EXAMPLE PROBLEMS AND SOLUTIONS

A-3-1. Simplify the block diagram shown in Figure 3-42.

> **Solution.** First, move the branch point of the path involving H_1 outside the loop involving H_2 , as shown in Figure 3-43(a). Then eliminating two loops results in Figure 3-43(b). Combining two blocks into one gives Figure 3-43(c).

A-3-2. Simplify the block diagram shown in Figure 3–44. Obtain the transfer function relating C(s) and R(s).











Figure 3-42 Block diagram of a

sysrem.







Figure 3-44 Block diagram of a system.



Figure 3–45 Reduction of the block diagram shown in Figure 3–44.

Solution. The block diagram of Figure 3-44 can be modified to that shown in Figure 3-45(a). Eliminating the minor feedforward path, we obtain Figure 3-45(b), which can be simplified to that shown in Figure 3-45(c). The transfer function C(s)/R(s) is thus given by

$$\frac{C(s)}{R(s)} = G_1 G_2 + G_2 + 1$$

The same result can also be obtained by proceeding as follows: Since signal X(s) is the sum of two signals $G_1R(s)$ and R(s), we have

$$X(s) = G_1 R(s) + R(s)$$

The output signal C(s) is the sum of $G_2X(s)$ and R(s). Hence

$$C(s) = G_2 X(s) + R(s) = G_2 [G_1 R(s) + R(s)] + R(s)$$

And so we have the same result as before:

$$\frac{C(s)}{R(s)} = G_1 G_2 + G_2 + 1$$

A-3-3. Simplify the block diagram shown in Figure 3–46. Then, obtain the closed-loop transfer function C(s)/R(s).



Figure 3–46 Block diagram of a system.



Figure 3–47 Successive reductions of the block diagram shown in Figure 3–46.



Solution. First move the branch point between G_3 and G_4 to the right-hand side of the loop containing G_3 , G_4 , and H_2 . Then move the summing point between G_1 and G_2 to the left-hand side of the first summing point. See Figure 3–47(a). By simplifying each loop, the block diagram can be modified as shown in Figure 3–47(b). Further simplification results in Figure 3–47(c), from which the closed-loop transfer function C(s)/R(s) is obtained as

$$\frac{C(s)}{R(s)} = \frac{G_1 G_2 G_3 G_4}{1 + G_1 G_2 H_1 + G_3 G_4 H_2 - G_2 G_3 H_3 + G_1 G_2 G_3 G_4 H_1 H_2}$$

A-3-4. Obtain transfer functions C(s)/R(s) and C(s)/D(s) of the system shown in Figure 3-48.

Solution. From Figure 3-48 we have

$$U(s) = G_f R(s) + G_c E(s)$$
 (3-88)

$$C(s) = G_p[D(s) + G_1U(s)]$$
(3-89)

$$E(s) = R(s) - HC(s)$$
 (3-90)



Figure 3–48 Control system with reference input and disturbance input. By substituting Equation (3-88) into Equation (3-89), we get

$$C(s) = G_p D(s) + G_1 G_p [G_f R(s) + G_c E(s)]$$
(3-91)

By substituting Equation (3–90) into Equation (3–91), we obtain

$$C(s) = G_p D(s) + G_1 G_p \{G_f R(s) + G_c [R(s) - HC(s)]\}$$

Solving this last equation for C(s), we get

$$C(s) + G_1 G_p G_c H C(s) = G_p D(s) + G_1 G_p (G_f + G_c) R(s)$$

Hence

$$C(s) = \frac{G_p D(s) + G_1 G_p (G_f + G_c) R(s)}{1 + G_1 G_p G_c H}$$
(3-92)

Note that Equation (3–92) gives the response C(s) when both reference input R(s) and disturbance input D(s) are present.

To find transfer function C(s)/R(s), we let D(s) = 0 in Equation (3-92). Then we obtain

$$\frac{C(s)}{R(s)} = \frac{G_{\iota}G_{\rho}(G_f + G_c)}{1 + G_{\iota}G_{\rho}G_cH}$$

Similarly, to obtain transfer function C(s)/D(s), we let R(s) = 0 in Equation (3-92). Then C(s)/D(s) can be given by

$$\frac{C(s)}{D(s)} = \frac{G_p}{1 + G_1 G_p G_c H}$$

A-3-5. Figure 3-49 shows a system with two inputs and two outputs. Derive $C_1(s)/R_1(s)$, $C_1(s)/R_2(s)$, $C_2(s)/R_1(s)$, and $C_2(s)/R_2(s)$. (In deriving outputs for $R_1(s)$, assume that $R_2(s)$ is zero, and vice versa.)





Solution. From the figure, we obtain

$$C_1 = G_1 (R_1 - G_3 C_2) \tag{3-93}$$

$$C_2 = G_4(R_2 - G_2C_1) \tag{3-94}$$

By substituting Equation (3-94) into Equation (3-93), we obtain

$$C_1 = G_1 [R_1 - G_3 G_4 (R_2 - G_2 C_1)]$$
(3-95)

By substituting Equation (3–93) into Equation (3–94), we get

$$C_2 = G_4 [R_2 - G_2 G_1 (R_1 - G_3 C_2)]$$
(3-96)

Solving Equation (3–95) for C_1 , we obtain

$$C_1 = \frac{G_1 R_1 - G_1 G_3 G_4 R_2}{1 - G_1 G_2 G_3 G_4}$$
(3-97)

Solving Equation (3-96) for C_2 gives

$$C_2 = \frac{-G_1 G_2 G_4 R_1 + G_4 R_2}{1 - G_1 G_2 G_3 G_4}$$
(3-98)

Equations (3-97) and (3-98) can be combined in the form of the transfer matrix as follows:

$$\begin{bmatrix} C_1 \\ C_2 \end{bmatrix} = \begin{bmatrix} \frac{G_1}{1 - G_1 G_2 G_3 G_4} & \frac{G_1 G_3 G_4}{1 - G_1 G_2 G_3 G_4} \\ -\frac{G_1 G_2 G_4}{1 - G_1 G_2 G_3 G_4} & \frac{G_4}{1 - G_1 G_2 G_3 G_4} \end{bmatrix} \begin{bmatrix} R_1 \\ R_2 \end{bmatrix}$$

Then the transfer functions $C_1(s)/R_1(s)$, $C_1(s)/R_2(s)$, $C_2(s)/R_1(s)$ and $C_2(s)/R_2(s)$ can be obtained as follows:

$$\frac{C_1(s)}{R_1(s)} = \frac{G_1}{1 - G_1 G_2 G_3 G_4}, \qquad \frac{C_1(s)}{R_2(s)} = -\frac{G_1 G_3 G_4}{1 - G_1 G_2 G_3 G_4}$$
$$\frac{C_2(s)}{R_1(s)} = -\frac{G_1 G_2 G_4}{1 - G_1 G_2 G_3 G_4}, \qquad \frac{C_2(s)}{R_2(s)} = \frac{G_4}{1 - G_1 G_2 G_3 G_4}$$

Note that Equations (3–97) and (3–98) give responses C_1 and C_2 , respectively, when both inputs R_1 and R_2 are present.

Notice that when $R_2(s) = 0$, the original block diagram can be simplified to those shown in Figures 3–50(a) and (b). Similarly, when $R_1(s) = 0$, the original block diagram can be simplified to those shown in Figures 3–50(c) and (d). From these simplified block diagrams we can also obtain $C_1(s)/R_1(s)$, $C_2(s)/R_1(s)$, $C_1(s)/R_2(s)$, and $C_2(s)/R_2(s)$, as shown to the right of each corresponding block diagram.



A-3-6. Show that for the differential equation system

$$\ddot{y} + a_1 \ddot{y} + a_2 \dot{y} + a_3 y = b_0 \ddot{u} + b_1 \dot{u} + b_2 \dot{u} + b_3 u$$
 (3-99)

state and output equations can be given, respectively, by

$$\begin{bmatrix} \dot{x}_{1} \\ \dot{x}_{2} \\ \dot{x}_{3} \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -a_{3} & -a_{2} & -a_{1} \end{bmatrix} \begin{bmatrix} x_{1} \\ x_{2} \\ x_{3} \end{bmatrix} + \begin{bmatrix} \beta_{1} \\ \beta_{2} \\ \beta_{3} \end{bmatrix} u$$
(3-100)

and

$$y = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \beta_0 u$$
 (3-101)

where state variables are defined by

$$x_1 = y - \beta_0 u$$

$$x_2 = \dot{y} - \beta_0 \dot{u} - \beta_1 u = \dot{x}_1 - \beta_1 u$$

$$x_3 = \ddot{y} - \beta_0 \ddot{u} - \beta_1 \dot{u} - \beta_2 u = \dot{x}_2 - \beta_2 u$$

and

$$\beta_{0} = b_{0}$$

$$\beta_{1} = b_{1} - a_{1}\beta_{0}$$

$$\beta_{2} = b_{2} - a_{1}\beta_{1} - a_{2}\beta_{0}$$

$$\beta_{3} = b_{3} - a_{1}\beta_{2} - a_{2}\beta_{1} - a_{3}\beta_{0}$$

Solution. From the definition of state variables x_2 and x_3 , we have

 α

1

$$\dot{x}_1 = x_2 + \beta_1 u \tag{3-102}$$

$$\dot{x}_2 = x_3 + \beta_2 u \tag{3-103}$$

To derive the equation for \dot{x}_3 , we first note from Equation (3–99) that

$$\ddot{y} = -a_1\ddot{y} - a_2\dot{y} - a_3y + b_0\ddot{u} + b_1\ddot{u} + b_2\dot{u} + b_3u$$

Since

$$x_3 = \ddot{y} - \beta_0 \ddot{u} - \beta_1 \dot{u} - \beta_2 u$$

we have

$$\begin{aligned} \dot{x}_{3} &= \ddot{y} - \beta_{0} \ddot{u} - \beta_{1} \dot{u} - \beta_{2} \dot{u} \\ &= (-a_{1} \dot{y} - a_{2} \dot{y} - a_{3} y) + b_{0} \ddot{u} + b_{1} \ddot{u} + b_{2} \dot{u} + b_{3} u - \beta_{0} \ddot{u} - \beta_{1} \ddot{u} - \beta_{2} \dot{u} \\ &= -a_{1} (\ddot{y} - \beta_{0} \ddot{u} - \beta_{1} \dot{u} - \beta_{2} u) - a_{1} \beta_{0} \ddot{u} - a_{1} \beta_{1} \dot{u} - a_{1} \beta_{2} u \\ -a_{2} (\dot{y} - \beta_{0} \dot{u} - \beta_{1} u) - a_{2} \beta_{0} \dot{u} - a_{2} \beta_{1} u - a_{3} (y - \beta_{0} u) - a_{3} \beta_{0} u \\ &+ b_{0} \ddot{u} + b_{1} \ddot{u} + b_{2} \dot{u} + b_{3} u - \beta_{0} \ddot{u} - \beta_{1} \ddot{u} - \beta_{2} \dot{u} \\ &= -a_{1} x_{3} - a_{2} x_{2} - a_{3} x_{1} + (b_{0} - \beta_{0}) \ddot{u} + (b_{1} - \beta_{1} - a_{1} \beta_{0}) \dot{u} \\ &+ (b_{2} - \beta_{2} - a_{1} \beta_{1} - a_{2} \beta_{0}) \dot{u} + (b_{3} - a_{1} \beta_{2} - a_{2} \beta_{1} - a_{3} \beta_{0}) u \\ &= -a_{1} x_{3} - a_{2} x_{2} - a_{3} x_{1} + (b_{3} - a_{1} \beta_{2} - a_{2} \beta_{1} - a_{3} \beta_{0}) u \\ &= -a_{1} x_{3} - a_{2} x_{2} - a_{3} x_{1} + \beta_{3} u \end{aligned}$$

Hence, we get

$$\dot{x}_3 = -a_3 x_1 - a_2 x_2 - a_1 x_3 + \beta_3 u \tag{3-104}$$

Combining Equations (3–102), (3–103), and (3–104) into a vector-matrix equation, we obtain Equation (3–100). Also, from the definition of state variable x_1 , we get the output equation given by Equation (3–101).

A-3-7. Obtain a state-space equation and output equation for the system defined by

$$\frac{Y(s)}{U(s)} = \frac{2s^3 + s^2 + s + 2}{s^3 + 4s^2 + 5s + 2}$$

Solution. From the given transfer function, the differential equation for the system is

$$\ddot{y} + 4\ddot{y} + 5\dot{y} + 2y = 2\ddot{u} + \ddot{u} + \dot{u} + 2u$$

Comparing this equation with the standard equation given by Equation (3-33), rewritten

$$\ddot{y} + a_1 \ddot{y} + a_2 \dot{y} + a_3 y = b_0 \ddot{u} + b_1 \ddot{u} + b_2 \dot{u} + b_3 u$$

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we find

$$a_1 = 4$$
, $a_2 = 5$, $a_3 = 2$
 $b_0 = 2$, $b_1 = 1$, $b_2 = 1$, $b_3 =$

2

Referring to Equation (3-35), we have

$$\beta_0 = b_0 = 2$$

$$\beta_1 = b_1 - a_1\beta_0 = 1 - 4 \times 2 = -7$$

$$\beta_2 = b_2 - a_1\beta_1 - a_2\beta_0 = 1 - 4 \times (-7) - 5 \times 2 = 19$$

$$\beta_3 = b_3 - a_1\beta_2 - a_2\beta_1 - a_3\beta_0$$

$$= 2 - 4 \times 19 - 5 \times (-7) - 2 \times 2 = -43$$

Referring to Equation (3-34), we define

$$x_{1} = y - \beta_{0}u = y - 2u$$

$$x_{2} = \dot{x}_{1} - \beta_{1}u = \dot{x}_{1} + 7u$$

$$x_{3} = \dot{x}_{2} - \beta_{2}u = \dot{x}_{2} - 19u$$

Then referring to Equation (3-36),

$$\dot{x}_1 = x_2 - 7u$$

$$\dot{x}_2 = x_3 + 19u$$

$$\dot{x}_3 = -a_3x_1 - a_2x_2 - a_1x_3 + \beta_3u$$

$$= -2x_1 - 5x_2 - 4x_3 - 43u$$

Hence, the state-space representation of the system is

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -2 & -5 & -4 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} -7 \\ 19 \\ -43 \end{bmatrix} u$$
$$y = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + 2u$$

This is one possible state-space representation of the system. There are many (infinitely many) others. If we use MATLAB, it produces the following state-space representation:

$$\begin{bmatrix} \dot{x}_{1} \\ \dot{x}_{2} \\ \dot{x}_{3} \end{bmatrix} = \begin{bmatrix} -4 & -5 & -2 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} x_{1} \\ x_{2} \\ x_{3} \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} u$$
$$\cdot$$
$$y = \begin{bmatrix} -7 & -9 & -2 \end{bmatrix} \begin{bmatrix} x_{1} \\ x_{2} \\ x_{3} \end{bmatrix} + 2u$$

See MATLAB Program 3-4. (Note that all state-space representations for the same system are equivalent.)

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MATLAB Program 3-4 $num = [2 \ 1 \ 1 \ 2];$ den = $[1 \ 4 \ 5 \ 2];$ [A,B,C,D] = tf2ss(num, den)A =-4 -5 -2 1 0 0 0 1 0 B =1 0 0 C = -7 -2 -9 D = 2

A-3-8. Obtain a state-space model of the system shown in Figure 3–51.

Solution. The system involves one integrator and two delayed integrators. The output of each integrator or delayed integrator can be a state variable. Let us define the output of the plant as x_1 , the output of the controller as x_2 , and the output of the sensor as x_3 . Then we obtain

Figure 3–51 Control system.

which can be rewritten as

$$sX_{1}(s) = -5X_{1}(s) + 10X_{2}(s)$$

$$sX_{2}(s) = -X_{3}(s) + U(s)$$

$$sX_{3}(s) = X_{1}(s) - X_{3}(s)$$

$$Y(s) = X_{1}(s)$$

By taking the inverse Laplace transforms of the preceding four equations, we obtain

$$\dot{x}_1 = -5x_1 + 10x_2$$
$$\dot{x}_2 = -x_3 + u$$
$$\dot{x}_3 = x_1 - x_3$$
$$\dot{y} = x_1$$

Thus, a state-space model of the system in the standard form is given by

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} -5 & 10 & 0 \\ 0 & 0 & -1 \\ 1 & 0 & -1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \\ 0 \end{bmatrix} u$$
$$y = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}$$

It is important to note that this is not the only state-space representation of the system. Many other state-space representations are possible. However, the number of state variables is the same in any state-space representation of the same system. In the present system, the number of state variables is three, regardless of what variables are chosen as state variables.

A-3-9. Obtain a state-space model for the system shown in Figure 3–52(a).

Solution. First, notice that $(as + b)/s^2$ involves a derivative term. Such a derivative term may be avoided if we modify $(as + b)/s^2$ as

$$\frac{as+b}{s^2} = \left(a+\frac{b}{s}\right)\frac{1}{s}$$

Using this modification, the block diagram of Figure 3-52(a) can be modified to that shown in Figure 3-52(b).

Define the outputs of the integrators as state variables, as shown in Figure 3-52(b). Then from Figure 3-52(b) we obtain

$$\frac{X_1(s)}{X_2(s) + a[U(s) - X_1(s)]} = \frac{1}{s}$$
$$\frac{X_2(s)}{U(s) - X_1(s)} = \frac{b}{s}$$
$$Y(s) = X_1(s)$$

which may be modified to

$$sX_{1}(s) = X_{2}(s) + a[U(s) - X_{1}(s)]$$

$$sX_{2}(s) = -bX_{1}(s) + bU(s)$$

$$Y(s) = X_{1}(s)$$

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Taking the inverse Laplace transforms of the preceding three equations, we obtain

$$\dot{x}_1 = -ax_1 + x_2 + au$$
$$\dot{x}_2 = -bx_1 + bu$$
$$y = x_1$$

Rewriting the state and output equations in the standard vector-matrix form, we obtain

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -a & 1 \\ -b & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} a \\ b \end{bmatrix} u$$
$$y = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

A-3-10. Obtain a state-space representation of the system shown in Figure 3-53(a).

Solution. In this problem, first expand (s + z)/(s + p) into partial fractions.

$$\frac{s+z}{s+p} = 1 + \frac{z-p}{s+p}$$

Next, convert K/[s(s + a)] into the product of K/s and 1/(s + a). Then redraw the block diagram, as shown in Figure 3–53(b). Defining a set of state variables, as shown in Figure 3–53(b), we obtain the following equations:

$$\dot{x}_{1} = -ax_{1} + x_{2}$$

$$\dot{x}_{2} = -Kx_{1} + Kx_{3} + Ku$$

$$\dot{x}_{3} = -(z - p)x_{1} - px_{3} + (z - p)u$$

$$y = x_{1}$$



(a)



igure 3-53 i) Control system; b) block diagram efining state ariables for the stem.



$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} -a & 1 & 0 \\ -K & 0 & K \\ -(z-p) & 0 & -p \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ K \\ z-p \end{bmatrix} u$$
$$y = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}$$

Notice that the output of the integrator and the outputs of the first-order delayed integrators [1/(s + a) and (z - p)/(s + p)] are chosen as state variables. It is important to remember that the output of the block (s + z)/(s + p) in Figure 3-53(a) cannot be a state variable, because this block involves a derivative term, s + z.

A-3-11. Obtain the transfer function of the system defined by

$$\begin{bmatrix} \dot{x}_{1} \\ \dot{x}_{2} \\ \dot{x}_{3} \end{bmatrix} = \begin{bmatrix} -1 & 1 & 0 \\ 0 & -1 & 1 \\ 0 & 0 & -2 \end{bmatrix} \begin{bmatrix} x_{1} \\ x_{2} \\ x_{3} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u$$
$$y = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_{1} \\ x_{2} \\ x_{3} \end{bmatrix}$$

Solution. Referring to Equation (3–29), the transfer function G(s) is given by

$$G(s) = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B} + D$$

In this problem, matrices A, B, C, and D are

$$\mathbf{A} = \begin{bmatrix} -1 & 1 & 0 \\ 0 & -1 & 1 \\ 0 & 0 & -2 \end{bmatrix}, \qquad \mathbf{B} = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}, \qquad \mathbf{C} = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix}, \qquad D = 0$$

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Hence

$$G(s) = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \begin{bmatrix} s+1 & -1 & 0 \\ 0 & s+1 & -1 \\ 0 & 0 & s+2 \end{bmatrix}^{-1} \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$
$$= \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \begin{bmatrix} \frac{1}{s+1} & \frac{1}{(s+1)^2} & \frac{1}{(s+1)^2(s+2)} \\ 0 & \frac{1}{s+1} & \frac{1}{(s+1)(s+2)} \\ 0 & 0 & \frac{1}{s+2} \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$
$$= \frac{1}{(s+1)^2(s+2)} = \frac{1}{s^3 + 4s^2 + 5s + 2}$$

A-3-12. Obtain a state-space representation of the system shown in Figure 3–54.**Solution.** The system equations are

$$m_1 \ddot{y}_1 + b \dot{y}_1 + k(y_1 - y_2) = 0$$

$$m_2 \ddot{y}_2 + k(y_2 - y_1) = u$$

The output variables for this system are y_1 and y_2 . Define state variables as

$$x_1 = y_1$$

$$x_2 = \dot{y}_1$$

$$x_3 = y_2$$

$$x_4 = \dot{y}_2$$

Then we obtain the following equations:

$$\begin{aligned} x_1 &= x_2 \\ \dot{x}_2 &= \frac{1}{m_1} \left[-b \dot{y}_1 - k (y_1 - y_2) \right] = -\frac{k}{m_1} x_1 - \frac{b}{m_1} x_2 + \frac{k}{m_1} x_3 \\ \dot{x}_3 &= x_4 \\ \dot{x}_4 &= \frac{1}{m_2} \left[-k (y_2 - y_1) + u \right] = \frac{k}{m_2} x_1 - \frac{k}{m_2} x_3 + \frac{1}{m_2} u \end{aligned}$$

Hence, the state equation is



Figure 3–54 Mechanical system.

and the output equation is

$$\begin{bmatrix} y_1 \\ y_2 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix}$$

A-3-13. Consider a system with multiple inputs and multiple outputs. When the system has more than one output, the command

produces transfer functions for all outputs to each input. (The numerator coefficients are returned to matrix NUM with as many rows as there are outputs.)

Consider the system defined by

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -25 & -4 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$
$$\begin{bmatrix} y_1 \\ y_2 \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$

This system involves two inputs and two outputs. Four transfer functions are involved: $Y_1(s)/U_1(s)$, $Y_2(s)/U_1(s)$, $Y_1(s)/U_2(s)$, and $Y_2(s)/U_2(s)$. (When considering input u_1 , we assume that input u_2 is zero and vice versa.)

Solution. MATLAB Program 3-5 produces four transfer functions.

MATLAB Program 3-5				
A = [0 B = [1 C = [1 D = [0 [NUM,0	1;-2 1;0 0;0 0;0 den] =	5 1] 1] 0] = ss2	-4]; ; ; ; 2tf(A,B,C,D,	1)
NUM =	:			
	0	1	4	
	0	0	-25	
den =				
	1	4	25	
[NUM,den] = ss2tf(A,B,C,D,2)				
NUM =				
0	1.0000		5.0000	
0	1.0000		-25.0000	
den =				
	1	4	25	

This is the MATLAB representation of the following four transfer functions:

$$\begin{aligned} \frac{Y_1(s)}{U_1(s)} &= \frac{s+4}{s^2+4s+25}, & \frac{Y_2(s)}{U_1(s)} &= \frac{-25}{s^2+4s+25} \\ \frac{Y_1(s)}{U_2(s)} &= \frac{s+5}{s^2+4s+25}, & \frac{Y_2(s)}{U_2(s)} &= \frac{s-25}{s^2+4s+25} \end{aligned}$$

A-3-14. Obtain the equivalent spring constants for the systems shown in Figures 3-55(a) and (b), respectively.

Solution. For the springs in parallel [Figure 3–55(a)] the equivalent spring constant k_{eq} is obtained from

$$k_1 x + k_2 x = F = k_{eq} x$$

or

$$k_{\rm eq} = k_1 + k_2$$

For the springs in series [Figure-55(b)], the force in each spring is the same. Thus

$$k_1 y = F, \qquad k_2 (x - y) = F$$

Elimination of y from these two equations results in

$$k_2\left(x-\frac{F}{k_1}\right)=F$$

or

$$k_2 x = F + \frac{k_2}{k_1} F = \frac{k_1 + k_2}{k_1} F$$

The equivalent spring constant k_{eq} for this case is then found as

$$k_{eq} \stackrel{\cdot}{=} \frac{F}{x} = \frac{k_1 k_2}{k_1 + k_2} = \frac{1}{\frac{1}{k_1} + \frac{1}{k_2}}$$



A-3-15. Obtain the equivalent viscous-friction coefficient b_{eq} for each of the systems shown in Figure 3-56(a) and (b).

Solution.

(a) The force f due to the dampers is

$$f = b_1 (\dot{y} - \dot{x}) + b_2 (\dot{y} - \dot{x}) = (b_1 + b_2)(\dot{y} - \dot{x})$$

In terms of the equivalent viscous friction coefficient b_{eq} , force f is given by

$$f = b_{\rm eq}(\dot{y} - \dot{x})$$

Hence

$$b_{\rm eq} = b_1 + b_2$$

(b) The force f due to the dampers is

$$f = b_1(\dot{z} - \dot{x}) = b_2(\dot{y} - \dot{z}) \tag{3-105}$$

where z is the displacement of a point between damper b_1 and damper b_2 . (Note that the same force is transmitted through the shaft.) From Equation (3–105), we have

$$(b_1 + b_2)\dot{z} = b_2\dot{y} + b_1\dot{x}$$

or

 $\dot{z} = \frac{1}{b_1 + b_2} (b_2 \dot{y} + b_1 \dot{x}) \tag{3-106}$

In terms of the equivalent viscous friction coefficient b_{eq} , force f is given by

$$f = b_{eq}(\dot{y} - \dot{x})$$

By substituting Equation (3-106) into Equation (3-105), we have

$$f = b_2(\dot{y} - \dot{z}) = b_2 \left[\dot{y} - \frac{1}{b_1 + b_2} (b_2 \dot{y} + b_1 \dot{x}) \right]$$
$$= \frac{b_1 b_2}{b_1 + b_2} (\dot{y} - \dot{x})$$

Thus,

$$f = b_{eq}(\dot{y} - \dot{x}) = \frac{b_1 b_2}{b_1 + b_2} (\dot{y} - \dot{x})$$

Hence,

$$b_{eq} = \frac{b_1 b_2}{b_1 + b_2} = \frac{1}{\frac{1}{b_1} + \frac{1}{b_2}}$$



Figure 3-56 (a) Two dampers connected in parallel; (b) two dampers connected in series.



A-3-16. Figure 3-57(a) shows a schematic diagram of an automobile suspension system. As the car moves along the road, the vertical displacements at the tires act as the motion excitation to the automobile suspension system. The motion of this system consists of a translational motion of the center of mass and a rotational motion about the center of mass. Mathematical modeling of the complete system is quite complicated.

> A very simplified version of the suspension system is shown in Figure 3-57(b). Assuming that the motion x_i at point P is the input to the system and the vertical motion x_o of the body is the output, obtain the transfer function $X_o(s)/X_i(s)$. (Consider the motion of the body only in the vertical direction.) Displacement x_o is measured from the equilibrium position in the absence of input x_i .

Solution. The equation of motion for the system shown in Figure 3-57(b) is

$$m\ddot{x}_{o} + b(\dot{x}_{o} - \dot{x}_{i}) + k(x_{o} - x_{i}) = 0$$

or

$$m\ddot{x}_{o} + b\dot{x}_{o} + kx_{o} = b\dot{x}_{i} + kx_{o}$$

Taking the Laplace transform of this last equation, assuming zero initial conditions, we obtain

$$(ms^{2} + bs + k)X_{o}(s) = (bs + k)X_{i}(s)$$

Hence the transfer function $X_o(s)/X_i(s)$ is given by

$$\frac{X_o(s)}{X_i(s)} = \frac{bs+k}{ms^2+bs+k}$$





Figure 3–57

A-3-17. Obtain the transfer function Y(s)/U(s) of the system shown in Figure 3-58. The input u is a displacement input. (Like the system of Problem **A-3-16**, this is also a simplified version of an automobile or motorcycle suspension system.)

Solution. Assume that displacements x and y are measured from respective steady-state positions in the absence of the input u. Applying the Newton's second law to this system, we obtain

$$m_1 \ddot{x} = k_2 (y - x) + b(\dot{y} - \dot{x}) + k_1 (u - x)$$

$$m_2 \ddot{y} = -k_2 (y - x) - b(\dot{y} - \dot{x})$$

Hence, we have

$$m_{1}\ddot{x} + b\dot{x} + (k_{1} + k_{2})x = b\dot{y} + k_{2}y + k_{1}u$$
$$m_{2}\ddot{y} + b\dot{y} + k_{2}y = b\dot{x} + k_{2}x$$

Taking Laplace transforms of these two equations, assuming zero initial conditions, we obtain

$$[m_1s^2 + bs + (k_1 + k_2)]X(s) = (bs + k_2)Y(s) + k_1U(s)$$
$$[m_2s^2 + bs + k_2]Y(s) = (bs + k_2)X(s)$$

Eliminating X(s) from the last two equations, we have

$$(m_1s^2 + bs + k_1 + k_2)\frac{m_2s^2 + bs + k_2}{bs + k_2}Y(s) = (bs + k_2)Y(s) + k_1U(s)$$

which yields

$$\frac{Y(s)}{U(s)} = \frac{k_1(bs+k_2)}{m_1m_2s^4 + (m_1+m_2)bs^3 + [k_1m_2 + (m_1+m_2)k_2]s^2 + k_1bs + k_1k_2}$$



Figure 3–58 Suspension system.

A-3-18. Obtain the transfer function of the mechanical system shown in Figure 3–59(a). Also obtain the transfer function of the electrical system shown in Figure 3–59(b). Show that the transfer functions of the two systems are of identical form and thus they are analogous systems.

Solution. In Figure 3–59(a) we assume that displacements x_i , x_o , and y are measured from their respective steady-state positions. Then the equations of motion for the mechanical system shown in Figure 3–59(a) are

$$b_1(\dot{x}_i - \dot{x}_o) + k_1(x_i - x_o) = b_2(\dot{x}_o - \dot{y})$$

$$b_2(\dot{x}_o - \dot{y}) = k_2 y$$

By taking the Laplace transforms of these two equations, assuming zero initial conditions, we have

$$b_1[sX_i(s) - sX_o(s)] + k_1[X_i(s) - X_o(s)] = b_2[sX_o(s) - sY(s)]$$
$$b_2[sX_o(s) - sY(s)] = k_2Y(s)$$

If we eliminate Y(s) from the last two equations, then we obtain

$$b_1[sX_i(s) - sX_o(s)] + k_1[X_i(s) - X_o(s)] = b_2sX_o(s) - b_2s\frac{b_2sX_o(s)}{b_2s + k_2}$$

or

$$(b_1s + k_1)X_i(s) = \left(b_1s + k_1 + b_2s - b_2s\frac{b_2s}{b_2s + k_2}\right)X_o(s)$$

Hence the transfer function $X_o(s)/X_i(s)$ can be obtained as

$$\frac{X_o(s)}{X_i(s)} = \frac{\left(\frac{b_1}{k_1}s + 1\right)\left(\frac{b_2}{k_2}s + 1\right)}{\left(\frac{b_1}{k_1}s + 1\right)\left(\frac{b_2}{k_2}s + 1\right) + \frac{b_2}{k_1}s}$$

For the electrical system shown in Figure 3–59(b), the transfer function $E_o(s)/E_i(s)$ is found to be

$$\frac{E_o(s)}{E_i(s)} = \frac{R_1 + \frac{1}{C_1 s}}{\frac{1}{(1/R_2) + C_2 s} + R_1 + \frac{1}{C_1 s}}$$
$$= \frac{(R_1 C_1 s + 1)(R_2 C_2 s + 1)}{(R_1 C_1 s + 1)(R_2 C_2 s + 1) + R_2 C_1 s}$$



Figure 3–59 (a) Mechanical system; (b) analogous electrical system. A comparison of the transfer functions shows that the systems shown in Figures 3-59(a) and (b) are analogous.

A-3-19. Obtain the transfer functions $E_o(s)/E_i(s)$ of the bridged T networks shown in Figures 3-60(a) and (b).

Solution. The bridged T networks shown can both be represented by the network of Figure 3-61(a), where we used complex impedances. This network may be modified to that, shown in Figure 3-61(b).

In Figure 3-61(b), note that

 $I_1 = I_2 + I_3, \qquad I_2 Z_1 = (Z_3 + Z_4)I_3$



Figure 3-60 Bridged *T* networks.



(a) Bridged T
network in terms of
complex impedances;
(b) equivalent
network.

Hence

$$I_2 = \frac{Z_3 + Z_4}{Z_1 + Z_3 + Z_4} I_1, \qquad I_3 = \frac{Z_1}{Z_1 + Z_3 + Z_4} I_1$$

Then the voltages $E_i(s)$ and $E_o(s)$ can be obtained as

$$E_{i}(s) = Z_{1}I_{2} + Z_{2}I_{1}$$

$$= \left[Z_{2} + \frac{Z_{1}(Z_{3} + Z_{4})}{Z_{1} + Z_{3} + Z_{4}}\right]I_{1}$$

$$= \frac{Z_{2}(Z_{1} + Z_{3} + Z_{4}) + Z_{1}(Z_{3} + Z_{4})}{Z_{1} + Z_{3} + Z_{4}}I_{1}$$

$$E_{o}(s) = Z_{3}I_{3} + Z_{2}I_{1}$$

$$= \frac{Z_{3}Z_{1}}{Z_{1} + Z_{3} + Z_{4}}I_{1} + Z_{2}I_{1}$$

$$= \frac{Z_{3}Z_{1} + Z_{2}(Z_{1} + Z_{3} + Z_{4})}{Z_{1} + Z_{3} + Z_{4}}I_{1}$$

Hence, the transfer function $E_o(s)/E_i(s)$ of the network shown in Figure 3-61(a) is obtained as

$$\frac{E_o(s)}{E_i(s)} = \frac{Z_3 Z_1 + Z_2 (Z_1 + Z_3 + Z_4)}{Z_2 (Z_1 + Z_3 + Z_4) + Z_1 Z_3 + Z_1 Z_4}$$
(3-107)

For the bridged T network shown in Figure 3-60(a), substitute

$$Z_1 = R, \qquad Z_2 = \frac{1}{C_1 s}, \qquad Z_3 = R, \qquad Z_4 = \frac{1}{C_2 s}$$

into Equation (3–107). Then, we obtain the transfer function $E_o(s)/E_i(s)$ to be

$$\frac{E_o(s)}{E_i(s)} = \frac{R^2 + \frac{1}{C_1 s} \left(R + R + \frac{1}{C_2 s}\right)}{\frac{1}{C_1 s} \left(R + R + \frac{1}{C_2 s}\right) + R^2 + R \frac{1}{C_2 s}}$$
$$= \frac{RC_1 RC_2 s^2 + 2RC_2 s + 1}{RC_1 RC_2 s^2 + (2RC_2 + RC_1)s + 1}$$

Similarly, for the bridged T network shown in Figure 3-60(b), we substitute

$$Z_1 = \frac{1}{Cs}, \qquad Z_2 = R_1, \qquad Z_3 = \frac{1}{Cs}, \qquad Z_4 = R_2$$

into Equation (3–107). Then the transfer function $E_o(s)/E_i(s)$ can be obtained as follows:

$$\frac{E_o(s)}{E_i(s)} = \frac{\frac{1}{Cs}\frac{1}{Cs} + R_1\left(\frac{1}{Cs} + \frac{1}{Cs} + R_2\right)}{R_1\left(\frac{1}{Cs} + \frac{1}{Cs} + R_2\right) + \frac{1}{Cs}\frac{1}{Cs} + R_2\frac{1}{Cs}}$$
$$= \frac{R_1CR_2Cs^2 + 2R_1Cs + 1}{R_1CR_2Cs^2 + (2R_1C + R_2C)s + 1}$$



Figure 3–62 Operationalamplifier circuit.

> **A-3-20.** Obtain the transfer function $E_o(s)/E_i(s)$ of the op-amp circuit shown in Figure 3-62. Solution. The voltage at point A is

$$e_A = \frac{1}{2} \left(e_i - e_o \right) + e_o$$

The Laplace-trasformed version of this last equation is

$$E_A(s) = \frac{1}{2} [E_i(s) + E_o(s)]$$

The voltage at point B is

$$E_B(s) = \frac{\frac{1}{Cs}}{R_2 + \frac{1}{Cs}} E_i(s) = \frac{1}{R_2 Cs + 1} E_i(s)$$

Since $[E_B(s) - E_A(s)]K = E_o(s)$ and $K \ge 1$, we must have $E_A(s) = E_B(s)$. Thus

$$\frac{1}{2} \left[E_i(s) + E_o(s) \right] = \frac{1}{R_2 C s + 1} E_i(s)$$

Hence

$$\frac{E_o(s)}{E_i(s)} = -\frac{R_2Cs - 1}{R_2Cs + 1} = -\frac{s - \frac{1}{R_2C}}{s + \frac{1}{R_2C}}$$

A-3-21. Obtain the transfer function $E_o(s)/E_i(s)$ of the op-amp system shown in Figure 3-63 in terms of complex impedances Z_1, Z_2, Z_3 , and Z_4 . Using the equation derived, obtain the transfer function $E_o(s)/E_i(s)$ of the op-amp system shown in Figure 3-62.

Solution. From Figure 3-63, we find

$$\frac{E_i(s) - E_A(s)}{Z_3} = \frac{E_A(s) - E_o(s)}{Z_4}$$

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Figure 3–63 Operationalamplifier circuit.

or

$$E_i(s) - \left(1 + \frac{Z_3}{Z_4}\right) E_A(s) = -\frac{Z_3}{Z_4} E_o(s)$$
(3-108)

Since

$$E_A(s) = E_B(s) = \frac{Z_1}{Z_1 + Z_2} E_i(s)$$
(3-109)

by substituting Equation (3-109) into Equation (3-108), we obtain

$$\left[\frac{Z_4Z_1+Z_4Z_2-Z_4Z_1-Z_3Z_1}{Z_4(Z_1+Z_2)}\right]E_i(s) = -\frac{Z_3}{Z_4}E_o(s)$$

from which we get the transfer function $E_o(s)/E_i(s)$ to be

$$\frac{E_o(s)}{E_i(s)} = -\frac{Z_4 Z_2 - Z_3 Z_1}{Z_3 (Z_1 + Z_2)}$$
(3-110)

To find the transfer function $E_o(s)/E_i(s)$ of the circuit shown in Figure 3–62, we substitute

$$Z_1 = \frac{1}{Cs}, \qquad Z_2 = R_2, \qquad Z_3 = R_1, \qquad Z_4 = R_5$$

into Equation (3-110). The result is

$$\frac{E_o(s)}{E_i(s)} = -\frac{R_1R_2 - R_1\frac{1}{Cs}}{R_1\left(\frac{1}{Cs} + R_2\right)} = -\frac{R_2Cs - 1}{R_2Cs + 1}$$

which is, as a matter of course, the same as that obtained in Problem A-3-20.

A-3-22. Obtain the transfer function $E_o(s)/E_i(s)$ of the operational-amplifier circuit shown in Figure 3-64. **Solution.** We will first obtain currents i_1, i_2, i_3, i_4 , and i_5 . Then we will use node equations at nodes A and B.

$$i_{1} = \frac{e_{1} - e_{A}}{R_{1}}; \qquad i_{2} = \frac{e_{A} - e_{o}}{R_{3}}, \qquad i_{3} = C_{1} \frac{de_{A}}{dt}$$
$$i_{4} = \frac{e_{A}}{R_{2}}, \qquad i_{5} = C_{2} \frac{-de_{o}}{dt}$$

At node A, we have $i_1 = i_2 + i_3 + i_4$, or

$$\frac{e_i - e_A}{R_1} = \frac{e_A - e_o}{R_3} + C_1 \frac{de_A}{dt} + \frac{e_A}{R_2}$$
(3-111)

At node *B*, we get $i_4 = i_5$, or

$$\frac{e_A}{R_2} = C_2 \frac{-de_o}{dt} \tag{3-112}$$

By rewriting Equation (3–111), we have

$$C_1 \frac{de_A}{dt} + \left(\frac{1}{R_1} + \frac{1}{R_2} + \frac{1}{R_3}\right)e_A = \frac{e_i}{R_1} + \frac{e_o}{R_3}$$
(3-113)

From Equation (3-112), we get

$$e_A = -R_2 C_2 \frac{de_o}{dt} \tag{3-114}$$

By substituting Equation (3-114) into Equation (3-113), we obtain

$$C_1\left(-R_2C_2\frac{d^2e_o}{dt^2}\right) + \left(\frac{1}{R_1} + \frac{1}{R_2} + \frac{1}{R_3}\right)\left(-R_2C_2\right)\frac{de_o}{dt} = \frac{e_i}{R_1} + \frac{e_o}{R_3}$$

Taking the Laplace transform of this last equation, assuming zero initial conditions, we obtain

$$-C_1C_2R_2s^2E_o(s) + \left(\frac{1}{R_1} + \frac{1}{R_2} + \frac{1}{R_3}\right)(-R_2C_2)sE_o(s) - \frac{1}{R_3}E_o(s) = \frac{E_i(s)}{R_1}$$

from which we get the transfer function $E_o(s)/E_i(s)$ as follows:

$$\frac{E_o(s)}{E_i(s)} = -\frac{1}{R_1 C_1 R_2 C_2 s^2 + [R_2 C_2 + R_1 C_2 + (R_1/R_3)R_2 C_2]s + (R_1/R_3)}$$



Figure 3--64 Operationalamplifier circuit. A-3-23. Consider the servo system shown in Figure 3-65(a). The motor shown is a servomotor, a dc motor designed specifically to be used in a control system. The operation of this system is as follows: A pair of potentiometers acts as an error-measuring device. They convert the input and output positions into proportional electric signals. The command input signal determines the angular position r of the wiper arm of the input potentiometer. The angular position r is the reference input to the system, and the electric potential of the arm is proportional to the angular position of the arm. The output shaft position determines the angular position c of the wiper arm of the input potentiometer the angular position r and the output potentiometer. The difference between the input angular position r and the output angular position c is the error signal e, or

$$e = r - c$$

The potential difference $e_r - e_c = e_v$ is the error voltage, where e_r is proportional to r and e_c is proportional to c; that is, $e_r = K_0 r$ and $e_c = K_0 c$, where K_0 is a proportionality constant. The error voltage that appears at the potentiometer terminals is amplified by the amplifier whose gain constant is K_1 . The output voltage of this amplifier is applied to the armature circuit of the dc motor. A fixed voltage is applied to the field winding. If an error exists, the motor develops a torque to rotate the output load in such a way as to reduce the error to zero. For constant field current, the torque developed by the motor is

$$T = K_2 i_c$$

where K_2 is the motor torque constant and i_a is the armature current.



Figure 3-65

(a) Schematic diagram of servo system; (b) block diagram for the system; (c) simplified block diagram.

When the armature is rotating, a voltage proportional to the product of the flux and angular velocity is induced in the armature. For a constant flux, the induced voltage e_b is directly proportional to the angular velocity $d\theta/dt$, or

$$e_b = K_3 \frac{d\theta}{dt}$$

where e_b is the back emf, K_3 is the back emf constant of the motor, and θ is the angular displacement of the motor shaft.

Obtain the transfer function between the motor shaft angular displacement θ and the error voltage e_v . Obtain also a block diagram for this system and a simplified block diagram when L_a is negligible.

Solution. The speed of an armature-controlled dc servomotor is controlled by the armature voltage e_a . (The armature voltage $e_a = K_1 e_v$ is the output of the amplifier.) The differential equation for the armature circuit is

$$L_a \frac{di_a}{dt} + R_a i_a + e_b = e_a$$

or

$$L_a \frac{di_a}{dt} + R_a i_a + K_3 \frac{d\theta}{dt} = K_1 e_v$$
(3-115)

The equation for torque equilibrium is

$$J_0 \frac{d^2\theta}{dt^2} + b_0 \frac{d\theta}{dt} = T = K_2 i_a$$
(3-116)

where J_0 is the inertia of the combination of the motor, load, and gear train referred to the motor shaft and b_0 is the viscous-friction coefficient of the combination of the motor, load, and gear train referred to the motor shaft.

By eliminating i_a from Equations (3–115) and (3–116), we obtain

$$\frac{\Theta(s)}{E_{\nu}(s)} = \frac{K_1 K_2}{s (L_a s + R_a) (J_0 s + b_0) + K_2 K_3 s}$$
(3-117)

We assume that the gear ratio of the gear train is such that the output shaft rotates n times for each revolution of the motor shaft. Thus,

$$C(s) = n\Theta(s) \tag{3-118}$$

The relationship among $E_{v}(s)$, R(s), and C(s) is

$$E_{v}(s) = K_{0}[R(s) - C(s)] = K_{0}E(s)$$
(3-119)

The block diagram of this system can be constructed from Equations (3-117), (3-118), and (3-119), as shown in Figure 3-65(b). The transfer function in the feedforward path of this system is

$$G(s) = \frac{C(s)}{\Theta(s)} \frac{\Theta(s)}{E_v(s)} \frac{E_v(s)}{E(s)} = \frac{K_0 K_1 K_2 n}{s[(L_a s + R_a)(J_0 s + b_0) + K_2 K_3]}$$

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When L_a is small, it can be neglected, and the transfer function G(s) in the feedforward path becomes

$$G(s) = \frac{K_0 K_1 K_2 n}{s [R_a (J_0 s + b_0) + K_2 K_3]}$$

= $\frac{K_0 K_1 K_2 n / R_a}{J_0 s^2 + (b_0 + \frac{K_2 K_3}{R_a})s}$ (3-120)

The term $[b_0 + (K_2K_3/R_a)]s$ indicates that the back emf of the motor effectively increases the viscous friction of the system. The inertia J_0 and viscous friction coefficient $b_0 + (K_2K_3/R_a)$ are referred to the motor shaft. When J_0 and $b_0 + (K_2K_3/R_a)$ are multiplied by $1/n^2$, the inertia and viscous-friction coefficient are expressed in terms of the output shaft. Introducing new parameters defined by

 $J = J_0/n^2$ = moment of inertia referred to the output shaft

$$B = [b_0 + (K_2 K_3 / R_a)] / n^2 = \text{viscous-friction coefficient referred to the output shaft}$$

$$K = K_0 K_1 K_2 / n R_a$$

the transfer function G(s) given by Equation (3-120) can be simplified, yielding

$$G(s) = \frac{K}{Js^2 + Bs}$$

or

$$G(s) = \frac{K_m}{s(T_m s + 1)}$$

where

$$K_m = \frac{K}{B}, \qquad T_m = \frac{J}{B} = \frac{R_a J_0}{R_a b_0 + K_2 K_3}$$

The block diagram of the system shown in Figure 3-65(b) can thus be simplified as shown in Figure 3-65(c).

A–3–24. Consider the system shown in Figure 3–66. Obtain the closed-loop transfer function C(s)/R(s).

Solution. In this system there is only one forward path that connects the input R(s) and the Output C(s). Thus,

$$P_1 = \frac{1}{C_1 s} \frac{1}{R_1} \frac{1}{C_2 s}$$





There are three individual loops. Thus,

$$L_{1} = -\frac{1}{C_{1}s}\frac{1}{R_{1}}$$
$$L_{2} = -\frac{1}{C_{2}s}\frac{1}{R_{2}}$$
$$L_{3} = -\frac{1}{R_{1}}\frac{1}{C_{2}s}$$

 $R_1 C_2 s$ Loop L_1 does not touch loop L_2 . (Loop L_1 touches loop L_3 , and loop L_2 touches loop L_3 .) Hence

Loop L_1 does not touch loop L_2 . (Loop L_1 touches loop L_3 , and loop L_2 touches loop L_3 .) Hence the determinant Δ is given by

$$\Delta = 1 - (L_1 + L_2 + L_3) + (L_1 L_2)$$

= 1 + $\frac{1}{R_1 C_1 s}$ + $\frac{1}{R_2 C_2 s}$ + $\frac{1}{R_1 C_2 s}$ + $\frac{1}{R_1 C_1 R_2 C_2 s^2}$

Since all three loops touch the forward path P_1 , we remove L_1, L_2, L_3 , and L_1L_2 from Δ and evaluate the cofactor Δ_1 as follows:

 $\Delta_1 = 1$

Thus we obtain the closed-loop transfer function to be

$$\frac{C(s)}{R(s)} = \frac{P_1 \Delta_1}{\Delta}$$

$$= \frac{\frac{1}{R_1 C_1 C_2 s^2}}{1 + \frac{1}{R_1 C_1 s} + \frac{1}{R_2 C_2 s} + \frac{1}{R_1 C_2 s} + \frac{1}{R_1 C_1 R_2 C_2 s^2}}$$

$$= \frac{R_2}{R_1 C_1 R_2 C_2 s^2 + (R_1 C_1 + R_2 C_2 + R_2 C_1)s + 1}$$

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A--3--25. Obtain the transfer function Y(s)/X(s) of the system shown in Figure 3–67.

> Solution. The signal flow graph shown in Figure 3-67 can be successively simplified as shown in Figures 3–68 (a), (b), and (c). From Figure 3–68(c), X_3 can be written as

$$X_3 = \frac{1}{s^2} X \div \frac{a_1 s + a_2}{s^2} X_3$$

This last equation can be simplified as

$$(s^2 + a_1s + a_2)X_3 = X$$

from which we obtain







Succesive

A-3-26. Figure 3-69 is the block diagram of an engine-speed control system. The speed is measured by a set of flyweights. Draw a signal flow graph for this system.

Solution. Referring to Figure 3–36(e), a signal flow graph for

$$\frac{Y(s)}{X(s)} = \frac{1}{s+140}$$

may be drawn as shown in Figure 3-70(a). Similarly, a signal flow graph for

$$\frac{Z(s)}{X(s)} = \frac{1}{s^2 + 140s + 100^2} = \frac{\frac{1}{s + 140} \frac{1}{s}}{1 + \frac{100^2}{s + 140} \frac{1}{s}}$$

may be drawn as shown in Figure 3-70(b).

Drawing a signal flow graph for each of the system components and combining them together, a signal flow graph for the complete system may be obtained as shown in Figure 3-70(c).







(a) Signal flow graph for Y(s)/X(s) = 1/(s + 140);
(b) signal flow graph for Z(s)/X(s) = 1/(s² + 140s + 100²);
(c) signal flow graph for the system shown in Fig. 3-69.

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$$z = x^2 + 4xy + 6y^2$$

in the region defined by $8 \le x \le 10, 2 \le y \le 4$.

Solution. Define

$$f(x, y) = z = x^2 + 4xy + 6y^2$$

Then

$$z = f(x, y) = f(\bar{x}, \bar{y}) + \left[\frac{\partial f}{\partial x}(x - \bar{x}) + \frac{\partial f}{\partial y}(y - \bar{y})\right]_{x = \bar{x}, y = \bar{y}} + \cdots$$

where $\bar{x} = 9$, $\bar{y} = 3$.

Since the higher-order terms in the expanded equation are small, neglecting these higherorder terms, we obtain

$$z-\bar{z}=K_1(x-\bar{x})+K_2(y-\bar{y})$$

where

$$K_{1} = \frac{\partial f}{\partial x} \bigg|_{x=\bar{x}, y=\bar{y}} = 2\bar{x} + 4\bar{y} = 2 \times 9 + 4 \times 3 = 30$$

$$K_{2} = \frac{\partial f}{\partial y} \bigg|_{x=\bar{x}, y=\bar{y}} = 4\bar{x} + 12\bar{y} = 4 \times 9 + 12 \times 3 = 72$$

$$\bar{z} = \bar{x}^{2} + 4\bar{x}\bar{y} + 6\bar{y}^{2} = 9^{2} + 4 \times 9 \times 3 + 6 \times 9 = 243$$

Thus

$$z - 243 = 30(x - 9) + 72(y - 3)$$

Hence a linear approximation of the given nonlinear equation near the operating point is

$$z - 30x - 72y + 243 = 0$$

B-3-1. Simplify the block diagram shown in Figure 3-71 and obtain the closed-loop transfer function C(s)/R(s).

 G_1

B-3-2. Simplify the block diagram shown in Figure 3-72 and obtain the transfer function C(s)/R(s).

B-3-3. Simplify the block diagram shown in Figure 3-73 and obtain the closed-loop transfer function C(s)/R(s).



Block diagram of a system.

B-3-4. Consider industrial automatic controllers whose control actions are proportional, integral, proportional-plus-integral, proportional-plus-derivative, and proportional-plus-integral-plus-derivative. The transfer functions of these controllers can be given, respectively, by

$$\frac{U(s)}{E(s)} = K_p$$

$$\frac{U(s)}{E(s)} = \frac{K_i}{s}$$

$$\frac{U(s)}{E(s)} = K_p \left(1 + \frac{1}{T_i s}\right)$$

$$\frac{U(s)}{E(s)} = K_p (1 + T_d s)$$

$$\frac{U(s)}{E(s)} = K_p \left(1 + \frac{1}{T_i s} + T_d s\right)$$

where U(s) is the Laplace transform of u(t), the controller output, and E(s) the Laplace transform of e(t), the actuat-



Figure 3–74 Closed-loop system.

ing error signal. Sketch u(t) versus t curves for each of the five types of controllers when the actuating error signal is

(a)
$$e(t)$$
 = unit-step function

(b) e(t) =unit-ramp function

In sketching curves, assume that the numerical values of K_{ρ} , K_i , T_i , and T_d are given as

$$K_p$$
 = proportional gain = 4
 K_i = integral gain = 2
 T_i = integral time = 2 sec
 T_d = derivative time = 0.8 sec

B-3-5. Figure 3-74 shows a closed-loop system with a reference input and disturbance input. Obtain the expression for the output C(s) when both the reference input and disturbance input are present.

B-3-6. Consider the system shown in Figure 3-75. Derive the expression for the steady-state error when both the reference input R(s) and disturbance input D(s) are present.

B-3-7. Obtain the transfer functions C(s)/R(s) and C(s)/D(s) of the system shown in Figure 3-76.



D(s)

Figure 3–75 Control system.



B-3-8. Obtain a state-space representation of the system shown in Figure 3-77.



Figure 3–77 Control system.

B-3-9. Consider the system described by

$$\ddot{y} + 3\ddot{y} + 2\dot{y} = u$$

Derive a state-space representation of the system.

B-3-10. Consider the system described by

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -4 & -1 \\ 3 & -1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 1 \end{bmatrix} u$$
$$y = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

Obtain the transfer function of the system.

B-3-11. Consider a system defined by the following statespace equations:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -5 & -1 \\ 3 & -1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 2 \\ 5 \end{bmatrix} u$$
$$y = \begin{bmatrix} 1 & 2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

Obtain the transfer function G(s) of the system.

B-3-12. Obtain the transfer matrix of the system defined by

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -2 & -4 & -6 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & 1 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$
$$\begin{bmatrix} y_1 \\ y_2 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}$$

B-3-13. Obtain the equivalent viscous-friction coefficient b_{eq} of the system shown in Figure 3-78.



Damper system.

B-3-14. Obtain mathematical models of the mechanical systems shown in Figure 3-79(a) and (b).



Figure 3–79 Mechanical systems.

B-3-15. Obtain a state-space representation of the mechanical system shown in Figure 3-80, where u_1 and u_2 are the inputs and y_1 and y_2 are the outputs.



Figure 3–80 Mechanical system.

B-3-16. Consider the spring-loaded pendulum system shown in Figure 3-81. Assume that the spring force acting on the pendulum is zero when the pendulum is vertical, or $\theta = 0$. Assume also that the friction involved is negligible and the angle of oscillation θ is small. Obtain a mathematical model of the system.





B-3-17. Referring to Examples 3-8 and 3-9, consider the inverted pendulum system shown in Figure 3-82. Assume that the mass of the inverted pendulum is m and is evenly distributed along the length of the rod. (The center of gravity of the pendulum is located at the center of the rod.) Assuming that θ is small, derive mathematical models for the system in the forms of differential equations, transfer functions, and state-space equations.



Figure 3--82 Inverted pendulum system.

B-3-18. Obtain the transfer functions $X_1(s)/U(s)$ and $X_2(s)/U(s)$ of the mechanical system shown in Figure 3-83.



Figure 3–83 Mechanical system.

B-3-19. Obtain the transfer function $E_o(s)/E_i(s)$ of the electrical circuit shown in Figure 3–84.



Electrical circuit.

B–3–20. Consider the electrical circuit shown in Figure 3–85. Obtain the transfer function $E_o(s)/E_i(s)$ by use of the block diagram approach.



Electrical circuit.

B-3-21. Derive the transfer function of the electrical circuit shown in Figure 3-86. Draw a schematic diagram of an analogous mechanical system.



Electrical circuit.

B-3-22. Obtain the transfer function $E_o(s)/E_i(s)$ of the op-amp circuit shown in Figure 3–87.



Figure 3–87 Operational-amplifier circuit.

B-3-23. Obtain the transfer function $E_o(s)/E_i(s)$ of the op-amp circuit shown in Figure 3-88.



Figure 3–88 Operational-amplifier circuit.

B-3-24. Using the impedance approach, obtain the transfer function $E_o(s)/E_i(s)$ of the op-amp circuit shown in Figure 3-89.



Figure 3–89 Operational-amplifier circuit.

B-3-25. Consider the system shown in Figure 3-90. An armature-controlled dc servomotor drives a load consisting of the moment of inertia J_L . The torque developed by the motor is *T*. The moment of inertia of the motor rotor is J_m . The angular displacements of the motor rotor and the load element are θ_m and θ , respectively. The gear ratio is $n = \theta/\theta_m$. Obtain the transfer function $\Theta(s)/E_i(s)$.



Figure 3–90 Armature-controlled dc servomotor system.

B-3-26. Obtain the transfer function Y(s)/X(s) of the system shown, in Figure 3-91.



Figure 3–91 Signal flow graph of a system.

B-3-27. Obtain the transfer function Y(s)/X(s) of the system shown in Figure 3-92.





B-3-28. Linearize the nonlinear equation

$$z = x^2 + 8xy + 3y^2$$

in the region defined by $2 \le x \le 4$, $10 \le y \le 12$.

B-3-29. Find a linearized equation for

$$y = 0.2x^{3}$$

about a point x = 2.