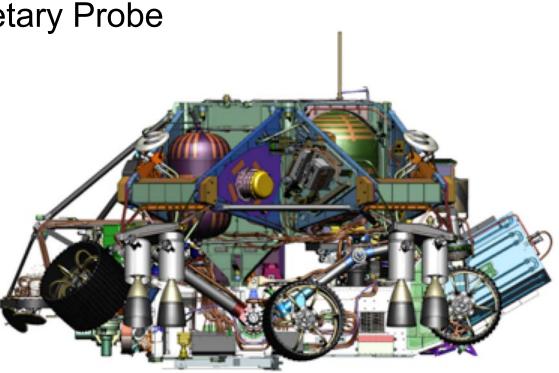


# Mars 2020 Entry, Descent, and Landing Update

## 15<sup>th</sup> International Planetary Probe Workshop

Boulder, CO June 2018

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### Mars 2020 Mission Overview

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#### LAUNCH

- Atlas V 541 Rocket
- Period: Jul-Aug 2020

#### CRUISE/APPROACH

- ~7 month cruise
- Arrive Feb 2021

#### ENTRY, DESCENT & LANDING

- MSL EDL System: guided entry, powered descent, and sky crane
- Augmented by *range trigger*: 16 x 14 km landing ellipse
- Augmented by *TRN*: enables safe landing at a greater number of scientifically valuable sites
- Access to landing sites ±30° latitude, ≤ - 0.5 km elevation
- Deliver a 1050 kg rover

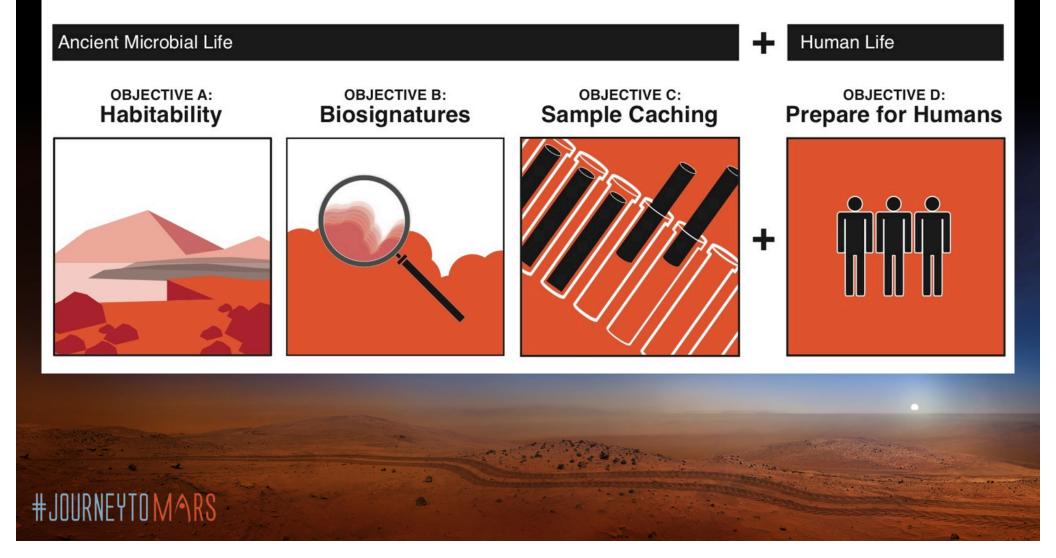
#### SURFACE MISSION

- Prime mission of 1.5 Mars years
- 20 km traverse distance capability
- Seeking signs of past life
- Returnable cache of samples
- Prepare for human exploration of Mars

## Mars 2020 Objectives



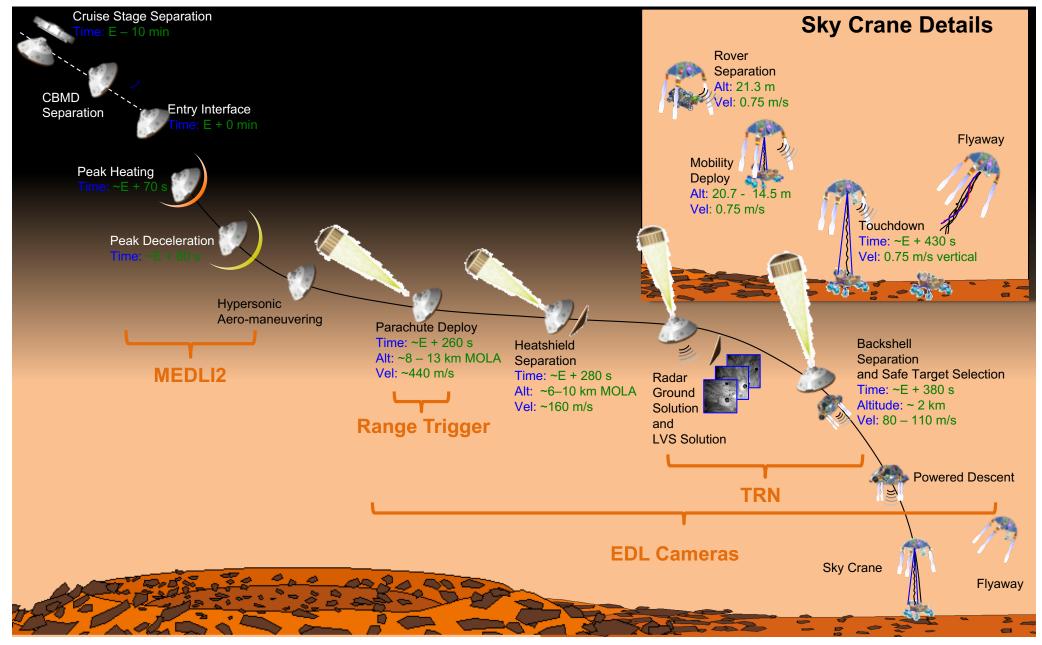
## Understanding the Possibilities for Life on Mars



## **EDL Timeline**



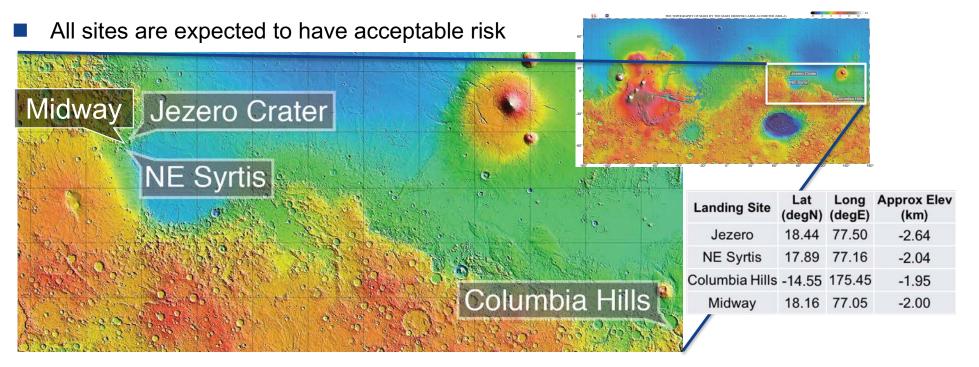
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## **Landing Site Selection**



- A fourth candidate site, Midway, has been added for evaluation by the engineering and science teams
- Terrain and Atmosphere reviews completed in the past year; several improvements to rock identification have been incorporated
- EDL is currently evaluating landing site safety in support of the 4th landing site workshop this Fall, with final site selection expected in early 2019 by NASA HQ





## **TRN Development Status**

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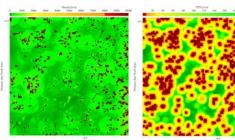
- Terrain Relative Navigation (TRN), Vision Compute Element (VCE), and VCE SW design reviews are complete
- Design has matured and hardware and software deliveries are in progress
  - □ VCE engineering dev units (EDU) delivered; EM and FM builds are in progress; LVS Camera (LCAM) EM1 delivered
  - Box-level software design is ongoing and expected to be part of 2019 field test
  - □ System-level software design is finalized and being tested
- Monte Carlo simulations have exercised the TRN capability using a Safe Targets Map (STM) and generated successful landing metrics based on the hazard map
- Full SW testing and delivery to ATLO happening Fall 2018
  - □ ATLO testing in Jan 2019
  - □ Field testing in Feb 2019

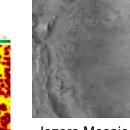






LCAM EM1 calibration image

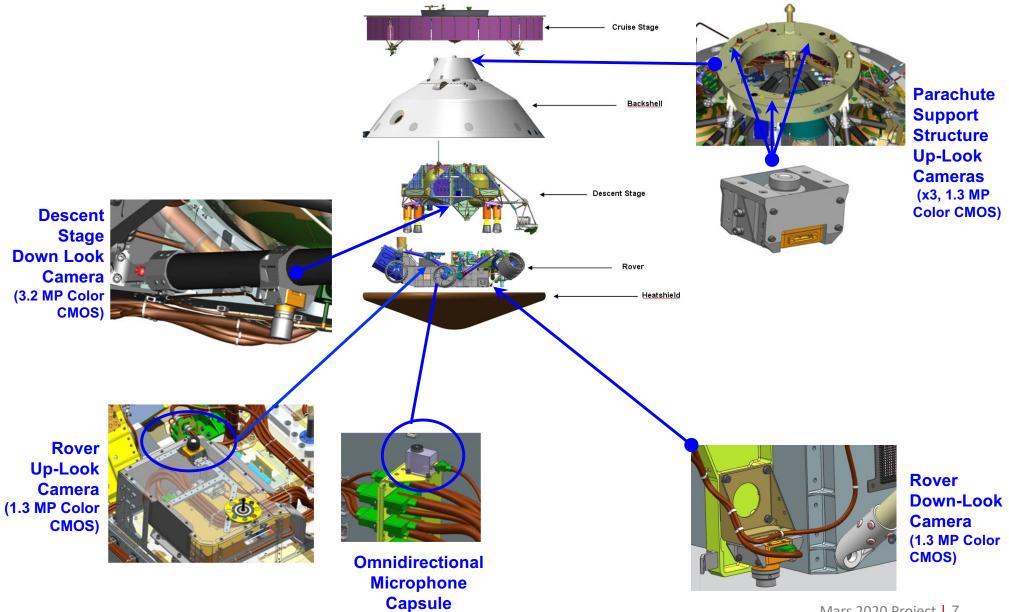




Jezero Mosaic, example of onboard appearance map for TRN

Jezero site: hazard map vs. STM

## **EDL Camera Overview**



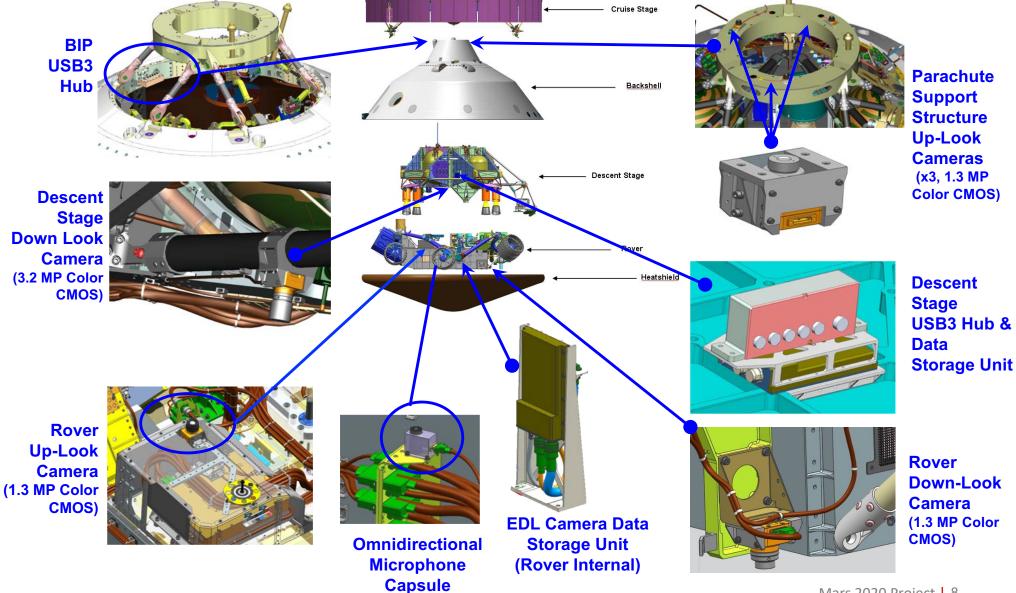
Mars 2020 Project 7

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## **EDL Camera Overview**

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Mars 2020 Project 8

## Mars Helicopter



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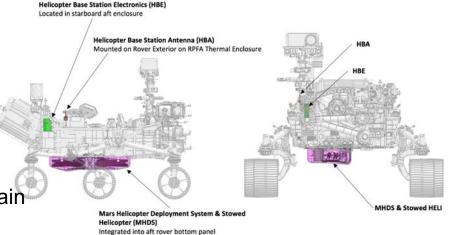
- Mars Helicopter is now officially part of the Mars 2020 Mission
- Overview:
  - Mass: ~1.8kg
  - □ Approx. 1.1m x 0.25m x 0.02m
  - Dual counter-rotating blades (L = 1.21m)
  - Solar powered with lithium-ion battery
  - B&W camera for nav; color camera for terrain

#### Concept of Operations

- The helicopter will be "dropped off" at a relatively flat location.
- 5 tech demo flights in 30-sol period of 30-120 sec each
- EDL Impact: due to its location, the helicopter increases the risk due to a rock strike by ~0.5-1%



Credits: NASA/JPL-Caltech Mars 2020 Project 9



### **ATLO Hardware Status**

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- Major hardware integration on the cruise and descent stage is complete (thermal, propulsion) or in progress (e.g. GNC sensor integration)
- First power on occurred in April
- System testing begins February 2019

#### **Descent Stage**



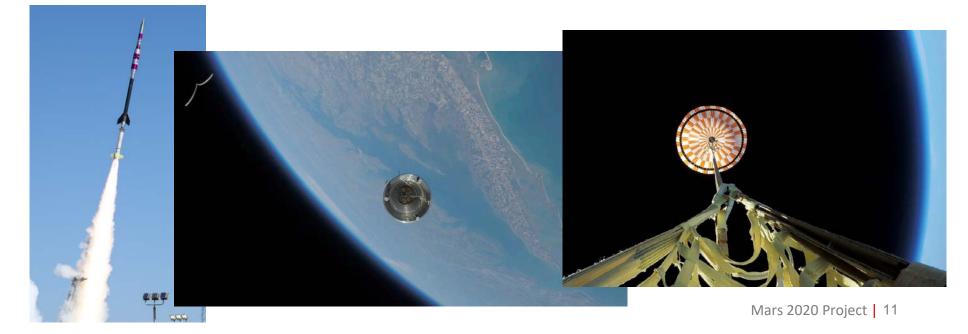
CS Propulsion Tanks and Power Dist Module Installed



Credits: NASA/JPL-Caltech



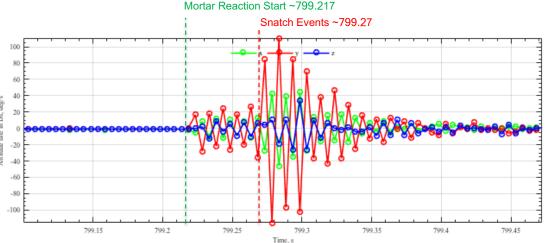
- Mars 2020 is doing supersonic parachute testing as a risk reduction activity for canopy stresses during the inflation process
- One flight test each of the MSL design and a strengthened Mars 2020 design have been successfully completed
  - □ 2<sup>nd</sup> flight was highest load ever survived by a supersonic parachute! (56 klbf/ 249 kN)
- A third flight test, 2<sup>nd</sup> of the strengthened design, at a margined load condition is planned for late July 2018; target load will be even higher than the prior flight



## Issues in Work (1/2)



- Parachute Deployment Snatch Forces Disturbance on Descent IMU
  - Recent analysis of the MSL flight data revealed an excitation that caused unexpected large rates (120°/s)
  - The timing suggests it is likely due to a snatch event associated with the mortar sabot and/or the confluence fitting (hardware connecting the triple bridle to the parachute single riser)



□ There are concerns this excitation could cause temporary IMU saturation and/or navigation errors which can lead to large EDL attitude errors and potentially loss of mission

#### Mitigations

- Mechanical design changes are being considered to reduce the magnitude or coupling of the disturbance; these include softer triple bridles, a lighter confluence fitting, adding mass to the descent stage panel where the IMU is mounted, etc.
- The primary GNC mitigation being considered is enabling a low pass filter in the IMU; there is also the potential to use the Rover IMU (lower quality data, less susceptible to saturation) as a sanity check on the descent IMU.



## Issues in Work (2/2)

- Heatshield Static Test Failure
  - The flight heatshield developed a circumferential crack during static testing in March
  - Root cause is still under investigation while Project is moving to build a replacement heatshield
  - Adequate schedule margin exists to build the new heatshield and still meet the Mars 2020 launch opportunity





- Mars 2020 Project development is making good progress and continues to be on track for the 2020 launch opportunity
- Landing Site Selection efforts will likely wrap up this year, leading to a final site selection by NASA HQ in early 2019
- The EDL team continues to work technical issues while supporting hardware deliveries, FSW testing, and the start of activities in ATLO