

ENCE 710

Design of Steel Structures

V. Lateral-Torsional Buckling of Beams

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Introduction

Following subjects are covered:

- Lateral Torsional Buckling (LTB)
- Flange Local Buckling (FLB)
- Web Local Buckling (WLB)
- Shear strength
- Lateral Bracing Design

Reading:

- Chapters 9 of Salmon & Johnson
- AISC LRFD Specification Chapters B (Design Requirements) and F (Design of Members for Flexure)

Introduction

A beam can fail by reaching the plastic moment and becoming fully plastic (see last section) or fail prematurely by:

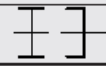



1. **LTB**, either elastically or inelastically
2. **FLB**, either elastically or inelastically
3. **WLB**, either elastically or inelastically

If the maximum bending stress is less than the proportional limit when buckling occurs, the failure is elastic. Else it is inelastic.

For bending $\phi_b M_n (\phi_b = 0.9)$

Design of Members for Flexure (about Major Axis)

TABLE User Note F1.1
Selection Table for the Application of Chapter F Sections

Section in Chapter F	Cross Section	Flange Slenderness	Web Slenderness	Limit States
F2		C	C	Y, LTB
F3		NC, S	C	LTB, FLB
F4		C, NC, S	C, NC	Y, LTB, FLB, TFY
F5		C, NC, S	S	Y, LTB, FLB, TFY

Lateral Torsional Buckling (LTB)

- Compact Members (AISC F2)
- Failure Mode
- Plastic LTB (Yielding)
- Inelastic LTB
- Elastic LTB
- Moment Gradient Factor C_b

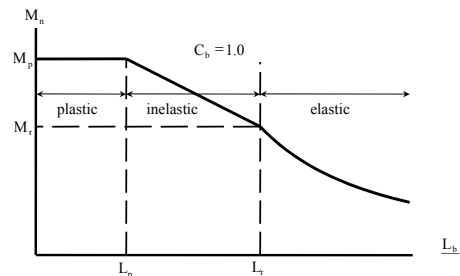
Lateral Torsional Buckling (cont.)

- Failure Mode
- A beam can buckle in a lateral-torsional mode when the bending moment exceeds the critical moment.



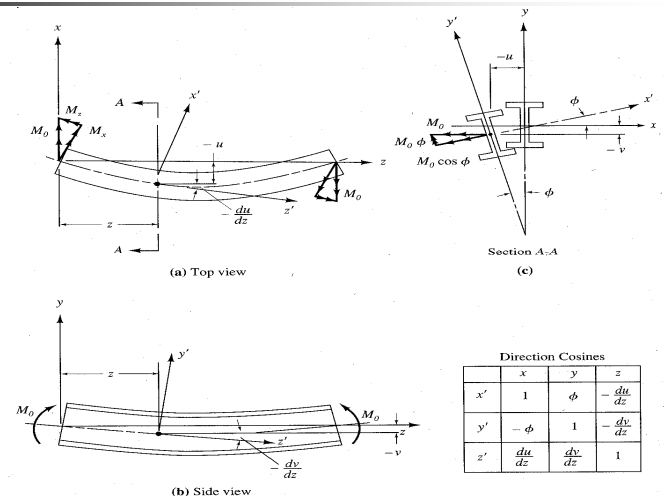
Lateral Torsional Buckling (cont.)

- Nominal Flexural Strength M_n
 - plastic when $L_b \leq L_p$ and $M_n = M_p$
 - inelastic when $L_p < L_b \leq L_r$ and $M_p > M_n \geq M_r$
 - elastic when $L_b > L_r$ and $M_n < M_r$



Lateral Torsional Buckling (cont.)

I-Beam
in a
Buckled
Position



Direction Cosines

	x	y	z
x'	1	ϕ	$-\frac{du}{dz}$
y'	$-\phi$	1	$-\frac{dv}{dz}$
z'	$\frac{du}{dz}$	$\frac{dv}{dz}$	1

Lateral Torsional Buckling (cont.)

Elastic LTB

- coupled differential equations for rotation and lateral translation

$$M_z = GJ \frac{d\phi}{dz} - EC_w \frac{d^3\phi}{dz^3} \quad (8.5.10)$$

where

- M_z = moment at location z along member axis
- z = axis along member length
- ϕ = angle of twist
- G = shear modulus
- J = torsional constant (AISC Table 1-1 for torsional prop.)
- E = modulus of elasticity
- C_w = warping constant (AISC Table 1-1 for warping)

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Lateral Torsional Buckling (cont.)

Plastic LTB (Yielding)

- Flexural Strength** $M_n = M_p = F_y Z$ (AISC F2-1)

where Z = plastic section modulus & F_y = section yield stress

Limits

- Lateral bracing limit $L_b < L_p = 1.76 r_y \sqrt{\frac{E}{F_y}}$ (AISC F2-5)
- Flange and Web width/thickness limit (AISC Table B4.1)

(Note: L_{pd} in Salmon & Johnson Eq. (9.6.2) is removed from AISC 13th Ed.)

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Lateral Torsional Buckling (cont.)

Inelastic LTB $L_p < L_b \leq L_r$

- Flexure Strength** (straight line interpolation)

$$M_n = C_b \left[M_p - (M_p - M_r) \left(\frac{L_b - L_p}{L_r - L_p} \right) \right] \leq M_p \quad (9.6.4)$$

or

$$M_n = C_b \left[M_p - (M_p - 0.7F_y S_x) \left(\frac{L_b - L_p}{L_r - L_p} \right) \right] \leq M_p \quad (\text{AISC F2-2})$$

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Lateral Torsional Buckling (cont.)

Elastic LTB $L_b > L_r$

- Flexure Strength**

$$M_n = F_{cr} S_x \leq M_p \quad (\text{AISC F2-3})$$

$$F_{cr} = \frac{C_b \pi^2 E}{\left(\frac{L_b}{r_s} \right)^2} \sqrt{1 + 0.078 \frac{Jc}{S_x h_o} \left(\frac{L_b}{r_s} \right)^2} \quad (\text{AISC F2-4})$$

(The square root term may be conservatively taken equal to 1.0)
(c in AISC F2-8a,b for doubly symmetric I-shape, and channel, respectively)

- Limit** $L_r = 1.95 r_s \frac{E}{0.7 F_y} \sqrt{\frac{Jc}{S_x h_o} \left[1 + \sqrt{1 + 6.76 \left(\frac{0.7 F_y S_x h_o}{E Jc} \right)^2} \right]}$ (AISC F2-6)

$$r_s^2 = \frac{\sqrt{I_y C_w}}{S_x} \quad (\text{AISC F2-7})$$

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Lateral Torsional Buckling (cont.)

Moment Gradient Factor C_b

- The moment gradient factor C_b accounts for the variation of moment along the beam length between bracing points. Its value is highest, $C_b=1$, when the moment diagram is **uniform** between adjacent bracing points.
- When the moment diagram is **not uniform**

$$C_b = \frac{12.5M_{\max}}{2.5M_{\max} + 3M_A + 4M_B + 3M_C} \quad (9.6.3) \quad (\text{AISC F1-1})$$

where

M_{\max} = absolute value of maximum moment in unbraced length
 M_A, M_B, M_C = absolute moment values at one-quarter, one-half, and three-quarter points of unbraced length

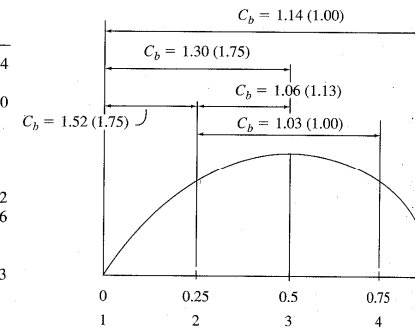
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C_b for a Simple Span Bridge

C_b FOR PARABOLIC SEGMENTS
 USING LRFD-F1.2a, FORMULA
 (C-F1-3), EQ. 9.6.11*

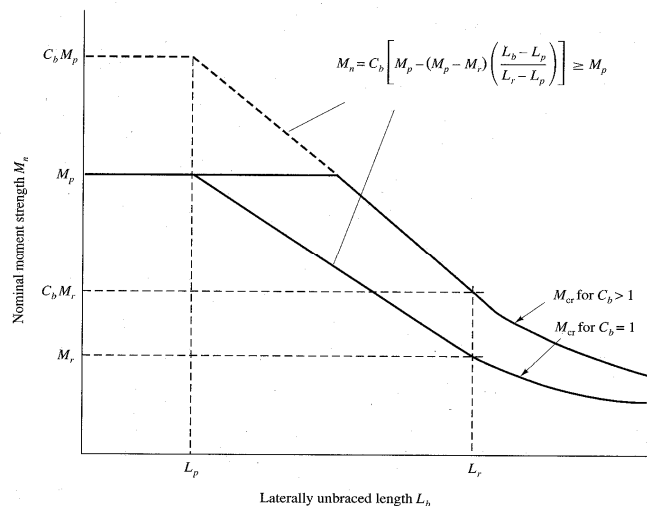
Case 1	Laterally braced at ends; points 1 and 5 only; M_{\max} at 3	$C_b = 1.14$
Case 2	Laterally braced at ends and midspan; points 1, 3, and 5 only; M_{\max} at 3	$C_b = 1.30$
Case 3	Laterally braced at end and 1st quarter point; bracing at points 1 and 2; M_{\max} at 2	$C_b = 1.52$
Case 4	Laterally braced at 1st and 2nd quarter points; bracing at points 2 and 3; M_{\max} at 3	$C_b = 1.06$
Case 5	Laterally braced at 1st and 3rd quarter points; bracing at points 2 and 4; M_{\max} at 3	$C_b = 1.03$

* Values from 1986 LRFD, Eq. 9.6.12 shown in parenthesis.



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Nominal Moment Strength M_u as affected by C_b



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Flange Local Buckling (FLB)

- Compact Web and Noncompact/Slender Flanges (AISC F3)
- Failure Mode
- Noncompact Flange
- Slender Flange
- Nominal Flexural strength, $M_n = \text{Min} (F2, F3)$

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Flange Local Buckling (cont.)

Failure Mode

The compression flange of a beam can buckle locally when the bending stress in the flange exceeds the critical stress.

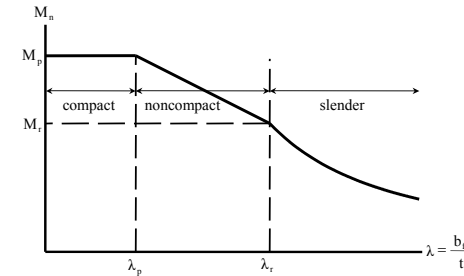


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Flange Local Buckling (cont.)

Nominal Flexural Strength M_n

- plastic when $b_f / 2t_f \leq \lambda_p$ and $M_n = M_p$
- inelastic when $\lambda_p \leq b_f / 2t_f \leq \lambda_r$ and $M_p > M_n \geq M_r$
- elastic when $b_f / 2t_f > \lambda_r$ and $M_n < M_r$



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Flange Local Buckling (cont.)

Noncompact Flange (straight line interpolation)

Flexure Strength

$$M_n = M_p - (M_p - 0.7F_y S_x) \left(\frac{\lambda - \lambda_{pf}}{\lambda_{rf} - \lambda_{pf}} \right) \quad (\text{AISC F3-1})$$

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Flange Local Buckling (cont.)

Slender Flange

Flexure Strength

$$M_n = \frac{0.9Ek_c S_x}{\lambda^2} \quad (\text{AISC F3-2})$$

$$k_c = \frac{4}{\sqrt{h/t_w}}$$

(k_c shall not be less than 0.35 and not greater than 0.76)

Limit (AISC Table B4.1)

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Web Local Buckling (WLB)

- Compact or Noncompact Webs (AISC F4)
- Failure Mode
- Compact Web (Yielding)
- Noncompact Web
- Slender Web
- Nominal Flexural Strength, $M_n = \min$ (compression flange yielding, LTB, compression FLB, tension flange yielding)

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Web Local Buckling (cont.)

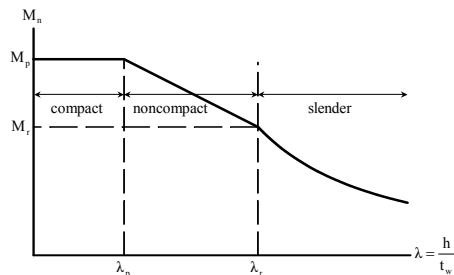
- Failure Mode
- The web of a beam can also buckle locally when the bending stress in the web exceeds the critical stress.



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Web Local Buckling (cont.)

- Nominal Flexural Strength M_n
 - plastic when $\lambda \leq \lambda_p$ and $M_n = M_p$
 - inelastic when $\lambda_p < \lambda \leq \lambda_r$ and $M_p > M_n \geq M_r$
 - elastic when $\lambda > \lambda_r$ and $M_n < M_r$



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Web Local Buckling (cont.)

- Compression Flange Yielding
 - Flexural Strength

$$M_n = R_{pc} M_{yc} = R_{pc} F_y S_{xc} \quad (\text{AISC F4-1})$$

where R_{pc} = web plasticification factor (AISC F4-9a, b) & F_y = section yield stress

- Limits (AISC Tables B4.1)

$$L_b < L_p = 1.1 r_t \sqrt{\frac{E}{F_y}}$$

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Web Local Buckling (cont.)

■ LTB (Inelastic) $L_p < L_b \leq L_r$

■ Flexure Strength

$$M_n = C_b \left[R_{pc} M_{yc} - (R_{pc} M_{yc} - F_L S_{xc}) \left(\frac{\lambda - \lambda_{pf}}{\lambda_{rf} - \lambda_{pf}} \right) \right] \leq M_p \quad (\text{AISC F4-12})$$

where F_L = a stress determined by AISC F4-6a, b

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Web Local Buckling (cont.)

■ LTB (Elastic) $L_b > L_r$

■ Flexure Strength

$$M_n = F_{cr} S_{xc} \leq R_{pc} M_{yc} \quad (\text{AISC F4-3})$$

$$F_{cr} = \frac{C_b \pi^2 E}{\left(\frac{L_b}{r_t} \right)^2} \sqrt{1 + 0.078 \frac{J}{S_x h_o} \left(\frac{L_b}{r_t} \right)^2} \quad (\text{AISC F4-5})$$

■ Limit (AISC Table B4.1)

$$L_r = 1.95 r_t \frac{E}{F_L} \sqrt{\frac{J}{S_x h_o}} \sqrt{1 + \sqrt{1 + 6.76 \left(\frac{F_L S_x h_o}{E J} \right)^2}} \quad (\text{AISC F4-8})$$

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Web Local Buckling (cont.)

■ Compression FLB (Noncompact Flange)

■ Flexure Strength

$$M_n = \left[R_{pc} M_{yc} - (R_{pc} M_{yc} - F_L S_{xc}) \left(\frac{\lambda - \lambda_{pf}}{\lambda_{rf} - \lambda_{pf}} \right) \right] \leq M_p \quad (\text{AISC F4-12})$$

■ Compression FLB (Slender Flange)

■ Flexure Strength

$$M_n = \frac{0.9 E k_c S_x}{\lambda^2} \quad (\text{AISC F4-13})$$

$$k_c = \frac{4}{\sqrt{h/t_w}}$$

(k_c shall not be less than 0.35 and not greater than 0.76)

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Web Local Buckling (cont.)

■ Tension Flange Yielding $S_{xt} < S_{xc}$

■ Flexure Strength

$$M_n = R_{pt} M_{yt} = R_{pt} F_y S_{xt} \quad (\text{AISC F4-14})$$

R_{pt} = web plastification factor to the tension flange yielding limit

$$(a) \quad h_d/t_w \leq \lambda_{pw} \quad R_{pt} = M_p/M_{yt} \quad (\text{AISC F4-15a})$$

$$(b) \quad h_d/t_w > \lambda_{pw} \quad (\text{AISC F4-15b})$$

$$R_{pt} = \left[\frac{M_p}{M_{yt}} - \left(\frac{M_p}{M_{yt}} - 1 \right) \left(\frac{\lambda - \lambda_{pw}}{\lambda_{rw} - \lambda_{pw}} \right) \right] \leq \frac{M_p}{M_{yt}}$$

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Shear Strength

- Failure Mode
- Shear-Buckling Coefficient
- Elastic Shear Strength
- Inelastic Shear Strength
- Plastic Shear Strength

For shear $\phi_v V_n$ ($\phi_v = 0.9$ except certain rolled I-beam $h/t_w \leq 2.24\sqrt{E/F_y}$, $\phi_v = 1.0$)

$$V_n = 0.6F_y A_w C_v \quad (\text{AISC G2-1})$$

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Shear Strength (cont.)

- Failure Mode

The web of a beam or plate girder buckles when the web shear stress exceeds the critical stress.

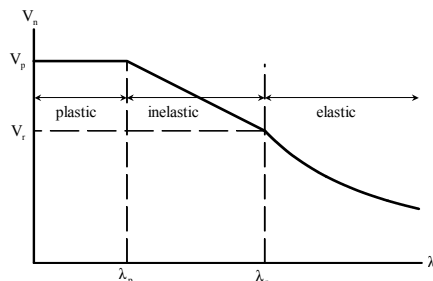


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Shear Strength (cont.)

- Nominal Shear Strength V_n ($\phi_v = 0.9$)

- plastic when $\lambda \leq \lambda_p$ and $\tau = \tau_y$
- inelastic when $\lambda \leq \lambda_r$ and $\tau = 0.8\tau_y$
- elastic when $\lambda > \lambda_r$ and $\tau = \tau_{cr}$



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Shear Strength (cont.)

- AISC G2 Nominal Shear Strength V_n

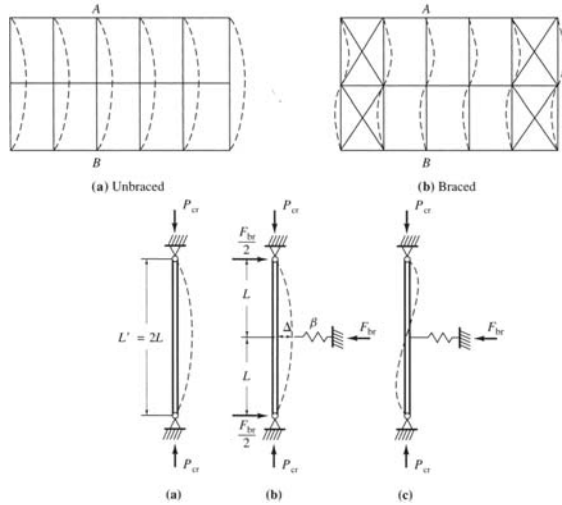
(a) For $\frac{h}{t_w} \leq 1.10\sqrt{\frac{k_v E}{F_{yw}}}$ $C_v = 1.0$ (AISC G2-3)

(a) For $1.10\sqrt{\frac{k_v E}{F_y}} \leq \frac{h}{t_w} \leq 1.37\sqrt{\frac{k_v E}{F_y}}$ $C_v = \left[\frac{1.10\sqrt{\frac{k_v E}{F_y}}}{h/t_w} \right]$ (AISC G2-4)

(a) For $1.37\sqrt{\frac{k_v E}{F_{yw}}} \leq \frac{h}{t_w}$ $C_v = \left[\frac{1.51Ek_v}{\left(\frac{h}{t_w}\right)^2 F_y} \right]$ (AISC G2-5)

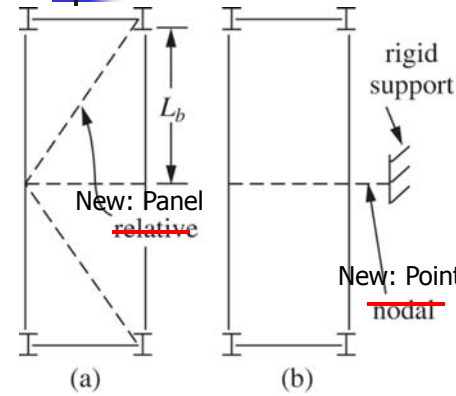
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Lateral Bracing Design



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Lateral Bracing Design



AISC Provisions – Stability Bracing Design for Beams

- 1. For stiffness β_{reqd}
 $\beta_{reqd} = 2 \beta_{ideal}$
- 2. For nominal strength F_{br}
 (a) $F_{br} = \beta_{ideal} (2\Delta_0)$;
 (b) $F_{br} = \beta_{ideal} (0.004L_b)$
 Where $\beta_{ideal} = P_{cr}/L_b$

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Lateral Bracing Design (cont.)

AISC Provisions – LRFD Stability Bracing Design for **Columns**

1. Panel bracing

$$Required V_{br} = 0.005P_r \quad (A-6-1)$$

$\Phi=0.75$

$$Required \beta_{rb} = \frac{1}{\phi} \left(\frac{2P_r}{L_b} \right) \quad (A-6-2a)$$

2. Point bracing

$$Required P_{rb} = 0.01P_r \quad (A-6-3)$$

$$Required \beta_{br} = \frac{1}{\phi} \left(\frac{8P_r}{L_b} \right) \quad (A-6-4a)$$

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Lateral Bracing Design (cont.)

AISC Provisions – LRFD Stability Bracing Design for **Beams**

1. Lateral Bracing

1a. Panel bracing

$$Required V_{br} = 0.01 \left(\frac{M_r C_d}{h_o} \right) \quad (A-6-5)$$

$\Phi=0.75$

$$Required \beta_{rb} = \frac{1}{\phi} \left(\frac{4M_r C_d}{L_b h_o} \right) \quad (A-6-6a)$$

1b. Point bracing

$$Required V_{br} = 0.02 \left(\frac{M_r C_d}{h_o} \right) \quad (A-6-7)$$

$$Required \beta_{br} = \frac{1}{\phi} \left(\frac{10M_r C_d}{L_b h_o} \right) \quad (A-6-8a)$$

2. Torsional Bracing

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