

Lynx Mission Concept Status

Jessica A. Gaskin (Lynx Study Scientist, NASA MSFC)



Presented on behalf of the Lynx Team

Community STDT

F. Özel, Arizona (Co-Chair)

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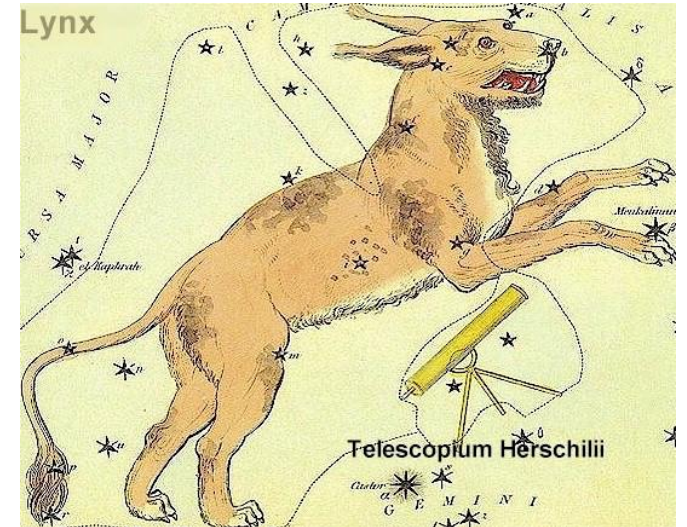
D. Pooley, Trinity

A. Ptak, GSFC

E. Quataert, Berkeley

C. Reynolds, UMD

D. Stern, JPL



- ❖ A symbol of great insight with the ability to see through solid objects to **reveal the true nature of things**.
- ❖ Much of the baryonic matter and the settings of the most active energy release in the Universe are visible primarily or exclusively in the X-rays

There are ~250 people across academia, industry, government, and non-US space agencies involved with the Lynx Concept Study

Lynx and the 2020 Astrophysics Decadal Survey

Lynx will revolutionize our view of the Universe by providing unique insight into the high-energy drivers that govern its formation and evolution.

- × 50 higher throughput while maintaining *Chandra's* angular resolution.
- × 16 larger solid angle for sub-arcsec imaging
- × 800 higher survey speed at the *Chandra* Deep Field limit
- Significant improvement in Resolving Power for grating spectroscopy
- High-resolution, spatially resolved spectroscopy on fine scales

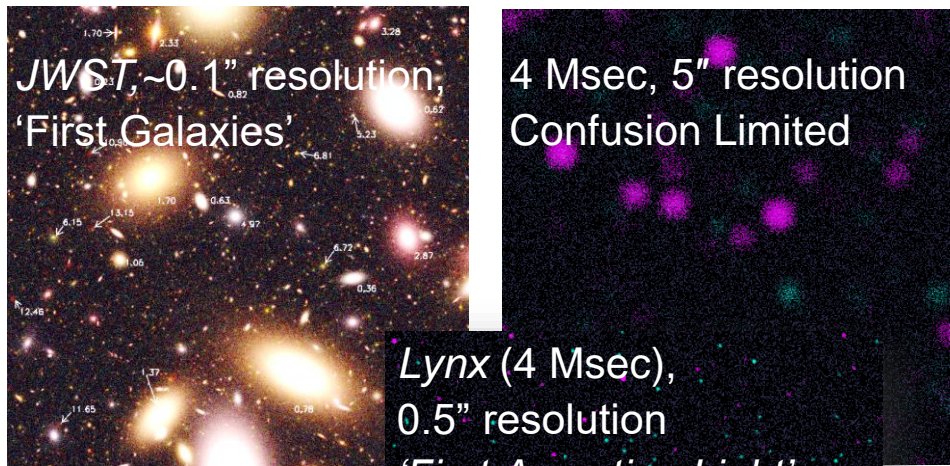
Astro2020 Decadal Study Input:

1. A **science case** for the mission
2. A **design reference mission and observatory**, including a report on any tradeoff analyses
3. A **technology assessment** including: current status, roadmap for maturation & resources
4. A **cost assessment** and listing of the top technical risks to delivering the science capabilities
5. A **top level schedule** including a notional launch date and top schedule risks.

The Dawn of Black Holes

Lynx will observe the birth of the first seed black holes at redshift up to 10 and provide a census of the massive black hole population in the local and distant universe, follow their growth and assembly across cosmic time, and measure the impact of their energy input on all scales.

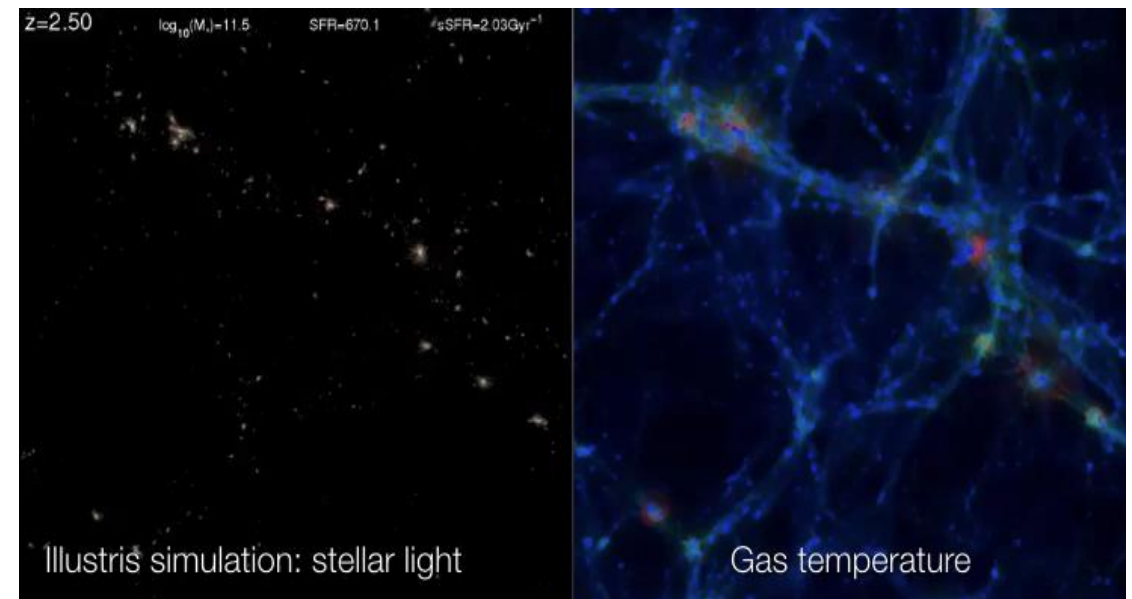
Simulated 2x2 arcmin deep fields



JWST will detect $\sim 2 \times 10^6$ gal/deg² at its sensitivity limit (Windhorst et al.). The $0.5''$ angular resolution of *Lynx* will minimize source confusion

The Invisible Drivers Behind Galaxy Formation and Evolution

The assembly, growth, and the state of visible matter in the cosmic structures is largely driven by violent processes that heats the gas in the CGM and IGM. The exquisite spectral and angular resolution of *Lynx* will make it a unique instrument for mapping the hot gas around galaxies and in the Cosmic Web.



Facility Class Observatory: Exploration Science with a Rich Community-Driven General Observer Program!

Lynx Optical Assembly

High-resolution X-ray Optical Assembly: 3 Viable Architectures – Trade Study

- Full Shell (K. Kilaru/USRA/MSFC, G. Pareschi/OAB)
- Adjustable Optics (P. Reid/SAO)
- Meta-Shell Si Optics (W. Zhang/GSFC)

OWG will make a formal recommendation to STDT: 6/1/18
STDT Finalizes their decision: 7/1/18

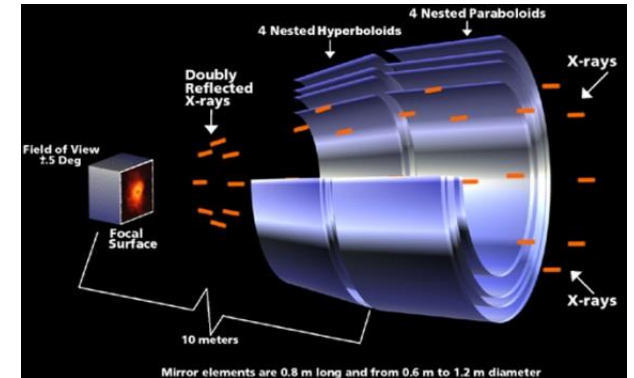
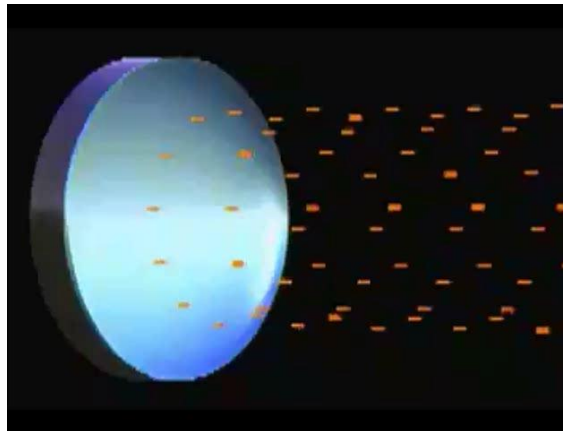
Up-select will be based on Science, Technical and Programmatic criteria (TBF)

- *Does the configuration Satisfy Science Requirements?*
- *Is there a feasible path for development?*
- *Are there existing X-ray measurements and/or analyses?*
- *Can it interface with the spacecraft and survive launch?*

Science Driven Requirements

Lynx Optical Assembly

Angular resolution (on-axis)	0.5 arcsec HPD (or better)
Effective area @ 1 keV	2 m ² (met with 3-m OD)
Off-axis PSF (grasp), A*(FOV for HPD < 1 arcsec)	≥ 2 m ² * 300 arcmin ²
Wide FOV sub-arcsec Imaging	10 arcmin radius

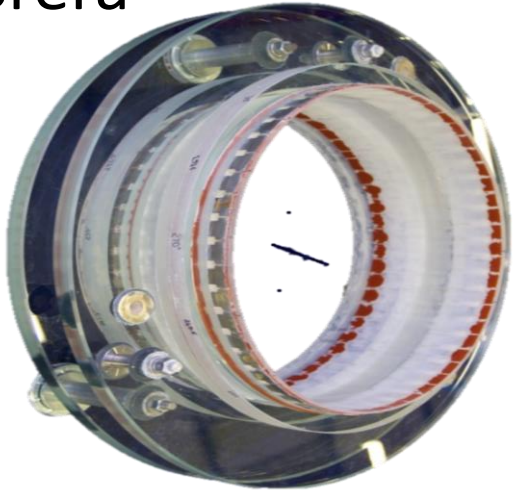


CXC/D. Berry

Full shell Optical Design

(G. Pareschi et al. OAB, K. Kilaru et al. USRA/MSFC)

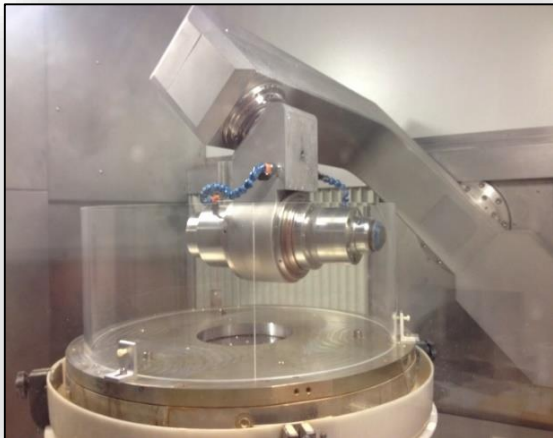
Brera



The mirror shell should have low density, low CTE, high modulus of elasticity and high yield strength. It should also be a material that is not too difficult to figure and polish.

- Fused Silica
- Readily available
- Fine-grinding to 1.5 μm OOR (P-V) + polish to 5-6 nm RMS microroughness
- Ion beam figuring corrections

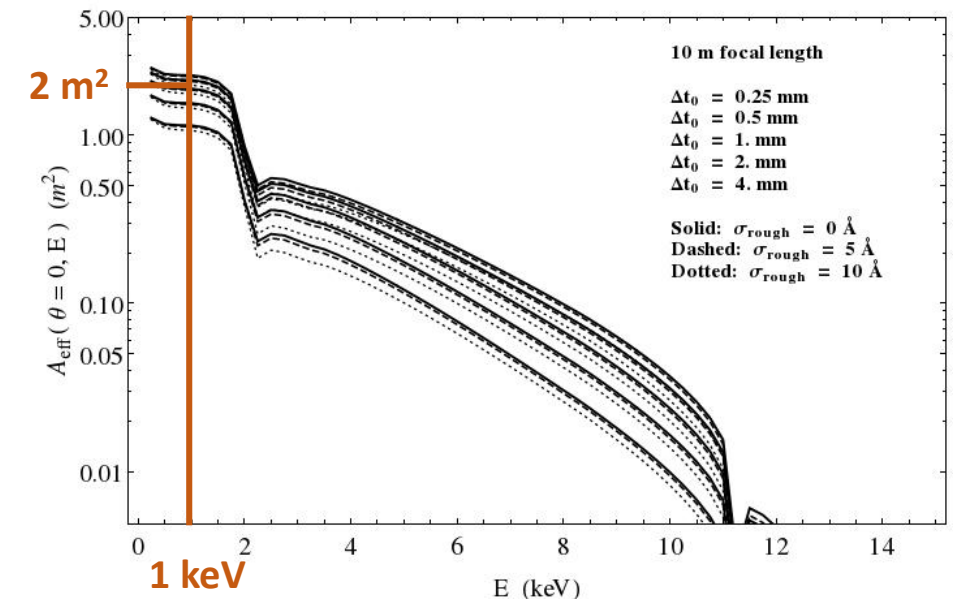
MSFC



- Be, BeAl
- Procure diamond-turned, heat treated, NiP alloy coated shells (<100 μm RMS)
- NiP is a hard material and can be figured, polished/superpolished
- Diamond-turn and Zeeko polish
- Differential deposition corrections

R. Eisner (NASA MSFC)

	(1)	(2)	(3)	(4)	(5)	(6)
Δt_0 (mm)	0.25	0.50	1.00	2.00	4.00	
	$f = 10 \text{ m}$					
N_{shells}	259	235	199	155	108	
Δt_{max} (mm)	0.6875	1.375	2.75	5.5	11.0	
$r_{i,max}$ (cm)	136.4	136.4	136.2	136.0	135.4	
$r_{i,min}$ (cm)	25.1	25.	25.1	25.0	25.2	
$r_{f,max}$ (cm)	137.4	137.4	137.2	137.0	136.4	
$r_{f,min}$ (cm)	25.1	25.1	25.2	25.1	25.3	
ℓ_{min} (cm)	9.498	9.486	9.515	9.473	9.563	
$A_{eff}(1 \text{ keV})^{1,2}$ (m^2)	2.270	2.126	1.890	1.551	1.142	
$A_{eff}(6 \text{ keV})^{1,2}$ (m^2)	0.213	0.190	0.157	0.117	0.078	
$A_{eff}(10 \text{ keV})^{1,2}$ (m^2)	0.037	0.031	0.024	0.016	0.010	
$M_{CE,tot}$ (Mg)	0.125	0.234	0.417	0.686	1.016	
$M_{Ni,tot}$ (Mg)	1.747	3.277	5.839	9.617	14.24	
$V(1 \text{ keV})^{2,3}$ (%)	9.762	9.674	9.545	9.403	9.264	
$\sigma_{D,module}(1 \text{ keV})^{2,4}$ (arcsec)	0.921	0.921	0.92	0.919	0.918	
$\alpha_{i,max}$ (arcmin)	116.5	116.5	116.4	116.1	115.7	
$\alpha_{i,min}$ (arcmin)	21.5	21.5	21.6	21.5	21.7	
$\Delta r_{i,max}$ (mm)	10.817	10.807	10.788	10.749	10.673	
$\Delta r_{i,min}$ (mm)	0.864	0.861	0.862	0.850	0.852	
$\Delta r_{f,max}$ (mm)	10.899	10.894	10.884	10.865	10.826	
$\Delta r_{f,min}$ (mm)	0.869	0.867	0.871	0.863	0.875	



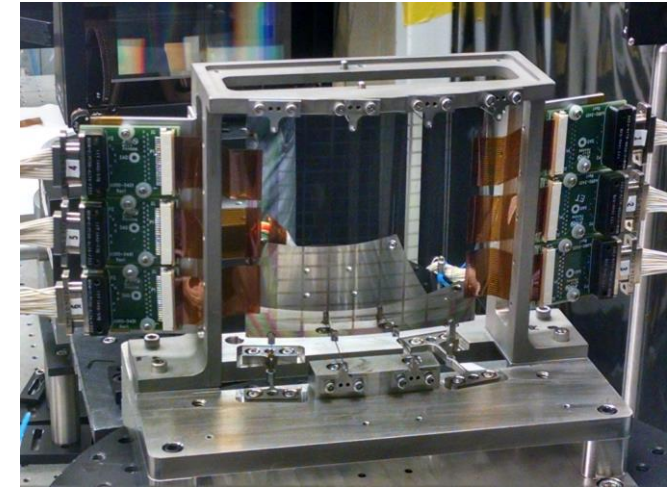
See Talks: 10399-30, 10399-31, 10399-32 , Wed 11:30 AM, Room 9

Adjustable Optics Status

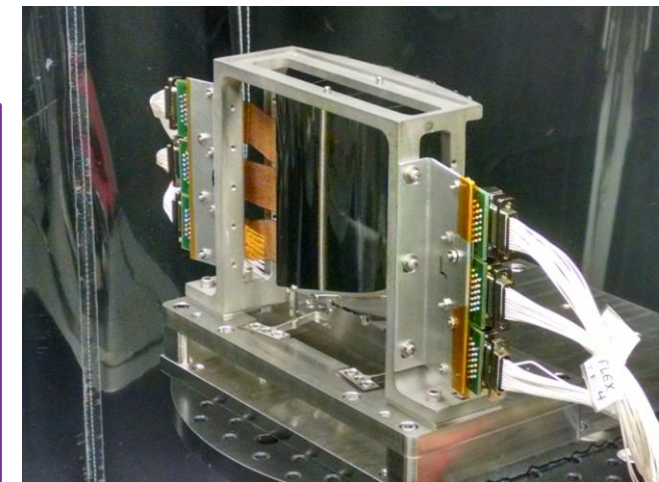
(Reid et al. SAO)

- 112 piezo-actuator cylindrical test mirror
 - Demonstrated ~ 7 arcsec HPD (single bounce, 1 keV) slumping performance (Cotroneo, 10399-33)
 - Demonstrated anisotropic conductive film electrical connectivity with minimal distortion (DeRoo, 10399-58)
 - Demonstrated 100 percent PZT yield (Walker, 10399-56)
 - Problem with 2nd to final step mounting (DeRoo)
- Piezo stress and stress compensation (Walker)
 - Identified major cause of large non-uniform stress (Rapid Thermal Annealer) and testing correction for future test optic

Back, convex, piezo side

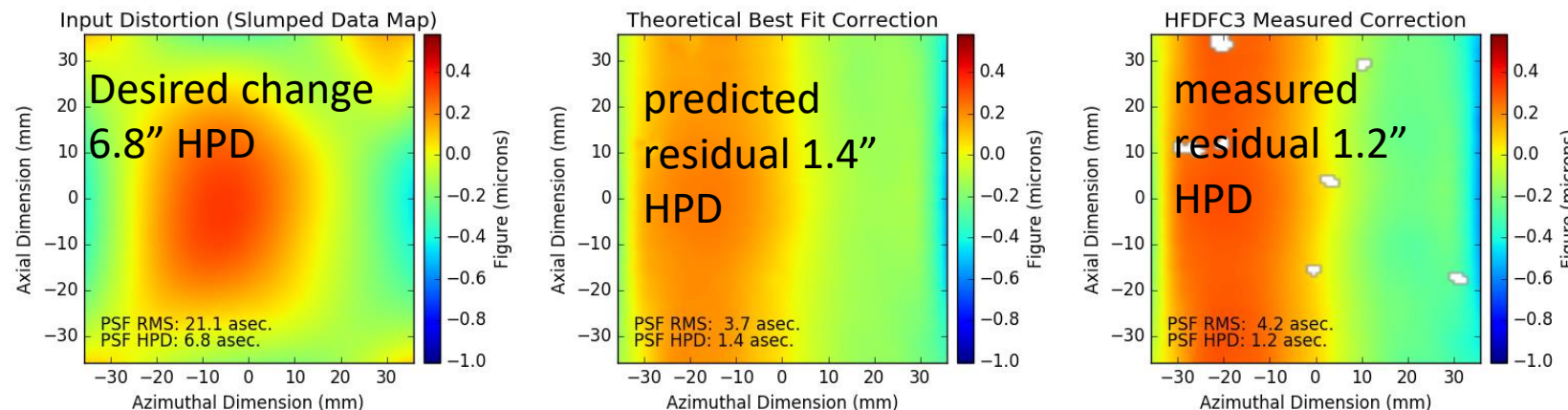


Front, concave, X-ray reflecting side



Major advance: successfully produced deterministic figure change via computer control of all 112 adjusters.

HFDFC3: Theoretical and Measured Corrections to Slumped Data Distortion, Figure Space



Prediction values (HPD, RMS) are computed at 1 keV for 0.375 deg graze for single reflection.

(DeRoo)

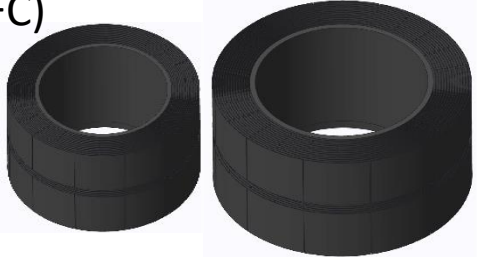
See Talks: 10399-56, 10399-58, Thurs 1:40 PM, 10399-33, Wed 2:00 PM, Room 9

Meta-Shell Status

(Zhang et al. NASA/GSFC)



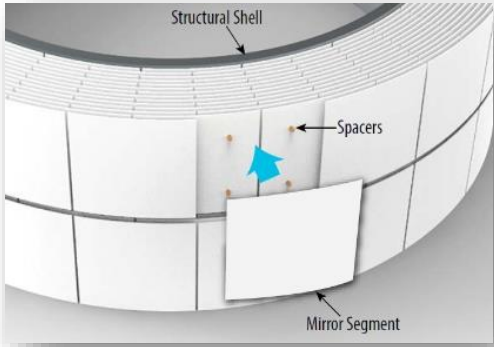
Segments



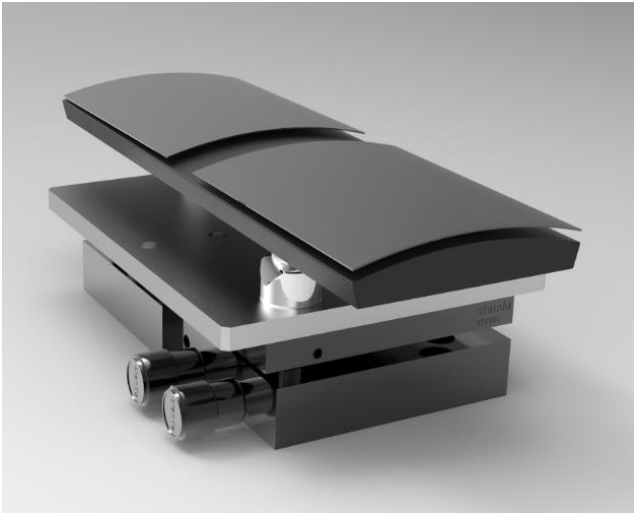
Meta-shells



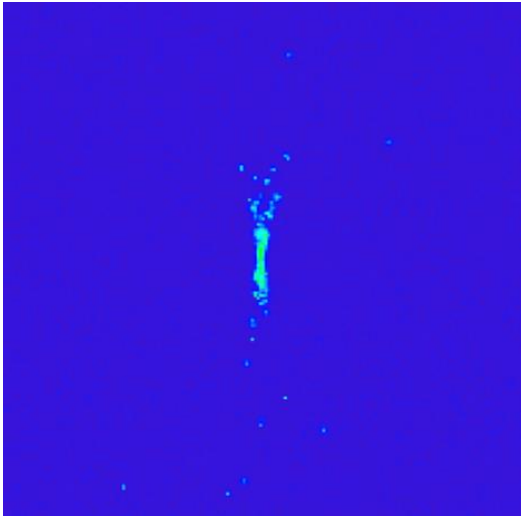
Assembly



Component	Contribution to HPD (")	
	Lynx Rqrmnt	Expected
Mirror segments	0.2	2.0
Alignment	0.1	1.5
Bonding	0.1	0.5
Thermal	0.2	0.2
Gravity release	0.1	0.1
Total	0.3	2.5



Single-Pair X-ray Test Module



**Full Illumination
4.5 keV X-rays
3.8" HPD**

See Talks: 10399-27, 10399-28, 10399-29, Wed 10:30 AM, Room 9

Lynx Optical Assembly

See Sessions:

Monocrystalline-Si Optics, Wed 10:30 AM

Fused-Silica Optics: Wed 11:30 AM

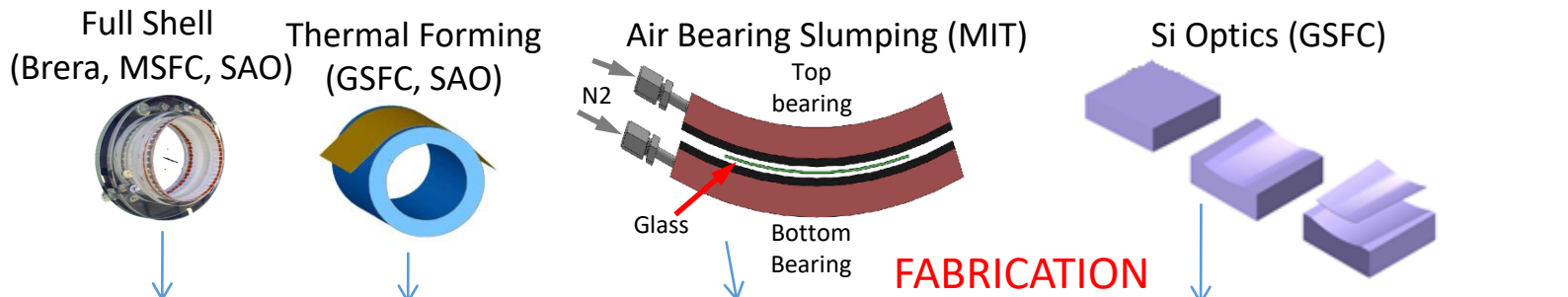
Slumped Optics, Wed 2:00 PM

Multilayer Coatings, Wed 4:10 PM

Coating Stress, Thurs 9:20 AM

Erosion/Deposition Figuring, Thurs 10:50 AM

Posters - Wednesday



CORRECTION

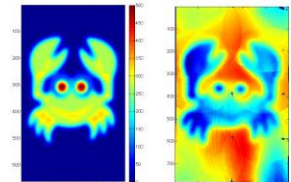
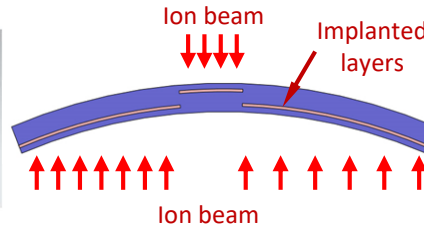
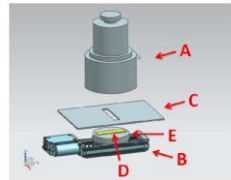
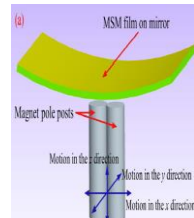
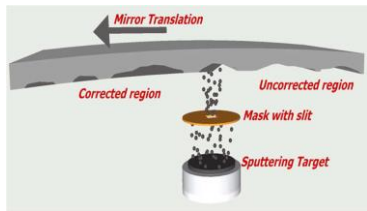
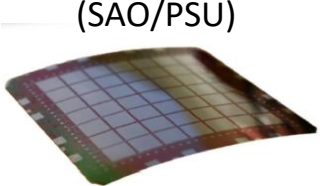
Piezo stress (SAO/PSU)

Deposition (MSFC, XRO)

Magnetic & deposition stress (NU)

Ion implant stress (MIT)

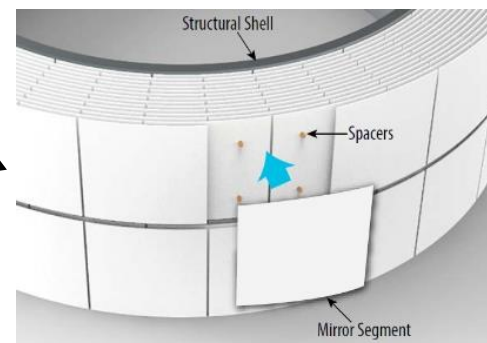
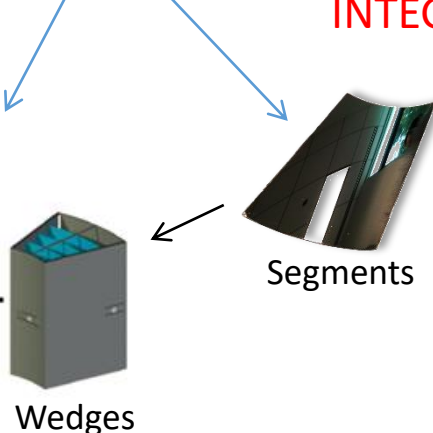
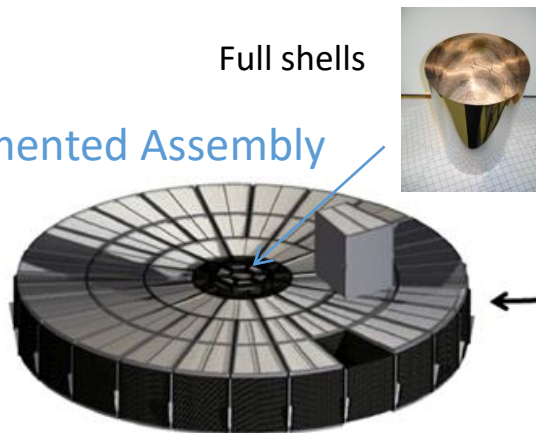
Ion beam figuring (OAB)



INTEGRATION

Segmented Assembly

Meta-Shell Assembly



Lynx Science Instruments

- High Definition X-ray Imager (HDXI)

- HDXI IWG Leads: M. Bautz (MIT), R. Kraft (SAO), A. Falcone (PSU)
- Instrument Design Study (On-going @ MSFC ACO)

[See Papers: 10397-1, 10397-2, 10397-3, 10397-5]

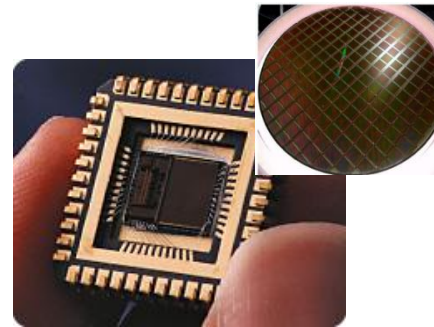
- X-Ray Grating Spectrometer (XGS)

- XGS IWG Leads: R. McEntaffer (PSU), Ralf Heilmann (MIT)
- Instrument Design Study (On-going @ MSFC ACO)

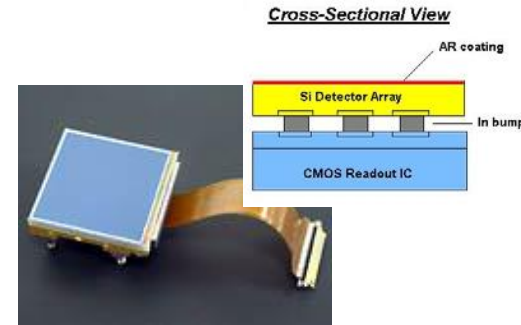
[See Talks: 10399-38, 10399-39, 10399-40, Wed 4:10 PM]

- Lynx X-ray Microcalorimeter (LXM)

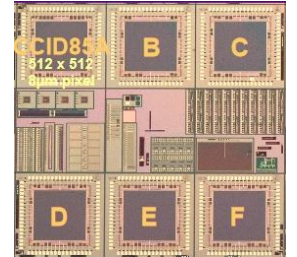
- LXM IWG Leads: S. Bandler (GSFC), E. Figueroa-Feliciano (Northwestern)
- Instrument Design Lab (Completed 1st IDL @ GSFC)



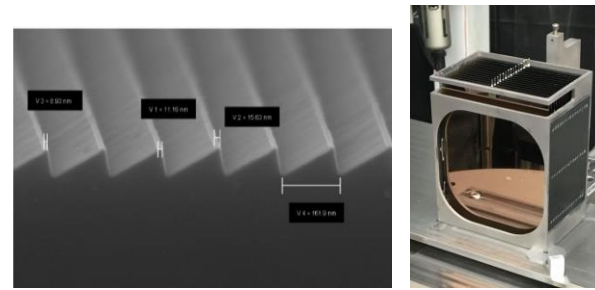
Monolithic CMOS



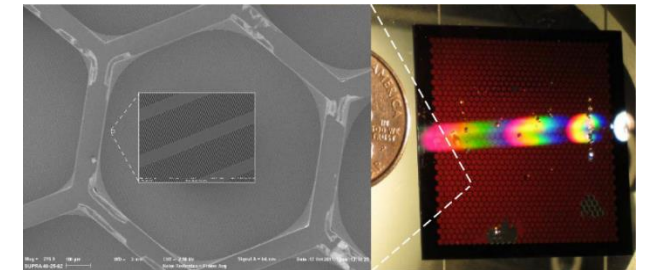
Hybrid CMOS



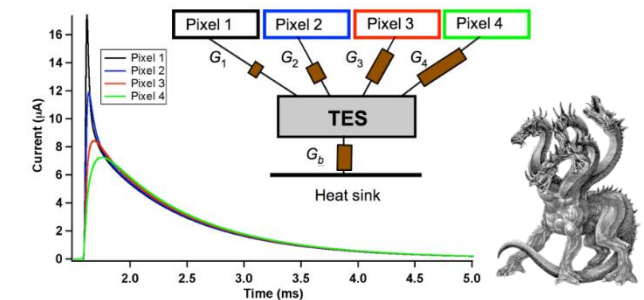
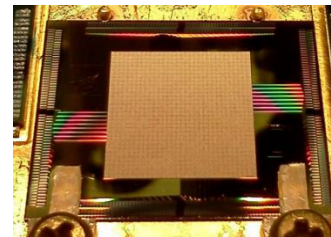
Digital CCD with CMOS readout



Off-Plane Grating Array

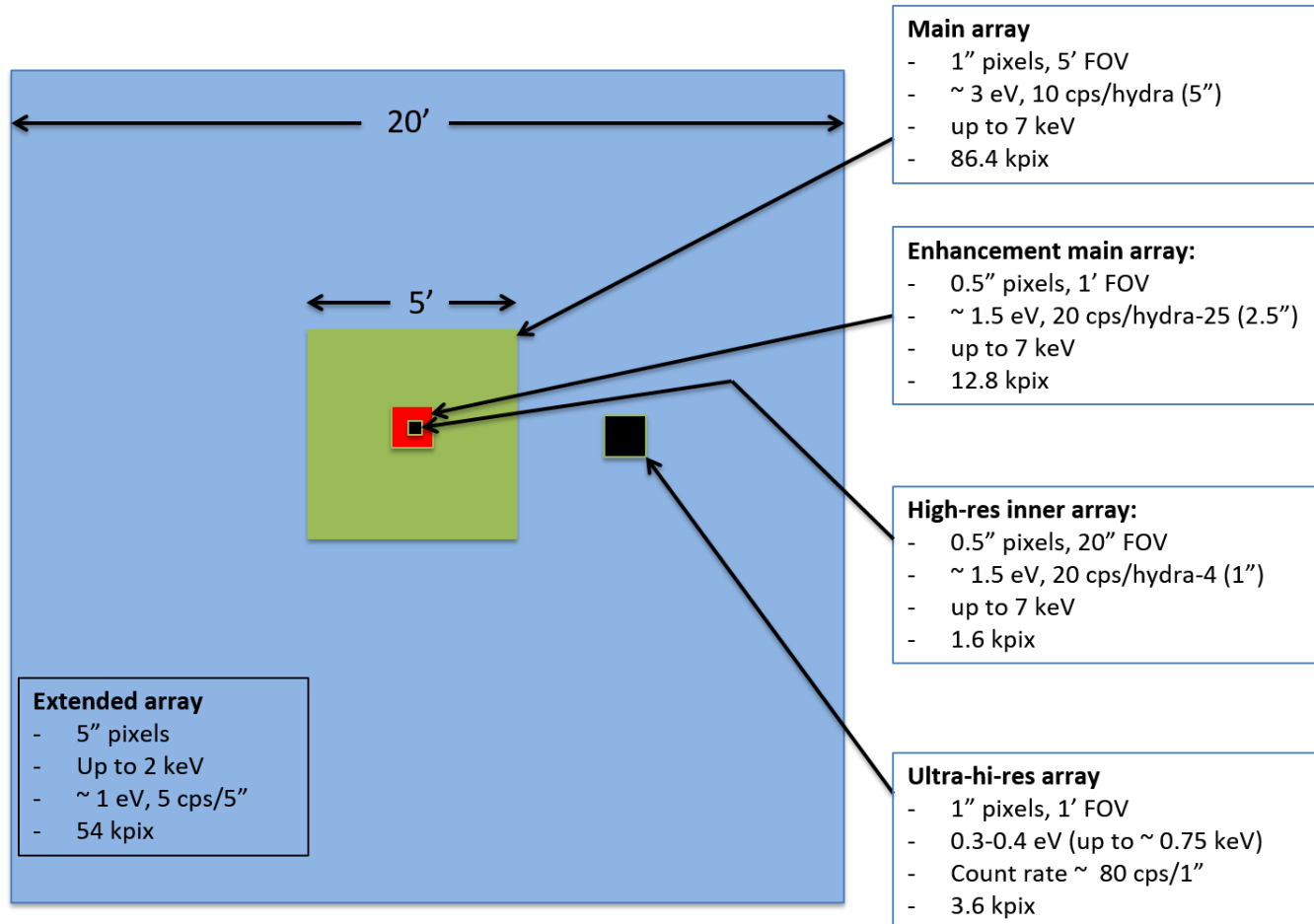


Critical Angle Transmission Grating Array



Lynx X-ray Microcalorimeter

Lynx Science Instruments: Requirements



Main array

- 1" pixels, 5' FOV
- ~ 3 eV, 10 cps/hydra (5")
- up to 7 keV
- 86.4 kpix

Enhancement main array:

- 0.5" pixels, 1' FOV
- ~ 1.5 eV, 20 cps/hydra-25 (2.5")
- up to 7 keV
- 12.8 kpix

High-res inner array:

- 0.5" pixels, 20" FOV
- ~ 1.5 eV, 20 cps/hydra-4 (1")
- up to 7 keV
- 1.6 kpix

Ultra-hi-res array

- 1" pixels, 1' FOV
- 0.3-0.4 eV (up to ~ 0.75 keV)
- Count rate ~ 80 cps/1"
- 3.6 kpix

Extended array

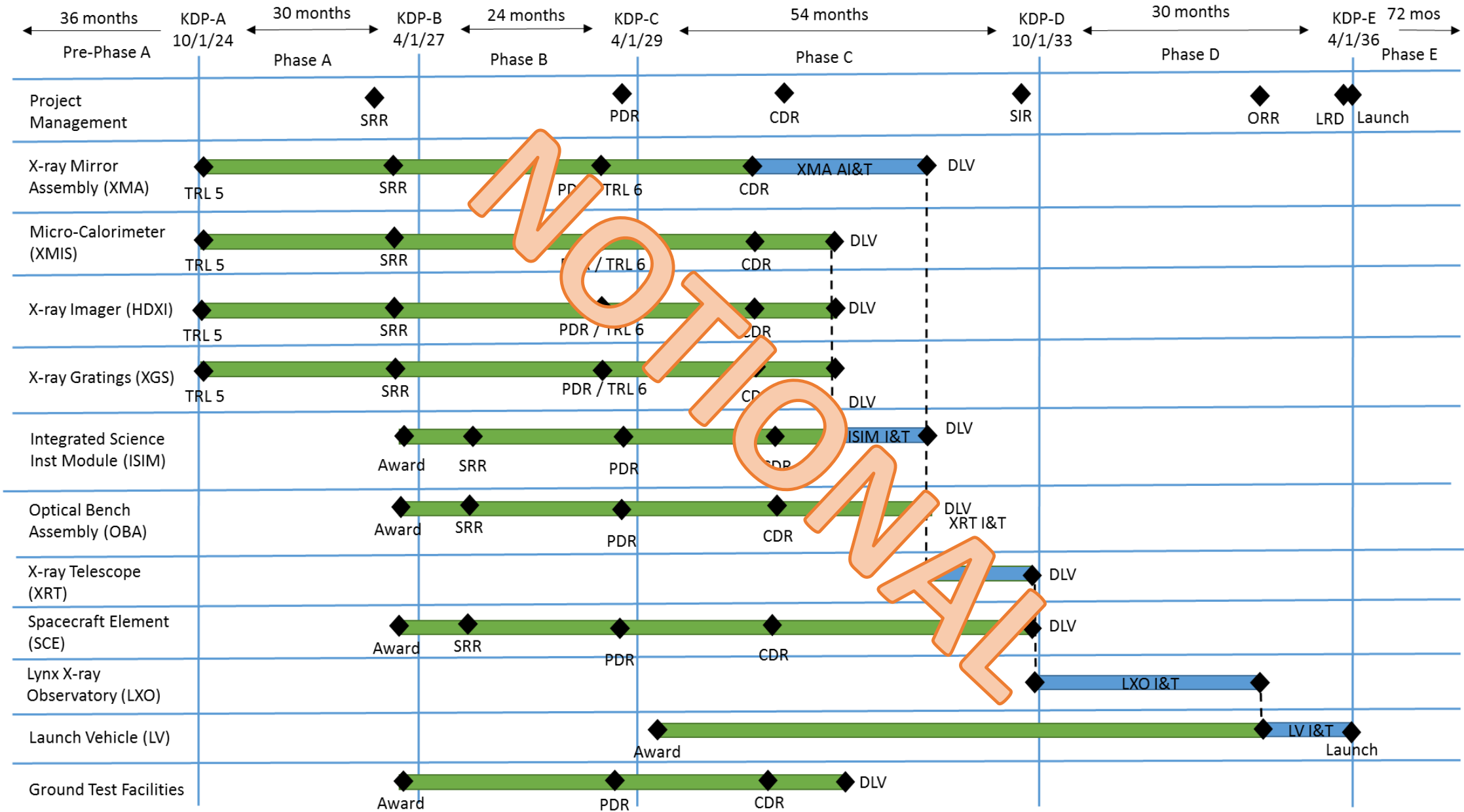
- 5" pixels
- Up to 2 keV
- ~ 1 eV, 5 cps/5"
- 54 kpix

High Definition X-ray Imager (<i>Notional</i>)	
Energy Range	0.2 – 10 keV QE > 90% (0.3-6 keV), QE > 10% (0.2-9 keV)
FOV	22' x 22' (4k x 4k pixels)
Pixel Size	< 16 x 16 μm (≤ 0.33")
Read Noise	≤ 4 e ⁻
Energy Resolution	37 eV @ 0.3 keV, 120 eV @ 6 keV (FWHM)
Frame Rate	> 100 frames/s (full frame) > 10000 frames/s (windowed region)
Radiation Tolerance	10 yrs at L2

X-ray Grating Spectrometer (<i>Notional</i>)	
Effective Area	~4000 cm ² @ 0.3 keV (63% azimuthal coverage)
Resolving Power, R	> 5,000
Energy Resolution	< 5 eV (FWHM)
Count Rate Capability	< 1 count/s/pixel
Array size	300 x 300 pixel array

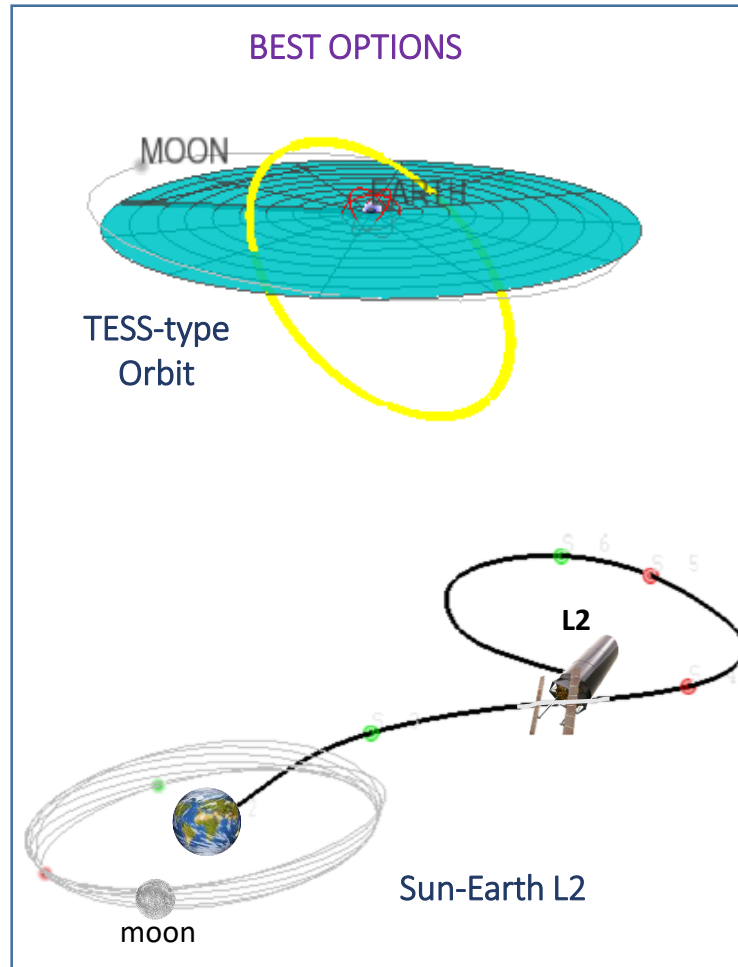
Lynx X-ray Microcalorimeter (<i>Notional</i>)	
Pixel Size	1" (50 μm pixels for 10-m focal length)
FOV	At least 5' x 5'
Energy Resolution	< 5 eV (FWHM)
Count Rate Capability	< 1 count/s/pixel
Array size	300 x 300 pixel array

Lynx X-ray Observatory – Notional Mission Lifecycle Schedule



Mission Design + Trades

- Structures [*Launch vehicle Trade*]
- Thermal
- Propulsion
- Avionics [*Comm Trade*]
- GNC [*Rapid response capability Trade*]
- Power
- Mechanisms [*Moveable optics vs. instrument table Trade*]
- Environments [*Orbit Trade*]

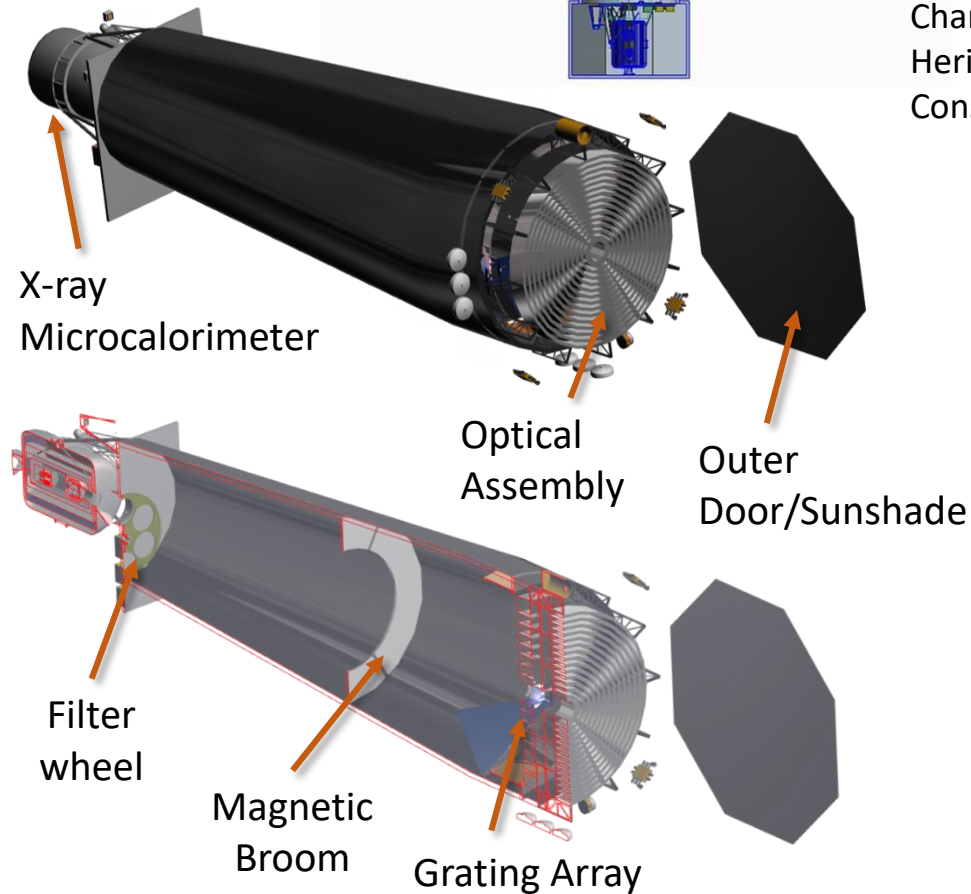


Orbit Trade (SE-L2 vs. TESS)

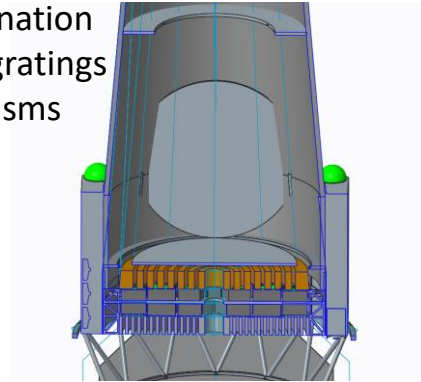
- Launch Vehicle (Both)
- Delta-V (SE-L2)
- Thermal Environment (SE-L2)
- Eclipsing (SE-L2, just)
- Communications (TESS)
- Meteoroid Environment (Both)
- Radiation Environment (average/worst case – TBD)
- Serviceability (SE-L2)
- Disposal (TESS)
- Station Keeping (TESS)
- Disturbance Environment (Both)

- **NASA MSFC Advanced Concept Office: J. Mulqueen, A. Schnell, R. Hopkins, R. Suggs, J. Garcia, S. Sutherlin, T. Boswell, A. Dominguez, P. Capizzo, J. Rowe, L. Fabisinski**

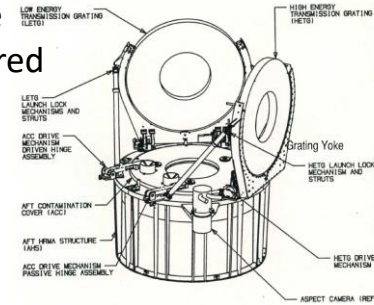
Mechanical Design



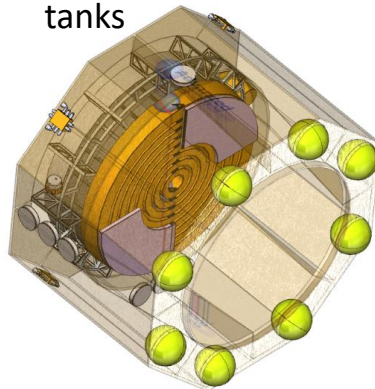
Internal contamination door & gratings mechanisms



Chandra Heritage Considered



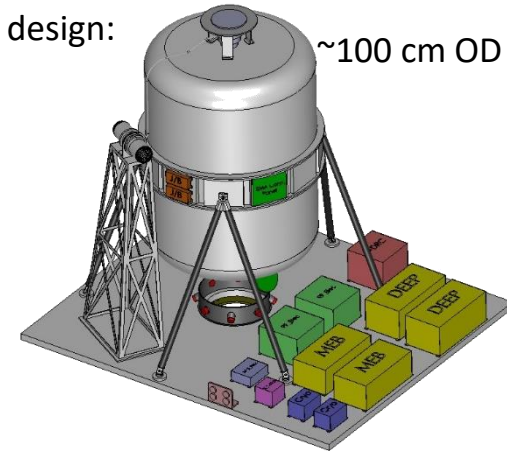
Mono-prop tanks



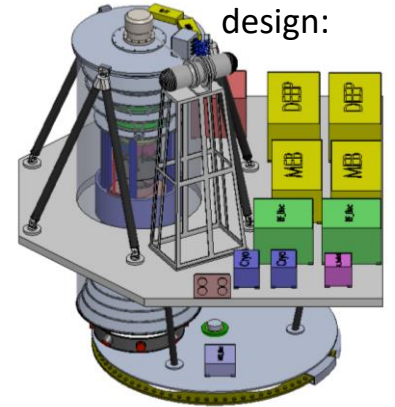
Integrated Science Instrument Module

X-ray Microcalorimeter Designs

Internal strut design:

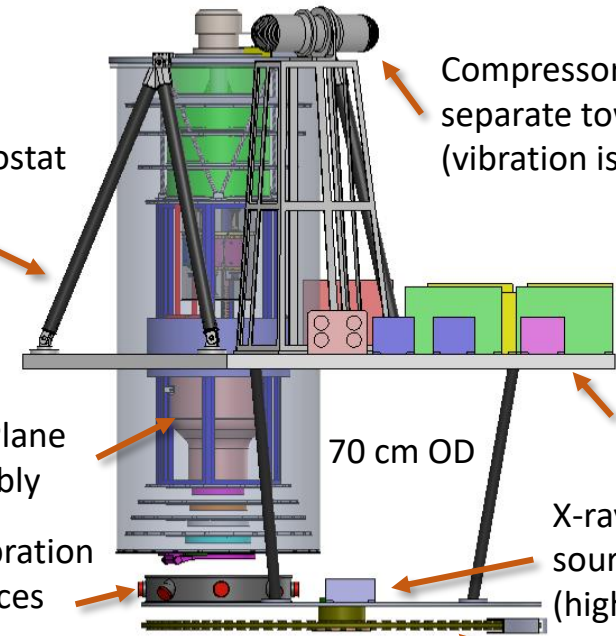


Thrust tube design:



Bipod cryostat supports

Compressor on separate tower (vibration isolation)



Deck: to be attached to movable table and focusing mechanisms

70 cm OD

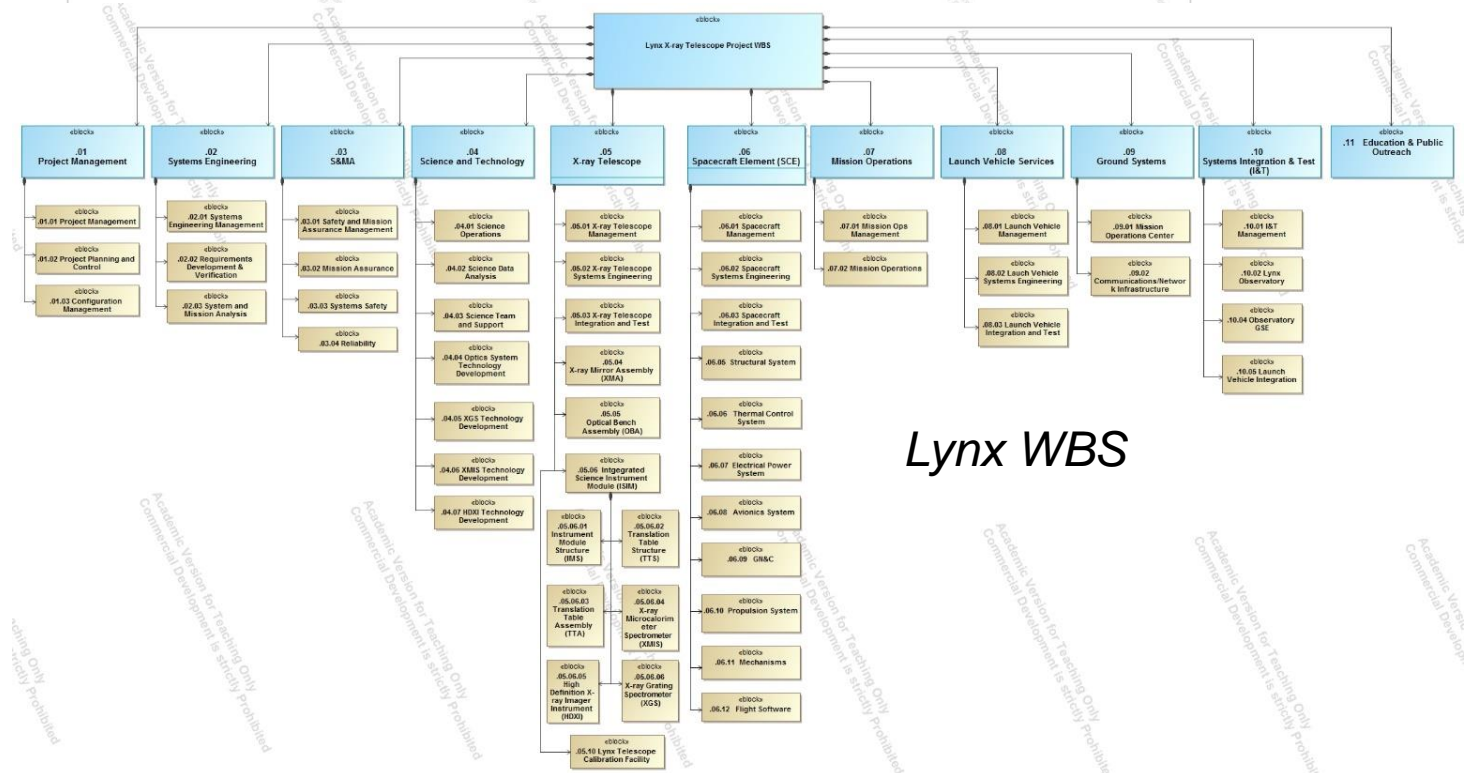
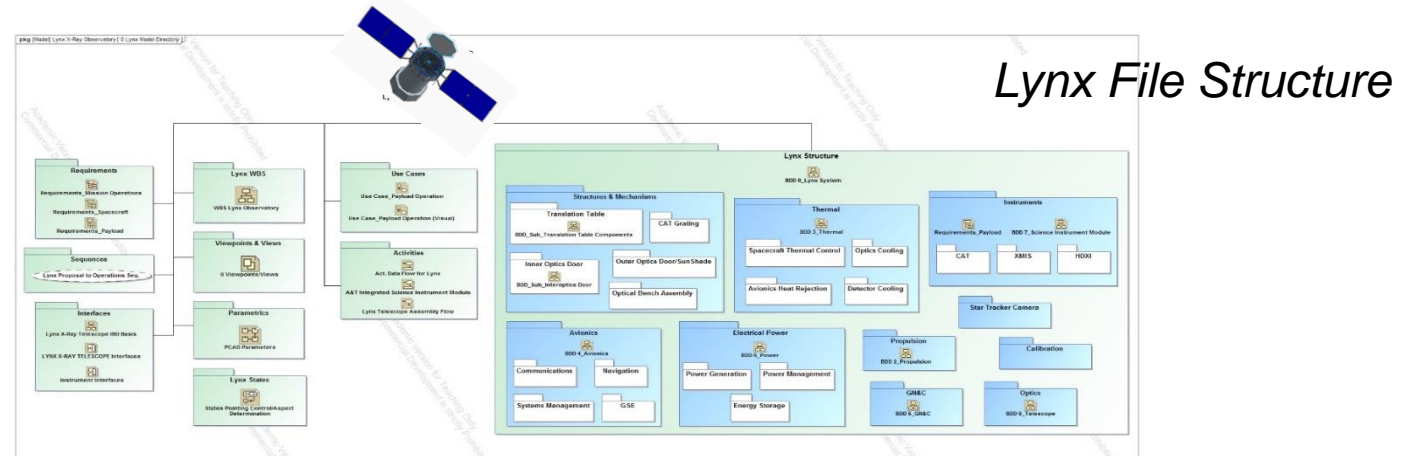
X-rays in from telescope

S. Bandler, NASA GSFC

Model Based Systems Engineering Approach

Focused on Concept of Operations

- WBS + dictionary
- Stakeholder Viewpoints
- System Block Diagrams
- System Interface Diagrams (internal and external)
- Use Diagrams
 - Manufacturing
 - AI&T (major sub-systems)
 - Ground Operations
 - Launch Integration
 - Launch Operations (launch, deployment, T&CO)
 - Mission Timeline
 - Science Operations
 - Off-nominal Operations
- Functional Block Diagrams
- System Requirements (Level 1 and 2)



Thank You!

For the latest Lynx news and events, and to sign up to the News Distribution visit us at:

<https://wwwastro.msfc.nasa.gov/lynx/>

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