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# Impeller Optimized Design of the Centrifugal Pump: A Numerical and Experimental Investigation

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**Abstract:** Combined numerical simulation with experiment, blade wrap angle, and blade exit angle are varied to investigate the optimized design of the impeller of centrifugal pump. Blade wrap angles are 122°, 126°, and 130°. Blade exit angles are 24°, 26°, and 28°. Based on numerical simulation, internal flow of the centrifugal pump with five different impellers under 0.6, 0.8, 1.0, 1.2, and 1.5  $Q_d$  are simulated. Variations of static pressure, relative velocity, streamline, and turbulent kinetic energy are analyzed. The impeller with blade wrap angle 126° and blade exit angle 24° are optimal. Distribution of static pressure is the most uniform and relative velocity sudden changes do not exist. Streamlines are the smoothest. Distribution scope of turbulent kinetic energy is the smallest. Based on performance experiments, head and efficiency of the centrifugal pump with the best impeller are tested. The values of head and efficiency are higher than that of the original pump. Centrifugal pump with the best impeller has better hydraulic performance than the original centrifugal pump.

Keywords: centrifugal pump; impeller; blade wrap angle; blade exit angle; optimized design

# 1. Introduction

A centrifugal pump is one kind of general machinery [1] which is widely and fully utilized in the industrial and agricultural fields [2], such as irrigation and water supply. It normally includes four different parts: suction pipe, impeller, volute, and exit pipe. The impeller is the core part and it converts the mechanical energy into pressure energy [3], which directly determines the transport capacity and the hydraulic performances of centrifugal pump. So, optimized design of the impeller is essential and significant for the efficient operation of a centrifugal pump [4,5].

One-dimensional flow theory, two-dimensional flow theory, and three-dimensional flow theory are three basic theories to optimally design the centrifugal pump [6,7]. Two-dimensional flow theory regards the distribution of meridian velocity across the cross-section as symmetrical. The flow is one kind of potential flow. Optimized design of hydraulic machinery in engineering fields, which is based on two-dimensional flow theory, is rare. It is only employed to optimally design the higher specific speed impellers of mixed-flow pump and runners of mixed-flow turbine. Three-dimensional flow theory, proposed by Wu [8], is also called  $S_1$  and  $S_2$  relative flow surface theory. In this method, the three-dimensional flow is converted into two two-dimensional flows in the surfaces  $S_1$  and  $S_2$ . Only in the ideal conditions of flow being inviscid, incompressible, and unsteady, can this method be applied to design the impeller successfully. This method is difficult and complex. Thus, it is not always

employed to optimally design an impeller for a centrifugal pump in the engineering field. However, one-dimensional flow theory, which is based on Euler equations, is convenient and simple. Flow in the meridional surface is symmetrical. Engineers coming from home and abroad factories widely employ one-dimensional flow theory to optimally design an impeller for a centrifugal pump [9–11].

Blade exit angle ( $\Phi$ ) and blade wrap angle ( $\beta_2$ ) are two significant parameters in the impeller optimized design process according to one-dimensional flow theory [12–15]. The hydraulic performance of a centrifugal pump is mainly determined by these parameters. In the actual optimized design process, blade exit angle often varies from 15° to 40° and blade wrap angle often varies from  $90^{\circ}$  to  $130^{\circ}$  [16]. Some scholars studied the effects of blade exit angle and blade wrap angle on the hydraulic performances of the centrifugal pump [17–19]. The selection of blade wrap angle and blade exit angle mainly depend on the specific speed of the centrifugal pump and the actual optimized design experience. If the blade wrap angle is too large, the impeller friction area becomes large, which is bad for the improvement of efficiency. If the blade wrap angle is too small, the impeller cannot control the flow of water effectively and the impeller cannot operate stably. For the blade exit angle, a moderate value could effectively improve the hydraulic performances of the centrifugal pump. Comprehensively considering the specific speed of the target centrifugal pump ( $n_s = 112$ ) and the effects of blade wrap angle and blade exit angle on the hydraulic performances of the centrifugal pump, the variation scope of the blade wrap angle and blade exit angle was selected carefully. The blade wrap angle was tested in the range of  $120^{\circ}$  to  $130^{\circ}$  and the values were  $\Phi = 122^{\circ}$ ,  $126^{\circ}$ , and  $130^{\circ}$ , respectively. The blade exit angle was tested in the range of 20° to 30° and the values were  $\beta_2 = 24^\circ$ , 26°, and 28°, separately.

#### 2. Basic Parameters of the Impeller

The inlet diameter of the impeller  $D_1 = 200$  mm and the outlet diameter  $D_2 = 420$  mm. The exit width  $b_2 = 34$  mm. The original blade wrap angle  $\Phi = 120^{\circ}$ . The original blade exit angle  $\beta_2 = 26^{\circ}$ . The basic parameters of the impeller are shown in Figure 1.



Figure 1. Two-dimensional model of the impeller.

## 3. Method to Modify the Impeller

The impeller shape is directly determined by the blade profiles. In the impeller optimization design process, all blade profiles must be smooth. The bend of the blade profiles should be in one single direction. An 'S' shape should not exist [16,20,21]. All blade profiles should be symmetrical. The blade profile differential equation is

$$d\theta = \frac{\mathrm{d}s}{r\tan\beta} \tag{1}$$

Here, *s* is the axial streamline.  $\theta$  denotes the axial angle. *r* represents the radius of one arbitrary point in the blade profile.  $\beta$  is the blade angle.

The speed triangle is shown in Figure 2. Here, *c* represents absolute velocity, *w* is the relative velocity, and *u* is the following velocity.  $c_m$  and  $w_m$  are the component of absolute velocity and relative velocity in the meridional plane, respectively.  $c_u$  is the component in the circumferential plane.

Based on the speed triangle,  $tan\beta$  could be described as

$$\tan\beta = \frac{c_m}{u - c_u} \tag{2}$$

Thus,

$$\frac{\mathrm{d}s}{r\mathrm{d}\theta} = \frac{c_m}{u - c_u} \tag{3}$$

So,

$$ds = \frac{c_m}{u - c_u} r d\theta \tag{4}$$

According to the above equations, blade profile optimized design has a closed relation with blade inlet angle ( $\beta_1$ ), blade exit angle ( $\beta_2$ ), and blade wrap angle ( $\Phi$ ). Three different methods are usually employed to modify the blade profiles. They are that

(1)  $\beta_1$  is the constant.  $\beta_2$  and  $\Phi$  are changed.

(2)  $\beta_1$  and  $\beta_2$  are fixed.  $\Phi$  is varied.

(3)  $\beta_2$  and  $\Phi$  are invariant.  $\beta_1$  is altered.



Figure 2. Speed triangle.

Method 1 ( $\beta_1$  is the constant.  $\beta_2$  and  $\Phi$  are changed.) was always employed to optimally design the impeller. As shown in Table 1, the blade exit angle of Schemes 1–3 was identical and the blade wrap angles varied from 122° to 130°, which were employed to investigate the effects of different blade wrap angles on the optimized design of the centrifugal pump. For Schemes 2, 4, and 5, the blade wrap angle was the same and the blade exit angle was changed, which was used to discuss the effects of diverse blade exit angles on the optimized design of the centrifugal pump.

Table 1. Optimized schemes of the blade profile.

Scheme	1	2	3	4	5
Φ	122°	$126^{\circ}$	$130^{\circ}$	$126^{\circ}$	126°
$\beta_2$	$26^{\circ}$	$26^{\circ}$	$26^{\circ}$	$28^{\circ}$	$24^{\circ}$

The impeller optimized design procedure is that, via the analysis of effects of blade wrap angle (Schemes 1–3) on the hydraulic performances of the centrifugal pump, one optimal scheme could be determined via computational fluid dynamics (CFD). According to the discussion of effects of blade exit angle (Schemes 2, 4, and 5) on the variations of head and efficiency, one optimal scheme could be

got by numerical simulation. Then, these two different schemes are compared and the best one could be obtained. The optimized procedure includes two main parts, as shown in Figure 3.



Figure 3. Flow chart of the optimized design of the impeller.

The physical models of the blades under different blade wrap angle and blade exit angle conditions are given in Figure 4.



Figure 4. Physical model of the blades.

# 4. Numerical Method

#### 4.1. Fundamental Equations

The fundamental equations [22] are employed to describe the flow characteristics in the centrifugal pump, which include two main parts: continuity equation and motion equation, corresponding with mass conservation law and momentum conservation law.

$$\frac{\partial \rho}{\partial t} + \frac{\partial (\rho u_i)}{\partial x_i} = 0 \tag{5}$$

$$\rho \frac{\partial u_i}{\partial t} + \rho u_j \frac{\partial u_i}{\partial x_j} = \rho f_i - \frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} \left[ \mu \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \right] - \frac{2}{3} \frac{\partial}{\partial x_i} \left( \mu \frac{\partial u_j}{\partial x_j} \right)$$
(6)

Here,  $\rho$  is the density of water. *t* represents the time. *u* denotes the velocity. *x* is the Cartesian coordinate. *f*<sub>i</sub> is the body force vector. *p* is the pressure.  $\mu$  is the dynamic viscosity.

#### 4.2. Turbulent Model

RNG *k*- $\varepsilon$  turbulent model proposed by Yakhot and Orzag [23] was employed to deal with turbulent flow. In the centrifugal pump, the impeller is the rotating part whose rotation effects could be fully dealt by the turbulent dissipation rate ( $\varepsilon$ ) equation in this turbulent model. On the other hand, the RNG *k*- $\varepsilon$  turbulent model has high-precision, which could guarantee the accuracy of the numerical results.

$$\frac{\partial(\rho k)}{\partial t} + \frac{\partial(\rho k u_i)}{\partial x_i} = \frac{\partial}{\partial x_j} \times \left[ (\alpha_k (\mu + \mu_t)) \frac{\partial k}{\partial x_j} \right] + G_k + \rho \varepsilon$$
(7)

$$\frac{\partial(\rho\varepsilon)}{\partial t} + \frac{\partial(\rho\varepsilon u_i)}{\partial x_i} = C_{1\varepsilon}\frac{\varepsilon}{k}G_kC_{2\varepsilon}\rho\frac{\varepsilon^2}{k} + \frac{\partial}{\partial x_j}\left[\alpha_\varepsilon(\mu+\mu_t)\frac{\partial\varepsilon}{\partial x_j}\right]$$
(8)

$$\mu_{\rm t} = \rho c_{\mu} \frac{k^2}{\varepsilon} \tag{9}$$

Here, *k* is the turbulent kinetic energy.  $\varepsilon$  is the turbulent dissipation rate.  $\mu_t$  is the turbulent viscosity. The five terms,  $C_{1\varepsilon}$ ,  $C_{2\varepsilon}$ ,  $\alpha_k$ ,  $\alpha_{\varepsilon}$ , and  $c_{\mu}$  are empirical coefficients and the values are 1.42, 1.68, 1.39, 1.39, and 0.09, separately.  $G_k$  is one generation term of turbulent kinetic energy which is caused by the mean velocity gradient.

## 5. Numerical Simulation Setup

#### 5.1. Physical Model

Three-dimensional (3D) single-stage and single-suction centrifugal pump was employed, as shown in Figure 5. The suction pipe was employed to keep the uniform of the flow. Exit pipe was utilized to avoid the backflow. The performance parameters were designed flow rate  $Q_d = 550 \text{ m}^3/\text{h}$ ; designed head  $H_d = 50 \text{ m}$ ; rated motor power P = 110 KW; and rated rotational speed n = 1480 rpm. The geometric parameters of the impeller and the volute were shown in Table 2.



Figure 5. Physical model of the centrifugal pump.

	Table 2. (	Geometric	parameters	of the	impeller	and v	olute
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Parameter	Value
Impeller inlet diameter $D_1$ , mm	200
Impeller outlet diameter $D_2$ , mm	420
Impeller exit width $b_2$ , mm	34
Number of blade Z	6
Base diameter of the volute $D_3$ , mm	435
Volute inlet width $b_3$ , mm	72
Volute outlet diameter $D_4$ , mm	250

To get accurate numerical simulation results a balance hole, front chamber, and back chamber were added to the 3D physical model of the centrifugal pump, as displayed in Figures 6 and 7. Diameter of the balance hole was 10mm and it was mainly employed to balance the axial force [24].



Figure 6. Physical model of the impeller.



Figure 7. Chamber of the centrifugal pump.

# 5.2. Mesh Generation

ANSYS-ICEM (15.0, ANSYS, Inc., Canonsburg, PA, USA) is employed to discrete the centrifugal pump computational domains. To guarantee the uniformity of the flow, structured meshes were employed to discretize the computational domains of the suction pipe and exit pipe, as displayed in Figure 8d–e. Fully considering the complex structure of the impeller and the volute, to get the better adaptability of the flow, unstructured meshes were employed to discrete the computational domains of impeller, front and back chamber, and volute, as shown in Figure 8a–c. Meshes in the leading edges of the impeller and balance hole were refined.



Figure 8. Cont.



Figure 8. Meshes of the computational domain of the centrifugal pump.

Mesh numbers of the five different impellers were 3,472,923, 3,473,052, 3,471,896, 3,472,816, and 3,473,011, respectively. All mesh quality was higher than 0.3. Mesh number and quality of front and back chamber, volute, suction pipe, and exit pipe were displayed in Table 3.

Table 3. Mesh number and quality

	Front and Back Chamber	Volute	Suction Pipe	Exit Pipe
Mesh number	934,607	2,238,179	498,506	327,624
Mesh quality	>0.4	>0.4	>0.8	>0.8

Head under designed flow rate condition was calculated to verify mesh independence. Results indicated that with the increase of mesh number, head increased firstly. Then, the differences of head were slight, as shown in Figure 9. Total mesh numbers of the five different centrifugal pumps were 7,139,826, 7,146,497, 7,143,016, 7,145,682, and 7,147,024, separately.



Figure 9. Meshes independence.

Reynolds averaged Naiver–Stokes (RANS) method was employed to simulate the internal flow of the centrifugal pump. At inlet, the velocity was set. At outlet, the free outflow was set. Near wall flow was treated by standard wall function. The interface was set between suction pipe and impeller, impeller and volute, volute and exit pipe. The SIMPLE scheme was used to solve the coupled equations of velocity. The PRESTO! was employed to compute pressure. The second order upwind scheme was used for the solving of momentum, turbulent kinetic energy, and turbulent dissipation rate. All residuals were less than  $1.0 \times 10^{-5}$ .

## 5.4. Verification of the Algorithm

The above-mentioned algorithm is employed to numerically simulate the internal flow of the original centrifugal pump and optimized centrifugal pump. To verify the reasonableness of the algorithm used, flow in one single-stage and single suction centrifugal pump which is shown in Figure 10 was simulated under different flow rate conditions. The rated flow rate is  $Q_d = 500 \text{ m}^3/\text{h}$  and the rated head is  $H_d = 53 \text{ m}$ . Rated rotating speed is n = 1480 rpm.



Figure 10. Experimental centrifugal pump.

The numerical results of head and efficiency had a good agreement with the experimental results, as displayed in Figure 11. The maximum relative error of the head was less than 3.7%. The largest fractional error of the efficiency was less than 2.2%. The algorithm designed was reasonable.



**Figure 11.** Comparison of the hydraulic performances of centrifugal pump. (**a**) Head (*H*)-flow rate (*Q*); (**b**) Efficiency ( $\eta$ )-flow rate (*Q*).

#### 6. Numerical Simulation Results Analysis

To measure the effects of blade wrap angle and blade exit angle on the performances of centrifugal pump, variations of static pressure, relative velocity, streamlines, and turbulent kinetic energy in the middle span of the impeller were analyzed under the typical flow rate conditions (low flow rate 0.6  $Q_d$  and 0.8  $Q_d$ , rated flow rate 1.0  $Q_d$ , and high flow rate 1.2  $Q_d$  and 1.5  $Q_d$ ). Also, the variations of head and efficiency under the above flow rate conditions were compared.

#### 6.1. Variation of Static Pressure

The overall static pressure variation law is that static pressure in different passages distributed uniformly. Differences of static pressure in all passages were relatively small, as shown in Figure 12. At the inlet of the impeller, static pressure was the lowest. Static pressure at the outlet of the impeller was the highest. For Impellers 1–5, static pressure increased overall with the growing of flow rate. Under 0.8  $Q_d$  condition, the increase of static pressure was more manifest. At outlet, static pressure attained the maximum under 1.5  $Q_d$  condition.



Figure 12. Cont.



Figure 12. Variation of static pressure in different impellers.

Static pressure in Impellers 1, 2, and 4 fluctuated more obviously and the pressure gradient was higher, which was caused by the intense rotor-stator interaction of the impeller and the volute [25]. Volumetric loss of the centrifugal pump is severe. Thus, the centrifugal pump could not operate efficiently. For Impeller 2, under low flow rate conditions, the distribution scope of lower pressure was much larger than that of other impellers. Centrifugal pumps operating under these conditions were unstable. Compared with Impellers 1, 2, 4, and 5, the lower pressure region in Impeller 3 was much larger under 0.8  $Q_d$  and 1.2  $Q_d$  conditions, mainly caused by secondary flow. Cavitation in Impeller 3 could occur easily, which could make the performances of the centrifugal pump decrease sharply [26]. In Impeller 5, the pressure gradient was the smallest and the lower pressure region was the smallest under all flow rate conditions, proving that the optimization of Impeller 5 is the best. Impeller with  $\Phi = 126^{\circ}$  and blade exit angles  $\beta_2 = 24^{\circ}$  could guarantee the safety operation of the centrifugal pump.

#### 6.2. Variation of Relative Velocity

At the inlet of the impeller, the relative velocity was small. At the outlet of the impeller, the relative velocity attained the maximum, which was in agreement with the experimental and theoretical results [27]. Relative velocity in the pressure surface was much smaller than that of suction pressure. With the increase of flow rate, the low relative speed zone became smaller, as shown in Figure 13.

Under low flow rate conditions, low relative velocity zone in Impeller 5 was smaller than that of other impellers, which could guarantee the stable operation of the centrifugal pump. Velocity gradient in Impellers 1–4 was larger and the flow in these impellers was disorderly. Under rated flow rate conditions, the low speed velocity zone in Impeller 1 was the most obvious. Although differences of low speed zones of Impellers 2–5 were smaller, the zone of Impeller 5 was smaller than that of Impellers 2–4. Under high flow rate conditions, the low velocity speed zone of Impeller 5 was the smallest. Velocity sudden change did not appear. The distribution of relative velocity was uniform, which reflected that blade wrap angle  $\Phi = 126^{\circ}$  and blade exit angle  $\beta_2 = 24^{\circ}$  could let the centrifugal pump operate stably. The hydraulic performances of Impeller 5 were better than others.



Figure 13. Cont.



Figure 13. Variation of absolute velocity in different impellers.

# 6.3. Variation of Streamlines

Under low flow rate conditions, streamlines in Impellers 1–4 were disorder and there were manifest vortices, especially for 0.6  $Q_d$  condition, which could cause severe energy loss and secondary flow. For Impeller 2, the vortices were the most obvious, as shown in Figure 14b. The centrifugal pumps under these two flow rate conditions operated unstably, which could induce severe vibrations and noises [28] and the efficiency of the centrifugal pumps decreased. However, the streamlines in Impeller 5 under these two conditions were smooth. The secondary flow could be avoided successfully. Compared with low flow rate conditions, streamlines in rated and high flow rate conditions became smooth. However, streamlines of Impellers 1–4 were more disorder than that of Impeller 5. In Impellers 1, 2, and 4, the low speed region on the pressure surface was dramatic, which could let the flow in suction surface was faster. Energy loss in this region could be caused easily. Overall, the streamlines in Impeller 5 were the smoothest, as displayed in Figure 14e, which could result in lower energy loss. The flow separation and back flow could be controlled. The centrifugal pump could operate stably.



Figure 14. Variation of streamlines in different impeller.

## 6.4. Variation of Turbulent Kinetic Energy

The distribution characteristics of turbulent kinetic energy in the middle span of the impeller were sharply different under diverse flow rate conditions, which were shown in Figure 15. The distribution scope became smaller gradually with the increase of flow rates. Under the 0.6  $Q_d$  condition, the distribution range of turbulent kinetic energy was the largest and under the 1.5  $Q_d$  condition, the scope was the smallest for all the impellers.

Under low flow rate conditions, the distribution scope of turbulent kinetic energy in Impeller 2 was larger than that of other impellers. Velocity vector had obvious change when water flows from the impeller to the volute. Centrifugal pump with Impeller 2 could not operate stably. Under rated flow rate conditions, the turbulent kinetic energy distribution scopes of Impellers 1, 3, and 4 were similar and they were slightly larger than that of Impeller 5. For high flow rate conditions, the distribution scope of turbulent kinetic energy was almost identical for all the impellers.

Based on the above analysis of turbulent kinetic energy and according to Equation (7), turbulent intensity l [29] of Impeller 5 was less intense than that of Impellers 1–4 overall. Thus, the energy loss in Impeller 5 was less than other impellers, which could guarantee the high transportation capacity of the centrifugal pump.

$$k = \frac{3}{2} (v_{\rm in} l)^2 \tag{10}$$

Here,  $v_{in}$  is the inlet velocity. *l* is the turbulent intensity.



Figure 15. Cont.



Figure 15. Variation of turbulent kinetic energy in different impellers.

# 6.5. Variations of Head and Efficiency

With the increase of flow rate, head decreased gradually and efficiency increased firstly then decreased, as exhibited in Figure 16. Differences of head were relatively slight. A head with  $\Phi = 122^{\circ}$  was higher than that of  $\Phi = 126^{\circ}$ . The head of the centrifugal pump with  $\Phi = 130^{\circ}$  was the highest one. Efficiency attained the maximum under designed flow rate. Efficiency with  $\Phi = 130^{\circ}$  was higher than that of  $\Phi = 122^{\circ}$  and  $126^{\circ}$ . Compared with Schemes 1 and 2, Scheme 3 was the optimal one.



Figure 16. Hydraulic performances of the centrifugal pump with different blade wrap angles.

Schemes 2, 4, and 5 reflected the effects of blade exit angles on hydraulic performances of the centrifugal pump, as displayed in Figure 17. Head decreased constantly and efficiency increased firstly then decreased with the increase of flow rate. The efficiency attained the maximum under the rated flow rate. Head with  $\beta_2 = 24^\circ$  was slightly higher than that of  $\beta_2 = 28^\circ$ . Differences of head with

 $\beta_2 = 26^\circ$  and  $28^\circ$  were sharp. The distinctions of efficiency were smaller. Efficiency of centrifugal pump with  $\beta_2 = 24^\circ$  was the highest. Compared with Schemes 2, 4, and 5, the centrifugal pump with  $\beta_2 = 24^\circ$  had the best hydraulic performances.



Figure 17. Hydraulic performances of the centrifugal pump with different blade exit angles.

In Scheme 3, blade wrap angle  $\Phi = 130^{\circ}$  was the optimal one. The corresponding blade exit angle  $\beta 2 = 26^{\circ}$ . In Scheme 5, blade exit angle  $\beta 2 = 24^{\circ}$  was the best one. The corresponding blade wrap angle  $\Phi = 126^{\circ}$ . Head and efficiency of Scheme 5 were higher than that of Scheme 3, as shown in Figure 18. So, Scheme 5 was selected to improve the hydraulic performances of the centrifugal pump.



Figure 18. Hydraulic performances of Schemes 3 and 5.

# 7. Experimental Analysis

Via numerical simulation in Part 6, Scheme 5 was determined to be the best one. The corresponding impeller was machined. Then, the performance experiments were done in the closed performance experiment rig of Shanghai Kaiquan Pump Group to verify that head and efficiency of the centrifugal pump with the Impeller 5 showing obvious improvement.

#### Experiment Setup

The impeller machining process was displayed in Figure 19. At the outlet of the impeller, the allowance was 5 mm, as shown in region A. Allowance of the impeller front shroud and back shroud was 3 mm, as exhibited in region B. At the inlet of the impeller, the allowance was 3 mm too, as displayed in region C. Allowance of the hub was 3 mm, as shown in region D.



Figure 19. The machining process of the impeller.

To measure the optimized design of the impeller of Scheme 5, 3D wax pattern impeller, given in Figure 20, was machined firstly in the HRPS-V rapid prototyping machine, as shown in Figure 21. Laminated object manufacturing (LOM) in this machine—which is based on the data coming from CAD laying model,  $CO_2$  laser beam, and hot pressing device—was employed to machine the 3D wax pattern impeller.



Figure 20. 3D wax pattern.



Figure 21. HRPS-V rapid prototyping machine.

Then, the casted impeller of Scheme 5 was machined, which was displayed in Figure 22. The material was 2Cr13. To guarantee that the passages of the impeller were smooth and clean, silt particles and cast joint flash in the passages were cleaned carefully.



Figure 22. The casted impeller.

The performance experiments of the centrifugal pump were done at the closed experiment rig of Shanghai Kaiquan Pump Group. The experiment rig, given in Figure 23, includes the model centrifugal pump, suction pipes, exit pipes, pressure gauges, valves, and flow meters. The type of pressure sensor is XU12087105 (1.0, Shanghai Automation Instrumentation Co. Ltd., Shanghai, China). The type of flow meter is DN300 (1.0, Tianjin Flow Meter Co. Ltd., Tianjin, China). The type of valve is ZA2.T. All testing precisions are national grade 1 (GB3216-2005 and ISO 9906-1999).



Figure 23. Performance experiment rig.

Figure 24 is the typical components of the closed performance experiment rig. It includes 16 different parts.



**Figure 24.** Components of the closed performance experiment rig. 1–electric motor; 2–torque and speed sensor; 3–centrifugal pump; 4–sluice valve; 5–inlet pressure sensor; 6–turbine flow meter; 7–outlet pressure sensor; 8–turbine flow meter; 9–expansion joint; 10–turbine flow meters; 11–regulating valve; 12–turbine flow meters; 13–cavitation tank; 14–separation tank of vapor and water; 15–ball valve; 16–electric motor; 17–vacuum pump.

Via the inlet pressure Sensor 5 and the outlet pressure Sensor 7,  $M_1$  and  $M_2$  were determined. According to Equations (11) and (12), the inlet pressure and the outlet pressure of the centrifugal pump could be got.

The inlet pressure is calculated by

$$\frac{p_1}{\rho g} = 102M_1 + h_1 \tag{11}$$

The outlet pressure is determined by

$$\frac{p_2}{\rho g} = 102M_2 - h_3 \tag{12}$$

where  $h_1$  is the vertical height between the center of inlet pressure sensor with axial lead of the centrifugal pump.  $h_3$  is the vertical height between the center of the outlet pressure sensor with the outlet pressure port.

The inlet flow rate ( $Q_1$ ) and outlet flow rate ( $Q_2$ ) of the centrifugal pump could be measured by turbine flow meter 6 and 8, respectively.

Thus, the inlet velocity of the centrifugal pump  $v_1$  could be calculated by

$$v_1 = \frac{Q_1}{A_1} \tag{13}$$

The outlet velocity of the centrifugal pump  $v_2$  could be got by

$$v_2 = \frac{Q_2}{A_2} \tag{14}$$

Here,  $A_1$  is the inlet cross-section area.  $A_2$  is the outlet cross-section area. Head (*H*) could be calculated by Equation (15)

$$H = \left(\frac{p_2}{\rho g} - \frac{p_1}{\rho g}\right) + \left(\frac{v_2^2}{2g} - \frac{v_1^2}{2g}\right) + (Z_2 - Z_1)$$
(15)

where  $Z_2$  and  $Z_1$  are the static head.

In the performance experiment rig,

$$Z_1 = 0$$
 (16)

For  $Z_2$ , it could be obtained by

$$Z_2 = h_2 + h_3 (17)$$

Here,  $h_2$  is the vertical height between the center of outlet pressure sensor with axial lead of the centrifugal pump.

In the performance experiment rig,

$$h_2 = h_1 \tag{18}$$

The Equation (15) could be modified. The new form was shown in Equation (20).

$$H = 102(M_2 - M_1) + \left(\frac{v_2^2}{2g} - \frac{v_1^2}{2g}\right)$$
(19)

Efficiency ( $\eta$ ) could be calculated by Equation (16)

$$\eta = \frac{\rho g Q H}{1000P} \times 100\% \tag{20}$$

Here, *P* is the inlet power, which could be measured by torque and speed sensor 2.

Hydraulic performance curves of the improved centrifugal pump and the original centrifugal pump were shown in Figure 25. Experimental results indicated that head and efficiency of the improved centrifugal pump were higher than that of the original centrifugal pump. Under low flow rate condition (0.8  $Q_d$ ), the head increased by 3.76 m and the efficiency increased by 3.84%. With rated flow rate condition (1.0  $Q_d$ ), the head increased by 2.74 m and the efficiency increased by 5.77%, separately. For high flow rate condition (1.2  $Q_d$ ), the head and efficiency increased by 2.67 m and 5.0%, respectively. The experimental results indicated that the optimized design is successful.



Figure 25. Hydraulic performance of improved and original centrifugal pump.

## 8. Conclusions

In this paper, effects of blade exit angle and blade wrap angle on the optimized design of the impeller were comprehensively investigated. Flows in the centrifugal pump with five different impellers under low flow rate, rated flow rate, and high flow rate were numerically simulated. Variations of static pressure, relative velocity, streamlines, and turbulent kinetic energy were analyzed. Head and efficiency of the centrifugal pump of different schemes were compared. The experimental head and efficiency of the centrifugal pump with the best impeller were compared with that of the original impeller. The main conclusions are as follows:

- (1) The impeller with blade wrap angle  $126^{\circ}$  and blade exit angle  $24^{\circ}$  was the best one.
- (2) For the best impeller, static pressure and relative velocity was the most uniform distribution. Streamlines were the smoothest and vortices did not exist. Compared with other impellers, the distribution scope of turbulent kinetic energy in the best impeller under all flow rate conditions was the smallest.
- (3) For the centrifugal pump with optimized impeller, head and efficiency were higher than that of the original pump. With low flow rate (0.8  $Q_d$ ), the head and efficiency increased by 3.76 m and 3.84%. With rated flow rate, the head and efficiency increased by 2.74 m and 5.77%. With high flow rate (1.2  $Q_d$ ), the head and efficiency increased by 2.67 m and 5.0%.

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