

Sheet Metal Forming

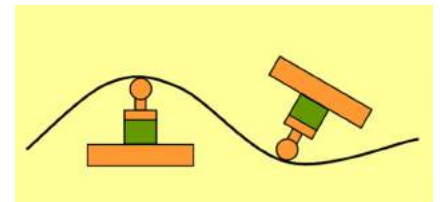
2.810

D. Cooper

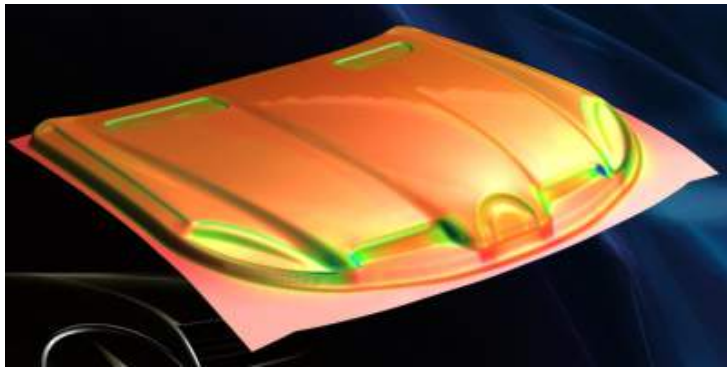
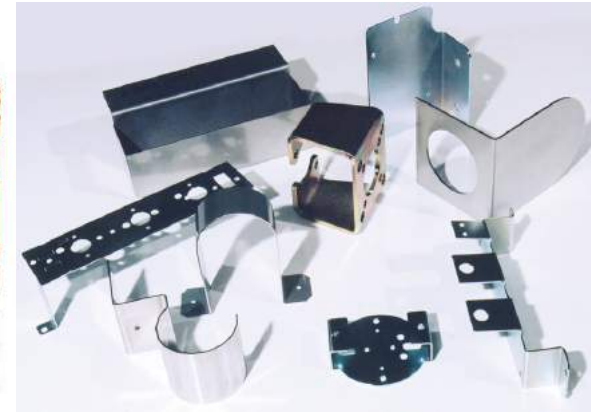
- ◆ “Sheet Metal Forming” Ch. 16 Kalpakjian
- ◆ “Design for Sheetmetal Working”,
Ch. 9 Boothroyd, Dewhurst and Knight



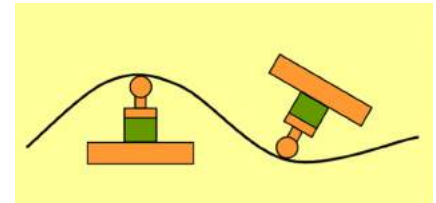
Massachusetts
Institute of
Technology



Examples-sheet metal formed



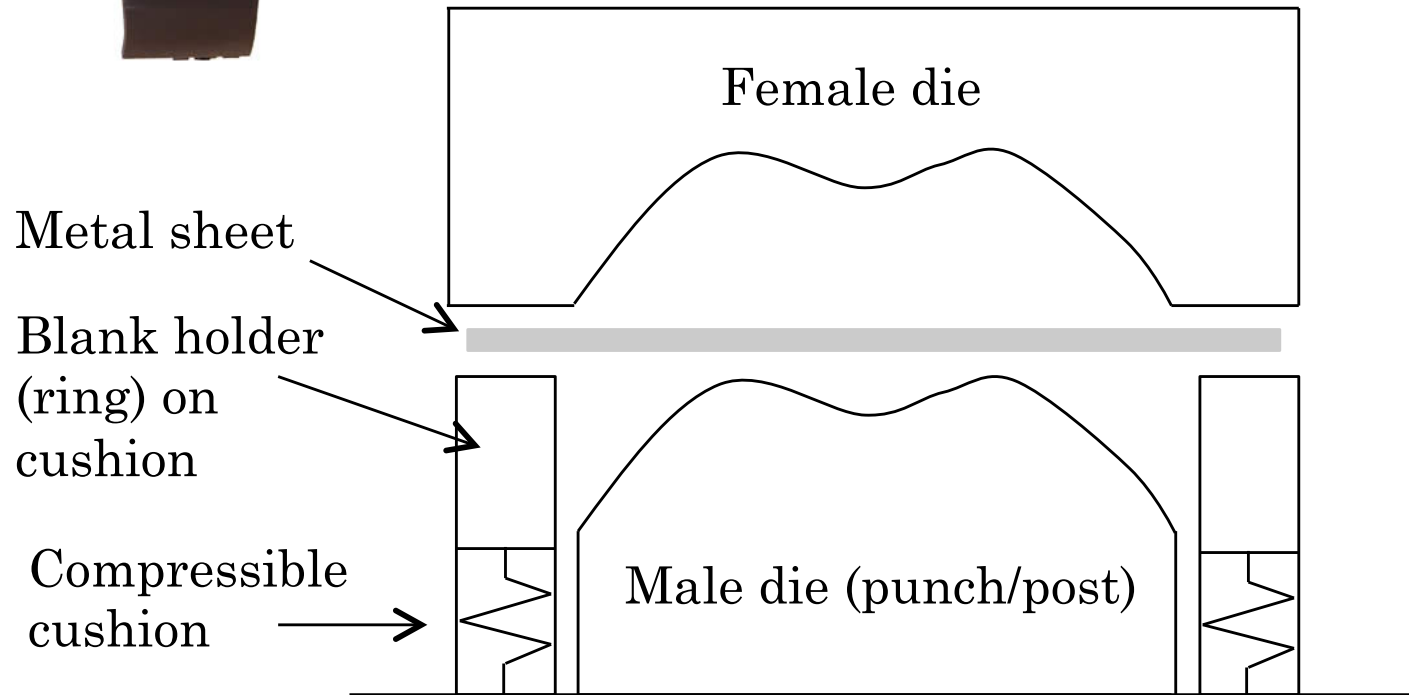
Massachusetts
Institute of
Technology



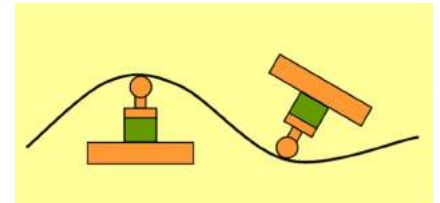
Sheet metal stamping/drawing – car industry



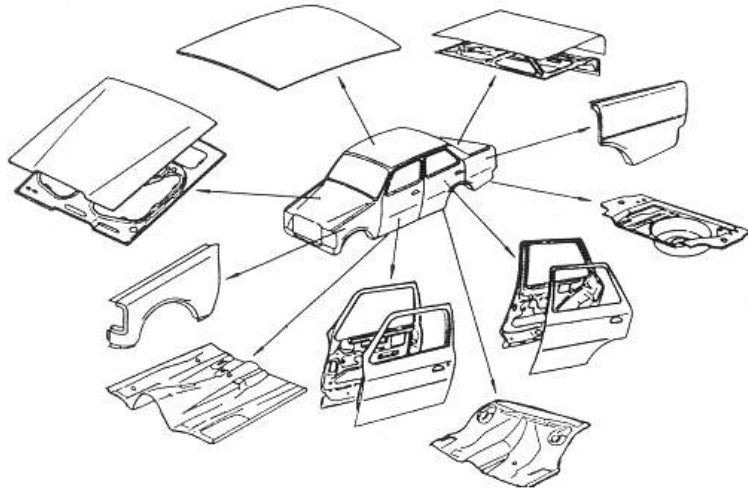
• **90million** cars and commercial vehicles produced worldwide in 2014



Massachusetts
Institute of
Technology

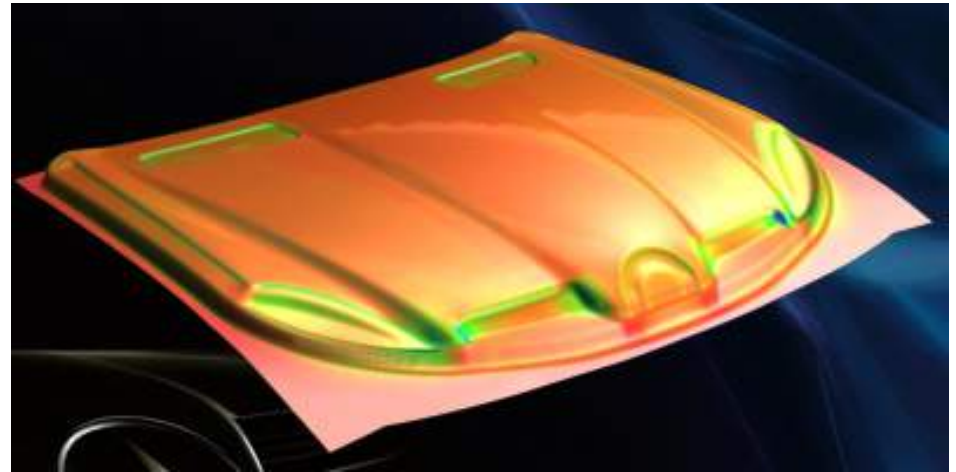


Stamping Auto body panels

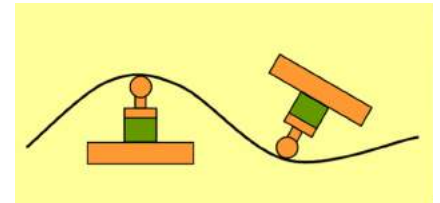


- 3 to 5 dies each
- Prototype dies ~ \$50,000
- Production dies ~ \$0.75-1 mil.

- Forming dies
- Trimming station
- Flanging station



Massachusetts
Institute of
Technology



Objectives

By the end of today you should be able to...

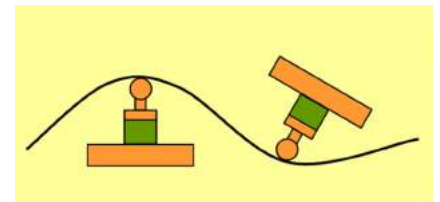
...**describe** different forming processes, when they might be used, and **compare** their production rates, costs and environmental impacts

...**calculate** forming forces, **predict** part defects (tearing, wrinkling, dimensional inaccuracy), and **propose** solutions

...**explain** current developments: opportunities and challenges



Massachusetts
Institute of
Technology

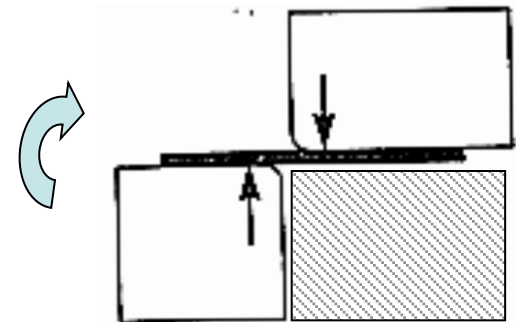
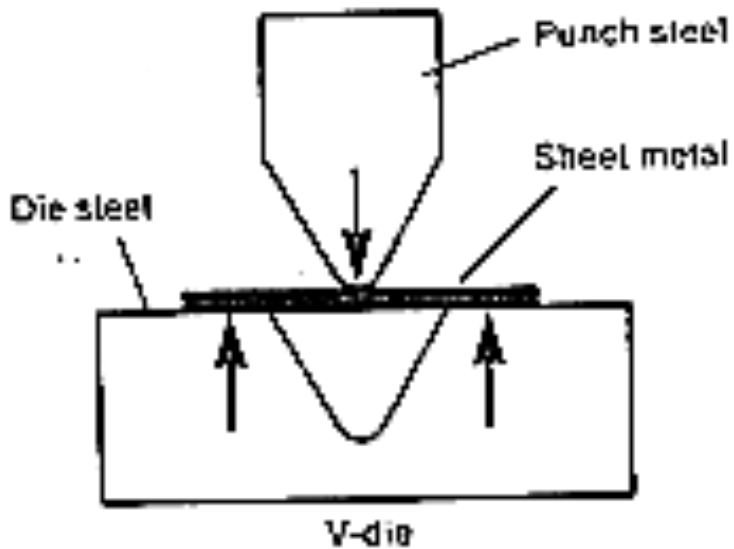


LMP Shop

Brake press



Finger brake



Technology – a brief review

Forming
Speed

Material drawn into shape

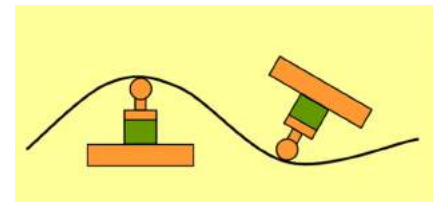
- **Conventional drawing/stamping** – expensive tooling, no net thinning, quick **20-1000pts/hr**
- **Hydro-forming** – cheap tooling, no net thinning, slow, high formability **7-13cycles/hr**

Material stretched into shape

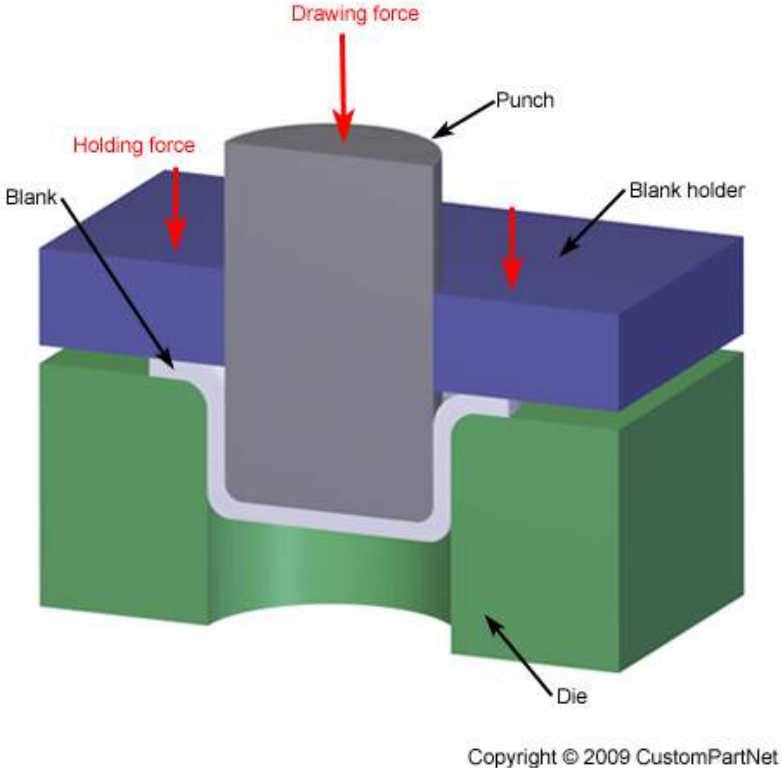
- **Stretch forming** – very cheap tooling, net thinning, slow, low formability **3-8pts/hr**
- **Super-plastic forming** – cheap tooling, net thinning, expensive sheet metal, slow, very high formability **0.3-4pts/hr**



Massachusetts
Institute of
Technology



Drawing – expensive tooling, no net thinning, quick



Deep-drawing

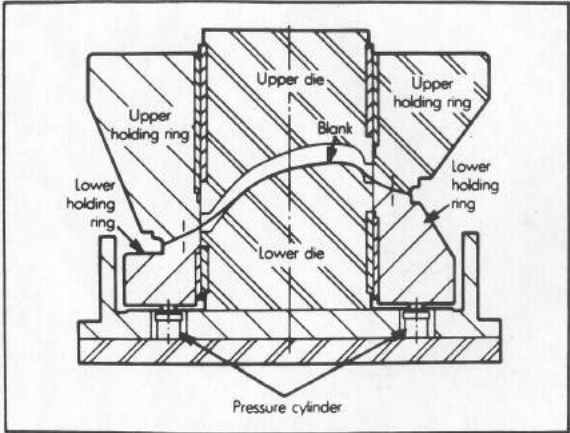
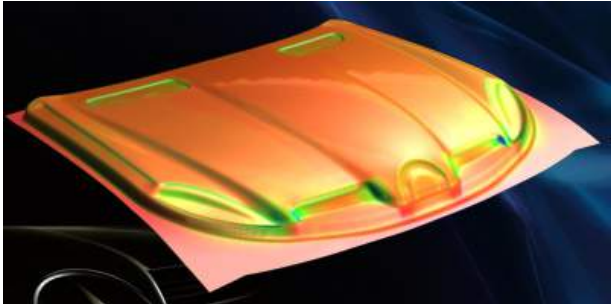


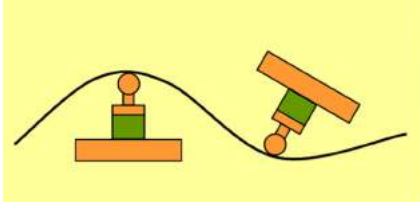
Fig. 7-23 Tooling for stretch-draw forming fenders from steel blanks. (Oldsmobile Div., General Motors Corp.)



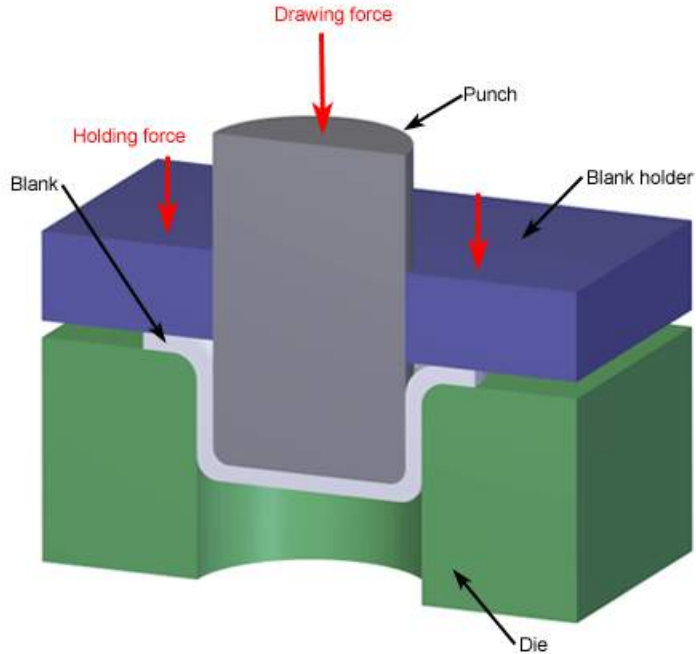
Shallow-drawing (stamping)



Massachusetts
Institute of
Technology

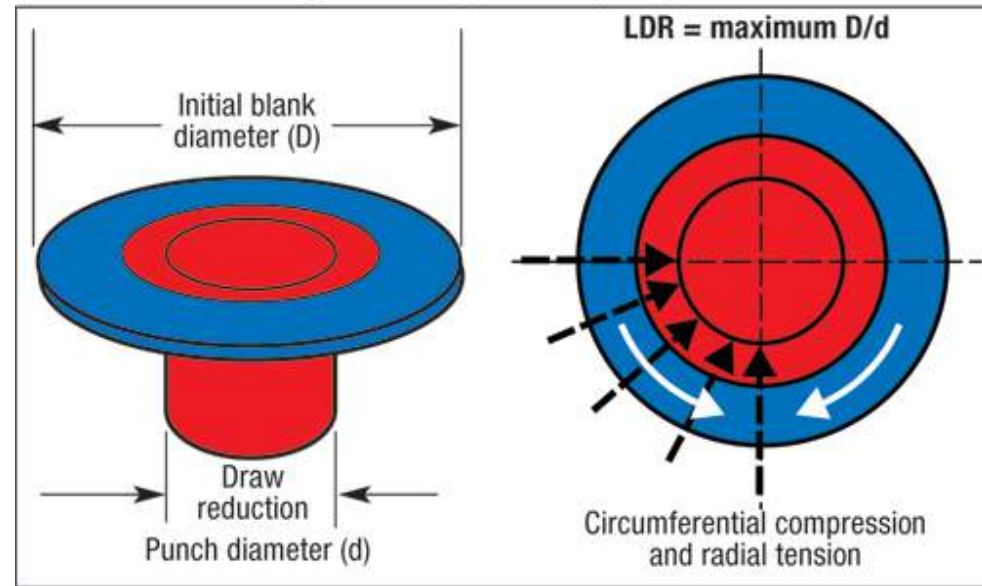


Deep-drawing



Copyright © 2009 CustomPartNet

Limiting Drawing Ratio (LDR) Defined

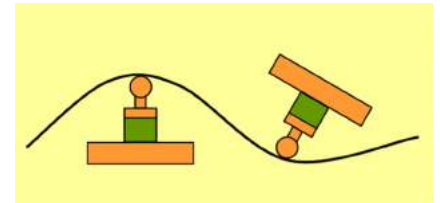


Blank holder helps prevent wrinkling and reduces springback

Blank holder not necessary if blank diameter / blank thickness is less than 25-40. Smaller values for deeper forming.



**Massachusetts
Institute of
Technology**

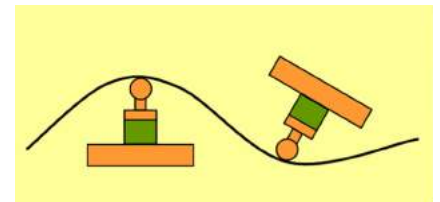




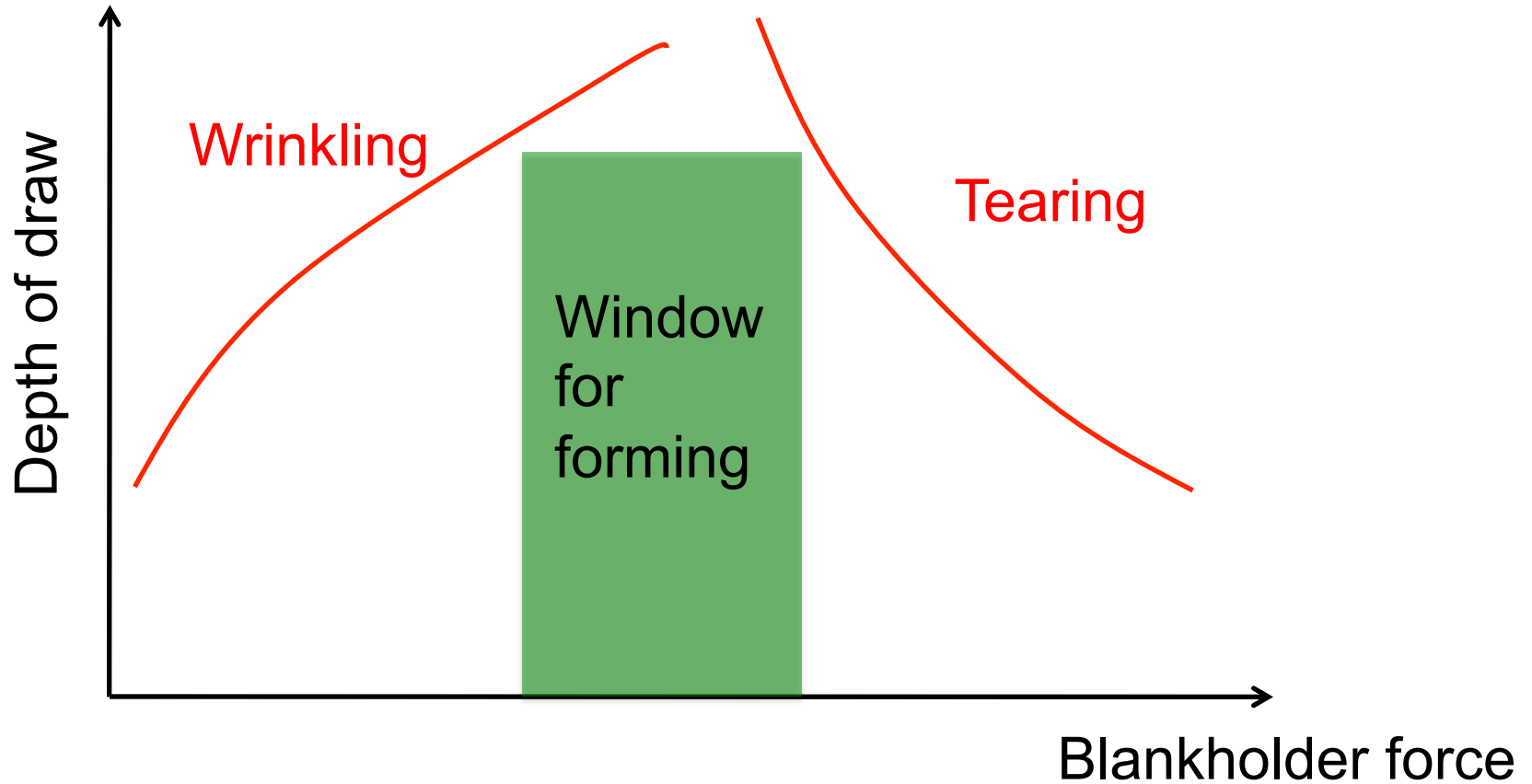
<http://www.thomasnet.com/articles/custom-manufacturing-fabricating/wrinkling-during-deep-drawing>



**Massachusetts
Institute of
Technology**



Blank holder force: forming window



Deep Drawing of drinks cans



Hosford and Duncan
(can making): <http://www.chymist.com/Aluminum%20can.pdf>

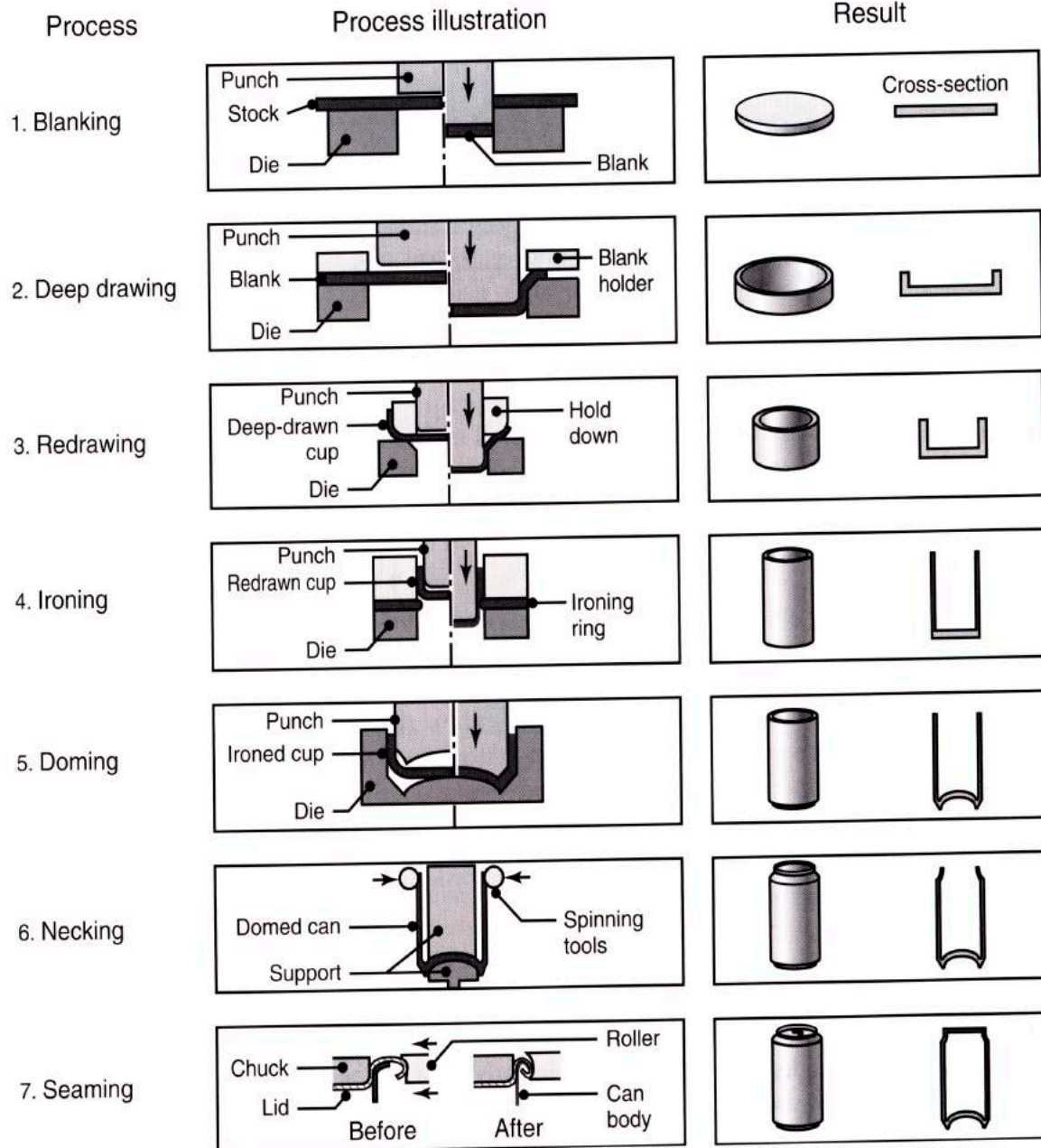
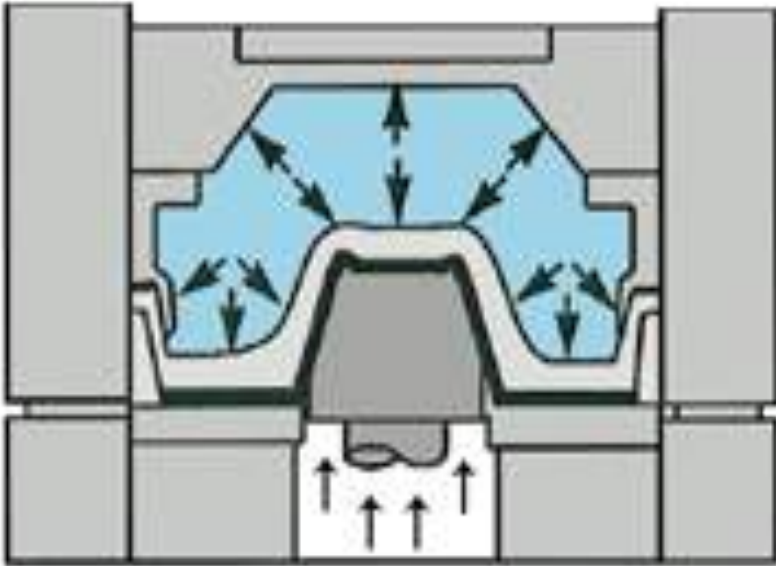


FIGURE 16.31 The metal-forming processes involved in manufacturing a two-piece aluminum beverage can.

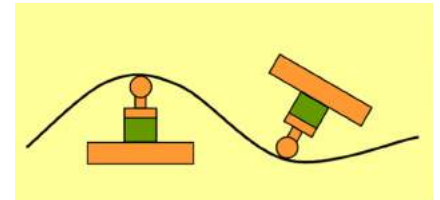
Hydro-forming – cheap tooling, no net thinning, slow(ish), high formability



Low volume batches



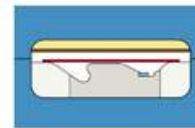
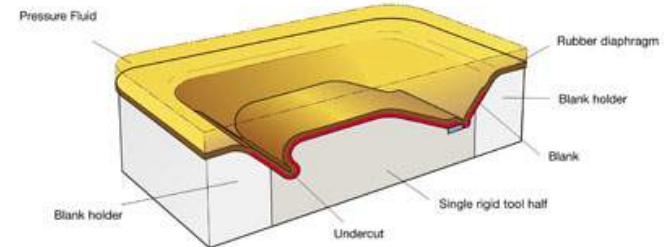
Massachusetts
Institute of
Technology



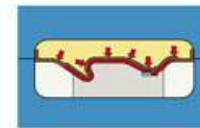
Hydro-forming – cheap tooling, no net thinning, slow(ish), high formability



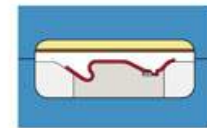
Flexform – Principle



Before



During

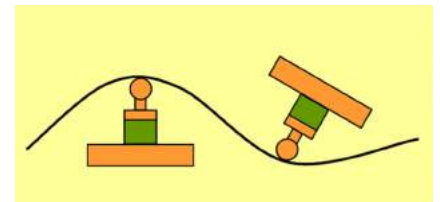


After

Low volume batches



Massachusetts
Institute of
Technology



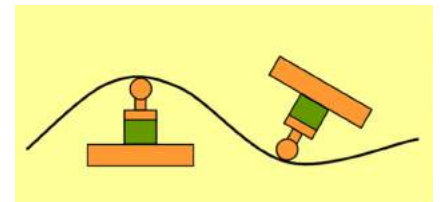
Hydro-forming – cheap tooling, no net thinning, slow, high formability



Small flexforming tool made by additive manufacturing

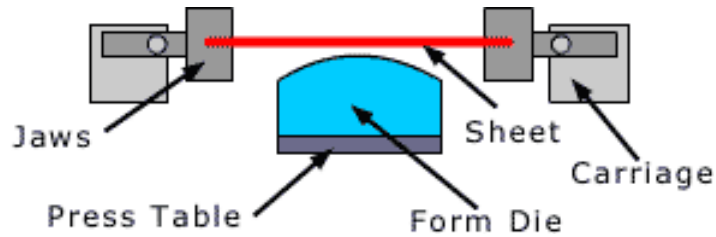


Massachusetts
Institute of
Technology

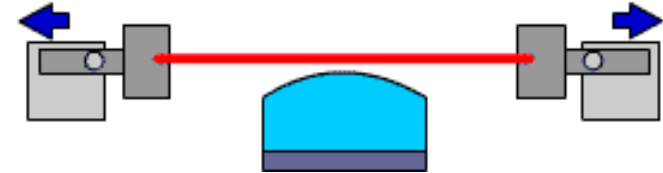


Stretch forming

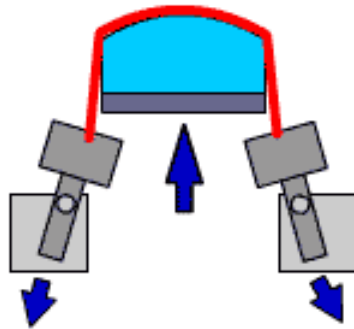
– very cheap tooling, net thinning, slow, low formability, sheet metal up to 15mx9m



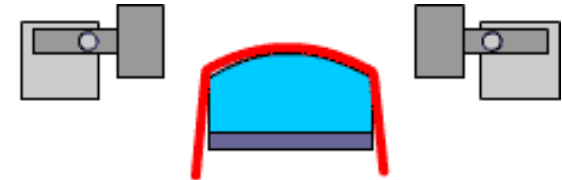
Loading



Pre-stretching



Wrapping

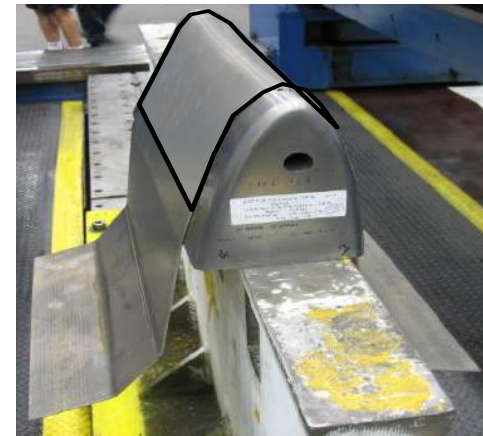


Release

* source: http://www.cyrilbath.com/sheet_process.html

Low volume batches

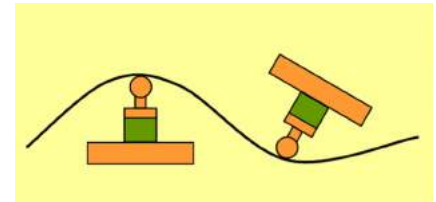
Stretch forming: Example parts



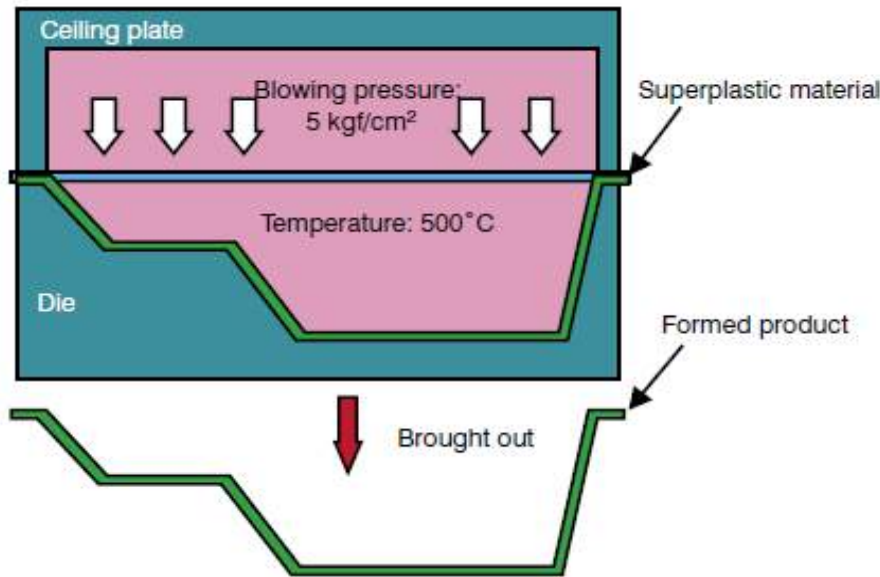
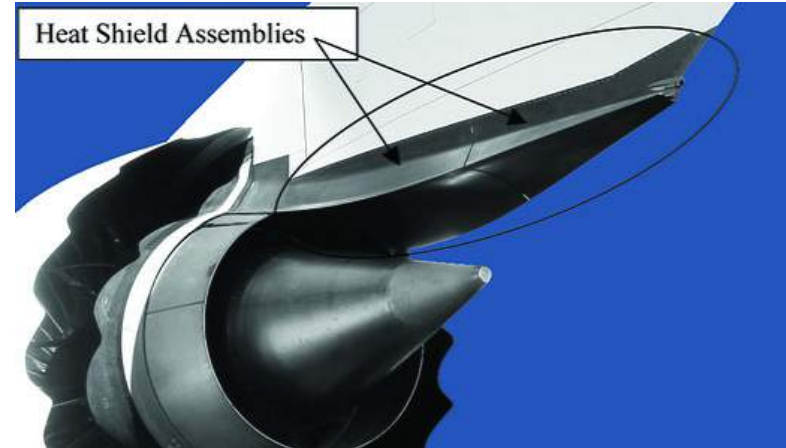
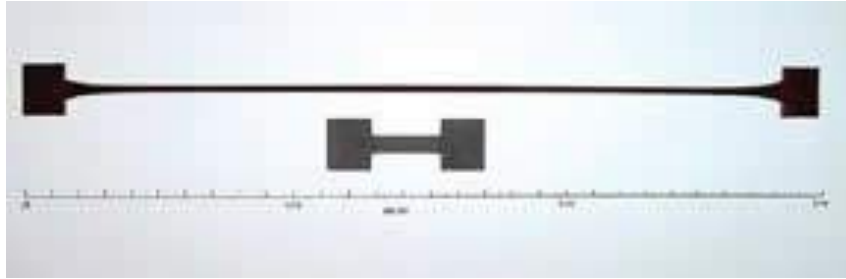
Higher aspect ratio, deeper parts



**Massachusetts
Institute of
Technology**



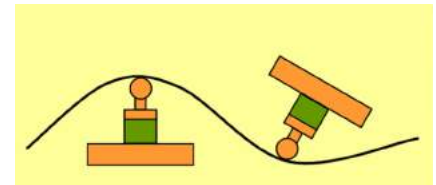
Super-plastic forming – cheap tooling, net thinning, slow, expensive sheet metal, very high formability



Low volume batches, 0.5-0.75 melting temp



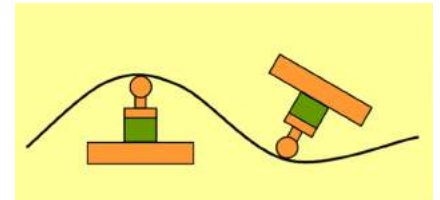
Massachusetts
Institute of
Technology



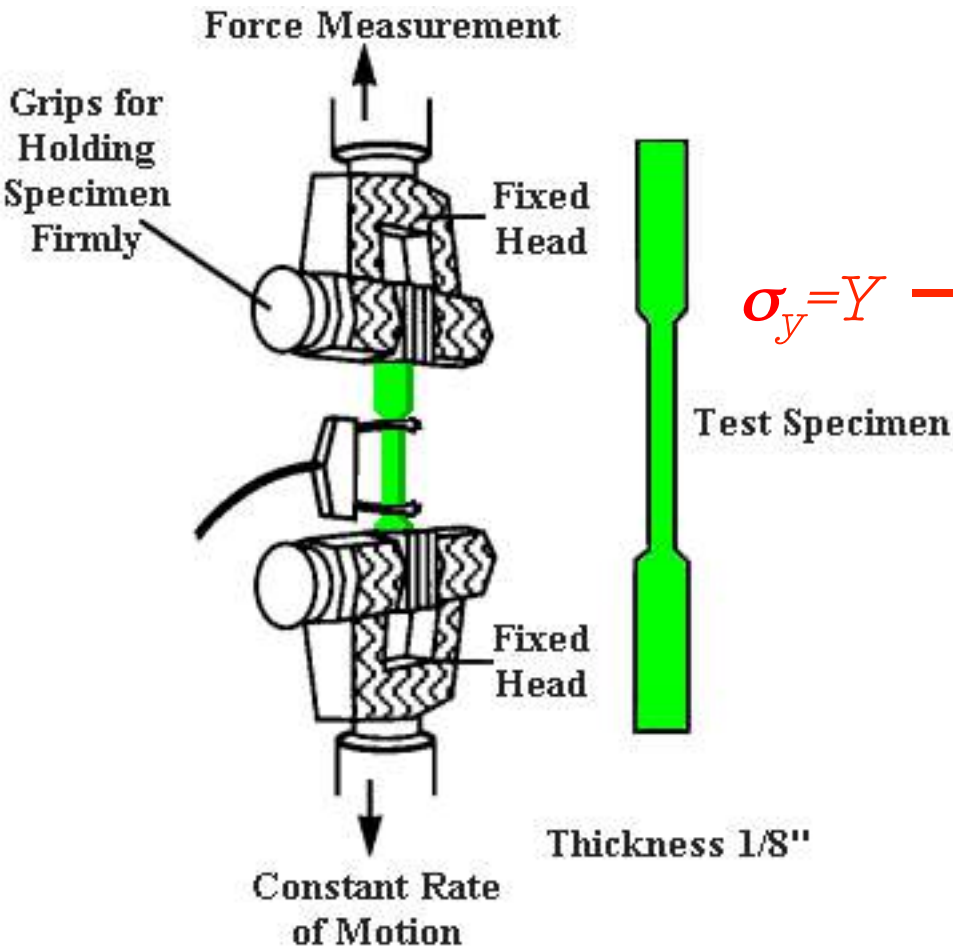
Forming forces and part geometry



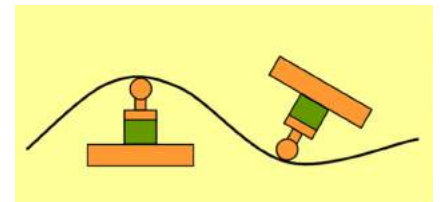
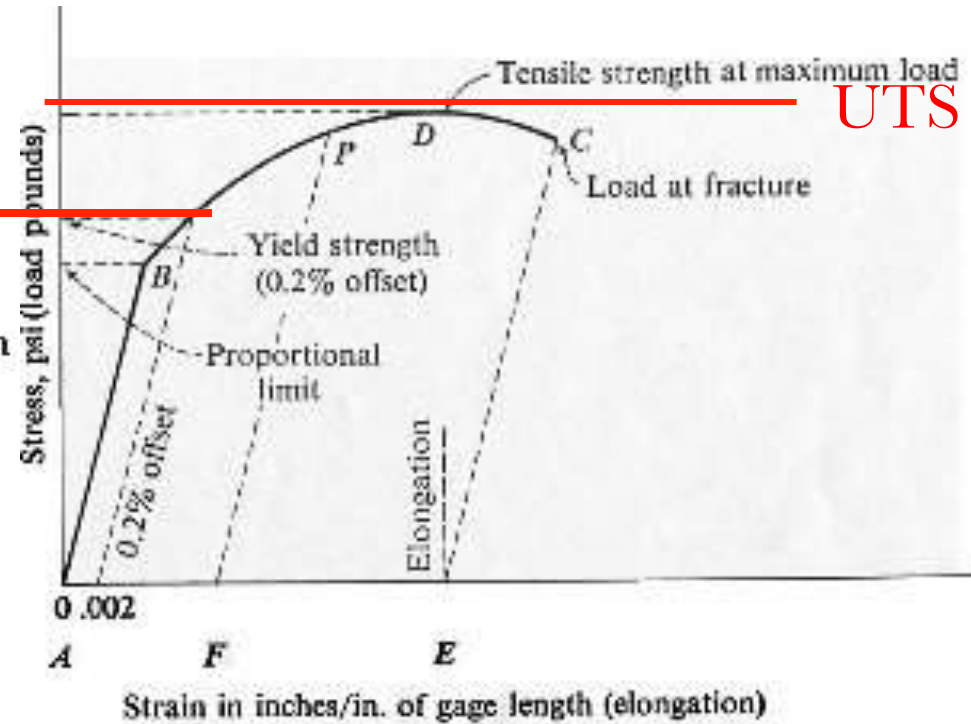
Massachusetts
Institute of
Technology

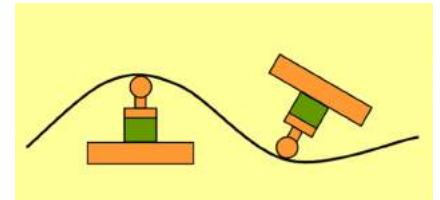
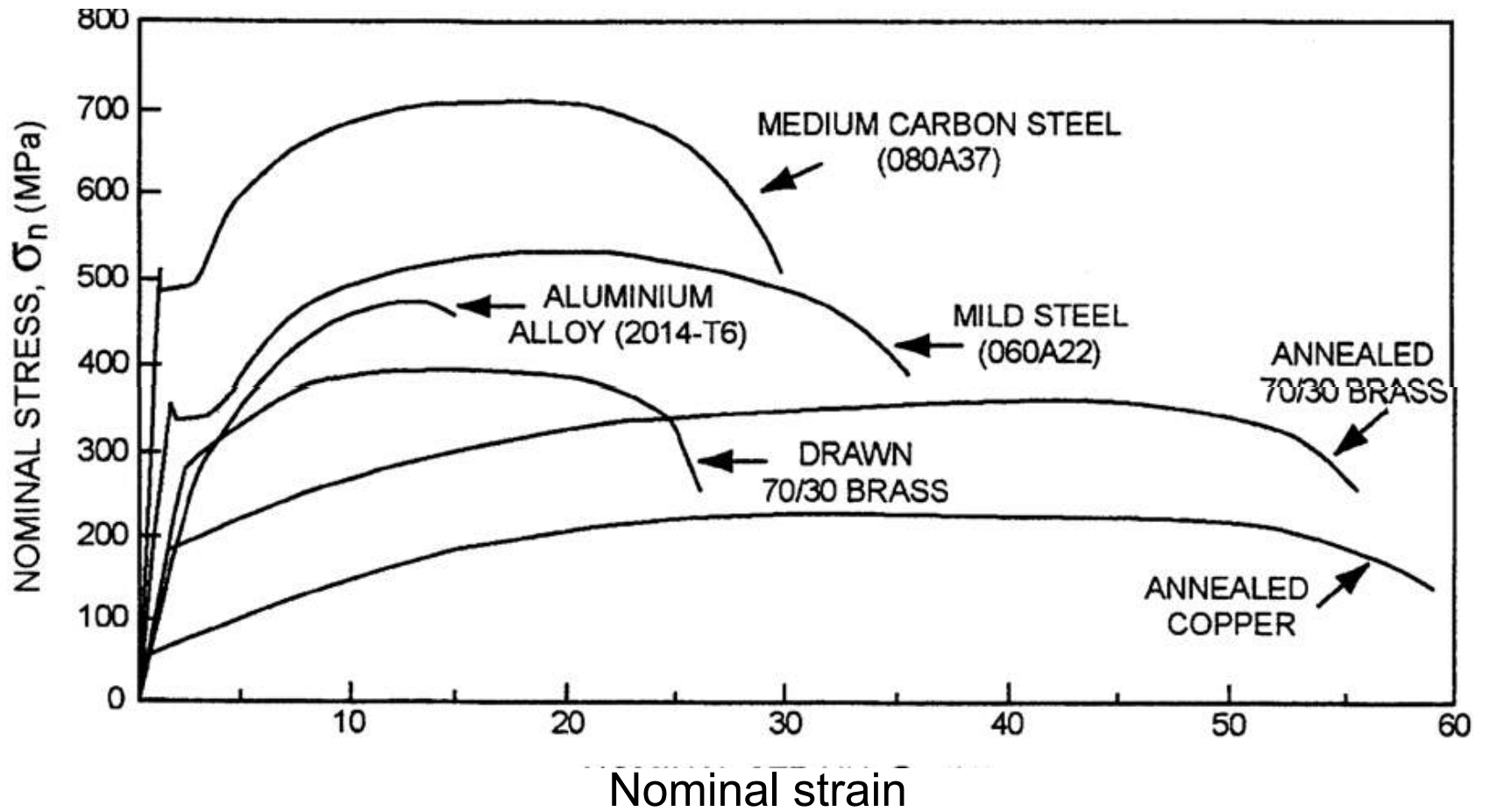


Tensile test – the Stress-strain diagram



$$\sigma_y = Y$$





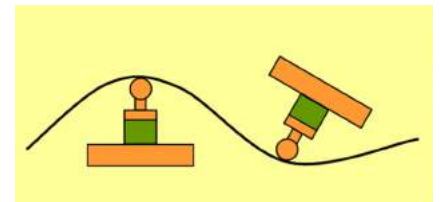
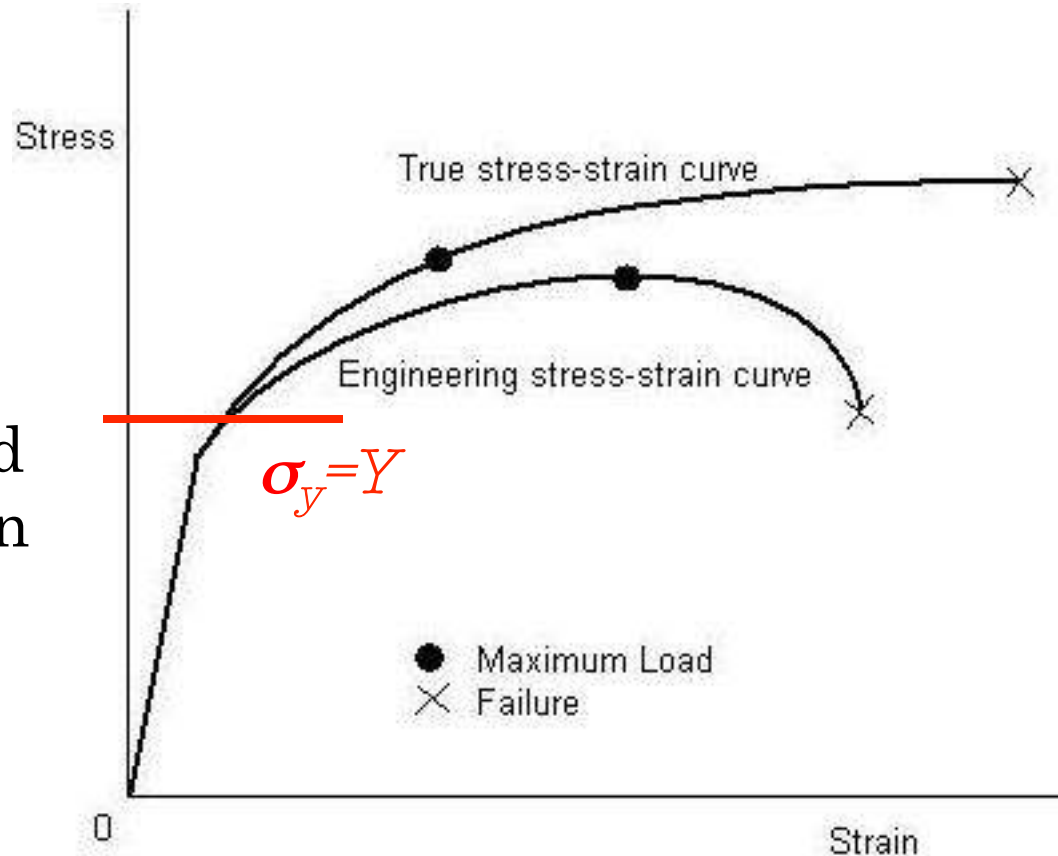
True stress & strain

$$\epsilon_{tr} = \ln(1 + \epsilon_{en})$$

$$\sigma_{tr} = \sigma_{en} (1 + \epsilon_{en})$$

True stress can be expressed using a power law (Hollomon equation):

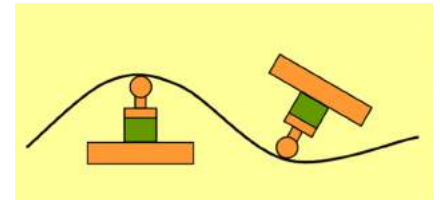
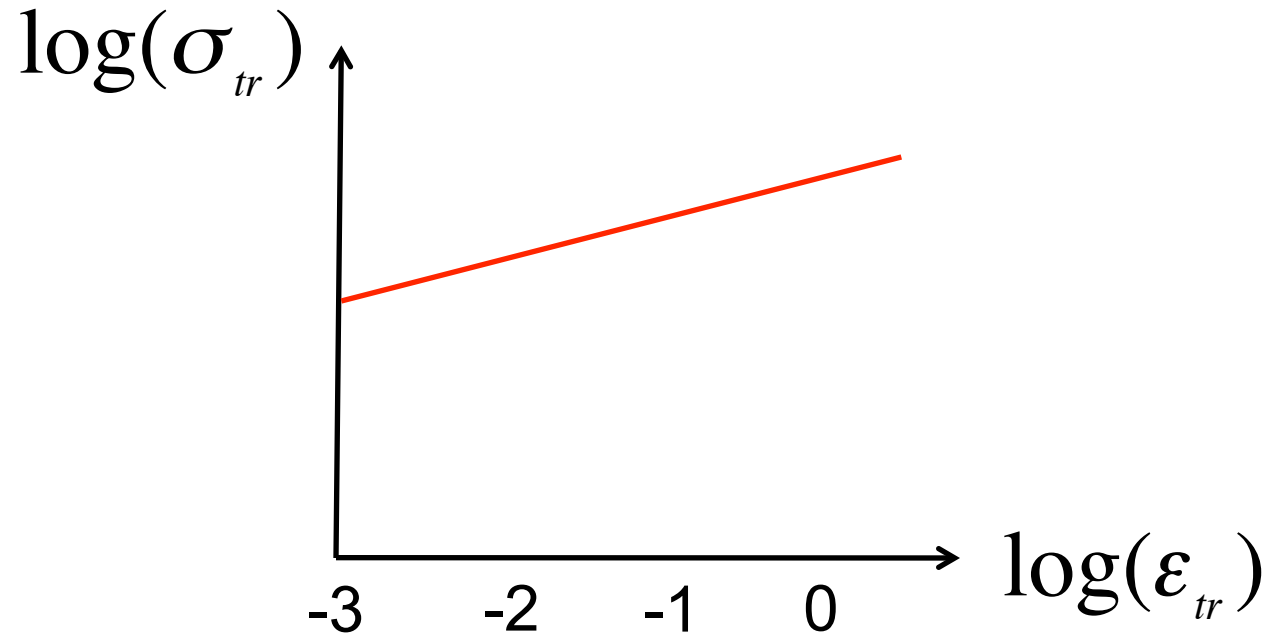
$$\sigma_{tr} = K \epsilon_{tr}^n$$



Power-Law Expression (Hollomon equation)

$$\sigma_{tr} = K \varepsilon_{tr}^n$$

Can be re-written: $\log(\sigma_{tr}) = n \log(\varepsilon_{tr}) + \log K$



Power-Law Expression (Hollomon equation)

$$\sigma_{tr} = K \varepsilon_{tr}^n$$

Can be re-written: $\log(\sigma_{tr}) = n \log(\varepsilon_{tr}) + \log K$

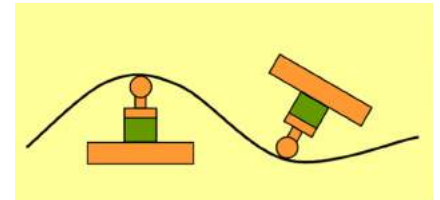
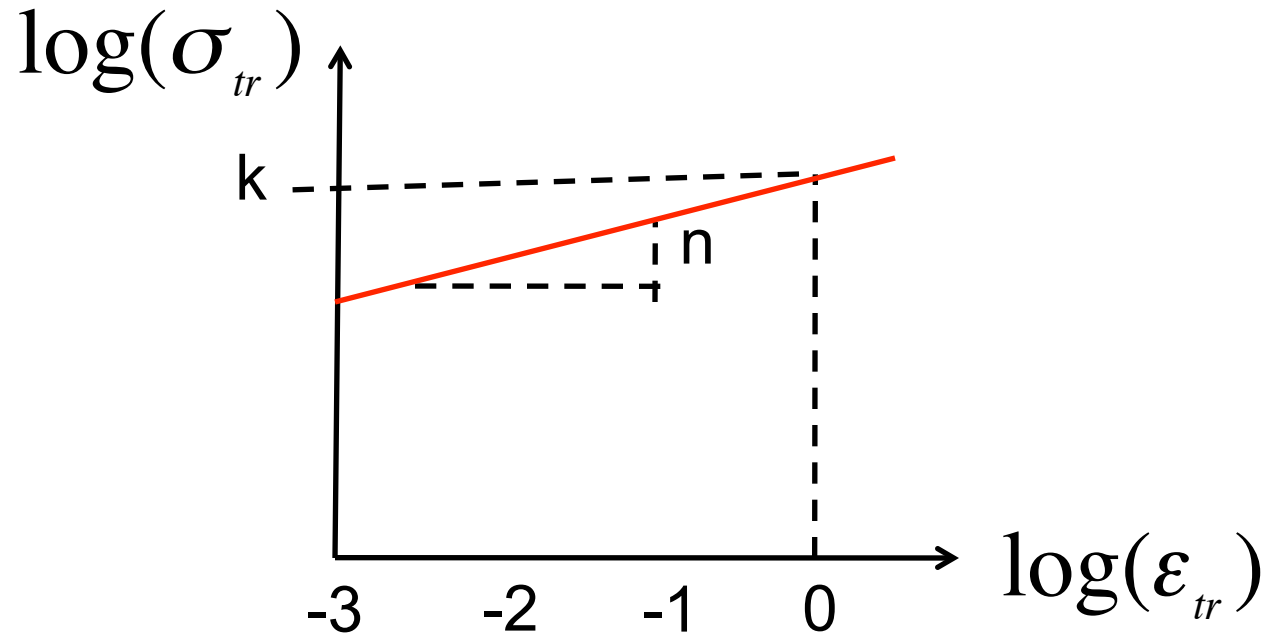
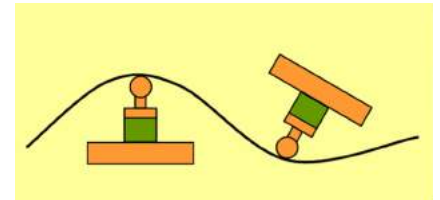
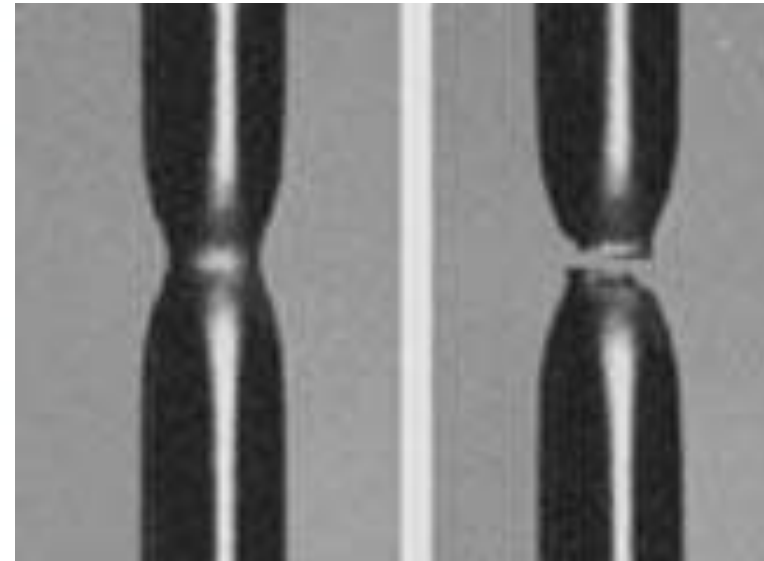
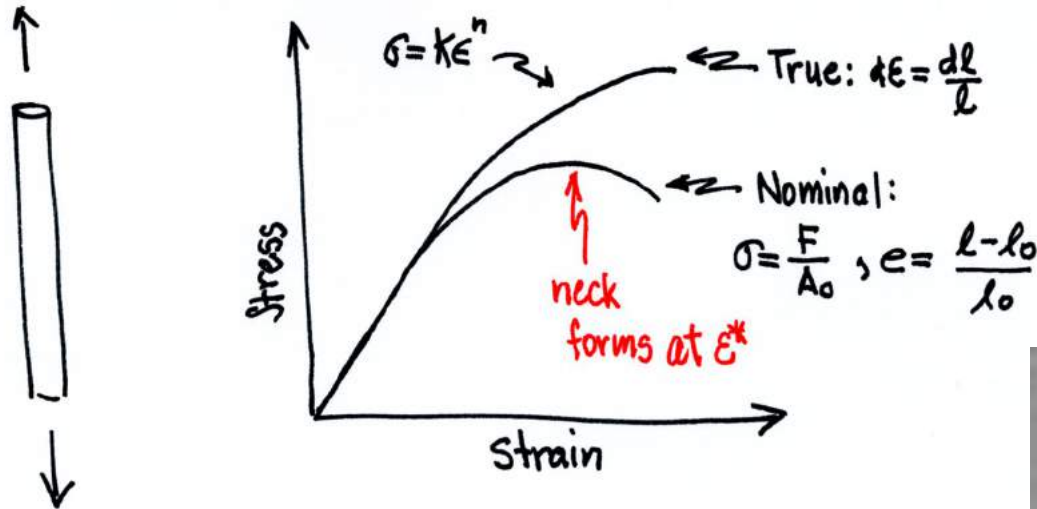


TABLE 2.3**Typical Values for K and n for Selected Metals**

Material	K (MPa)	n
Aluminum		
1100-O	180	0.20
2024-T4	690	0.16
5052-O	202	0.13
6061-O	205	0.20
6061-T6	410	0.05
7075-O	400	0.17
Brass		
70-30, annealed	900	0.49
85-15, cold rolled	580	0.34
Cobalt-based alloy, heat treated	2070	0.50
Copper, annealed	315	0.54
Steel		
Low-C, annealed	530	0.26
1020, annealed	745	0.20
4135, annealed	1015	0.17
4135, cold rolled	1100	0.14
4340, annealed	640	0.15
304 stainless, annealed	1275	0.45
410 stainless, annealed	960	0.10
Titanium		
Ti-6Al-4V, annealed, 20°C	1400	0.015
Ti-6Al-4V, annealed, 200°C	1040	0.026
Ti-6Al-4V, annealed, 600°C	650	0.064
Ti-6Al-4V, annealed, 800°C	350	0.146



Tensile instability - necking



Tensile instability (1-D)

$$F = \sigma A; \text{ so } dF = \sigma dA + A d\sigma = 0 \text{ at max load}$$

$$\frac{d\sigma}{\sigma} = -\frac{dA}{A} = d\epsilon$$

$$\frac{d\sigma}{d\epsilon} = \sigma$$

$$\text{With } \sigma = k\epsilon^n: \quad \frac{d\sigma}{d\epsilon} = n k \epsilon^{n-1} = \sigma = k\epsilon^n$$

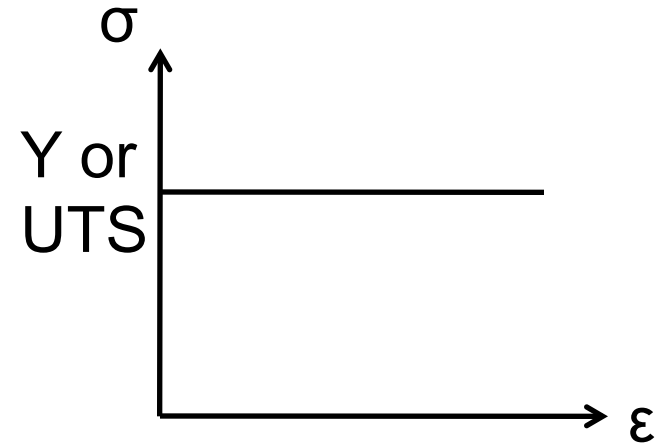
$$\Rightarrow \boxed{\epsilon^* = n}$$

Useful assumptions

Only interested in plastic effects:

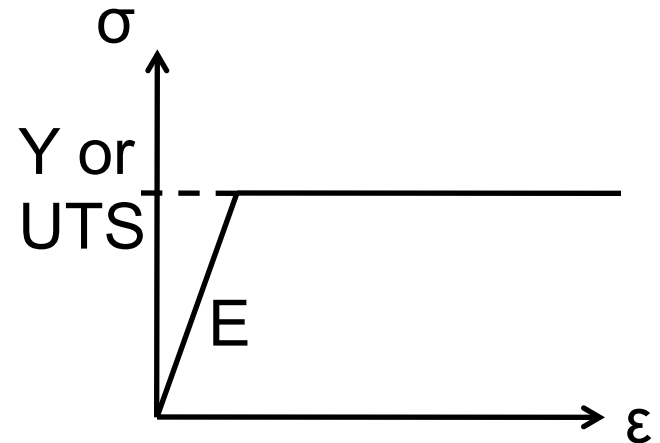
Perfectly plastic material

At Y , material deforms (‘flows’) in compression and fails in tension

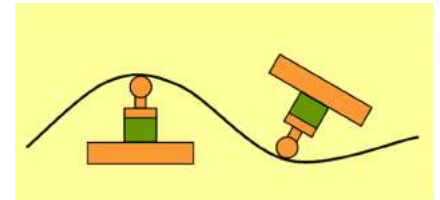


Interested in elastic and plastic effects:

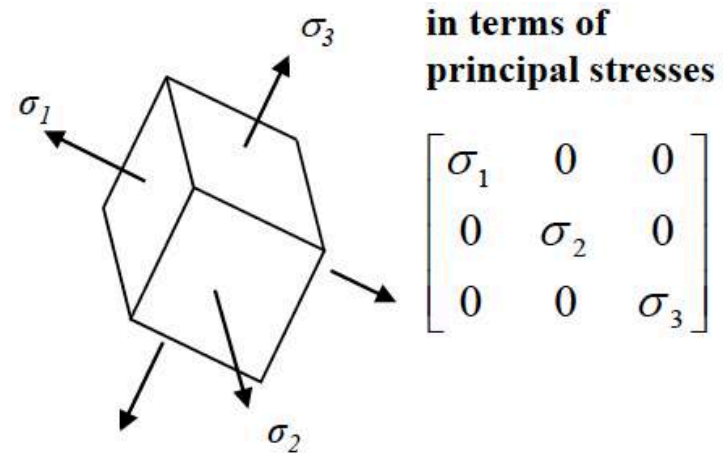
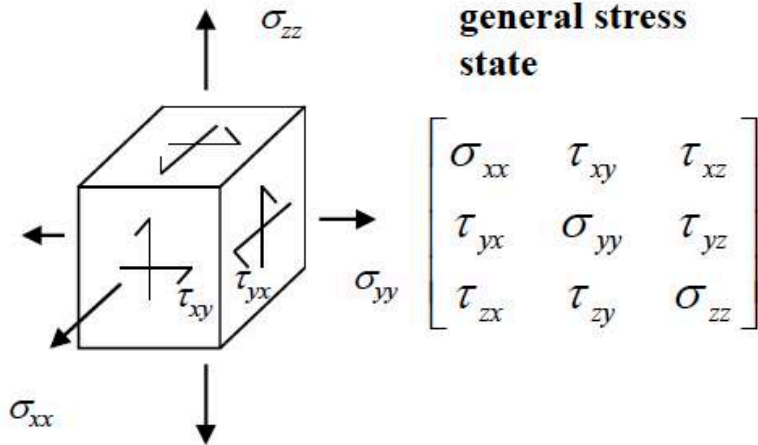
Elastic-perfectly plastic material



Massachusetts
Institute of
Technology

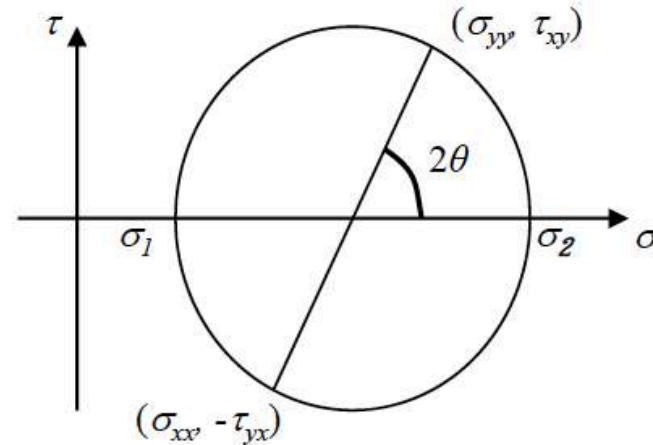


3D Problems



For any general stress state we can find a set of *principal axes*. The stress tensor for these axes contains no off-diagonal (shear) terms – only three principal stresses along the three axes.

Mohr's circle allows rotation of axes in two dimensions about one principal axis



In 1-D, $\sigma = K\varepsilon^n$ assuming perfectly plastic, yielding at: $\sigma = Y$

In 3-D, $\sigma_{eff} = K\varepsilon_{eff}^n$ assuming perfectly plastic, yielding at:

$$\sigma_{eff} = Y$$

3D Yield Criteria

Tresca: Yielding occurs at a maximum shear stress

Effective stress (in principal directions):

$$\sigma_{eff} = \left[\sigma_i - \sigma_j \right]_{\substack{\text{max,} \\ i \neq j}}$$

Yield criterion:

$$\sigma_{eff} = Y$$

$$\tau_{max} = k = \frac{Y}{2}$$

Effective strain:

$$\varepsilon_{eff} = \left(\varepsilon_i \right)_{\text{max}}$$

Von Mises: Yielding at maximum distortion strain energy

Effective stress (in principal directions):

$$\sigma_{eff} = \sqrt{\frac{1}{2} \times \left[(\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 + (\sigma_1 - \sigma_2)^2 \right]}$$

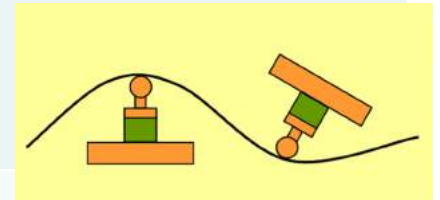
Yield criterion:

$$\sigma_{eff} = Y$$

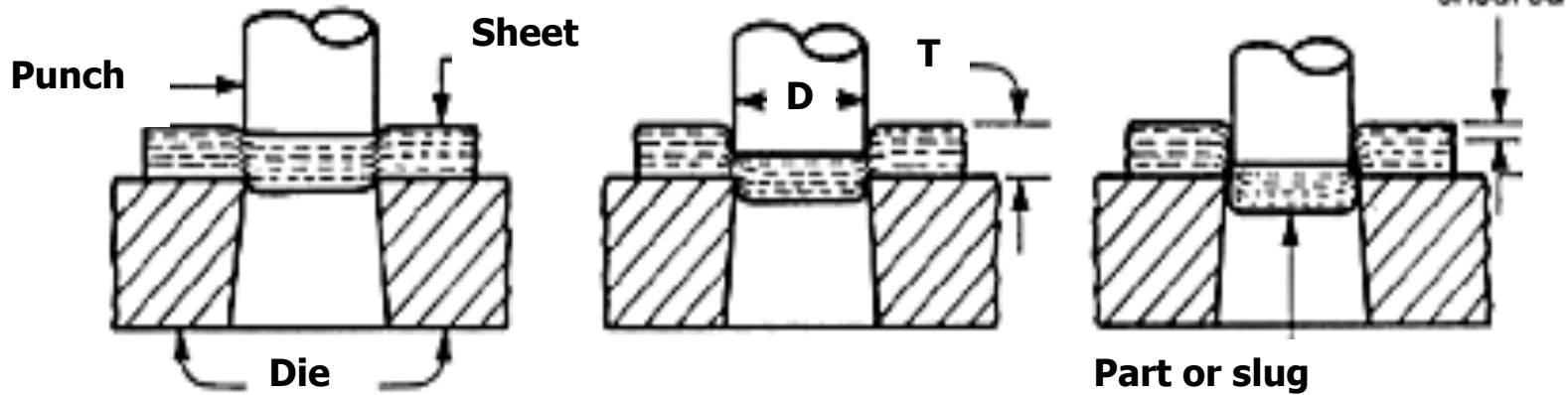
$$Y = \sqrt{3}k$$

Effective strain:

$$\varepsilon_{eff} = \sqrt{\left(\frac{2}{3} \right) \times \left(\varepsilon_1^2 + \varepsilon_2^2 + \varepsilon_3^2 \right)}$$



Shearing



$$F = 0.7 T L (\text{UTS})$$

T = Sheet Thickness

L = Total length Sheared

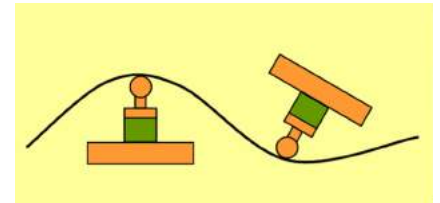
UTS = Ultimate Tensile Strength
of material



Shear press - LMP Shop



Massachusetts
Institute of
Technology



Side Note: For a general state of stress use “effective stress”

2-6 EFFECTIVE STRESS

With either yield criterion, it is useful to define an effective stress denoted as $\bar{\sigma}$ which is a function of the applied stresses. If the *magnitude* of $\bar{\sigma}$ reaches a critical value, then the applied stress state will cause yielding; in essence, it has reached an effective level. For the von Mises criterion,

$$\bar{\sigma} = \frac{1}{\sqrt{2}} [(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2]^{1/2} \quad (2-16)$$

while for the Tresca criterion,

$$\bar{\sigma} = \sigma_1 - \sigma_3 \quad \text{where} \quad \sigma_1 > \sigma_2 > \sigma_3 \quad (2-17)$$

Yielding occurs when $\sigma_{\text{effective}} = Y$

Origin of effective strain

2-7 EFFECTIVE STRAIN

Effective strain is *defined* such that the incremental work per unit volume is

$$dw = \bar{\sigma} d\bar{\epsilon} = \sigma_1 d\epsilon_1 + \sigma_2 d\epsilon_2 + \sigma_3 d\epsilon_3 \quad (2-18)$$

For the von Mises criterion, the effective strain is given by

$$d\bar{\epsilon} = \frac{\sqrt{2}}{3} [(d\epsilon_1 - d\epsilon_2)^2 + (d\epsilon_2 - d\epsilon_3)^2 + (d\epsilon_3 - d\epsilon_1)^2]^{1/2} \quad (2-19)$$

which may be expressed in a simpler form as

$$d\bar{\epsilon} = \left[\frac{2}{3} (d\epsilon_1^2 + d\epsilon_2^2 + d\epsilon_3^2) \right]^{1/2} \quad (2-20)$$

If the straining is proportional (with a constant ratio of $d\epsilon_1 : d\epsilon_2 : d\epsilon_3$), the total effective strain may be expressed in terms of the total strains as

$$\bar{\epsilon} = \left[\frac{2}{3} (\epsilon_1^2 + \epsilon_2^2 + \epsilon_3^2) \right]^{1/2} \quad (2-21)$$

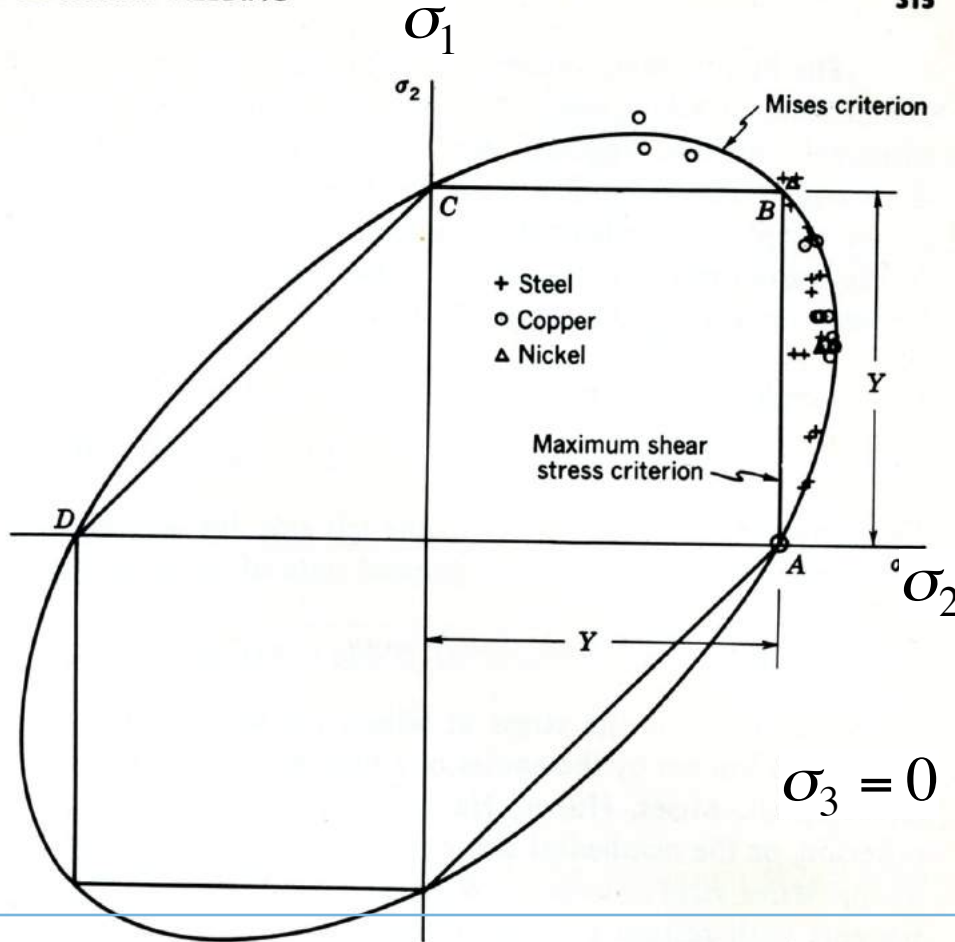
If the strain path is not constant, $\bar{\epsilon}$ must be found from a path integral of $d\bar{\epsilon}$. In

$$\bar{\sigma} = K \bar{\epsilon}^n$$

3D Yield Effective stress

SEC. 5.11 CRITERIA FOR INITIAL YIELDING

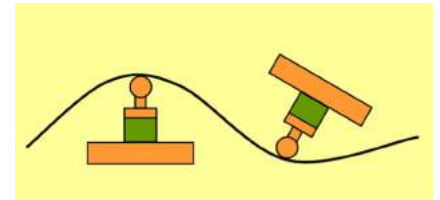
315



Tresca predicts 'flow' for lower stresses than von Mises



Massachusetts
Institute of
Technology



Forming Limit Diagrams

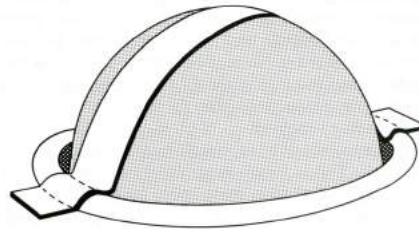
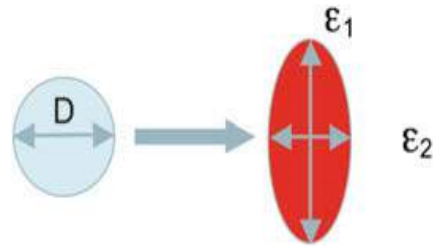
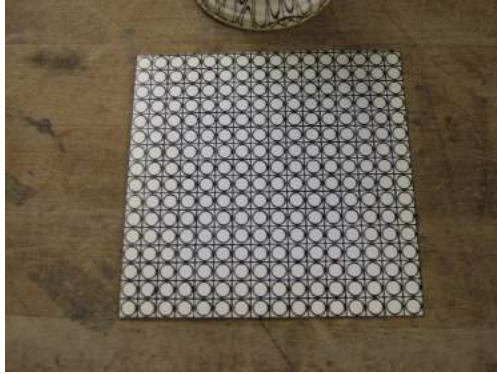
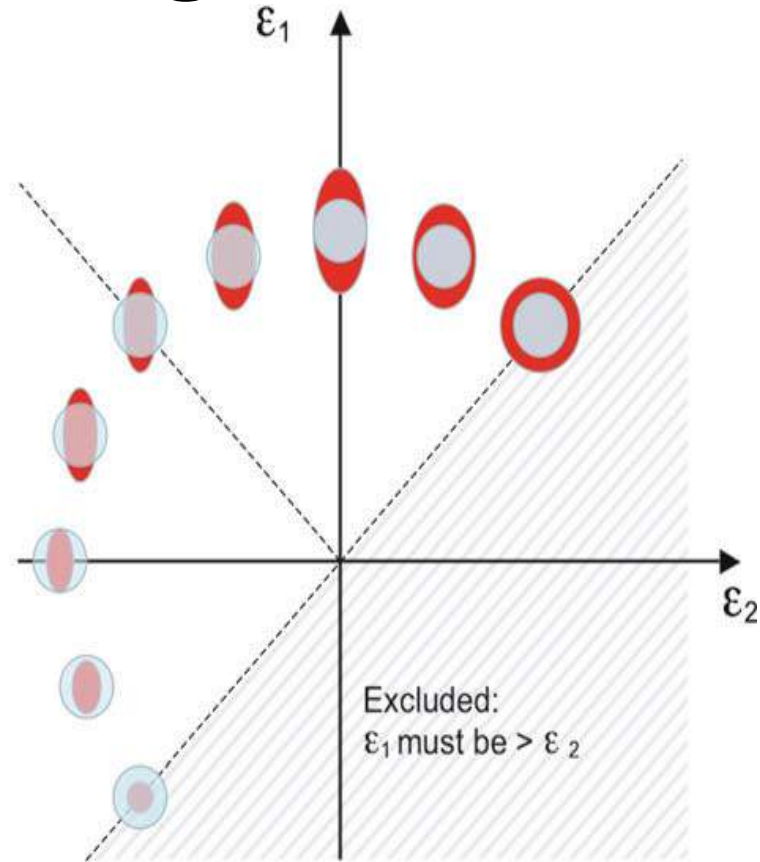
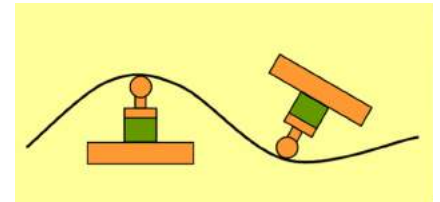
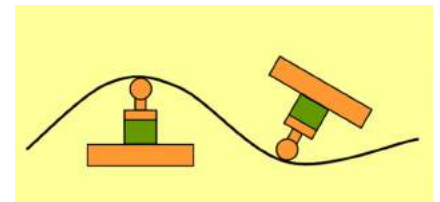
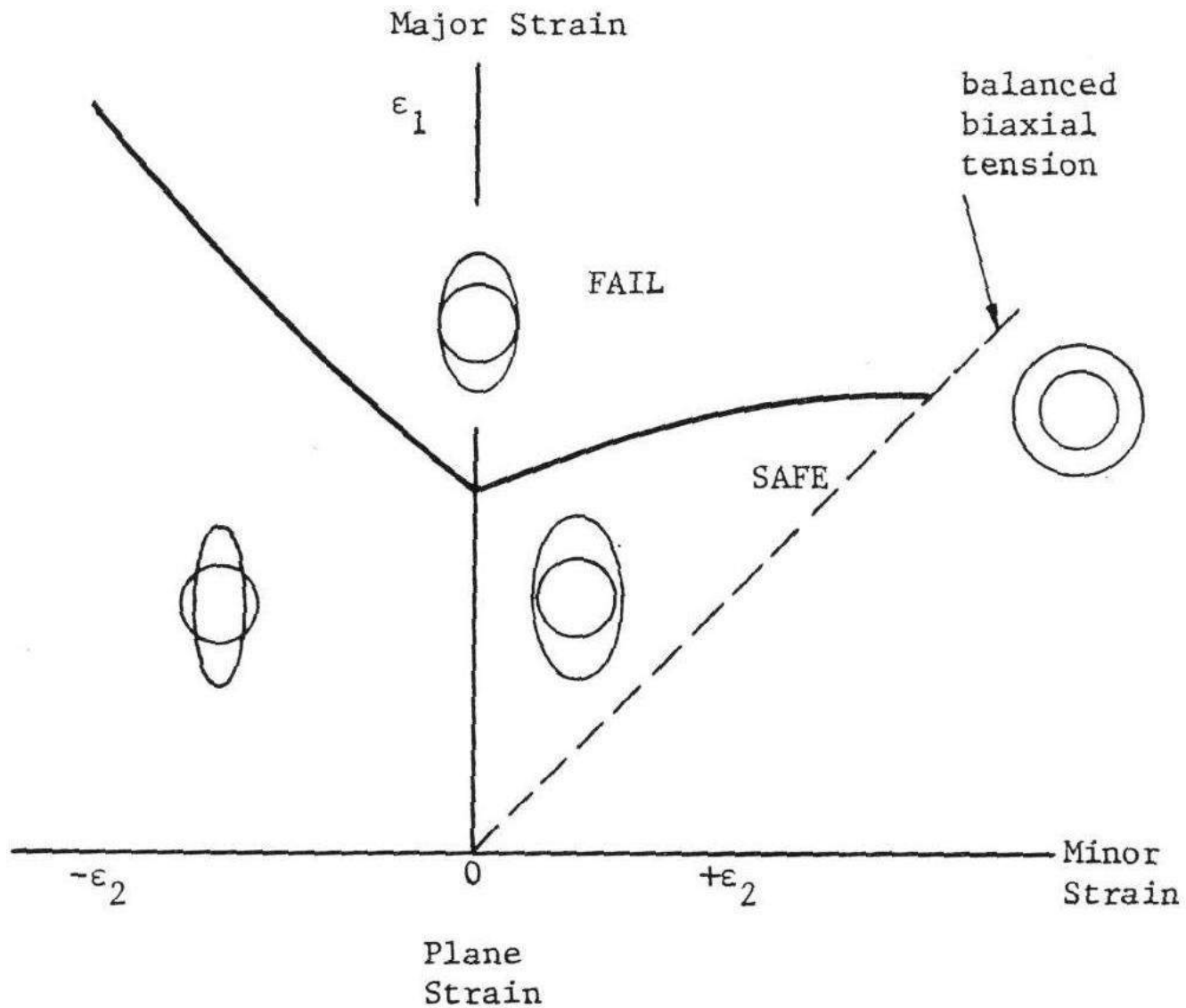


Figure 15-8 Strips of varying width are stretched to obtain different e_2/e_1 ratios.



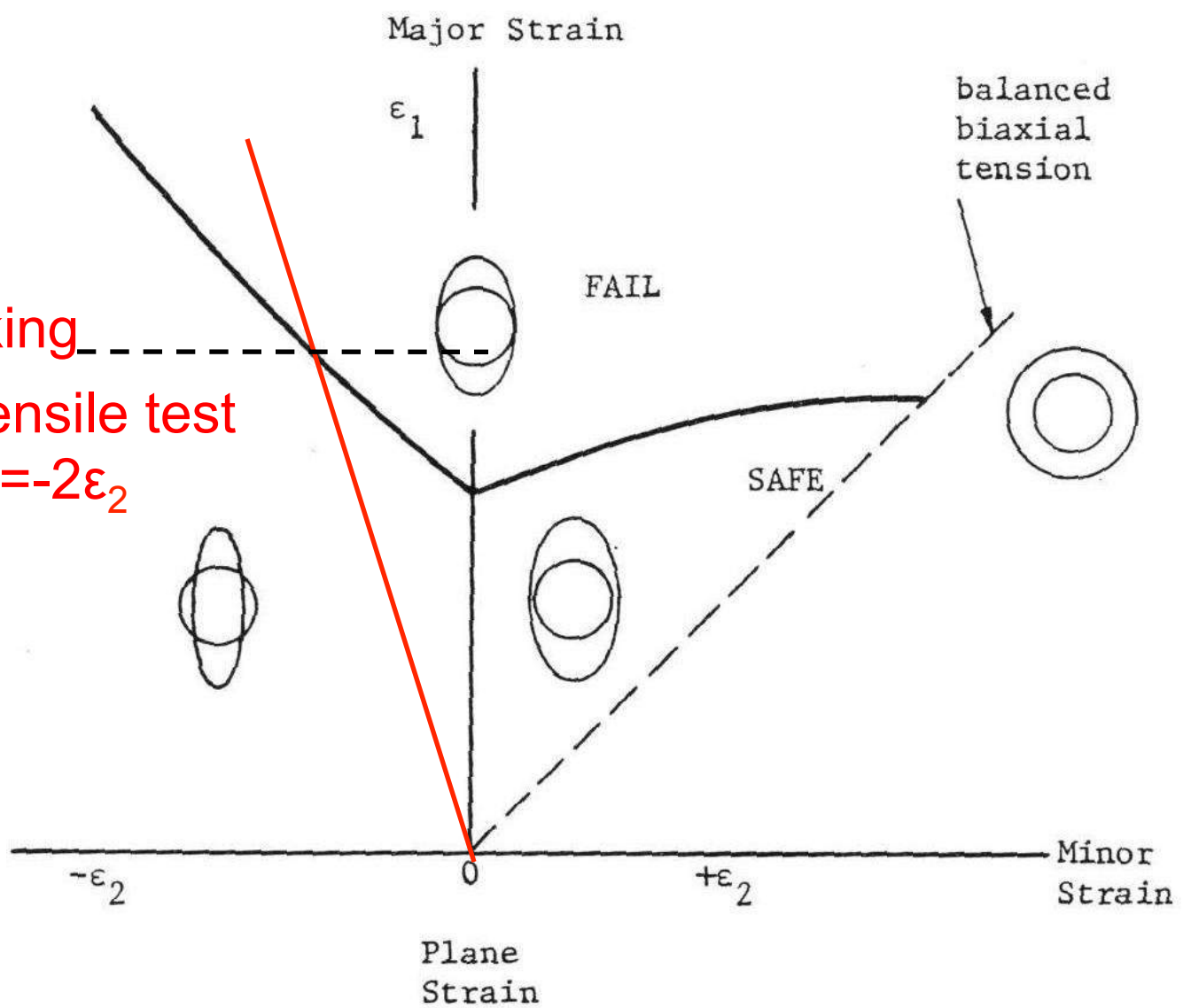
Massachusetts
Institute of
Technology



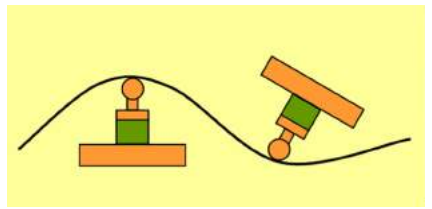


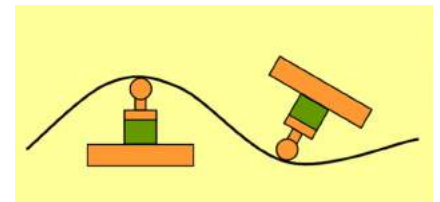
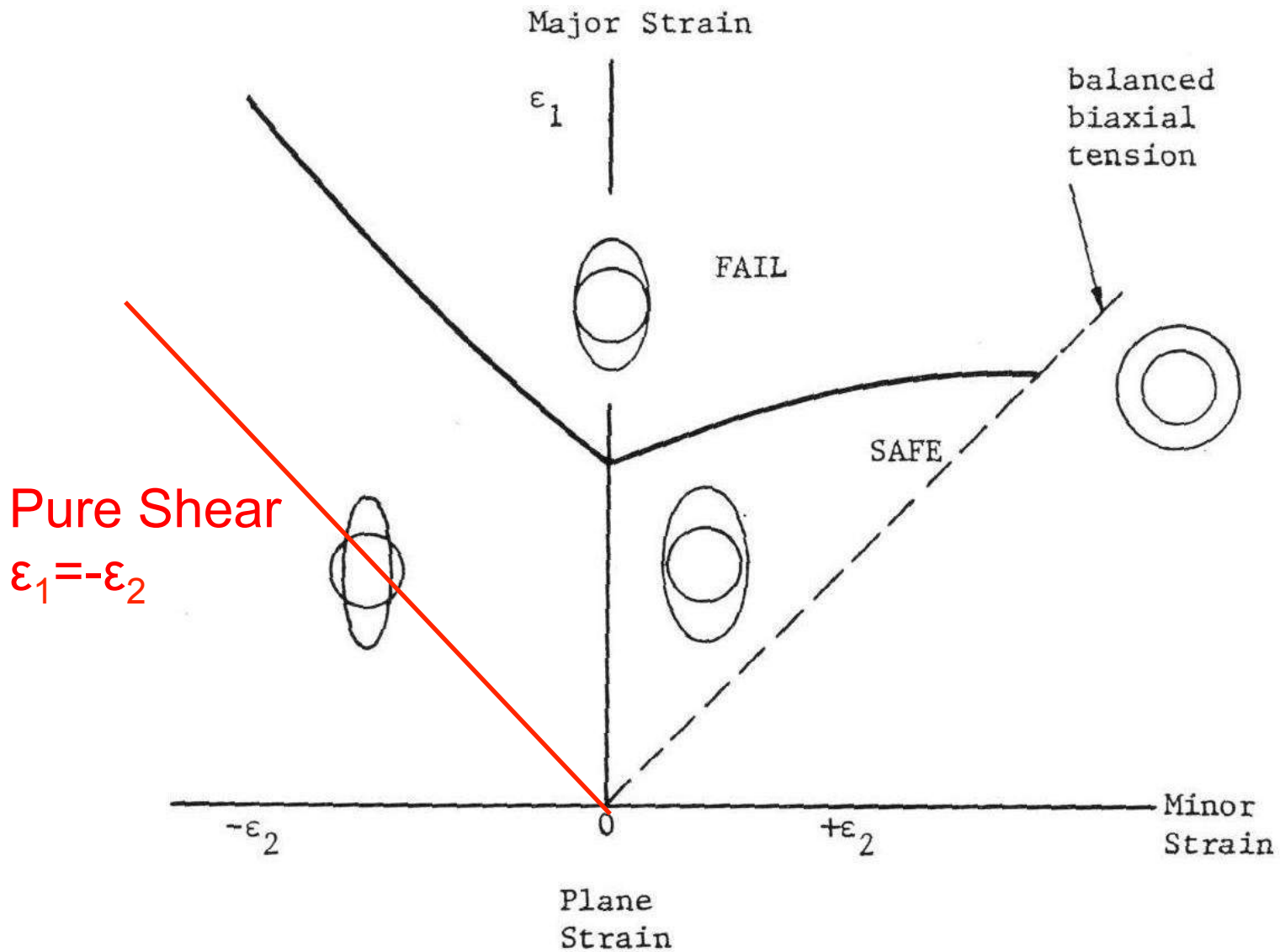
$\epsilon_1 = n = \text{necking}$

Tensile test
 $\epsilon_1 = -2\epsilon_2$

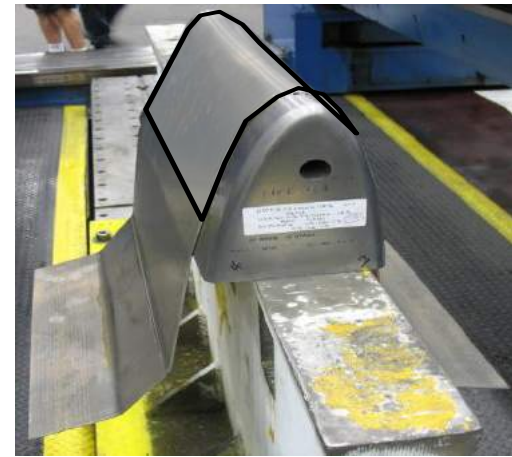


Massachusetts
Institute of
Technology





Stretch forming: **Forming force**



$$F = (Y_s + UTS)/2 * A$$

F = stretch forming force (lbs)

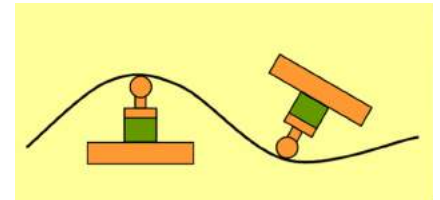
Y_s = material yield strength (psi)

UTS = ultimate tensile strength of the material (psi)

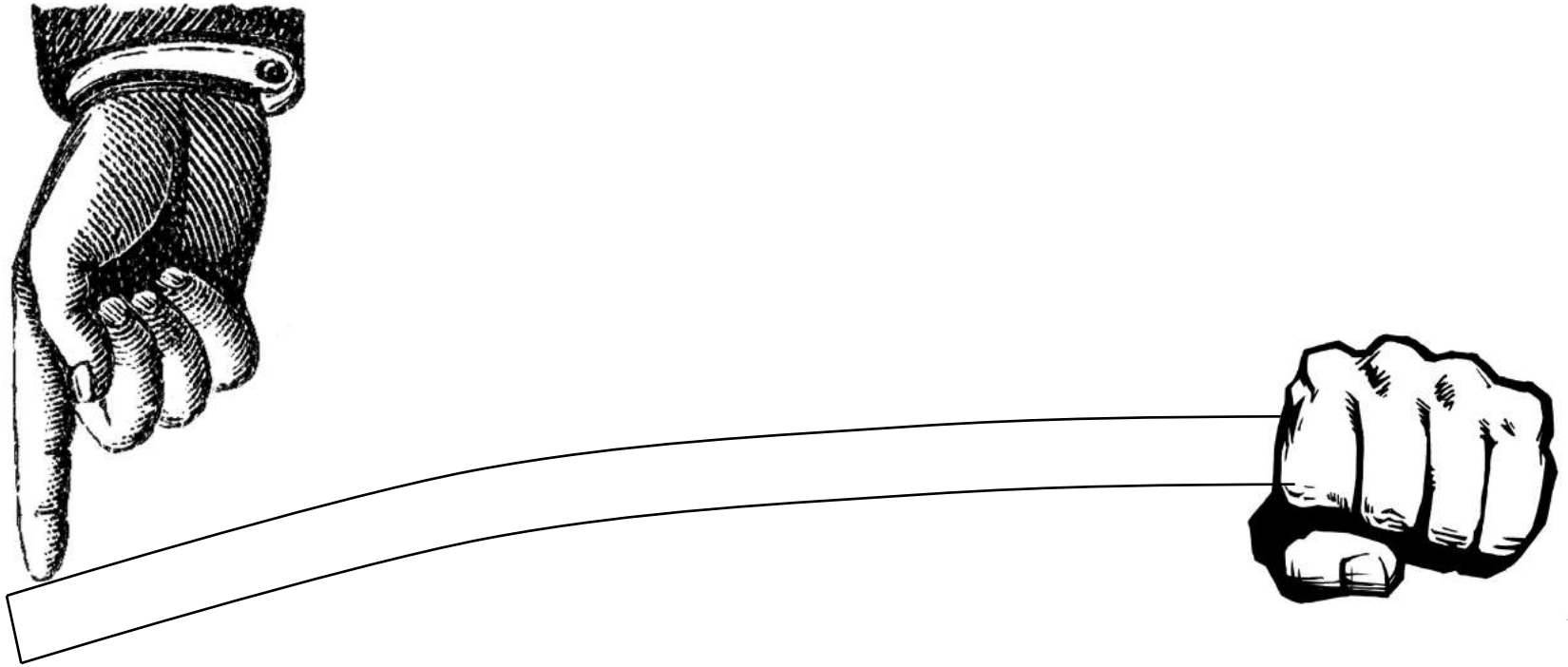
A = Cross-sectional area of the workpiece (in²)



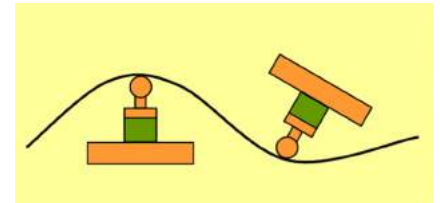
**Massachusetts
Institute of
Technology**



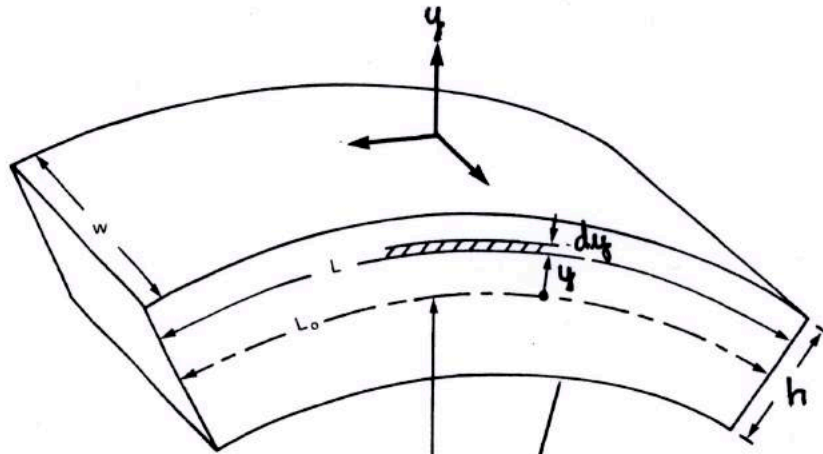
Forces needed to bend sheet metal



Massachusetts
Institute of
Technology



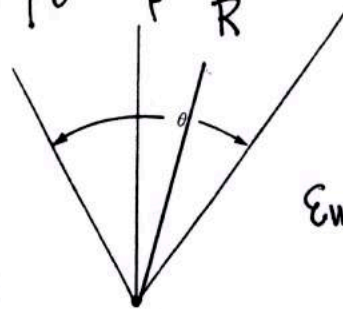
Bending



note: $L > L_0$

$$\Delta L = (L - L_0) = (R + y)\theta - R\theta = y\theta$$

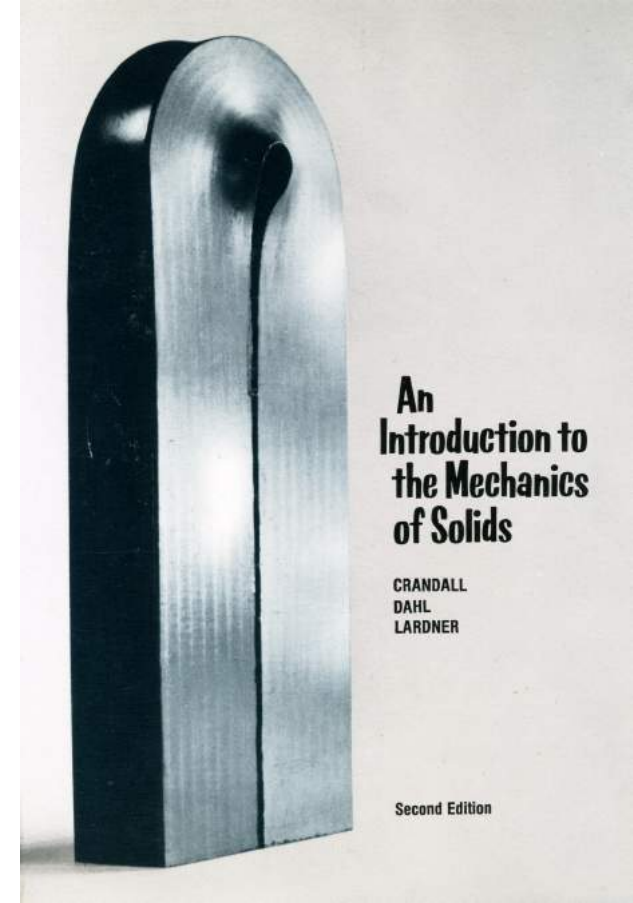
$$\epsilon = \frac{\Delta L}{L_0} = \frac{y\theta}{R\theta} = \frac{y}{R}$$



if $R = R + \frac{h}{2}$

$$\epsilon_{max} = \frac{h/2}{R + h/2} = \frac{1}{\frac{2R}{h} + 1}$$

Figure Coordinate system for analysis of bending.



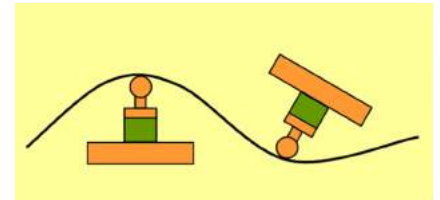
An Introduction to the Mechanics of Solids

CRANDALL
DAHL
LARDNER

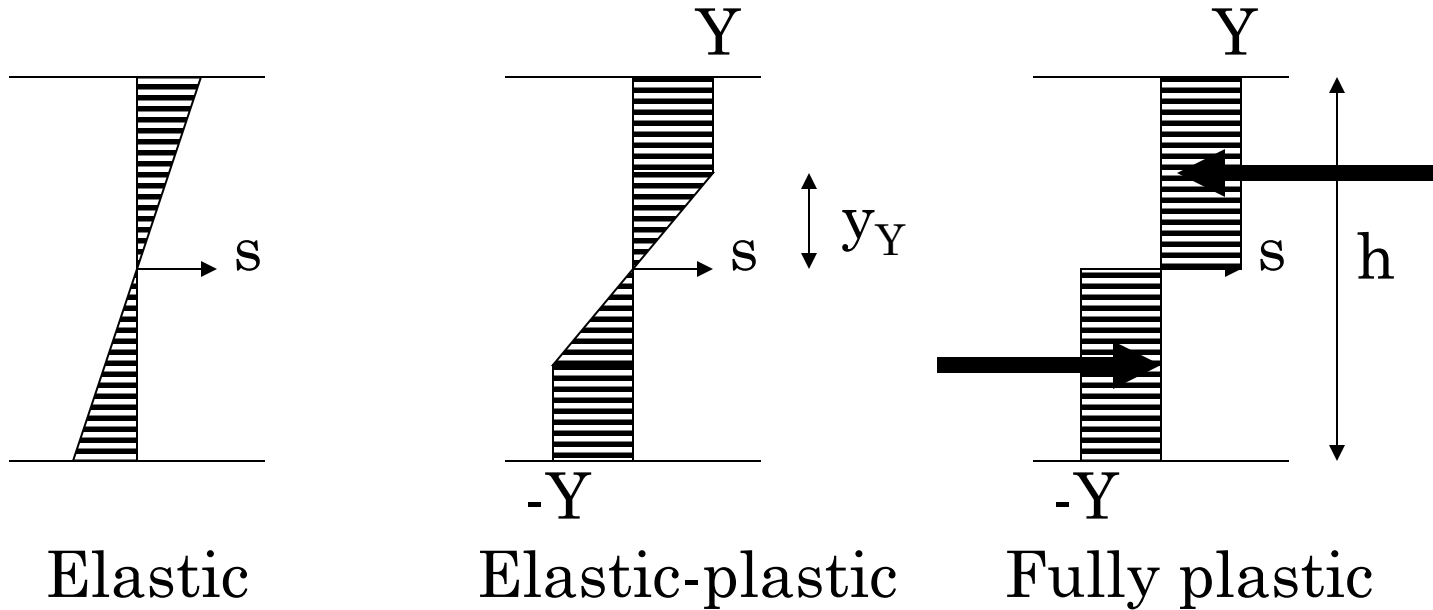
Second Edition



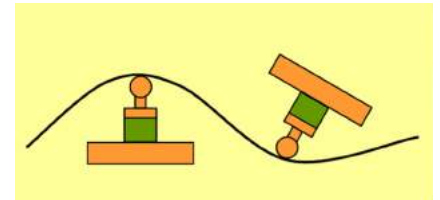
Massachusetts
Institute of
Technology



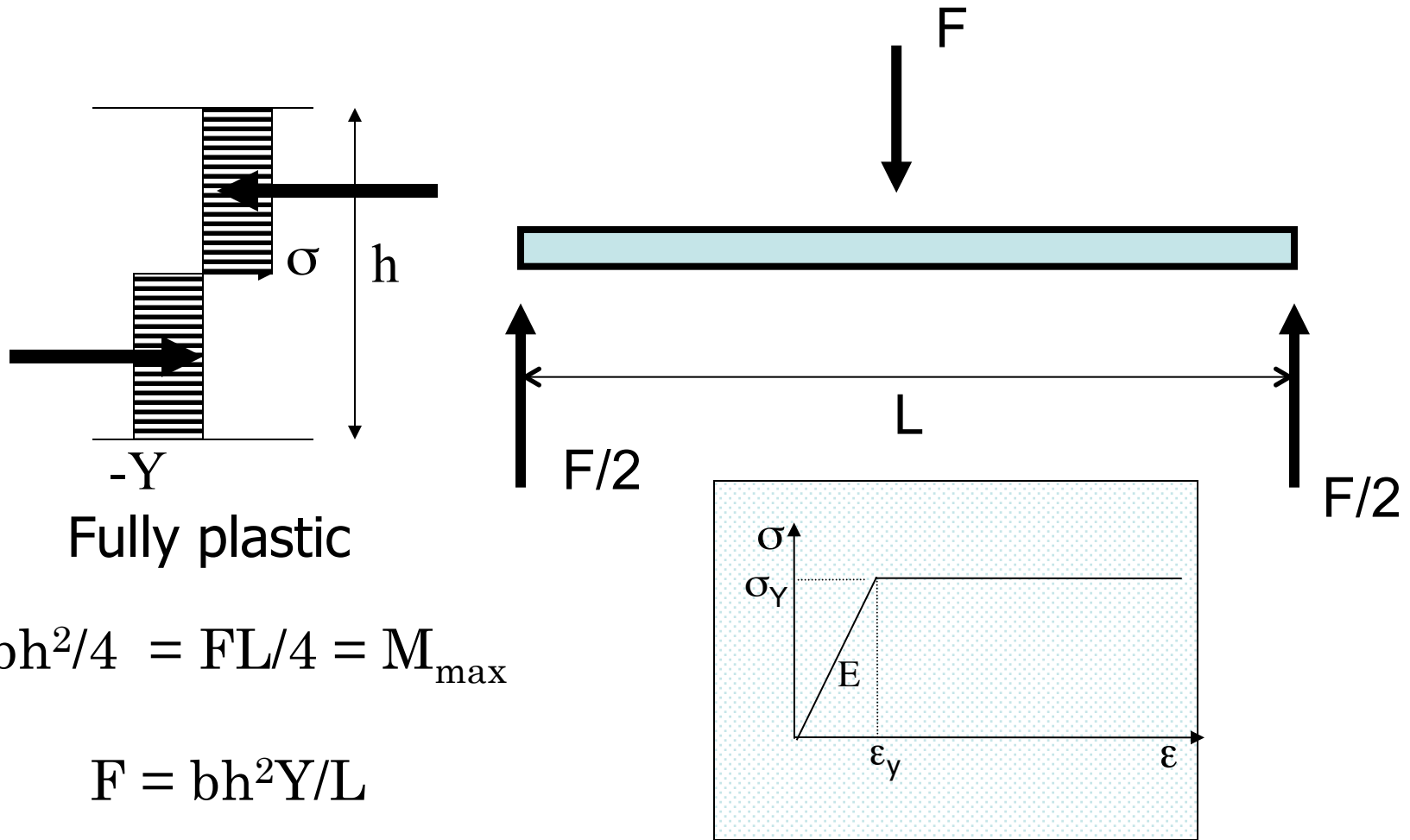
Stress distribution through the thickness of the part



Fully Plastic Moment, $M = Y (b h/2) h/2 = Ybh^2/4$

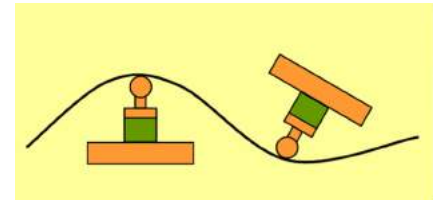


Balance external and internal moments

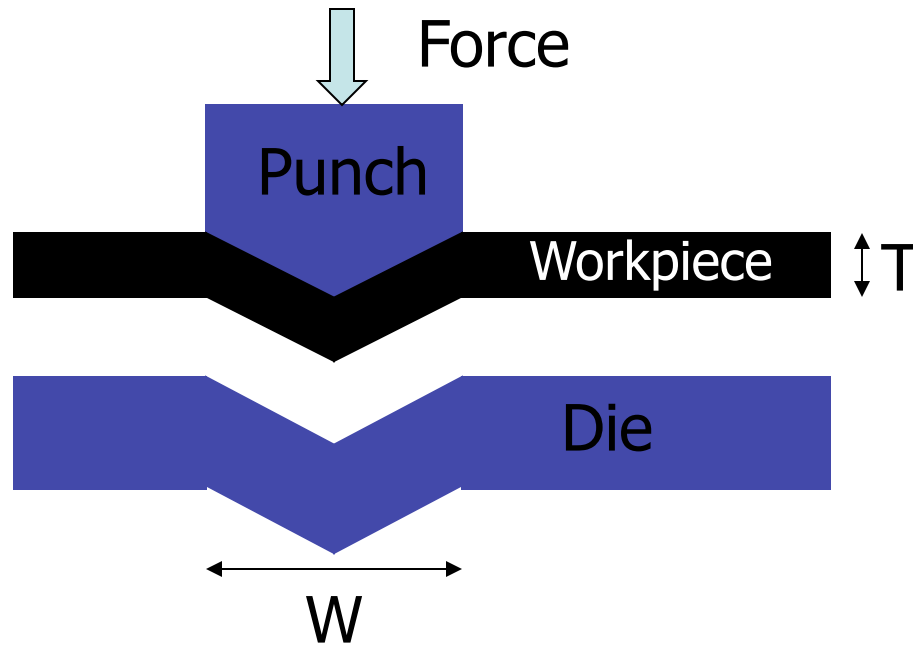


$$Ybh^2/4 = FL/4 = M_{\max}$$

$$F = bh^2Y/L$$



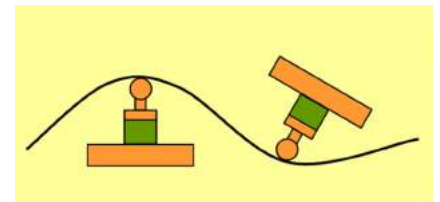
Bending Force Requirement



$$F = \frac{LT^2}{W}(UTS)$$

T = Sheet Thickness
W = Width of Die Opening
L = Total length of bend
(into the page)
UTS = Ultimate Tensile
Strength of material

Note: the notation used in the text (L, W) differs from that used in the previous development (b, L).

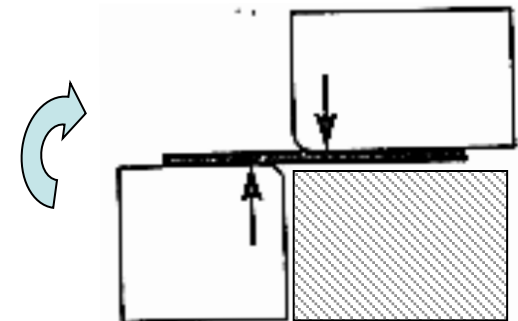
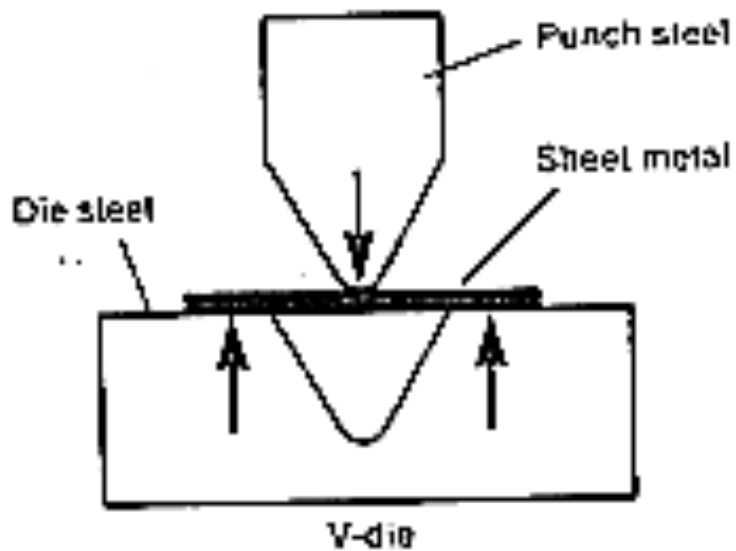


LMP Shop

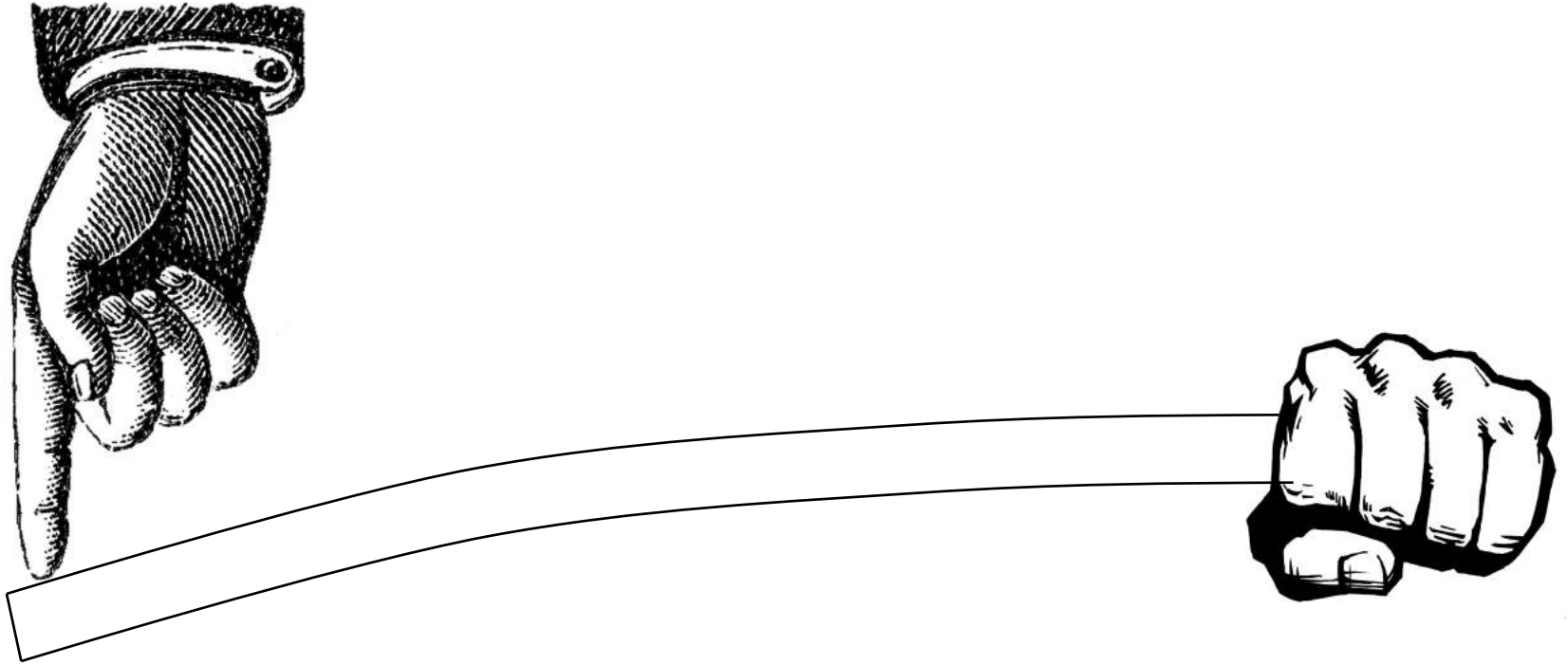
Brake press



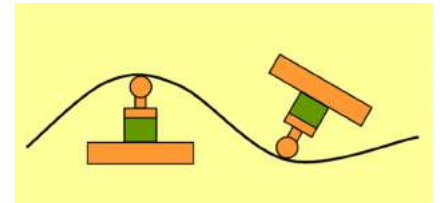
Finger brake



What shape have we created?

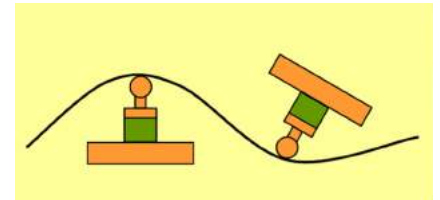
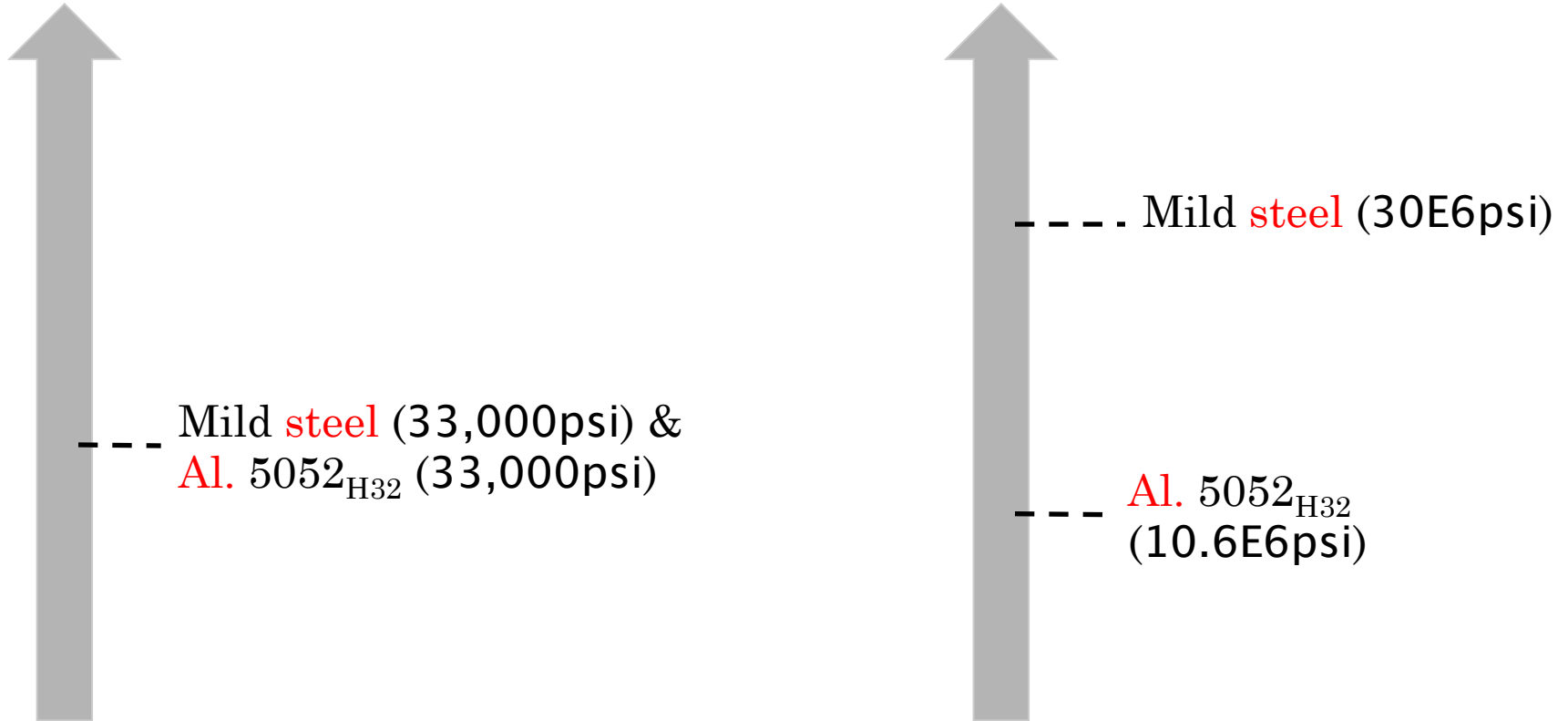


Massachusetts
Institute of
Technology



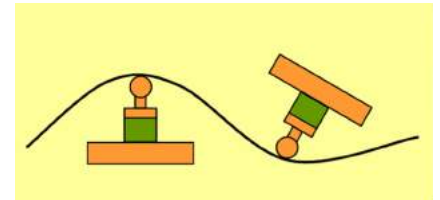
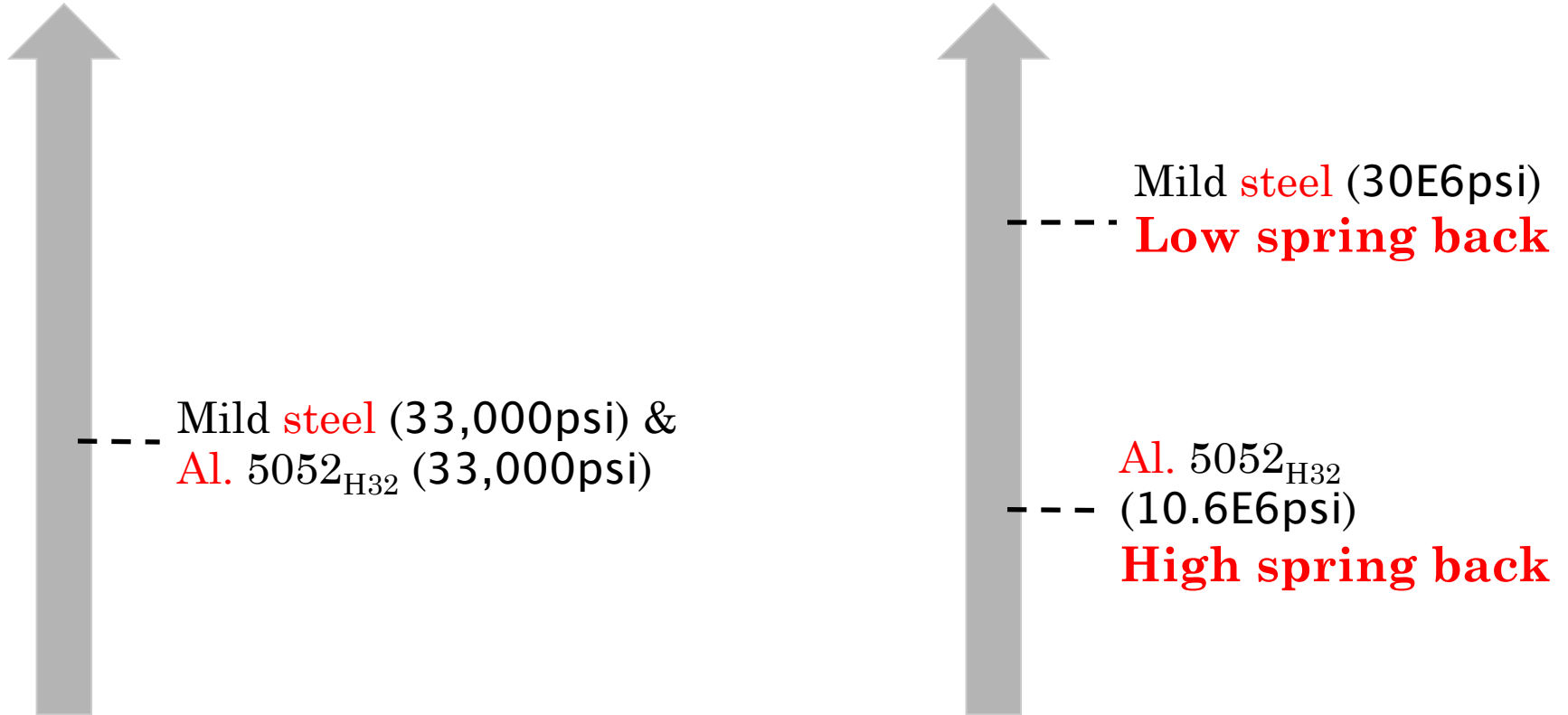
Steel versus aluminum...

Strength (σ_y) versus Stiffness (E)



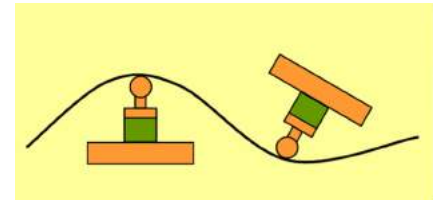
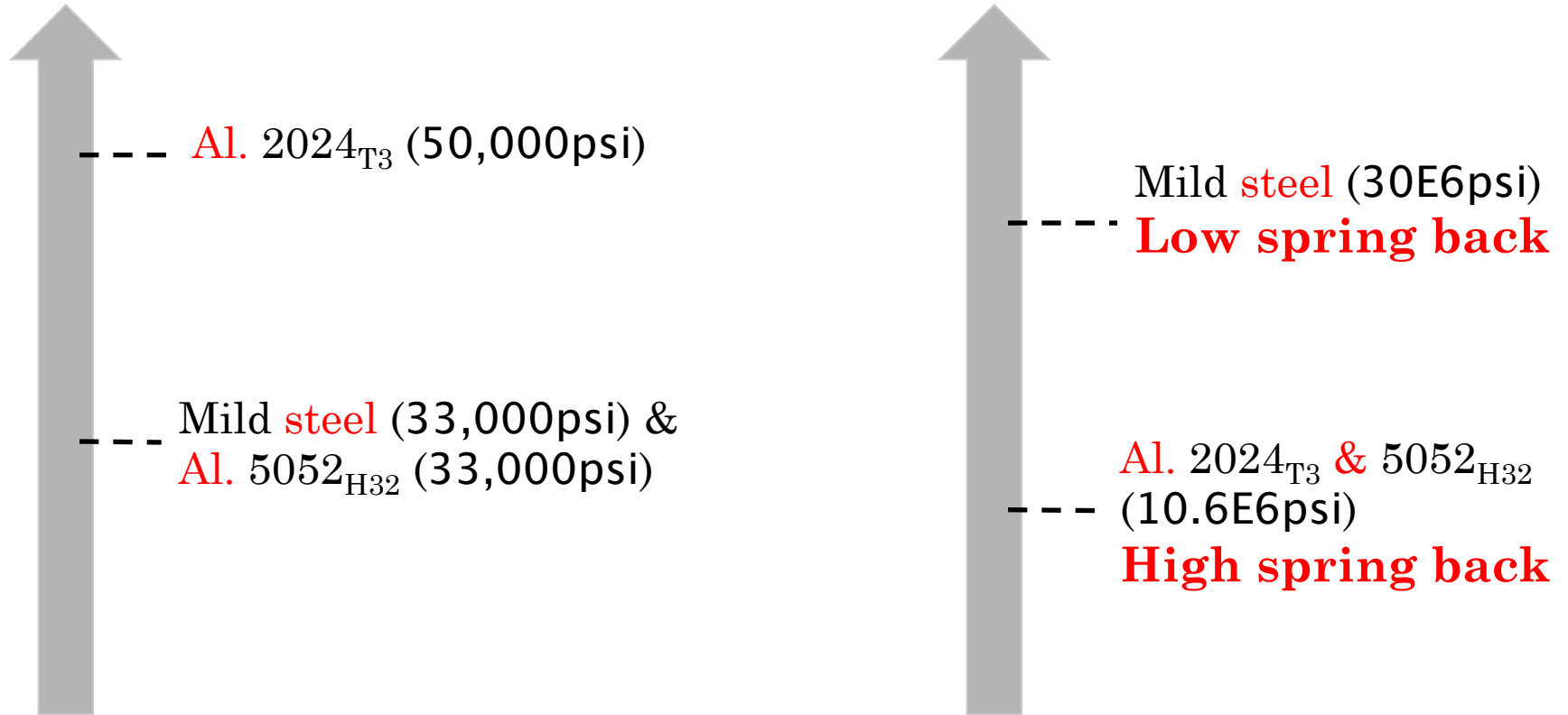
Steel versus aluminum...

Strength (σ_y) versus Stiffness (E)



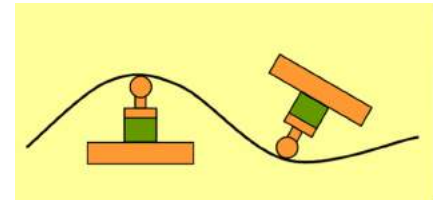
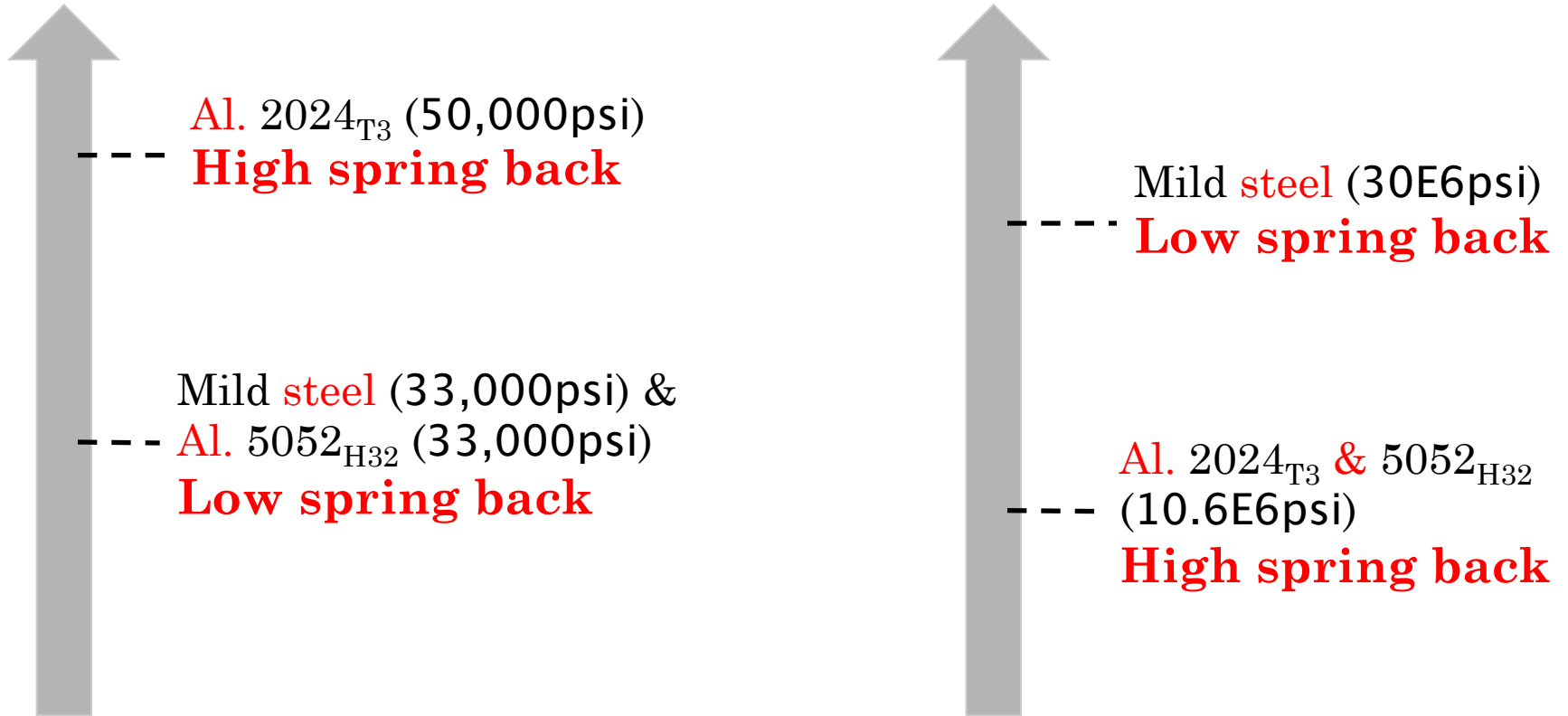
Steel versus aluminum...

Strength (σ_y) versus Stiffness (E)

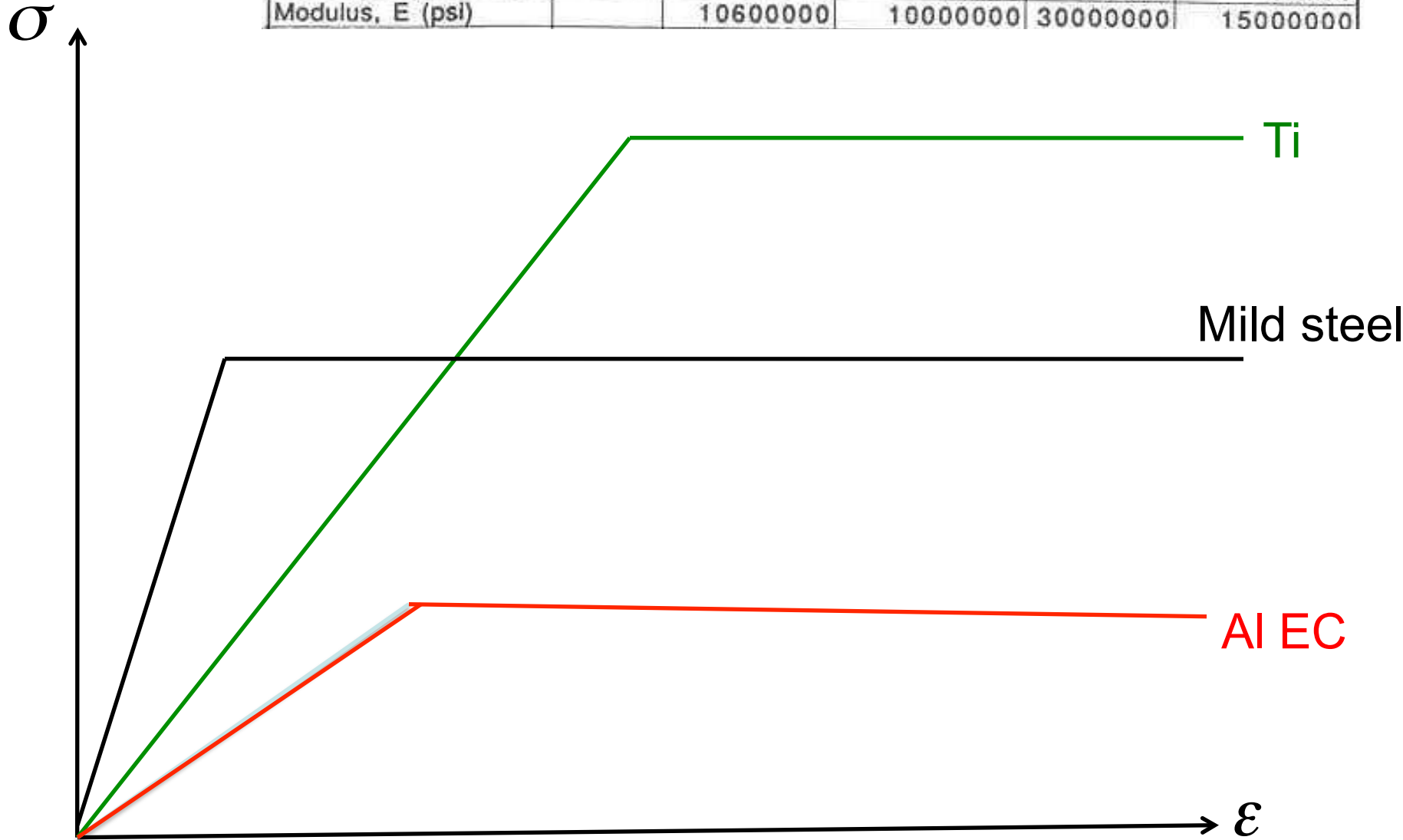


Steel versus aluminum...

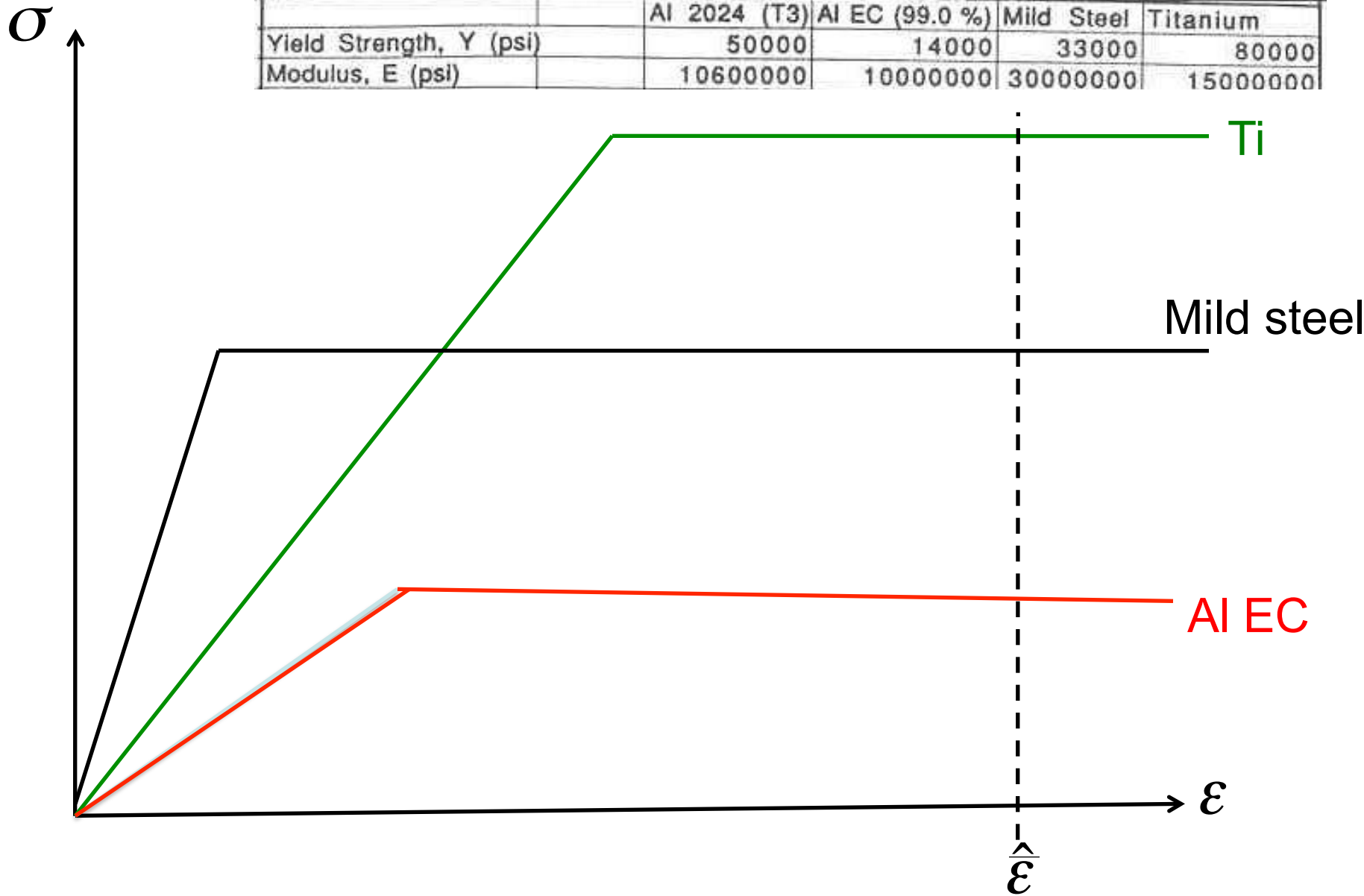
Strength (σ_y) versus Stiffness (E)



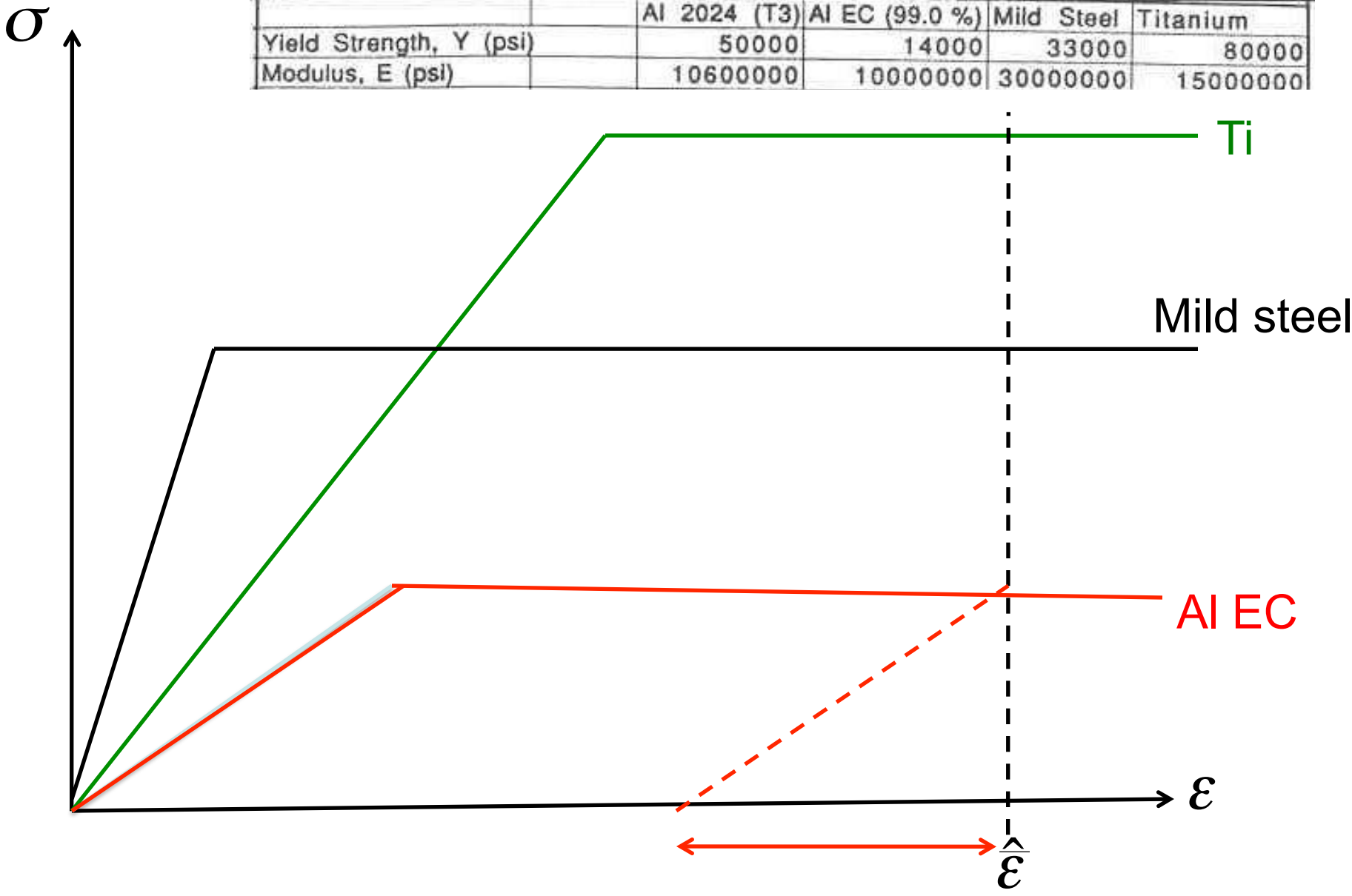
	Al 2024 (T3)	Al EC (99.0 %)	Mild Steel	Titanium
Yield Strength, Y (psi)	50000	14000	33000	80000
Modulus, E (psi)	10600000	10000000	30000000	15000000



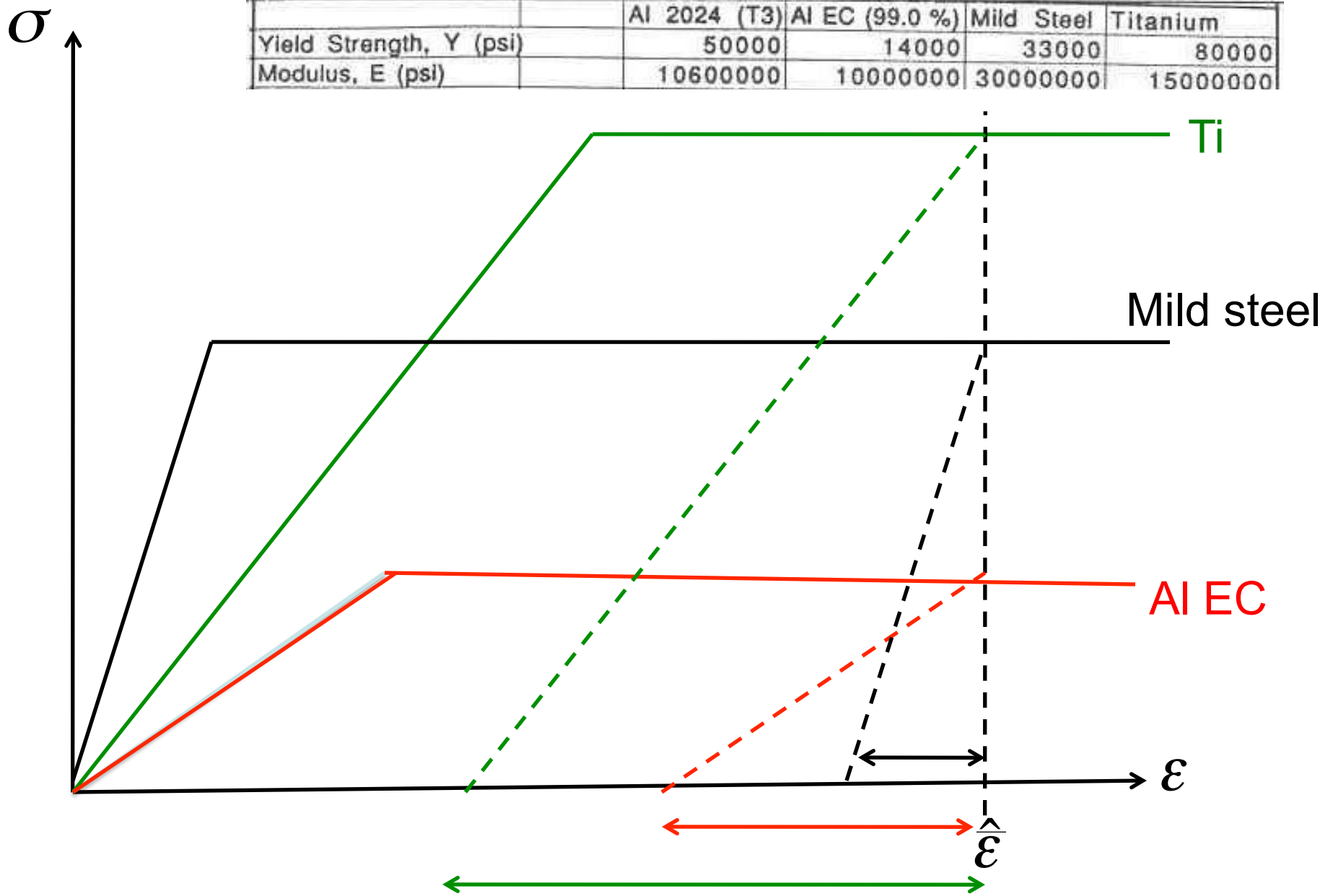
	Al 2024 (T3)	Al EC (99.0 %)	Mild Steel	Titanium
Yield Strength, Y (psi)	50000	14000	33000	80000
Modulus, E (psi)	10600000	10000000	30000000	15000000



	Al 2024 (T3)	Al EC (99.0 %)	Mild Steel	Titanium
Yield Strength, Y (psi)	50000	14000	33000	80000
Modulus, E (psi)	10600000	10000000	30000000	15000000

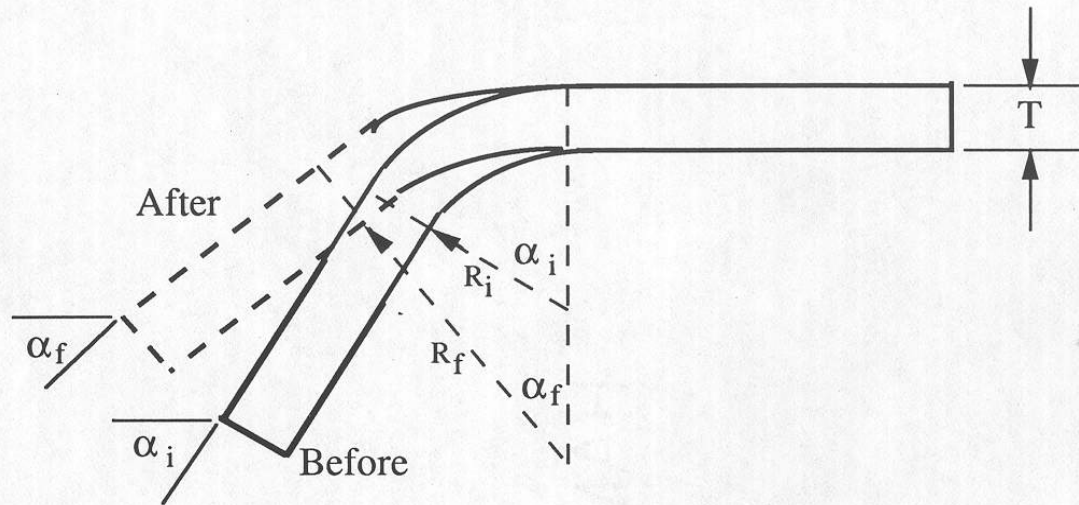


	Al 2024 (T3)	Al EC (99.0 %)	Mild Steel	Titanium
Yield Strength, Y (psi)	50000	14000	33000	80000
Modulus, E (psi)	10600000	10000000	30000000	15000000

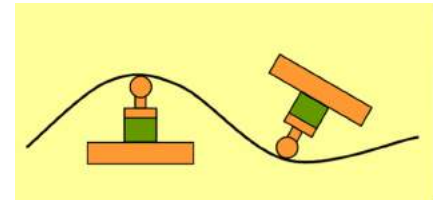


Springback

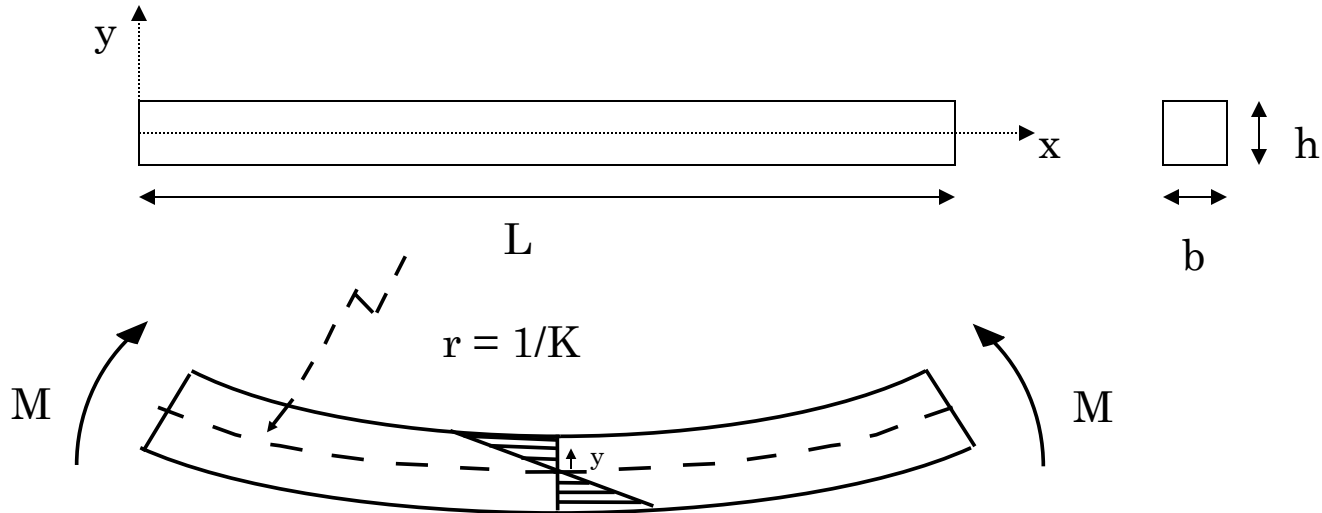
note R in the figure below is mislabeled, should go to the centerline of the sheet



Springback:
$$\frac{R_i}{R_f} = 4 \left(\frac{R_i Y}{ET} \right)^3 - 3 \left(\frac{R_i Y}{ET} \right) + 1$$



Elastic Springback Analysis

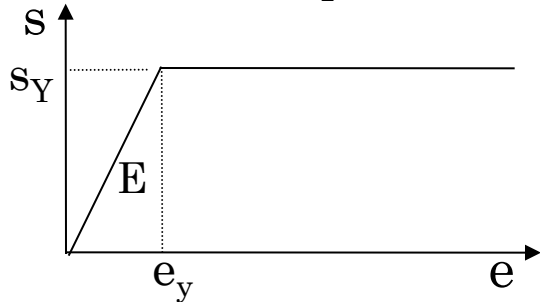


1. Assume plane sections remain plane:

$$e_y = -y/r$$

(1)

2. Assume elastic-plastic behavior for material



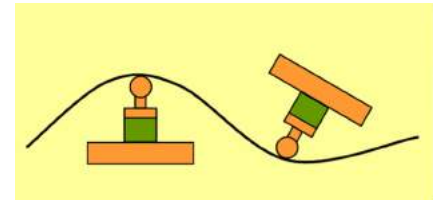
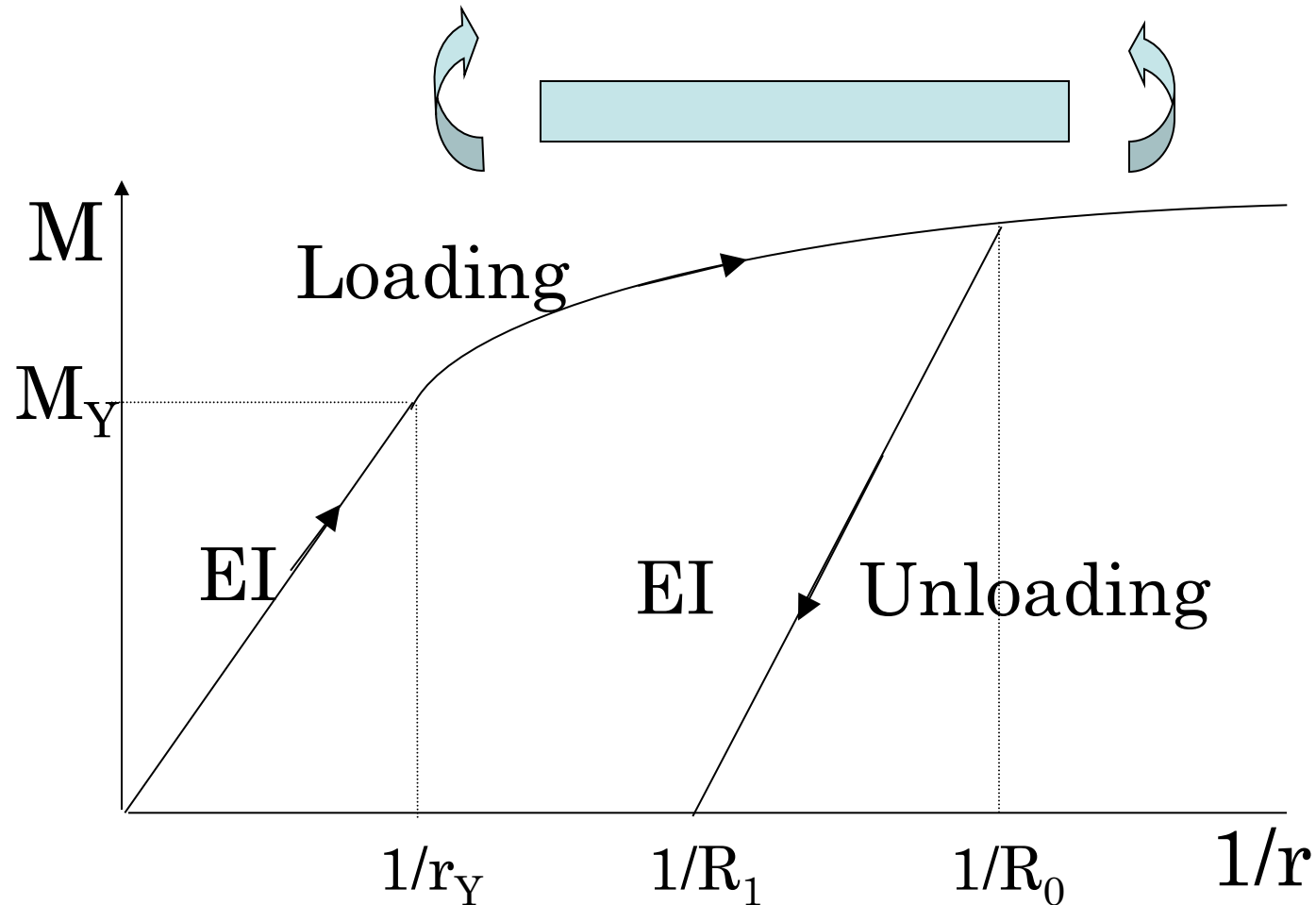
$$\sigma = E e$$

$$e < e_y$$

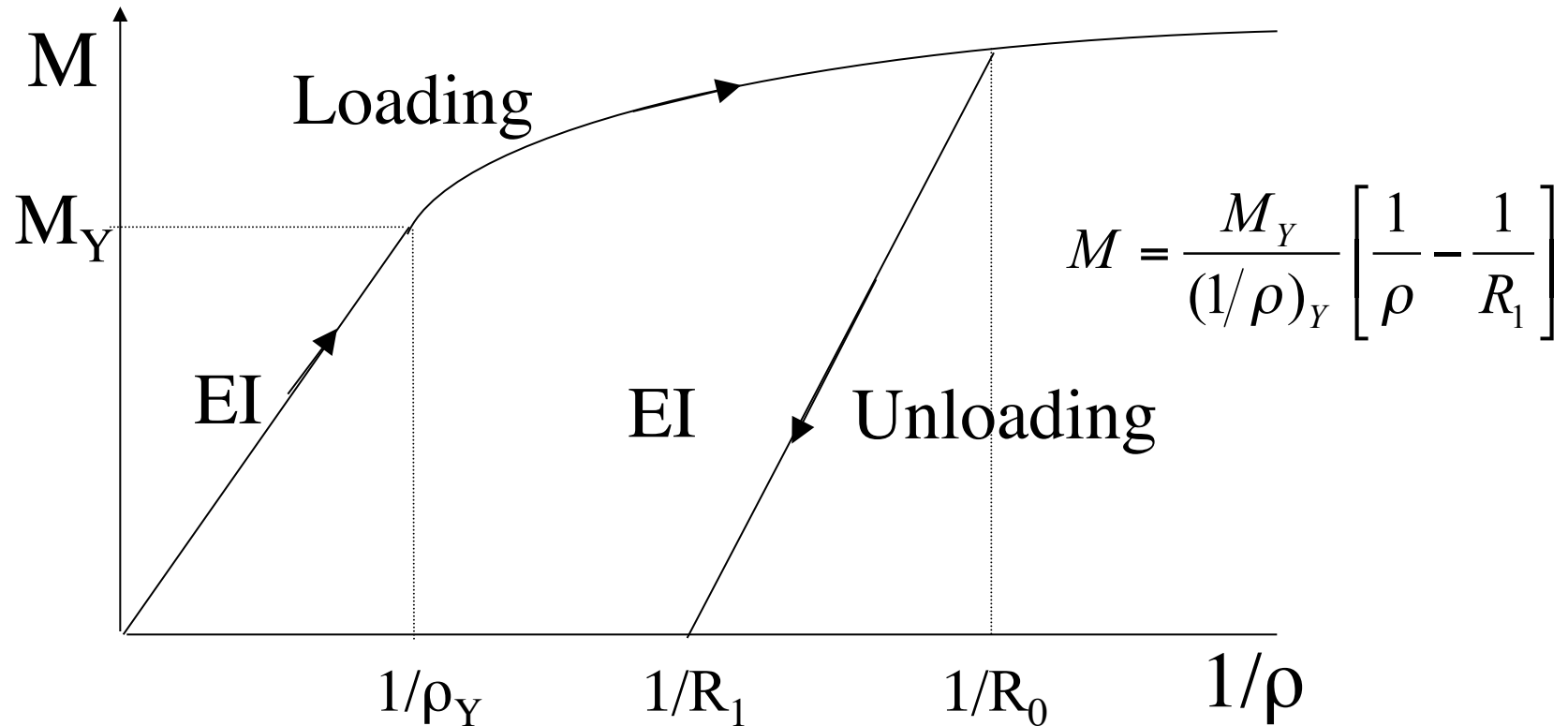
$$\sigma = \sigma_Y$$

$$e \geq e_y$$

Bending Moment – Curvature



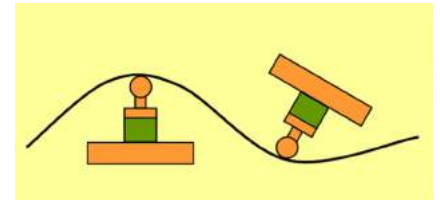
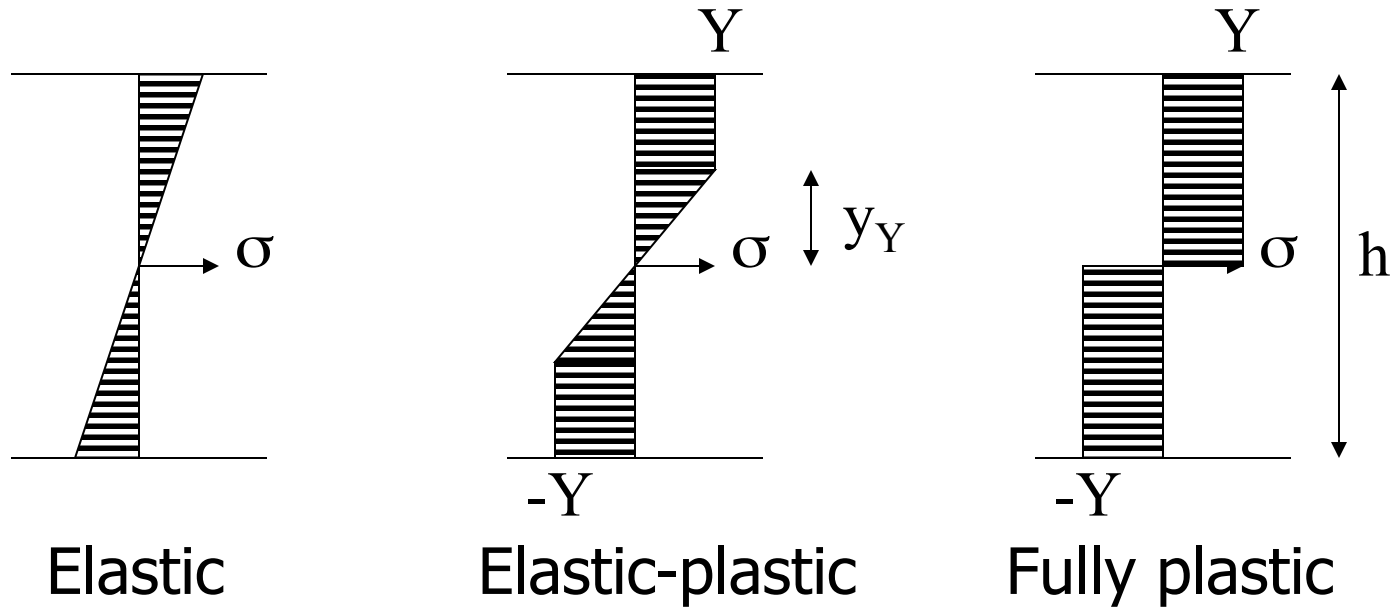
3. We want to construct the following
Bending Moment “M” vs. curvature “1/ρ” curve



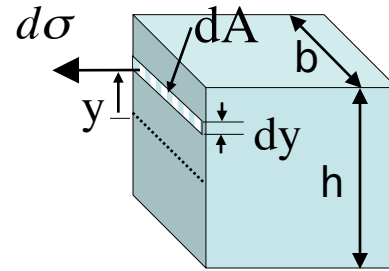
Springback is measured as
Permanent set is

$$\frac{1/R_0 - 1/R_1}{1/R_1} \quad (2)$$

4. Stress distribution through the thickness of the beam



$$5. M = \int_A \sigma y dA$$



Elastic region

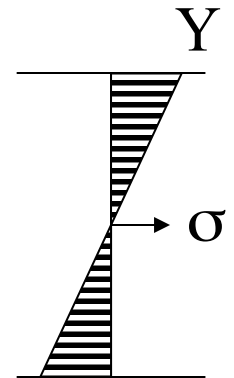
$$M = \int \sigma y dA = -E \int \frac{y^2}{\rho} dA = -\frac{EI}{\rho} \quad (3)$$

At the onset of plastic behavior

$$\sigma = -y/\rho E = -h/2\rho E = -Y \quad (4)$$

This occurs at

$$1/\rho = 2Y/hE = 1/\rho_Y \quad (5)$$

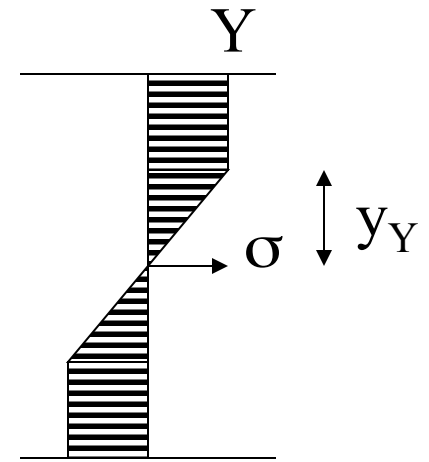


Substitution into eqn (3) gives us the moment at on-set of yield, M_Y

$$M_Y = -EI/\rho_Y = EI 2Y/hE = 2IY/h \quad (6)$$

After this point, the M vs $1/r$ curve starts to “bend over.”
Note from $M=0$ to $M=M_Y$ the curve is linear.

In the elastic – plastic region



$$M = \int \sigma y b dy = 2 \int_{y_Y}^{h/2} Y b y dy + 2 \int_0^{y_Y} \frac{y}{y_Y} Y b y dy$$

$$= 2Yb \frac{y^2}{2} \Big|_{y_Y}^{h/2} + 2 \frac{Y}{y_Y} b \frac{y^3}{3} \Big|_0^{y_Y}$$

$$= Yb \left(\frac{h^2}{4} - y_Y^2 \right) + \frac{2}{3} y_Y^2 Yb$$

$$M = \frac{bh^2}{4} Y \left[1 - \frac{1}{3} \left(\frac{y_Y}{h/2} \right)^2 \right] \quad (7)$$

Note at $y_Y = h/2$, you get on-set at yield, $M = M_Y$

And at $y_Y = 0$, you get fully plastic moment, $M = 3/2 M_Y$

To write this in terms of M vs $1/\rho$ rather than M vs y_Y , note that the yield curvature $(1/\rho)_Y$ can be written as (see eqn (1))

$$\frac{1}{\rho_Y} = \frac{\varepsilon_Y}{h/2} \quad (8)$$

Where ε_Y is the strain at yield. Also since the strain at y_Y is $-\varepsilon_Y$, we can write

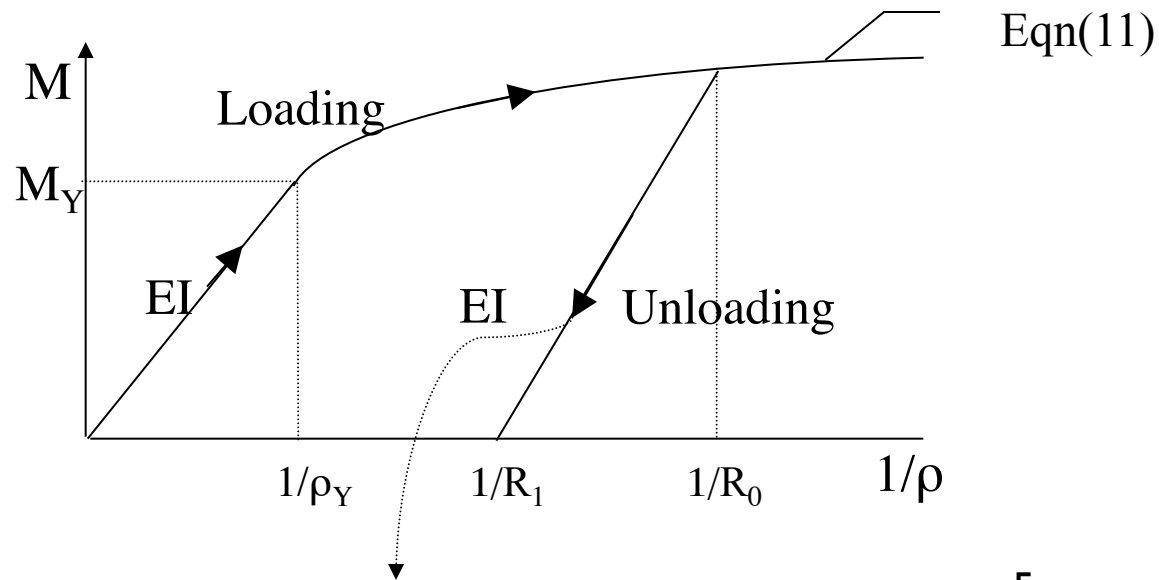
$$\frac{1}{\rho} = \frac{\varepsilon_Y}{y_Y} \quad (9)$$

Combining (8) and (9) gives

$$\frac{y_Y}{h/2} = \frac{(1/\rho)_Y}{1/\rho} \quad (10)$$

Substitution into (7) gives the result we seek:

$$M = \frac{3}{2} M_Y \left[1 - \frac{1}{3} \left(\frac{(1/\rho)_Y}{1/\rho} \right)^2 \right] \quad (11)$$



Elastic unloading curve

$$M = \frac{M_Y}{(1/\rho)_Y} \left[\frac{1}{\rho} - \frac{1}{R_1} \right] \quad (12)$$

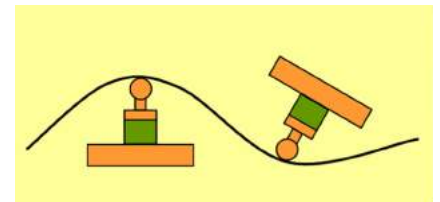
Now, eqn' s (11) and (12) intersect at $1/\rho = 1/R_0$

Hence,

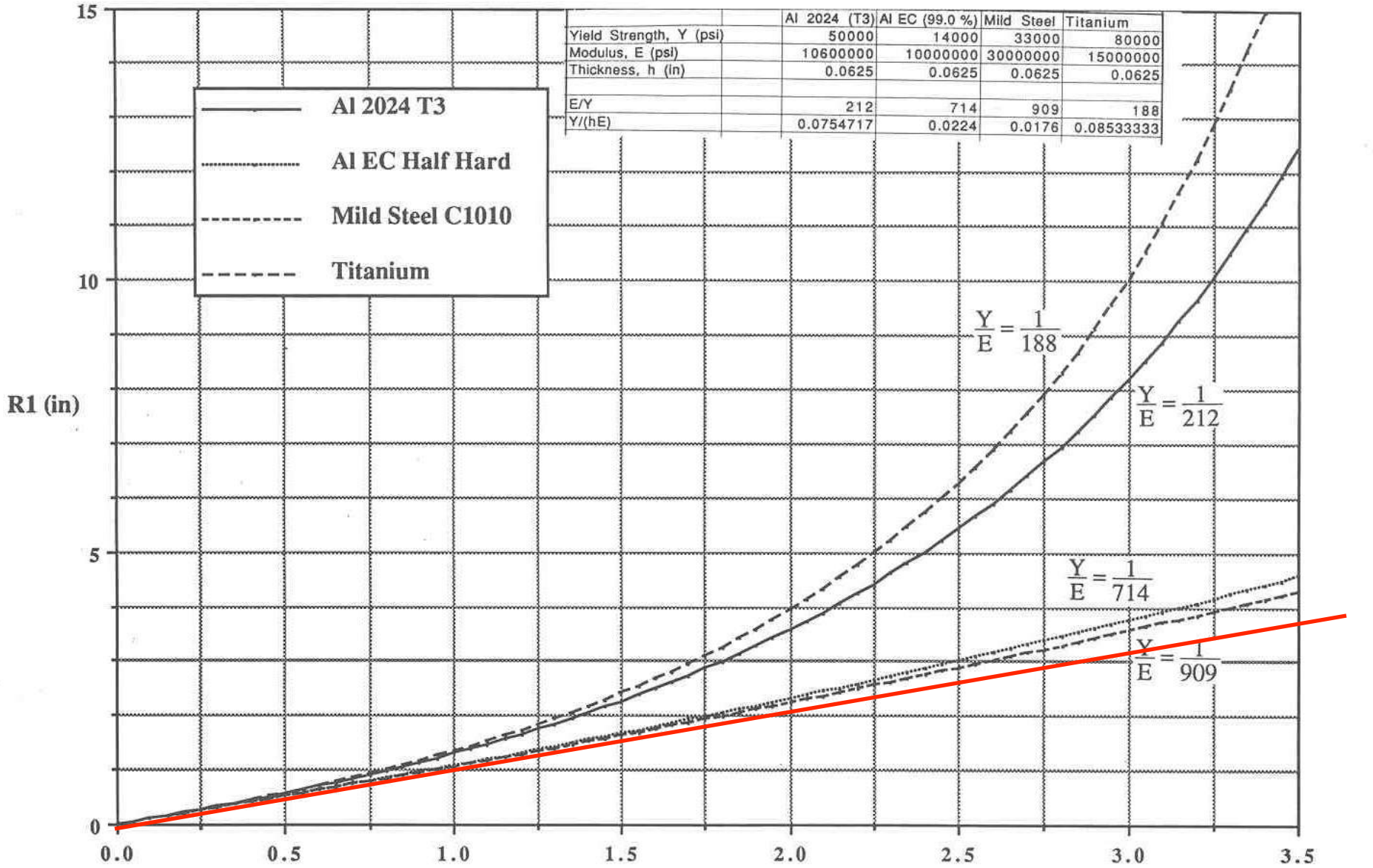
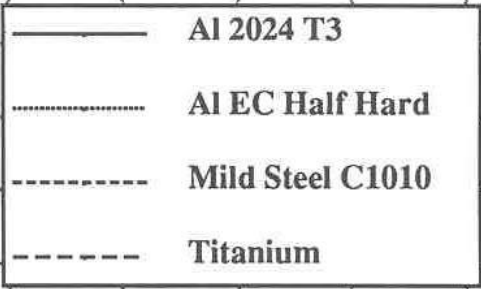
$$\frac{M_Y}{(1/\rho)_Y} \left[\frac{1}{R_0} - \frac{1}{R_1} \right] = \frac{3}{2} M_Y \left[1 - \frac{1}{3} \left(\frac{(1/\rho)_Y}{1/R_0} \right)^2 \right]$$

Rewriting and using $(1/\rho)_Y = 2Y / hE$ (from a few slides back), we get

$$\left[\frac{1}{R_0} - \frac{1}{R_1} \right] = 3 \frac{Y}{hE} - 4R_0^2 \left(\frac{Y}{hE} \right)^3 \quad (13)$$



	Al 2024 (T3)	Al EC (99.0 %)	Mild Steel	Titanium
Yield Strength, Y (psi)	50000	14000	33000	80000
Modulus, E (psi)	10600000	10000000	30000000	15000000
Thickness, h (in)	0.0625	0.0625	0.0625	0.0625
E/Y	212	714	909	188
Y/(hE)	0.0754717	0.0224	0.0176	0.08533333



$$\frac{Y}{E} = \frac{1}{188}$$

$$\frac{Y}{E} = \frac{1}{212}$$

$$\frac{Y}{E} = \frac{1}{714}$$

$$\frac{Y}{E} = \frac{1}{909}$$

R0=R1

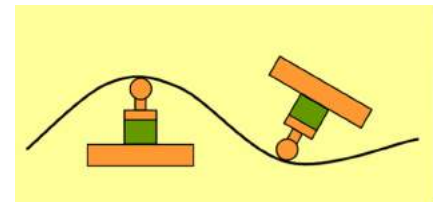
Ro (in)

Thickness, h = 0.0625 in.

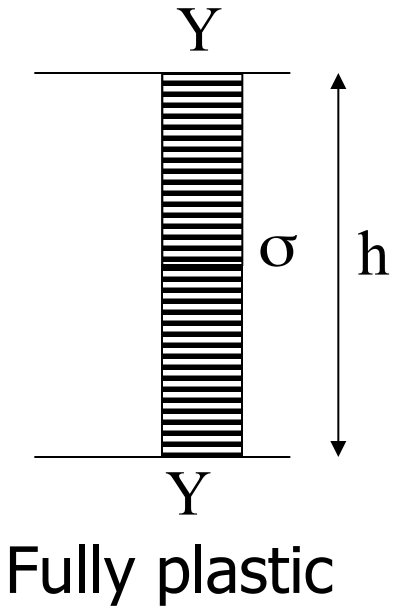
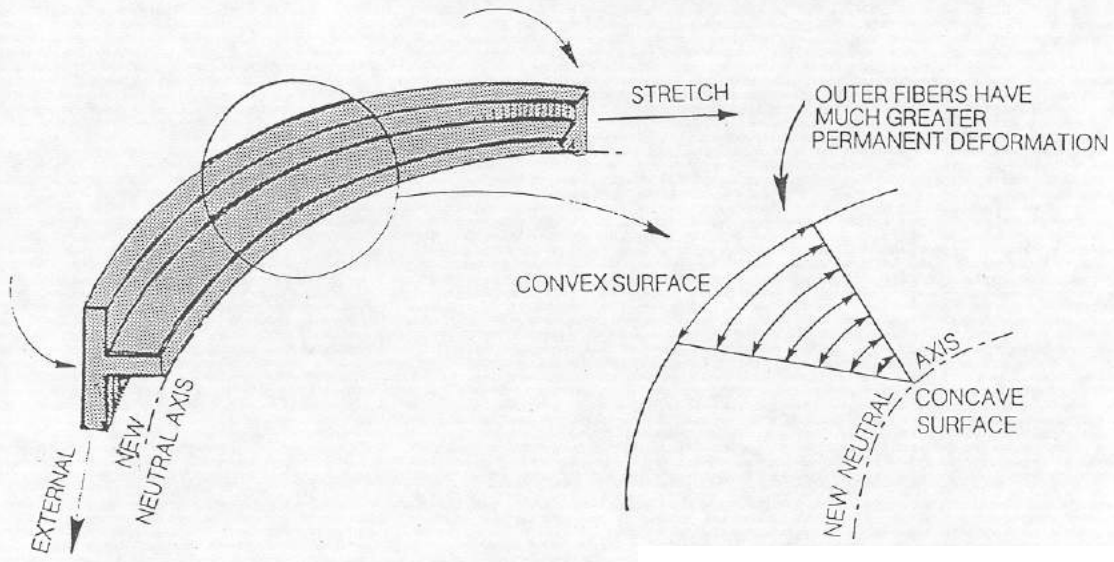
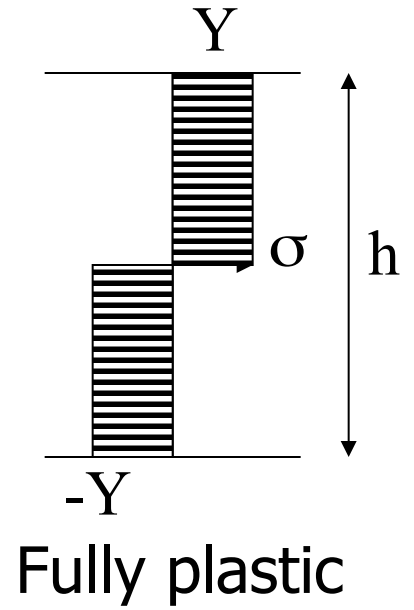
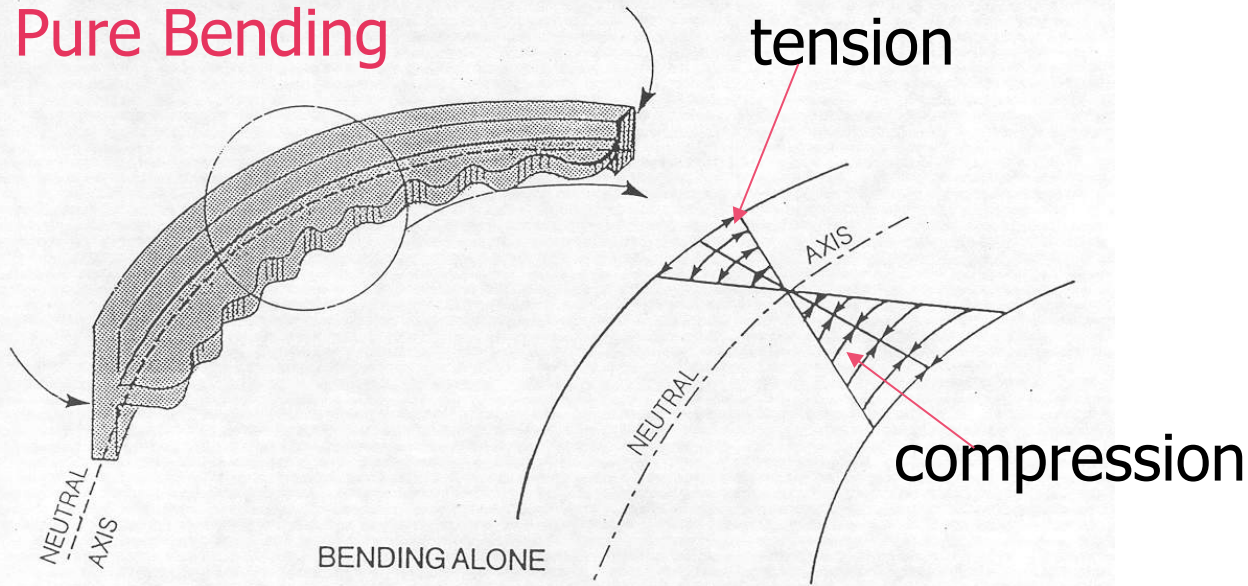
$$\frac{1}{R_0} - \frac{1}{R_1} = 3 \frac{Y}{hE} - 4R_0^2 \left(\frac{Y}{hE} \right)^3$$

Methods to reduce springback

- Smaller Y/E
- Larger thickness
- Over-bending
- Stretch forming
- “coining” or bottoming the punch



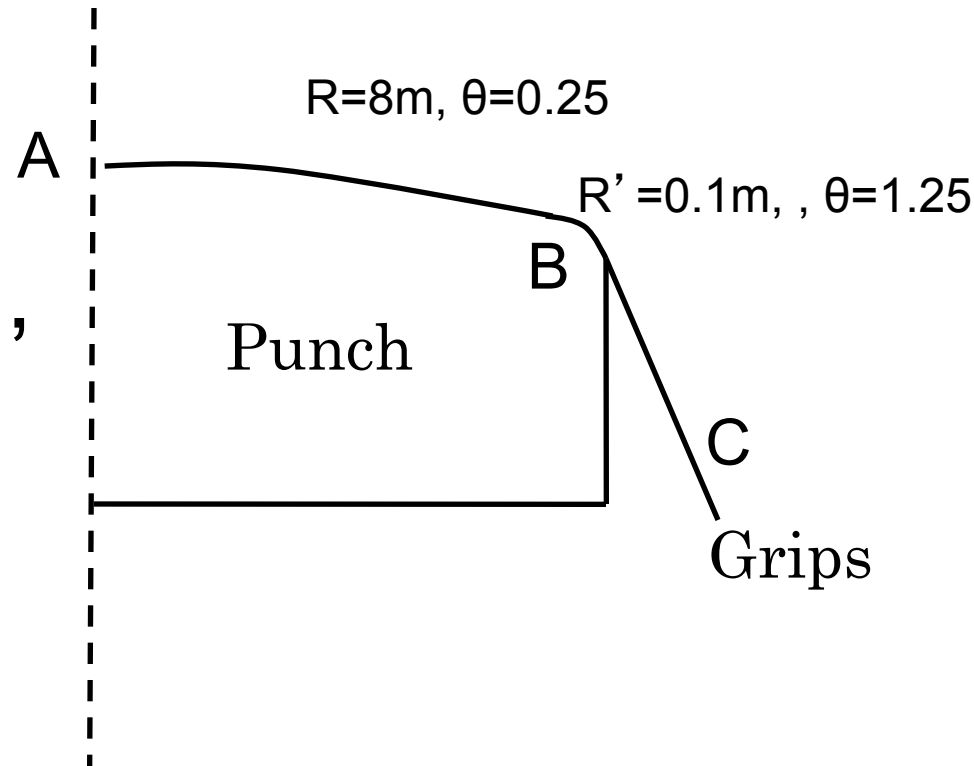
Pure Bending



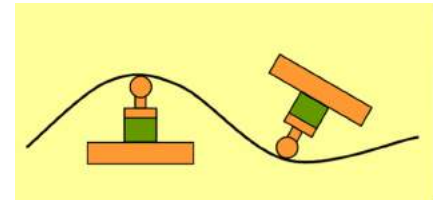
Bending & Stretching

Stretch forming: can we achieve a strain of 0.035 at A?

Sheet thickness 1mm,
 $\mu=0.1$
Material:
 $\sigma=520\varepsilon^{0.18}\text{MPa}$



Massachusetts
Institute of
Technology



Can we achieve a strain of 0.035 at A?

Sheet thickness 1mm,

$\mu=0.1$

Material: $\sigma=520\varepsilon^{0.18}\text{MPa}$

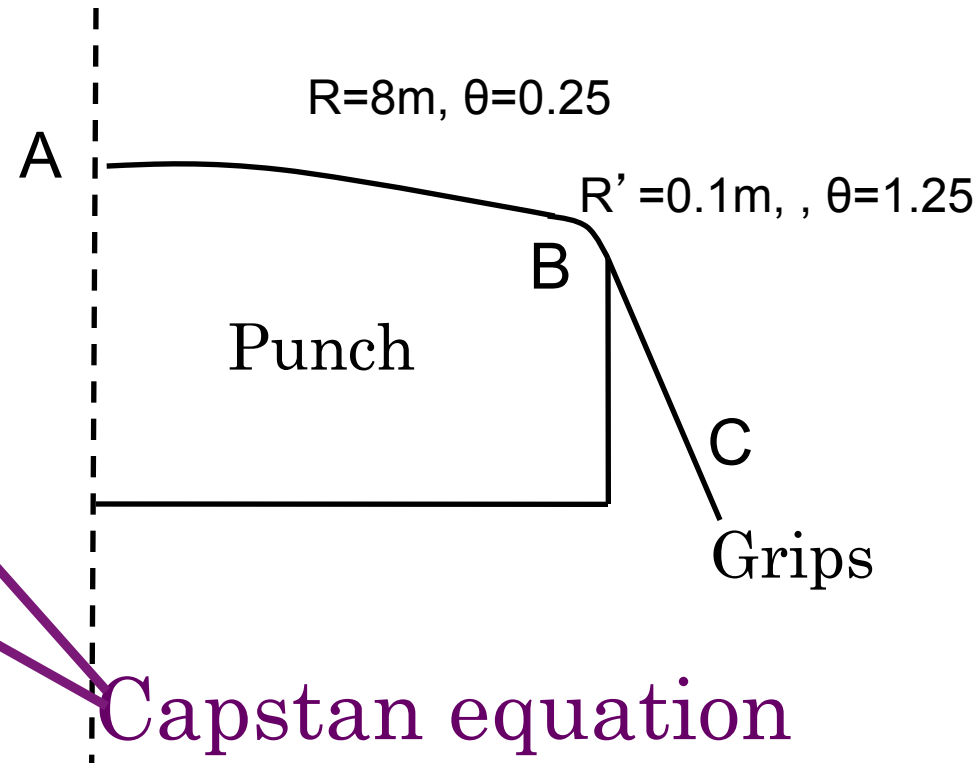
$$F_A = 0.001 \cdot 520 \cdot (0.035)^{0.18} = 284 \text{ kN/m}$$

$$F_B = F_A \cdot \exp(0.1 \cdot 0.25) = 292 \text{ kN/m}$$

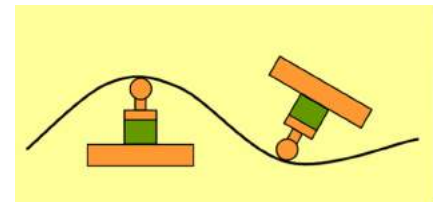
$$F_C = F_B \cdot \exp(0.1 \cdot 1.05) = 323 \text{ kN/m}$$

Max allowable force

$$= 0.001 \cdot 520 \cdot (0.18)^{0.18} = 381 \text{ kN/m}$$



Massachusetts
Institute of
Technology

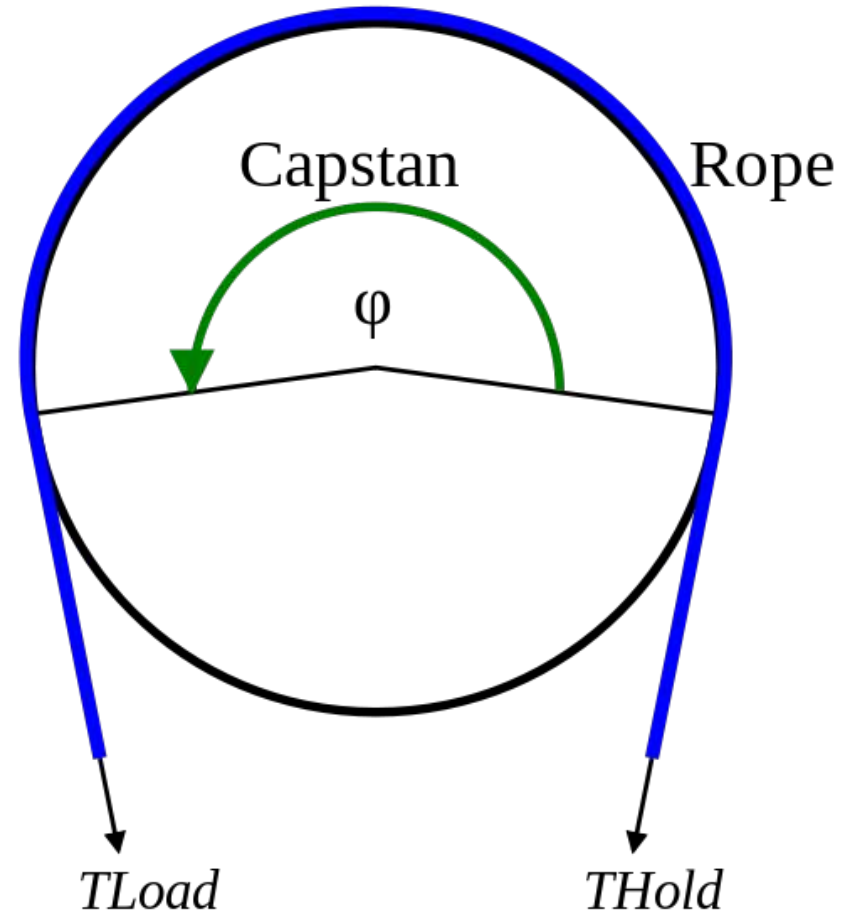


Friction and the capstan equation

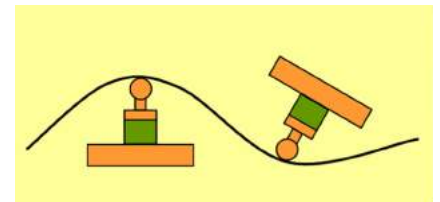
Typical stamping lubricants:

- Oil-based lubricants
- Aqueous lubricants
- Soaps and greases
- Solid films

$$T_{load} = T_{hold} \times \exp(\mu\theta)$$



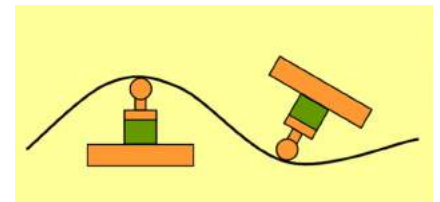
Massachusetts
Institute of
Technology



Research opportunities and challenges: reducing cost and environmental impacts



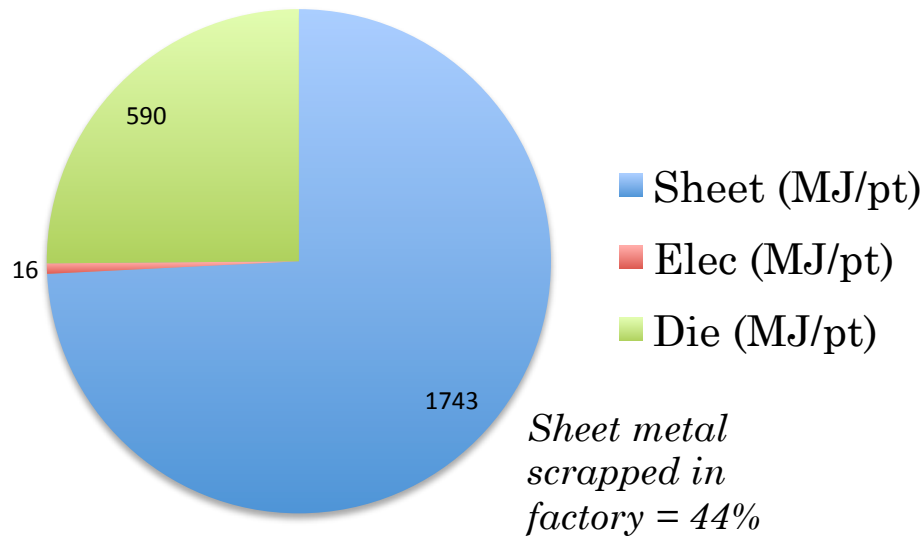
Massachusetts
Institute of
Technology



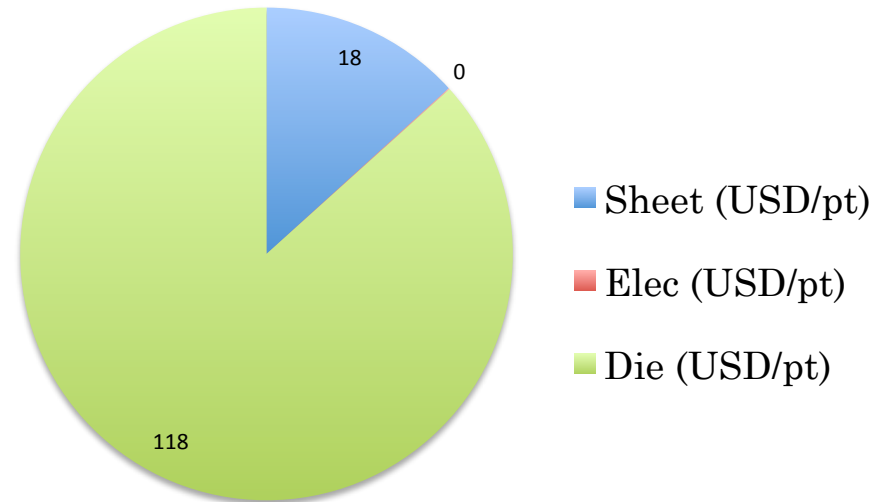
Energy & cost: Stamping alum car hoods

- Final part = 5.4kgs
- Total number of parts made = 400
- Die material: cast and machined zinc alloy

Energy. 2.3GJ/pt. Stamping alum. car hoods. 5.4kgO/P. (400pts)



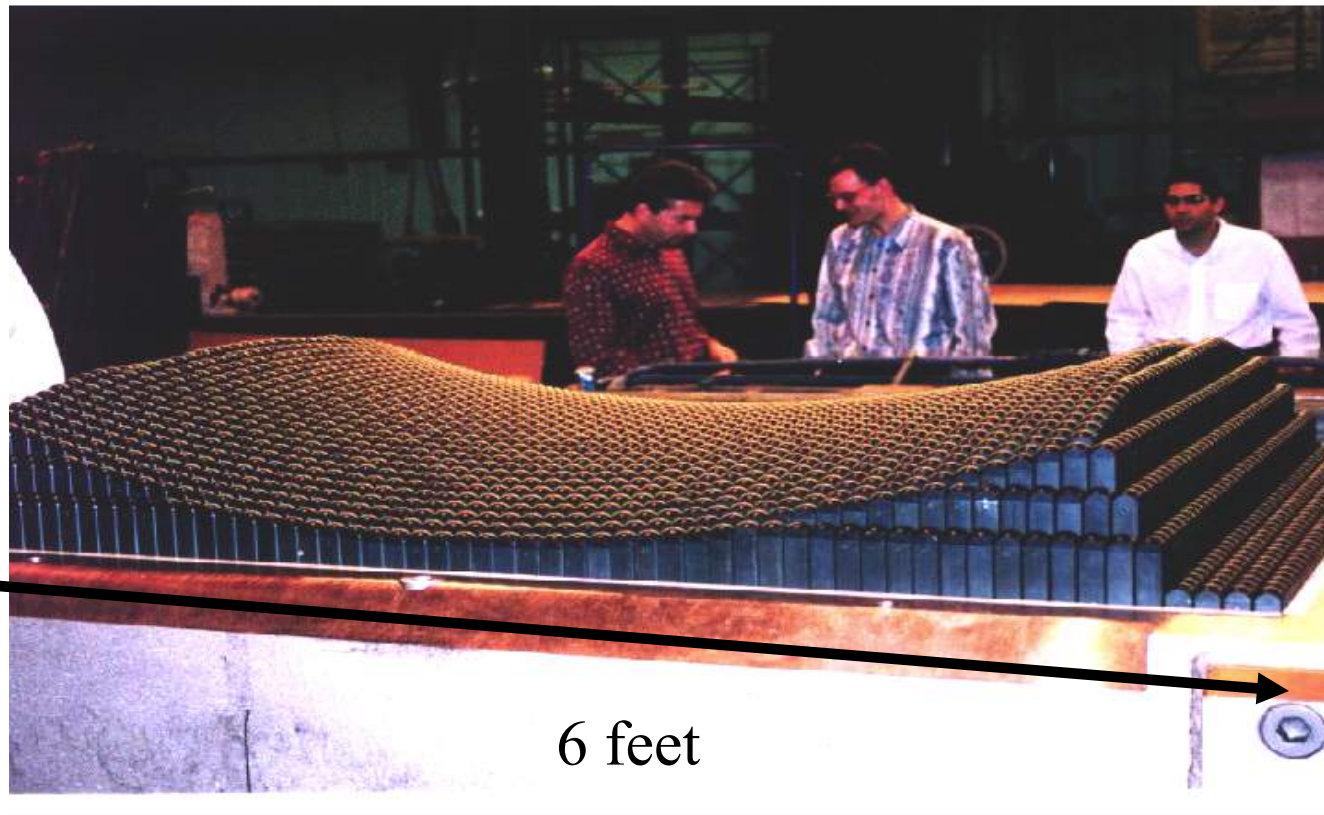
Cost. 136USD/pt



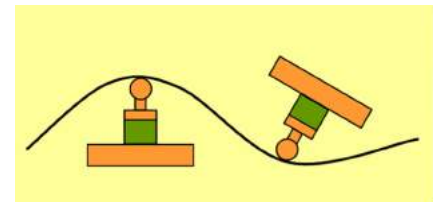
Source: Unpublished work: Cooper, Rossie, Gutowski (2015)

Excludes equipment depreciation and labor during forming

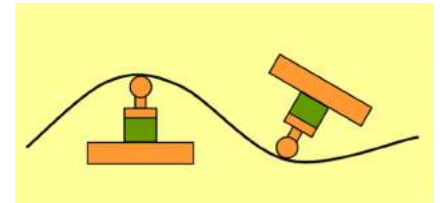
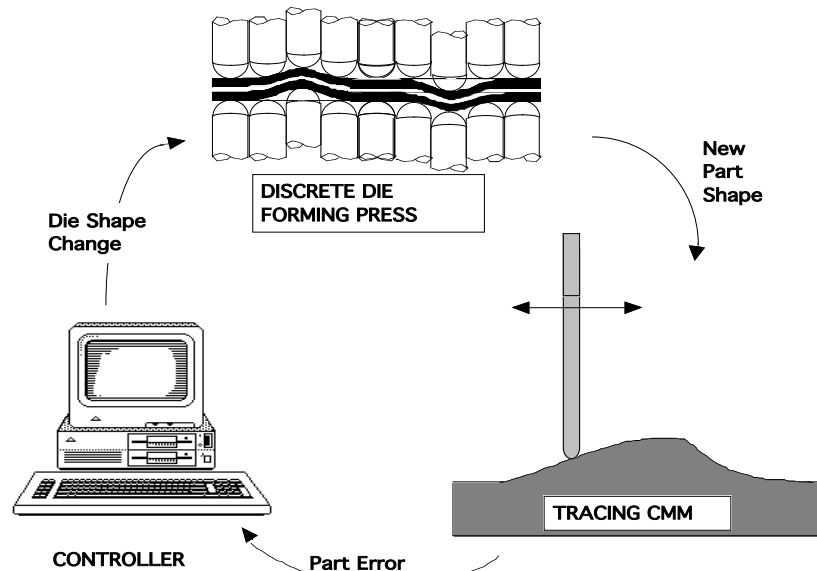
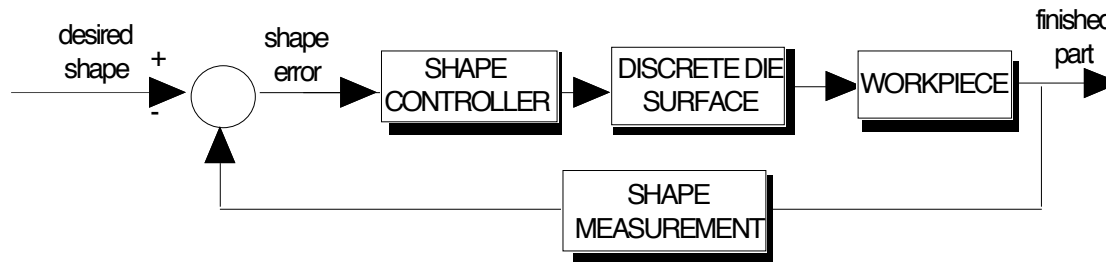
60 Ton Discrete Die Press (LMP - Hardt)



Massachusetts
Institute of
Technology



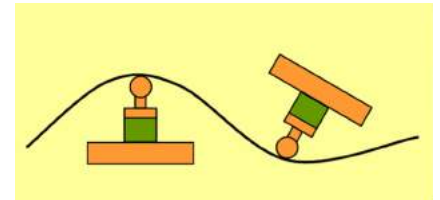
The Shape Control Concept



Stretch Forming with Reconfigurable Tool @ Northrop Grumman



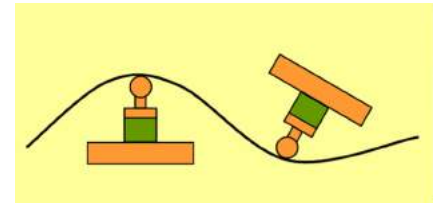
Massachusetts
Institute of
Technology



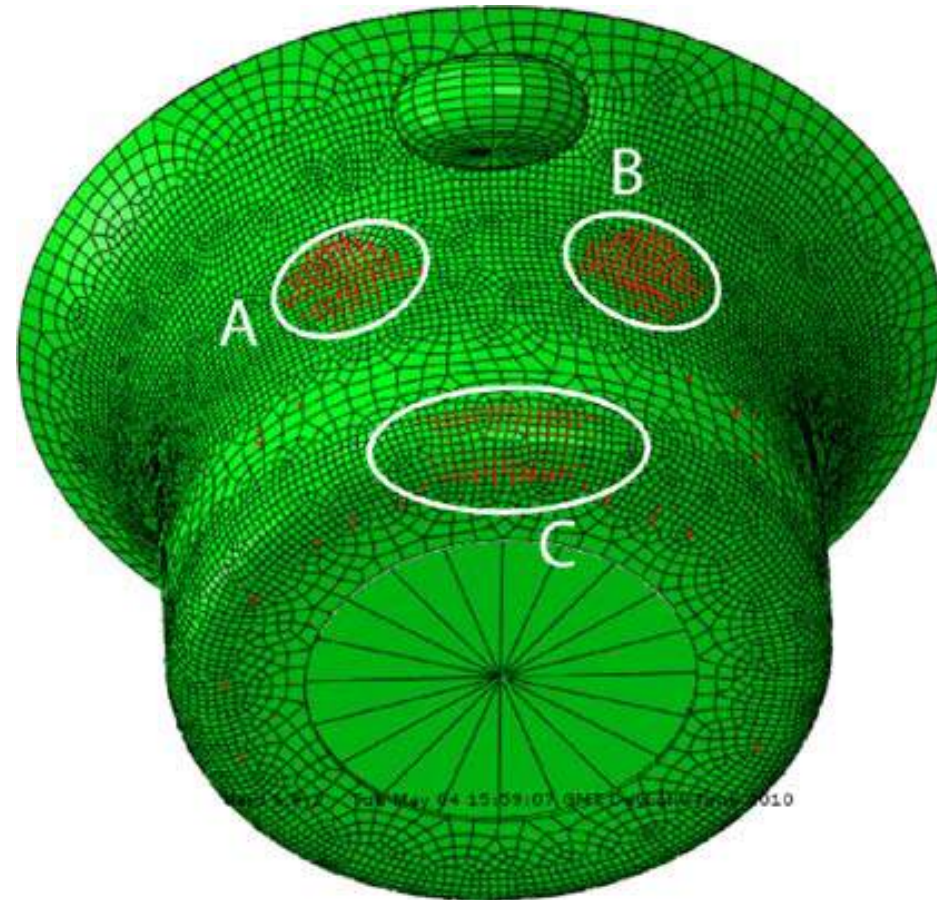
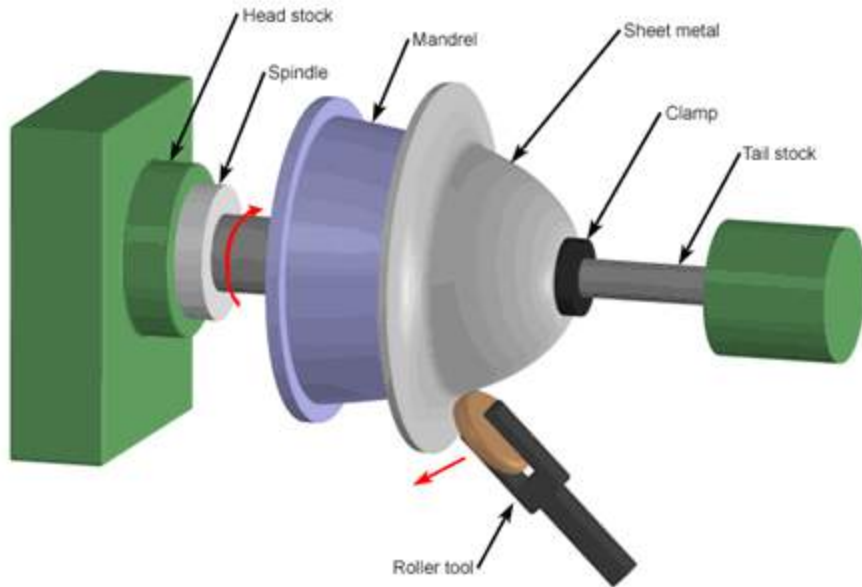
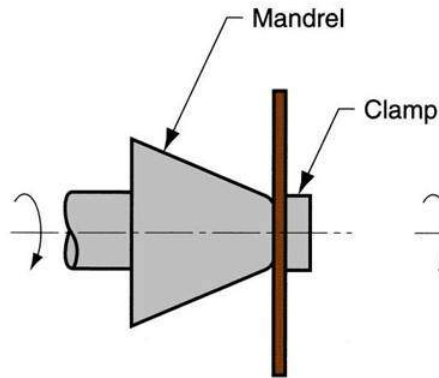
Flexible Forming at Ford



Massachusetts
Institute of
Technology

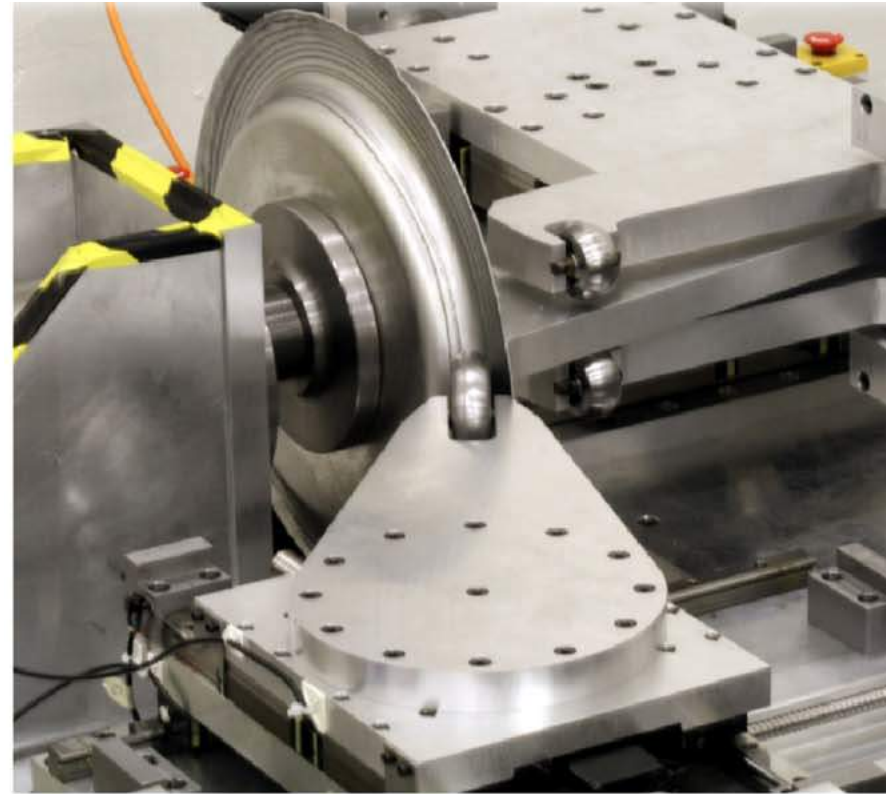
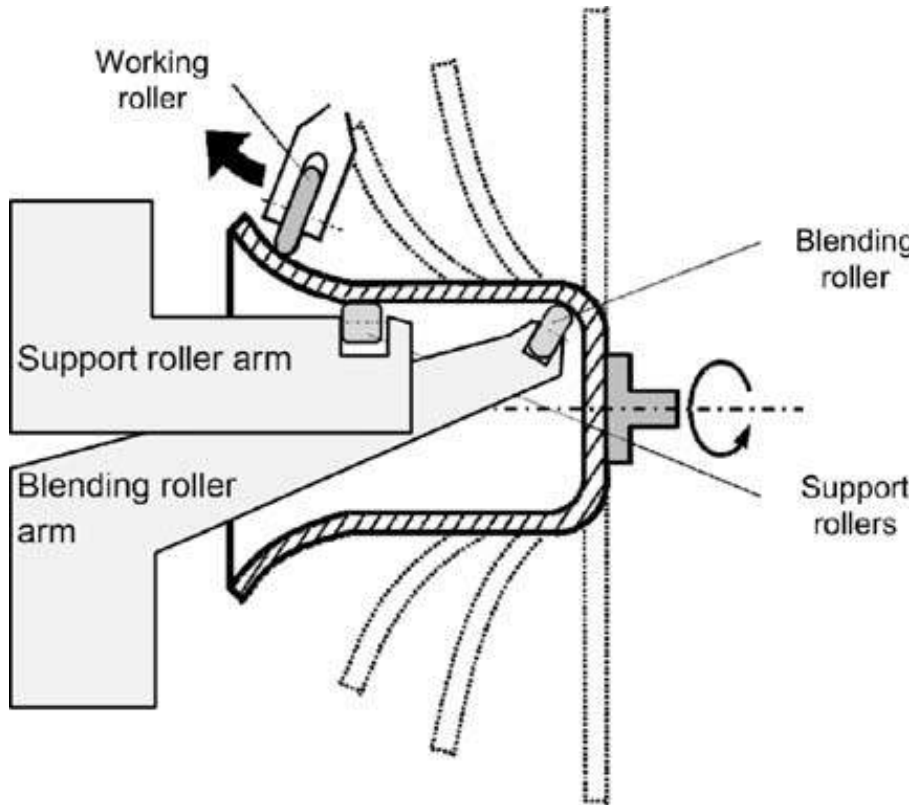


Conventional Spinning



Copyright © 2009 CustomPartNet

Flexible Spinning



(b) Machine in operation



Circular cup



Elliptical cup

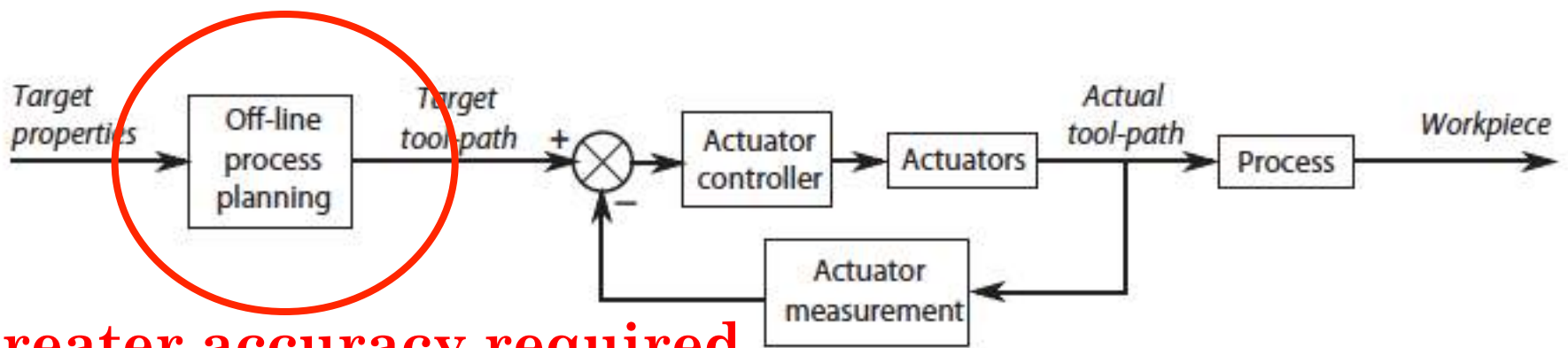


Rectangular cup



Kidney bean

Music, O., & Allwood, J. M. (2011). Flexible asymmetric spinning. *CIRP Annals - Manufacturing Technology*, 60(1), 319–322. doi:10.1016/j.cirp.2011.03.136



Greater accuracy required

Fig. 1. A system diagram for open-loop control of metal forming.

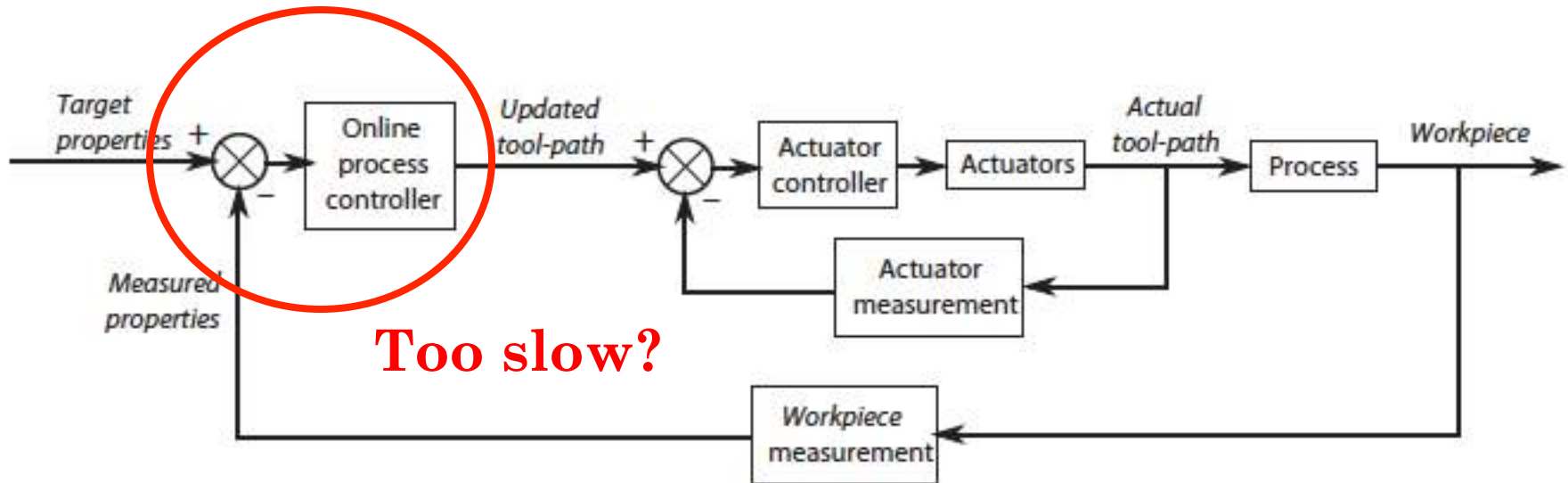
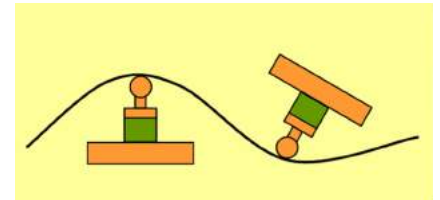


Fig. 2. A system diagram for closed-loop control of metal forming.



**Massachusetts
Institute of
Technology**

Polyblank, J. a., Allwood, J. M., & Duncan, S. R. (2014). Closed-loop control of product properties in metal forming: A review and prospectus. *Journal of Materials Processing Technology*, 214(11), 2333–2348. doi:10.1016/j.jmatprotec.2014.04.014



Thank you

Resourceful Manufacturing & Design Group

<http://remade.engin.umich.edu>



Massachusetts
Institute of
Technology

