EML4551-2



TEAM: 516 JACOB HACKETT CALEB JANSEN NOAH LANG KYLE NULTY HANNAH RODGERS



NASA Marshall Space Flight Center LSS Assembly Tool Team 516



Jacob Hackett Controls Engineer



Caleb Jansen Communications Engineer



Noah Lang Systems Engineer



Kyle Nulty Logic and Processing Design Engineer



Hannah Rodgers Mechanical Design Engineer



Advisor and Sponsor





Faculty Advisor: Dr. Christian Hubicki FAMU-FSU College of Engineering Project Sponsor: Justin Rowe NASA Marshall Space Flight Center







Project Scope:

Project Description:

- NASA wants the ability to relocate their lunar bases to different areas on the surface of the Moon.
- NASA's proposed solution is an assembly tool that can relocate their life support system (LSS) to nearby locations on the lunar surface.



Project Scope:

Objective: Move the LSS payload around the lunar surface

Develop a full-scale simulation and scaled prototype of the assembly tool to transport modules of the life support system on the surface of the Moon.



Identify methodology for scalability of the LSS Assembly Tool.



Create a full-scale simulation

Traverse obstacles present on the lunar surface

Create scaled prototype to move scaled payloads of NASA's equipment

Controlled via remote control

SAE level 1 autonomy

Jacob Hackett

01

02

03

04

05



Project Scope:

Assumptions:

Responsible only for the design and production of the scaled mechanism prototype to move the LSS or other payloads.

 Not responsible for the transportation of the LSS assembly tool to the lunar surface. All testing and verification will be conducted under Earth's atmosphere and on terrain analogous to lunar conditions. Existing software and hardware components will be utilized as needed.









Project Background

10

Design Requirements

- Comparable to current infrastructure machine
- Payload Equipment for a unique environment

Lunar Rover Design

- Comparable to current rover technology
- Focus on ability to move payload
- Combustion power not viable
- Current Technology-
 - Mars Exploration Rovers Spirit and Opportunity
 - ATHLETE Rover

FAMU-FSU Engineering





Project Background

Preparation for Unique Obstacles

- Lunar Environment Studies
 - "Zap Pits" Micro-meteor impact sites
 - Regolith
 - Atmospheric Conditions / Dangers
- Lunar Architecture/NASA Goals
 - Plans for long term base on lunar surface
- Full-Scale Simulations
 - Need application to simulate full force simulations
 - Simscape with additional toolboxes









Questions

How big is the payload we will be lifting?

Interpreted Responses

Need to lift a 300kg load



Caleb Jansen



Questions

- How big is the payload we will be lifting?
- What scale of a model do you expect?

FAMU-FSU Engineering

(Y)9

Interpreted Responses

- Need to lift a 300kg load
- Various scaled models



14

Caleb Jansen

Questions

- How big is the payload we will be lifting?
- What scale of a model do you expect?
- How detailed of a simulation would you like?



Interpreted Responses

- Need to lift a 300kg load
- Various scaled models
- A full scaled model for simulation

Caleb Jansen

Questions

- How big is the payload we will be lifting?
- What scale of a model do you expect?
- How detailed of a simulation would you like?
- Does the assembly tool need to be fully assembled when it arrives on the moon?

Interpreted Responses

- Need to lift a 300kg load
- Various scaled models



- A full scaled model for simulation
- The assembly tool will be fully assembly, excluding possible hand tool assembly



Caleb Jansen



17



FAMU-FSU Engineering

Functional Decomposition





Caleb Jansen



(FAMU-FSU Engineering

Functional Decomposition

Caleb Jansen

(1) FAMU-FSU Engineering

Functional Decomposition

Jacob Hackett

Jacob Hackett

🕑 🚱 FAMU-FSU Engineering

EML4551-2

TEAM: 516 JACOB HACKETT CALEB JANSEN NOAH LANG KYLE NULTY HANNAH RODGERS

References

National Aeronautics and Space Administration . (1996, June 21). Structural Deisgn and Test Factors of Safety For Spaceflight Hardware. Huntsville, Alabama, USA.

Stone, R. B., & Wood, K. L. (2000). Development of a Functional. *Journal of Mechanical Design*, 359-370.

Shuttleworth, J. (2019, January 7). SAE Standards News: J3016 automated-driving graphic update. Retrieved from SAE International: https://www.sae.org/news/2019/01/sae-updates-j3016-automated-driving-graphic

Williams, D. R. (2016, May 19). The Apollo Lunar Roving Vehicle. Retrieved from The Apollo Program (1963 - 1972): https://nssdc.gsfc.nasa.gov/planetary/lunar/apollo_lrv.html

Meyer, C. (2003). Lunar Regolith. Retrieved from NASA Lunar Petrographic Educational Thin Section Set: https://curator.jsc.nasa.gov/lunar/letss/regolith.pdf

FAMU-FSU Engineering

(**Y**)

SAE Level 1 Autonomy Graphic

24

Existing Technology

ATHELE Rover from JPL

Lunar Rover Vehicle from Apollo Missions

Regolith

Properties

- Thickness of about 5 m to 10 m depending on location
- Fine gray soil, with rock fragments throughout
- Constantly bombarded by micrometeorites and solar wind irradiation
 - Glass can be found at the bottom of craters

Customer Need Table

Questions	Response	Interpretation		
1. How big is the payload we	"300 kg in earth's gravity"	The LSS assembly tool will		
will be lifting?		need to lift a payload with a		
		mass of 300 kilograms.		
2. What are the size	"4m x 4m"	The LSS assembly tool will		
constraints of the system?		need to fit in an area of 16		
		meters squared, fully		
		assembled.		
3. What scale of a model do	"I would like this to be truly	The simulation will need to		
you expect?	parametric which I can use a	have various scaled models		
	slider gain to scale down the	available for the customer.		
	model."			
4. How detailed of a	"As a customer I would	A full physics model is		
simulation?	answer this with I want a full	needed for the simulation- it		
	animated simulation	must lift and transport the		
	including full physics model."	payload.		
5. Do we need to worry about	"This is going to be heavy	Powering the LSS assembly		
how to power the system?	machinery on the moon so I	tool will be the responsibility		
	am looking for you to	of the team.		
	determine how this would be			
	powered. I assume solar."			
6. Do you want it to be fully	"As a customer of course, I	Upon arrival to the Moon, the		
assembled when we get	want it fully assembled. My LSS assembly tool			
there?	expectation would be that the able to begin			
	any assembly needed would	lifting/transporting payloads,		
	only require the hand tools."	disregarding minor hand tool		
		adjustments.		

7. Will the operator be on the	"The operator would be on	The operator of the LSS	
Moon or on Earth?	same 'planet' as the machine''	assembly tool will be	
		relatively close to the system.	
8. Do we have mass	"Less than 805 kg. No	The LSS assembly tool will	
constraints? Material	specific requirements on	be less than 805 kilograms of	
requirements?	materials."	mass. There are no specific	
		materials that need be used.	
9. Will the system need to lift			
the payload and then attach it			
to another part of the lunar			
base (a docking mechanism)?			
10. Besides lifting and			
transporting the payload,			
should the LSS assembly tool			
do anything else?			
11. What range do you			
desire?			
12. How high is the platform			
that we will be moving the			
payload from/to?			
13. Are you concerned about	"TBD, assume yes until	Yes, the design will account	
regolith?	clarified"	for locomotion over regolith.	
14. Is there a specific	"One that is industry friendly	Until further notice, the team	
program or software package	and can be shared if	will use the simulation tool	
the simulation should be done	necessary."	recommended by our faculty	
in?		adviser.	
15. Is there a preferred	"No."	The control system to be used	
controller for the "driver" to		by the driver is at the	
use?		discretion of the team.	
17. Is there a concern for the	"Yes, but this will be	This is not the current focus	
time needed to move a	determined later."	of the design.	
payload?			

FAMU-FSU Engineering

Functional Decomposition

System Functional Decomposition					
Function	Measure	Transfer	Control Magnitude	Provide	Convert
Transmit Power		+			
Store Power				+	
Receive Power		+			
Regenerate Power			+		+
Send Communication Signals		+			
Broadcast Signal		+	+		
Receive Signal		+			
Process Signal		+	+		+
Identify Signal	+	+			
Detect Signal	+	+			
Translate Vehicle		+			
Rotate Vehicle		+			
Convert Electricity to Translational Motion					+
Convert Electricity to Rotational Motion					+
Traverse Terrain		+			
Take Angle Input	+				
Indicate Angle Change		+			
Translate Payload		+			
Secure Payload		+			
Rotate Payload		+			
Convert Electricity to Payload Rotation					+
Convert Electricity to Payload Translation					+
Lift Payload		+			

Flow Chart of Motion

Flow Chart of Energy

Flow Chart of Payload

Torque

---- Force

Flow Chart of Communication

FAMU-FSU Engineering

Current Progress

Target	
(Simulation)	
4kW	
1-hour max stress operation / 8-hours normal operations	
16-hour recharge time	
100 m	
100 m	
100 m	
0.250 ms (Response Time)	
0.250 ms (Response Time)	
100 m	
100 m	
360°	
500 Nm	
5 km ²	
0-360°	
0-360°	
2 m	
1500 N	
360°	
500 Nm	
500 Nm	
300kg	
16m ²	
100m	
SAE Level 1	
120V/230V	

	Attributes	Metric (Simulation/Prototype)	
	Transmit Power	Simscape/Multimeter. Based off requirement for electric	
		motor of typical forklift and requirement to lift 300 kg	
		payloads	
	Store Power	Simscape/Multimeter, Clock	
	Receive Power	Simscape/Multimeter, Clock	
	Send Communication Signal	Simscape/Test signal at varies range until no signal is found	

