



Flugskóli Íslands

TERMON

TF-IFA

P2002JF



Introduction

SPECIFICATION AND DESCRIPTION

P2002JF

This document applies only to the Tecnam P2002JF and is published for the purpose of providing general information for the evaluation of design, powerplant, performance and equipment.



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GENERAL DESCRIPTION

P2002JF

Low Wing High Pleasure

Construction

P2002JF

The P2002JF is a two-seater side by side, low wing aircraft. The P2002JF features superlative performance and flying qualities, now confirmed by hundreds of P2002 ultralights, Light Sport and VLA aircraft sold throughout the world and validated in 15 countries other than Europe. The ease of piloting and maintenance make this aircraft an excellent solution for training in flight schools. It is also an ideal platform for surveillance and as well as, of course, for pure recreational and private use. The option to use 100LL AVGAS or unleaded automotive fuel (with up to 10% ethanol content) makes this aircraft even more flexible and cost effective. The P2002JF encompasses the latest developments of Tecnam aircraft. The use of advanced software for design, structural and fluidynamics analysis, and experience in building aeroplanes with all types of materials results in continuous aircraft improvement. Due to the tapered laminar aerofoil and the slotted flaps the P2002JF is an outstanding aircraft, a perfect mix of aerodynamics, performance, and structural efficiency.

Many flight schools in Europe and all over the world rely on P2002JF (certified according to the CS-VLA and validated in several foreign countries) for students initial training. Many of them continue their training up to the ATPL with the Tecnam P2006T twin making Tecnam the ideal one-stopshop for flight training aircraft all over the world. The Tecnam P2002JF structure is based on a steel tubing cabin truss covered by aluminium sheets. The wing is all aluminium made and built with a single spar and full metal torsion box. The wing's leading edges are easily detachable for repairs and also incorporate the fuel tanks (50 Lt - 13.2 US gallons each). They are separated from the cabin in order to maximize passive protection. The sliding canopy allows 360° of vision in the cockpit and has full rollover protection tested via inverted drop tests.

The stabilator, horizontal tail design, provides remarkable longitudinal hands-off stability along with minimum drag and weight penalty. This provides balanced two finger flight control. The wide slotted flaps, electrically activated, allow stall speed lower than 40 Kts and allows the aircraft to perform steep approaches and easier landings. The all aluminium ailerons are effective and ensure a quick roll rate without being overly sensitive. All control surfaces are made out of aluminium and all of them, except for flaps and tab, are massbalanced.

Landing Gear

The main landing gear are constructed of spring steel. This provides a main gear that is robust enough for unimproved landing strips and requires no service. The trailing link nose gear uses a rubber shock absorber system that was designed for the rigours of the training environment. The main landing gear wheels and brakes are 5.00x5 providing ability to use multiple different tyre brands that can be chosen in relation with the mission-type and expected landings per hour. The brake control is activated by a single central lever located between the seats or, alternatively, by toe brakes which are also available as an option. A parking brake valve is located on the console between the seats.

Introduction

Powerplant and Propeller

The top and bottom engine cowls are quickly and easily removable making any maintenance procedure faster to accomplish. The top cowl has 2 large hinged gull-wing style doors for easy access and effective pre-flight inspections of the entire engine compartment.

The engine is set low and the cowling slopes down from the windshield, so forward visibility is outstanding even with a fully equipped instrument panel. The steel firewall is soundproofed.

The power plant is a Rotax 912S2 series fourcylinder, four-stroke engine. The engine is liquid and air cooled with an integrated 1:2.4286 reduction gear. The use of liquid cooled heads and air cooled cylinders allows the engine to maintain safe operating temperatures even if a rapid descent is performed immediately after a prolonged climb.

A fixed pitch wood, composite wrapped Hoffmann propeller comes standard while the hydraulic variable pitch propeller from Hoffman is also available as an option. An electrical fuel pump is installed to provide an effective back-up to the mechanical one. Circuit breakers are standard. The battery is located in the rear of the fuselage with easy access through an external hinged door. An external power socket allows for engine start, tests, and avionics management/training without the use of internal battery.

Avionics

The largest selection of avionics choices are available on the P2002JF in order to allow almost any type of operations: basic VFR-DAY equipment, VFR-NIGHT equipment and glass cockpits are available. With an extremely wide choice of rackmounted avionics, such as the latest Garmin radios and GPS, IFR training procedures (not in IMC conditions) can be possible via dual VOR indicators and radios, ADF and DME options.

Special Hand Controls version

On March, 27th, 2014, the P2002JF aeroplane was approved by EASA to incorporate full integrated hand control kits for disabled pilots. This makes the P2002JF the first worldwide factory-built VLA certified aircraft equipped with hand controls. This version of the Tecnam P2002JF aircraft architecture is very simple and flexible. Flight instruction will be allowed by a third throttle control and second slip indicator on the RH side, while whomever is seated on the left side (student or disabled pilot flying solo) will:

control the stick (pitch and roll) and the brakes with the left hand;

control the rudder, throttle and flap with the right hand on the central control.

No flight control operations other than the radio and altimeter settings, will need to separate the hands from controls making this solution safe and ergonomic.

In addition to the flight control modifications, several improvements have been made to make the access to disabled pilots easier: strengthened leading edges (both LH and RH), four additional grab handles to help step inside and outside the cabin and, finally, an increased canopy opening.

Certification

The Tecnam P2002JF is delivered in full compliance with the requirements of EASA CS-VLA.





INTERIOR AND EXTERIOR

Seats are adjustable in flight and increase in height as they are moved forward.

The luggage area allowing for 44 pounds/20 kg of weight is located behind the seats with ample room for several travel bags. The interior is spacious, ergonomic and comfortable.

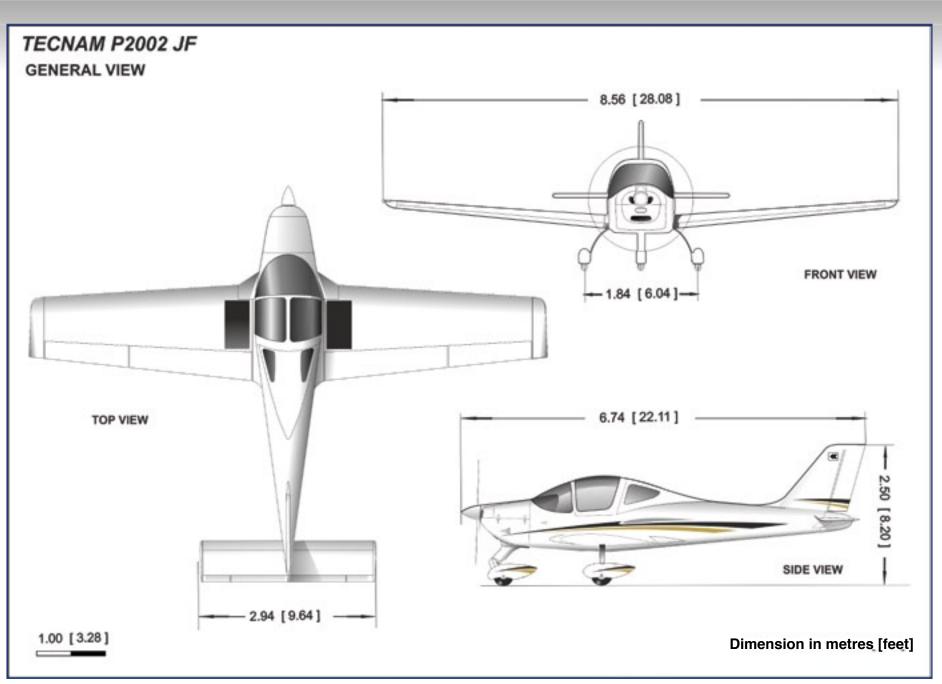
Cabin	ft m	
Height	3	0,91
Width	3.6	1,1

Baggage Compartment

Width	2.62 ft	0,80 m
Length	1.48 ft	0,45 m
Volume	5.74cu.ft	162lt
Max. permissible load	44lb	20kg



EXTERIOR



Views

DIMENSIONS

DIFILINGIONG	ft	m
Overall Height	7.9	2,4
Overall Length	21.7	6,63

ft

DESIGN WEIGHT AND LOADING	P2002-JF FP		
	kg	lb	
Maximum Take Off Weight	620	1,367	
Empty Weight, VFR Standard	380	838	
Useful Load	240	906	
Baggage allowance	20	44	

WING	ft	m
Span (overall)	28.2 8,6	
Area	123.8 ft ² 11,5 s	
Dihedral	5°	
Aspect ratio	6.4	

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PERFORMANCE	P2002-JF Fixed Pitch Propeller Variable Pitch Propeller			
			ch Propeller	
Max Cruise Speed KTAS	122 kts	226 km/h	128 kts	237 km/h
Stall Speed (Flaps Down Power Off) KCAS	41 kts	76 km/h	41 kts	76 km/h
Practical ceiling	14000 ft	4267 m	14000 ft	4267 m
Take off run	777 ft	237 m	630 ft	192 m
Take off distance	1286 ft	392 m	1083 ft	330 m
Landing run	538 ft	164 m	446 ft	136 m
Landing distance	1056 ft	322 m	1099 ft	335 m
Rate of climb	874 ft/min	4,4 m/sec	950 ft/min	4,8 m/sec
Range	568 NM	1502 km	568 NM	1502 km

All estimated performance data are based on aeroplane weights at MTOW; standard atmospheric conditions; level, hard surface, dry runways, no wind.



POWERPLANT & ACCESSORIES

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ROTAX 912 S2/3

- 4-cylinders
- 4-stroke liquid-/air-cooled engine with opposed cylinders
- Dry sump forced lubrication with separate oil tank, automatic adjustment by hydraulic valve tappet
- Mechanical fuel pump
- Dual electronic ignition
- Propeller speed reduction unit
- · Air intake system
- Gearbox Reduction Ratio 2.43:1





STANDARD EQUIPMENT

FLIGHT INSTRUMENTS AND INDICATORS

Magnetic Compass Airspeed Ind., Kts Altimeter (In) Vertical Speed Attitude Gyro Directional Gyro Turn And Bank Indicator Flaps Indicator Pitot System Static System Stabilator Trim Position Indicator

P2002JF

ENGINE INSTRUMENTS

RPM Indicator Hour Recorder Oil Press Oil Temp. Head Temp. Fuel Press. Voltmeter Ammeter LH + RH Fuel Qty

OTHER INSTRUMENTS / WARNING

Chronometer O.A.T. Indicator Generator Warning Light Vacuum Suction Gauge

FLIGHT CONTROLS

Hydraulic Brakes **Parking Brake Electrical Flaps Dual Flight Controls** Steerable Nose Wheel Stabilator Trim (electric actuated from stick) **Engine Controls** _ Throttle, Two Carburettor Heat Choke Flight Trim Controls Stabilator With Indicator Fuel Control Selector with On/Off Panel Switches: _ Starter Fuel Pump Engine LH And RH Ignition Switches

ELECTRICAL SYSTEM

12 VOLT 18AMP. Battery 12 VOLT Alternator 20 AMP. Switches _Nav. Lights _Landing Light _Strobe Light External Power Supply Receptable Circuit Breaker Panel

FUEL SYSTEM

Two Integral Fuel Tanks with 100 litres total capacity

Engine Driven Fuel Pump Auxiliary Fuel Pumps, electric Fuel Quick Drain 1 X Shut Off And Fuel Selector Valve ANDAIR

INTERIOR

Pilot And Copilot Seats _ adjustable fore and aft Seat Belts & Shoulder Harness, all seats Wall To Wall Carpeting Luggage Compartments Fire Extinguisher Radio Call Plate Soundproofing First Aid Kit Emergency Hammer

EXTERIOR

Epoxy Corrosion Proofing, all structure Sliding Canopy with Lock And Key Rear Window Tie Down Rings Main Wheels, 5,00 X 5 Cleveland Nose Wheel, 5,00 X 5

EXTERIOR LIGHTS

Nav. Lights LED with strobe AVEO Full LED TSO Taxi Light LED

CABIN CONFORT SYSTEM

Windshield Defroster Ventilator Adjustable, 2 Place Heating System

POWERPLANT AND PROPELLER

Engine - 1 ROTAX 912S2 100 HP, 4 Cylinders Liquid/air cooled, integrated reduction gear Dual Ignition System Throttle Control LH/RH Tubular Steel Engine Mount Propeller - Hoffmann, 2 Blade Fixed Pitch Propeller Spinner Air Filter Oil Filter Oil And Water Coolers Carburettor Heat with Manual Control Thermostat Valves Oil and Water

PRODUCT SUPPORT/DOCUMENTS

Manufacturer Full Two Year Limited Warranty Pilots Operation Handbook Maintenance Manual Parts Catalog Aircraft Log Book Engine Log Book

STANDARD GARMIN AVIONIC PACKAGE

GMA 340 Audio Panel GNC 255A COM/NAV GTX 335 Transponder ADS-B OUT ELT 406 Mhz KANNAD Antennas:

- Transponder
- VHF
- Marker Beacon
- ELT

Speakers Microphone Stick Push-To-Talk Switch-Pilot/Copilot Mic & Phone Jacks-Pilot/Copilot

AVIONICS OPTIONS

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GARMIN.

P2002JF

STANDARD AVIONICS



STANDARD GARMIN AVIONIC PACKAGE

- GMA 340 Audio Panel
- GNC 255A COM/NAV
- GTX 335 Transponder ADS-B OUT
- ELT 406 Mhz KANNAD
- Antennas:
- Transponder
- VHF
- Marker Beacon
- ELT
- Speakers
- Microphone
- Stick Push-To-Talk Switch-Pilot/Copilot
- Mic& Phone Jacks-Pilot/Copilot

Avionics Options



AVIONICS OPTION 1





VFR NIGHT VERSION

Includes the following equipment:

- Heated Pitot
- GILL 25A Battery
- Instrument Light
- Map Light
- Dimmer
- Aux Alternator

Non-Additive. Replaces all Standard Avionics

Avionics Options

GARMIN.



AVIONICS OPTION 2



GLASS COCKPIT G500

Includes the following equipment:

- GARMIN G500 PFD
- GTN 650 Com/Nav/Gps with Antennas and installation exchange GNC225A
- · Airspeed Indicator and Altimeter back up TSO
- Aux Alternator

Non-Additive. Replaces all Standard Avionics.

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Avionics Options

GARMIN



AVIONICS OPTION 3

GLASS COCKPIT G500 + VFR NIGHT *Includes the following equipment:*

- GARMIN G500 PFD
- GTN 650 Com/Nav/Gps with Antennas and installation - exchange GNC225A
- · Airspeed Ind. and Altimeter back up TSO
- Heated Pitot
- GILL 25A Battery
- Instrument Light
- Map Light
- Dimmer
- Aux Alternator

Non-Additive. Replaces all Standard Avionics.

Export certification requirements may require additional equipment and charges. Options also pictured.



VP OPTION 1



P2002 JF VP CS/VLA

Includes the following equipment:

- Rotax 912 S3 100 hp Engine with Governor
- Central Quadrant with single throttle and pitch lever
- Hoffmann Variable Pitch Propeller
- Manifold Pressure Indicator
- Attitude and Directional Electric

Non-Additive. Replaces all Standard Avionics.

Standard

Paint

PAINT SCHEMES

St1.1_Colour Stripes St1.2 __ Colour Stripes St1.3_Colour Stripes 521 INTENSE BLUE 502 BLACK 519 MEDIUM RED 546 SLVER 546 SILVER 546 SILVER 521 INTERSE BLUE 547 GOLD 502 BLACK St2.1_ Colour Stripes St2.2 __ Colour Stripes St2.3 Colour Stripes 528 INTENSE BLUE 502 BLACK 519 RED 546 SILVER 547 GOLD 502 BLACK 519 RED 546 SILVER 546 SILVER Special Paints Sp1.1 __ Paints __ Stripes Sp1.2 __ Paints __ Stripes Sp1.3 __ Paints __ Stripes RED LIGHT GRAY 502 BLACK 528 VIVID BLUE NKVY BLUE 547 GOLD 547 GOLD 548 SILVER 546 SILVER Paints __Stripes Sp2.2 __Paints __Stripes Sp2.3 __Paints __Stripes Sp2.1_ I-XXXX 502 BLACK 502 BLACK 546 SILVER RED LIGHT GRAY NAVY BLUE



OPTIONS

Code	Description		
SPECIAL CON	SPECIAL CONTROLS		
1006	Rudder, throttle and brakes hand controls		
INSTRUMENTS			
114/A	Turn & Slip Ind. 2 1/4"		
RADIO & NAVI	GATION EQUIPMENT		
GARMIN-COM/	/NAV/GPS		
120/A	GTR225 COM 25 Mhz with Antenna and Installation		
120/B	GTR225A COM 8.33 Khz with Antenna and Installation		
119	MD200 VOR Indicator Only for GNC255A		
121	GTN 650 Com/Nav/Gps with Antennas, Triplex & Installation with GI106A Indicator		
121/A	GTN 650 Com/Nav/Gps with Antennas, Triplex & Installation with GI106A Indicator (Exch. for Std. GNC255A)		
122	GTN 750 Com/Nav/Gps with Antennas, Triplex & Installation with GI106A Indicator.		
122/A	GTN 750 Com/Nav/Gps with Antennas, Triplex & Installation with GI106A Indicator (Exch for Std. GNC255A)		
GARMIN-GPS			
128	AERA 500 with Antenna, Panel Support and Installation		
132	795 with Antenna, Panel Support and Installation		
BENDIX KING			
137	KR 87 ADF with KI227 Indicator		
139	DME KN63-14 with KDI 572 Indicator		
OTHERS			
157	Head Sets, Two		
157/A	BOSE A 20 Head Sets, Two		

Code	Description		
AIRCRAFT EQ	AIRCRAFT EQUIPMENT		
174	Tinted Windows		
176	Toe Brakes (see note # 1)		
178	Central Quadrant with single throttle lever		
182	Fuselage Cover		
185	Battery Gill G25 (Exchange for standard battery)		
209	Control Lock		
210	Towing Bar		
EXTERIOR	EXTERIOR		
201/A (JF)	Special Paint Two Colours		
ENGINE and P	ENGINE and PROPELLER EQUIPMENT		
187	Aux. Alternator		

#1 - 176 TOE BRAKES

includes:

- New Pedals
- 4 Brake Pumps
- Parking Brake Selector



P2002-JF trainer for Argentinian Air Force

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