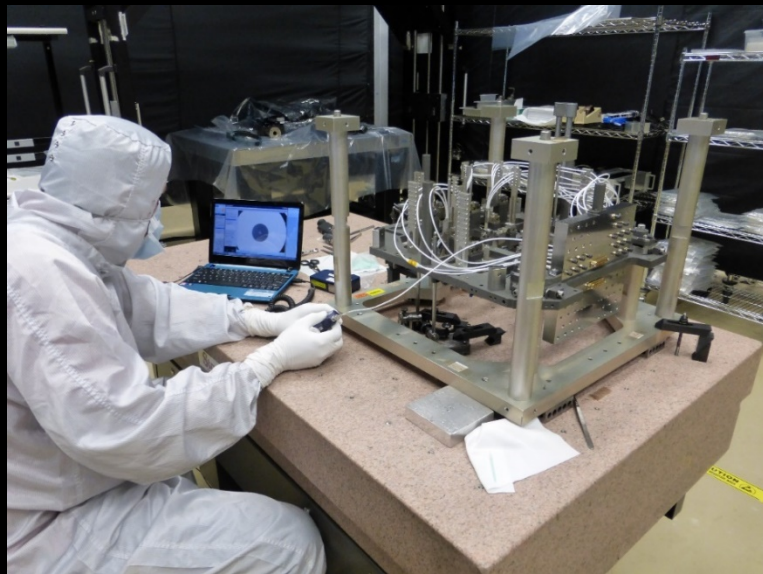
Science, Rovers and Communications

Communications: Multimode and Singlemode;

- Express Logistics Carrier on International Space Station. – Qualification of transceivers, fiber optic assemblies (2006 – 2010)
- Lunar Laser Communications Demonstration for MIT LL (2010)
- Communications for Cloud Aerosol Transport System; cats.gsfc.nasa.gov (2014) w/ FiberTek, Micropac
- Laser Communications Relay Demonstration; Screening and qualification (laser diodes & photonic components) (2014); Gooch & Housego

Science: Infrared, and/or polarization maintaining, single and multimode, thermal vacuum and cryogenic applications:

- James Webb Space Telescope; Ball Aerospace, Johnson Space Center & GSFC. (2008-2018)



Rob S. @ Ball installs cryo assemblies



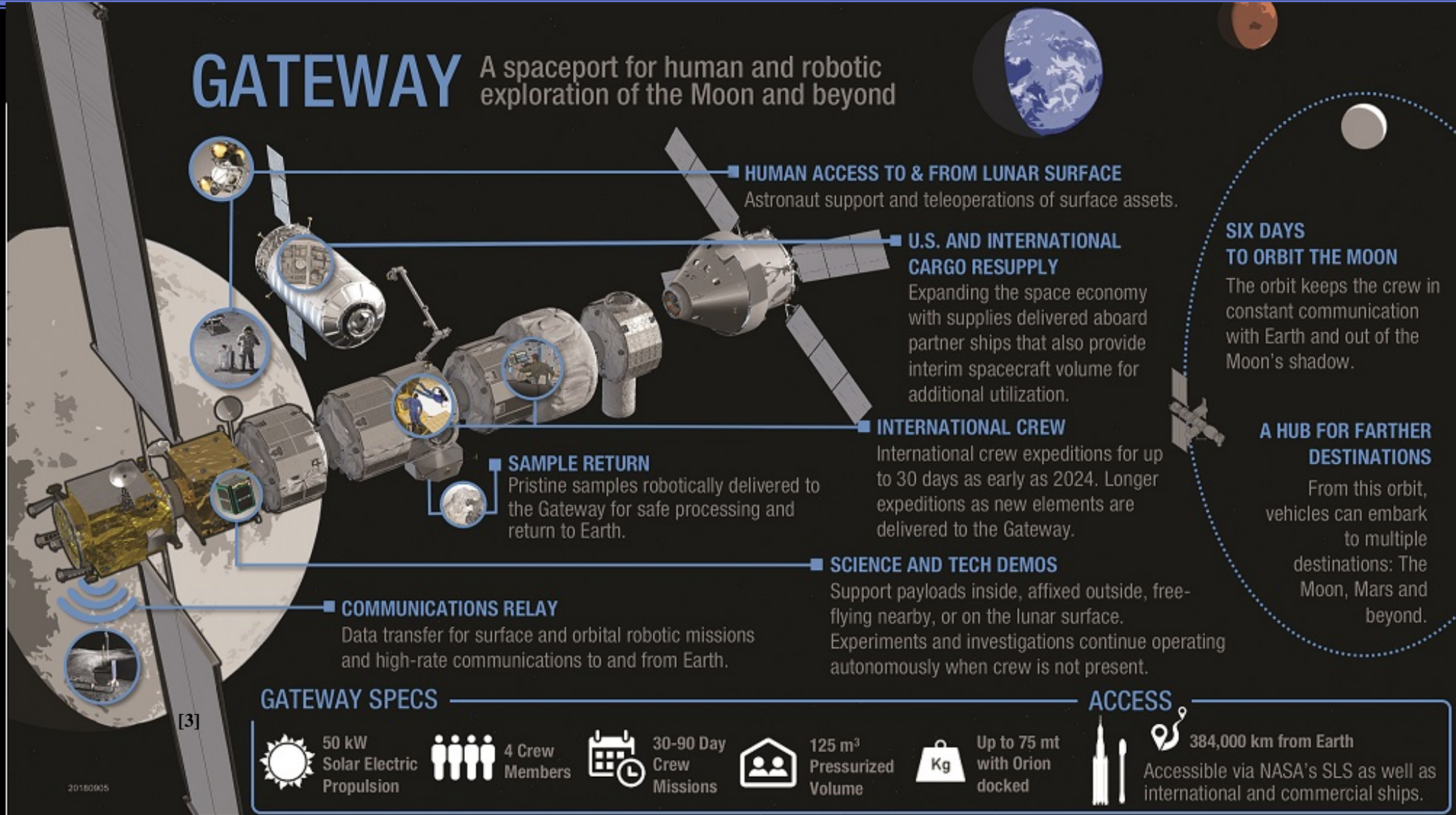
Eleanya Onuma installs vacuum feedthroughs



**Rob Switzer and Melanie Ott,
ELC integration @ Kennedy Space Center**

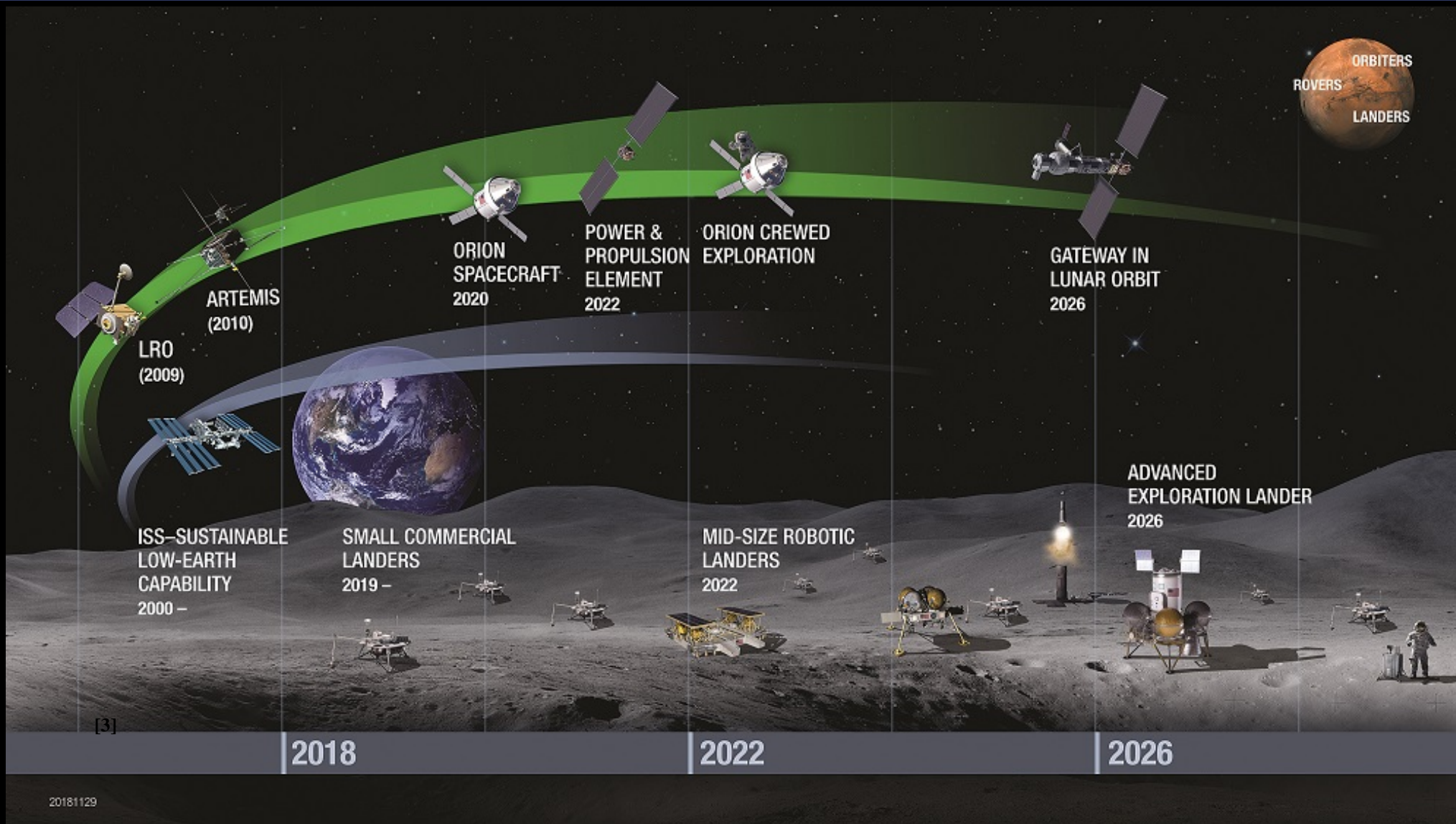
The Future Perspective

<https://spacenews.com/is-the-gateway-the-right-way-to-the-moon/>

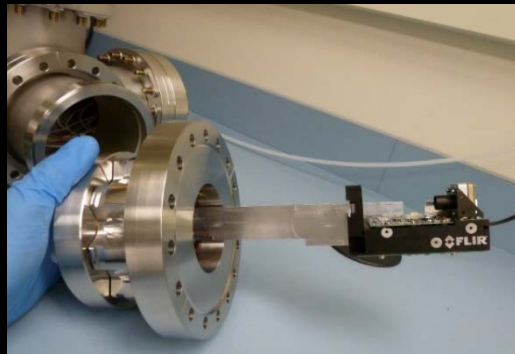
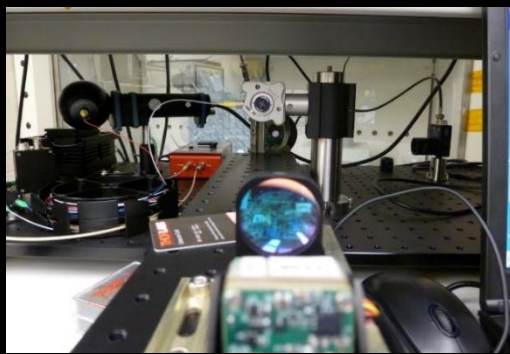


Gateway Roadmap

<https://spacenews.com/is-the-gateway-the-right-way-to-the-moon/>



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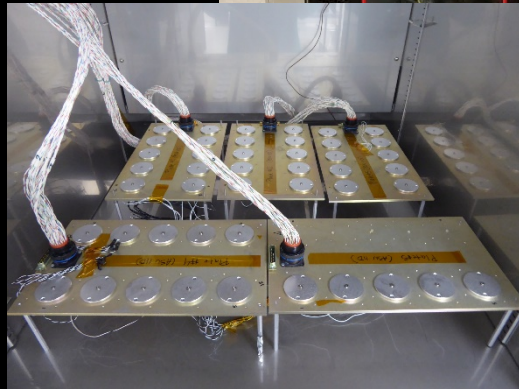
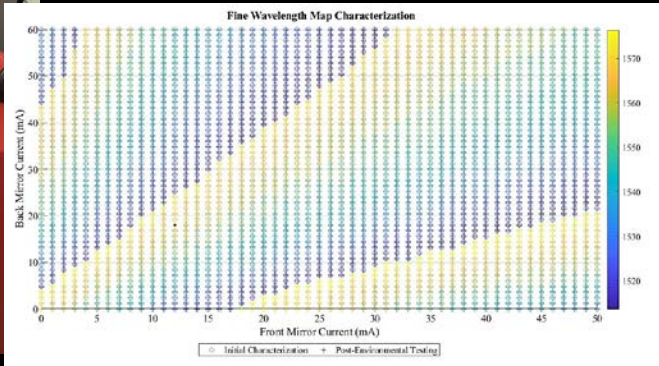
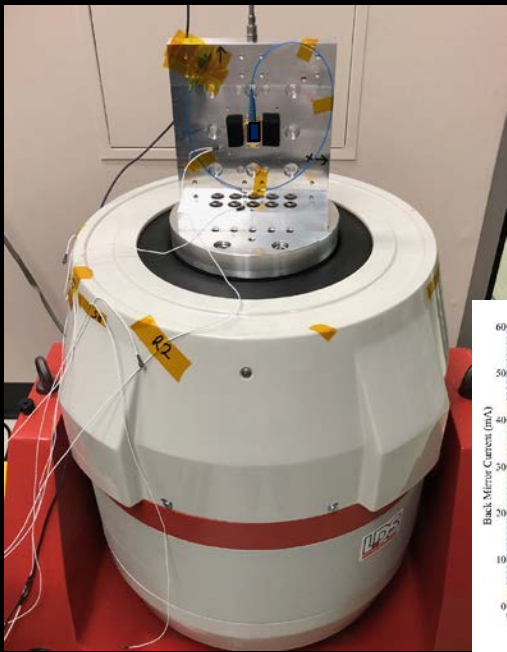
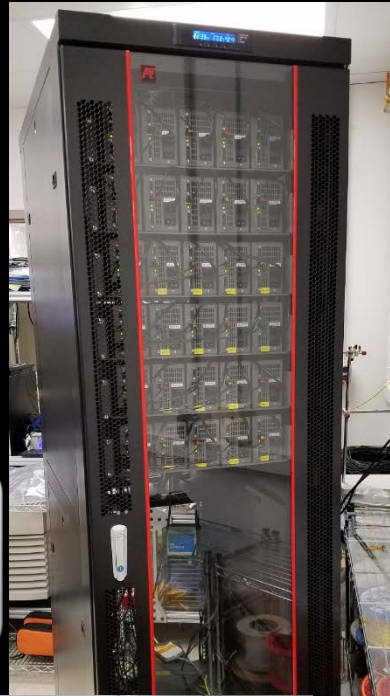


COTS LiDARs
 for **Lander**
 Autonomy

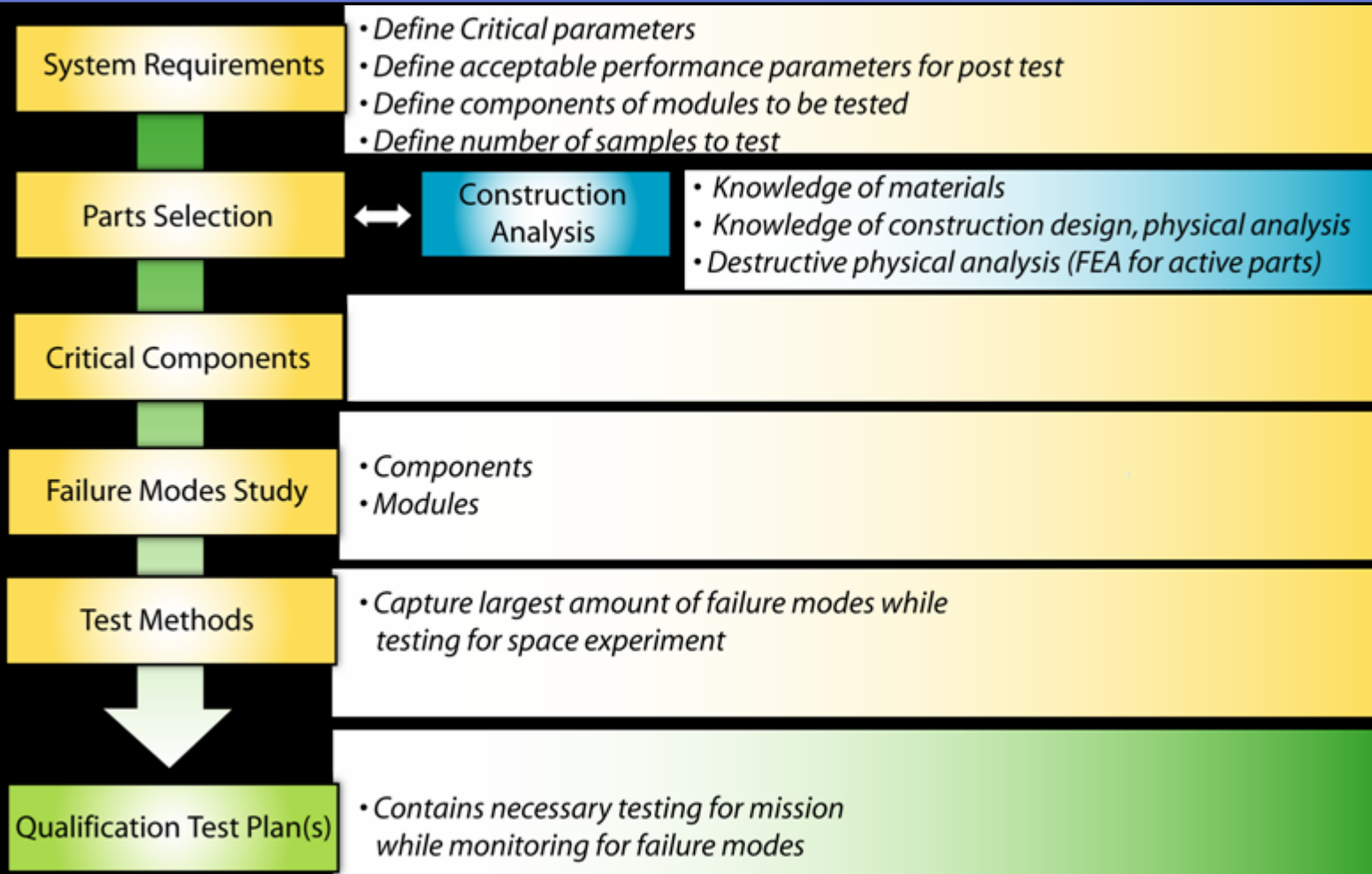
**Qualifying
 Optoelectronics
 & Photonics for
 Space**

Tunable Lasers
 for **Orbiter**
 Communications

Detectors for
Rover
 Spectroscopy

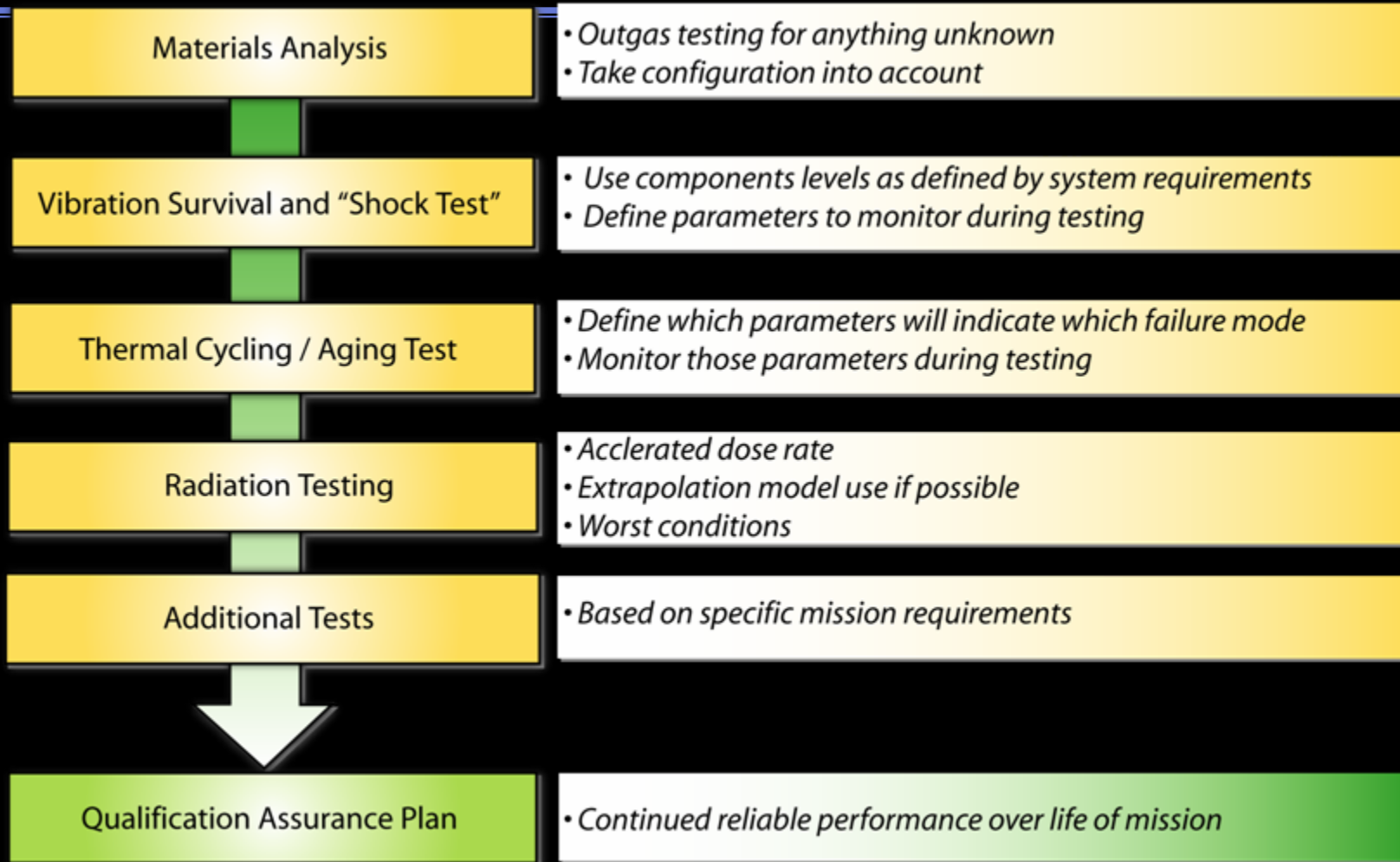


COTS Technology Assurance Approach



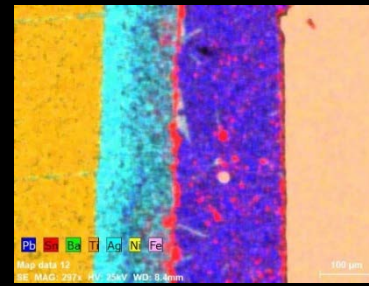
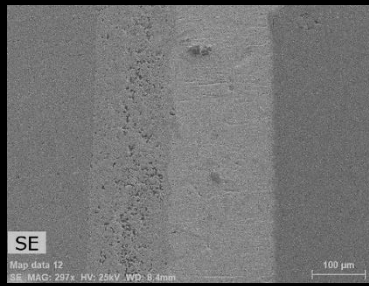
* *Photonic Components for Space Systems*, M. Ott, Presentation for Advanced Microelectronics and Photonics for Satellites Conference, 23 June 2004.

COTS Space Flight “Qualification”

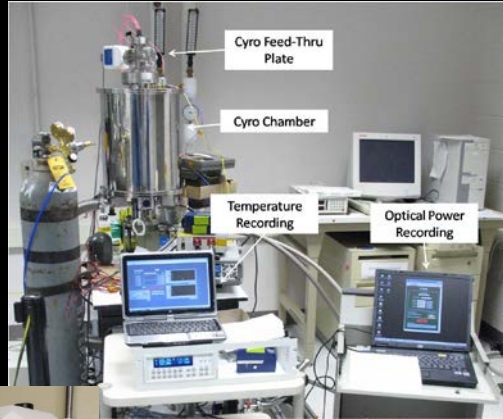


* *Photonic Components for Space Systems*, M. Ott, Presentation for Advanced Microelectronics and Photonics for Satellites Conference, 23 June 2004.

We perform selection, test and qualification of laser components the way the Parts Lab supports EEE parts.



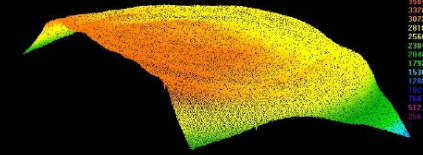
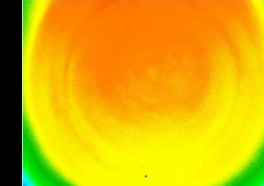
10 k X Mag SEM & Material Identification



Cryogenic Test Facility

ASTM-E595:
 300 mg, 125°C,
 10⁻⁶ Torr,
 24 Hr

Materials
 Screening /
 Construction
 Analysis



LED Beam Profile

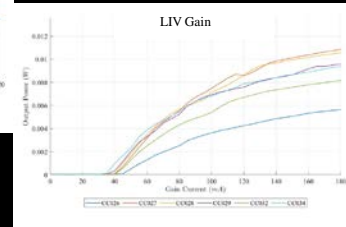
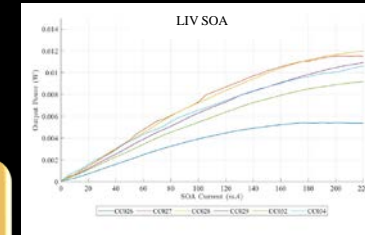
Additional
 Testing?

Optical
 Inspection &
 Screening

Optical Power,
 Current, Voltage
 Characterization



Performance
 Characterization



Radiation
 Testing



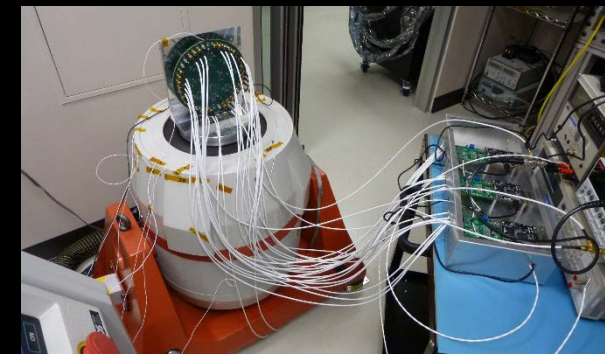
Radiation Test Equipment



White Light LED Testing in Environmental Chamber

Thermal
 Cycling /
 Vacuum

Vibration /
 "Shock"
 Testing

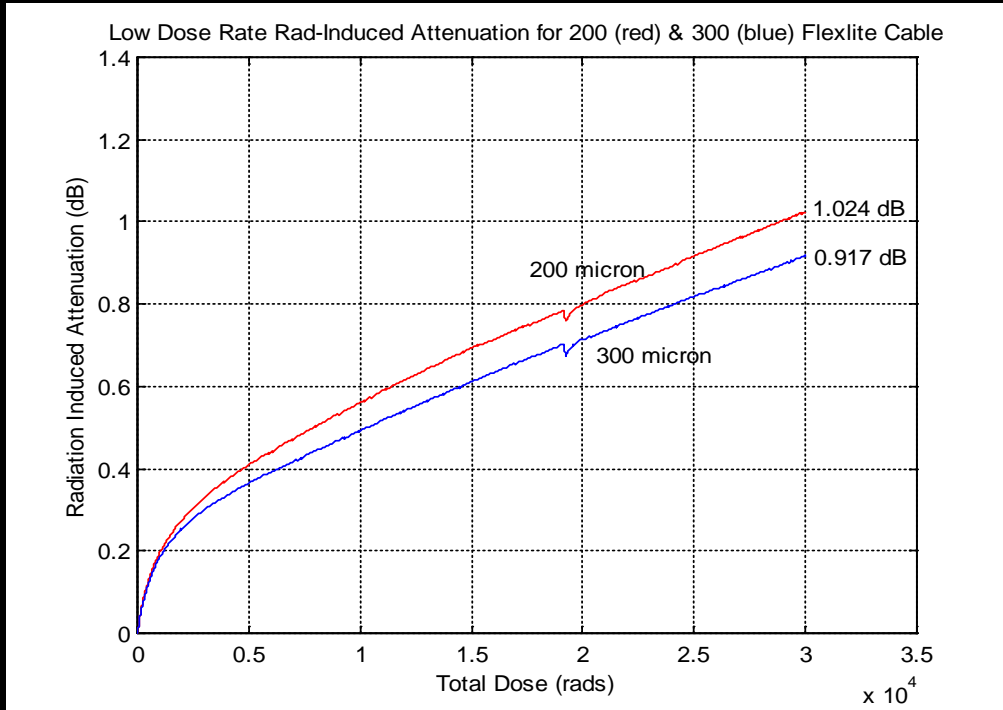
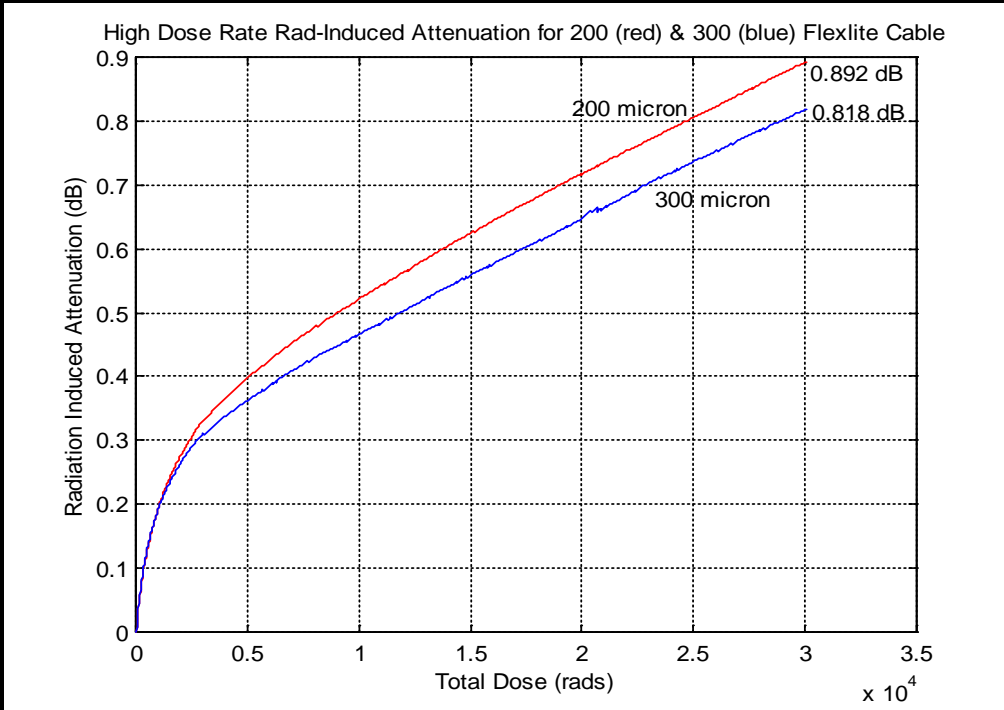


Random Vibration Test & Shock Equipment

Total Dose Radiation Performance

Not usually a detriment but for calibration purposes is always necessary;

Example Mercury Laser Altimeter Optical Fiber Radiation Data



Flexlite Radiation Test, 22.7 rads/min at -18.3°C

Flexlite Radiation Test, 11.2 rads/min at -24.1°C

**Radiation Conclusion: < .07 dB, using 11.2 rads/min, -24.1°C, 26.1 in, “dark”
Results for 10 m, at 30 Krads, -20°C, 850 nm, 23 rads/min ~ 1 dB or 0.10 dB/m**



Acronyms

- ATLAS = Advanced Topographic Laser Altimeter System
- COTS = Commercial Off The Shelf
- EEE = Electrical, Electronic, and Electromechanical
- ESD = Electrostatic Discharge
- FEA = Failure Effect Analysis
- GEDI = Global Ecosystem Dynamics Investigation
- ICESat-2 = Ice, Cloud, and land Elevation Satellite-2
- LED = Light Emitting Diode
- LiDAR = Light Detection and Ranging
- LOLA = Lunar Orbiter Laser Altimeter
- LRO = Lunar Reconnaissance Orbiter
- MESSENGER = Mercury Laser Altimeter on Mercury Surface, Space Environment, Geochemistry and Ranging
- MIT LL = Massachusetts Institute of Technology, Lincoln Laboratory
- MLA = Mercury Laser Altimeter
- MOLA = Mars Orbiter Laser Altimeter
- SEM = Scanning Electron Microscope
- TRL 9 = Technical Readiness Level 9
- TVAC = Thermal Vacuum

Thank You to Our Partners! (not all are listed here)



And thank you for your time.