

# **State of the Art: Lasercom Systems Engineering and Challenges**

**Emily Clements**

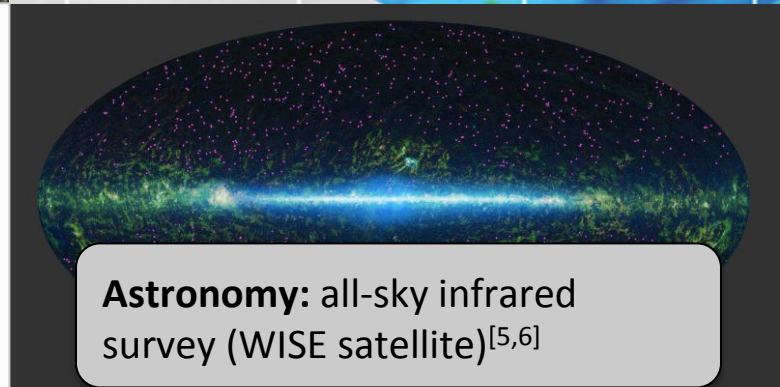
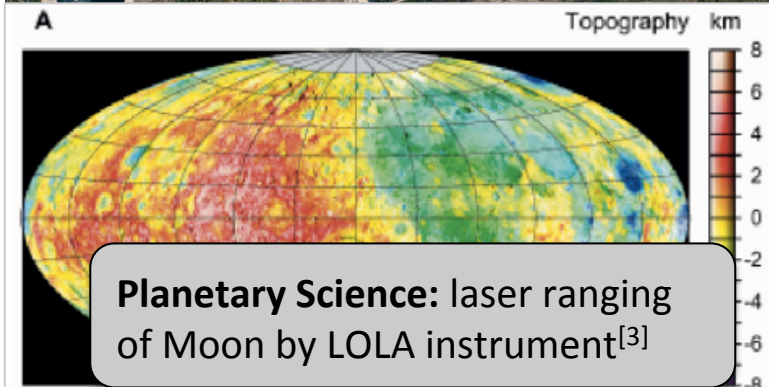
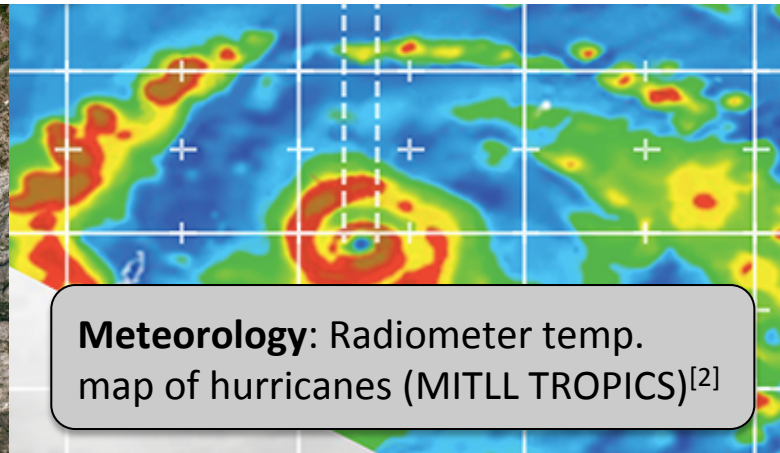
**KISS Workshop on Lasercom for Small Satellites**

# Outline

- Introduction
  - Motivation: Small Satellite Missions
  - Lasercom Advantages and Challenges
- Design of a Lasercom System
  - System Block Diagram
  - Link Performance Modeling
- Operations: Challenges and Opportunities
- Conclusion

# Missions can benefit from lasercom

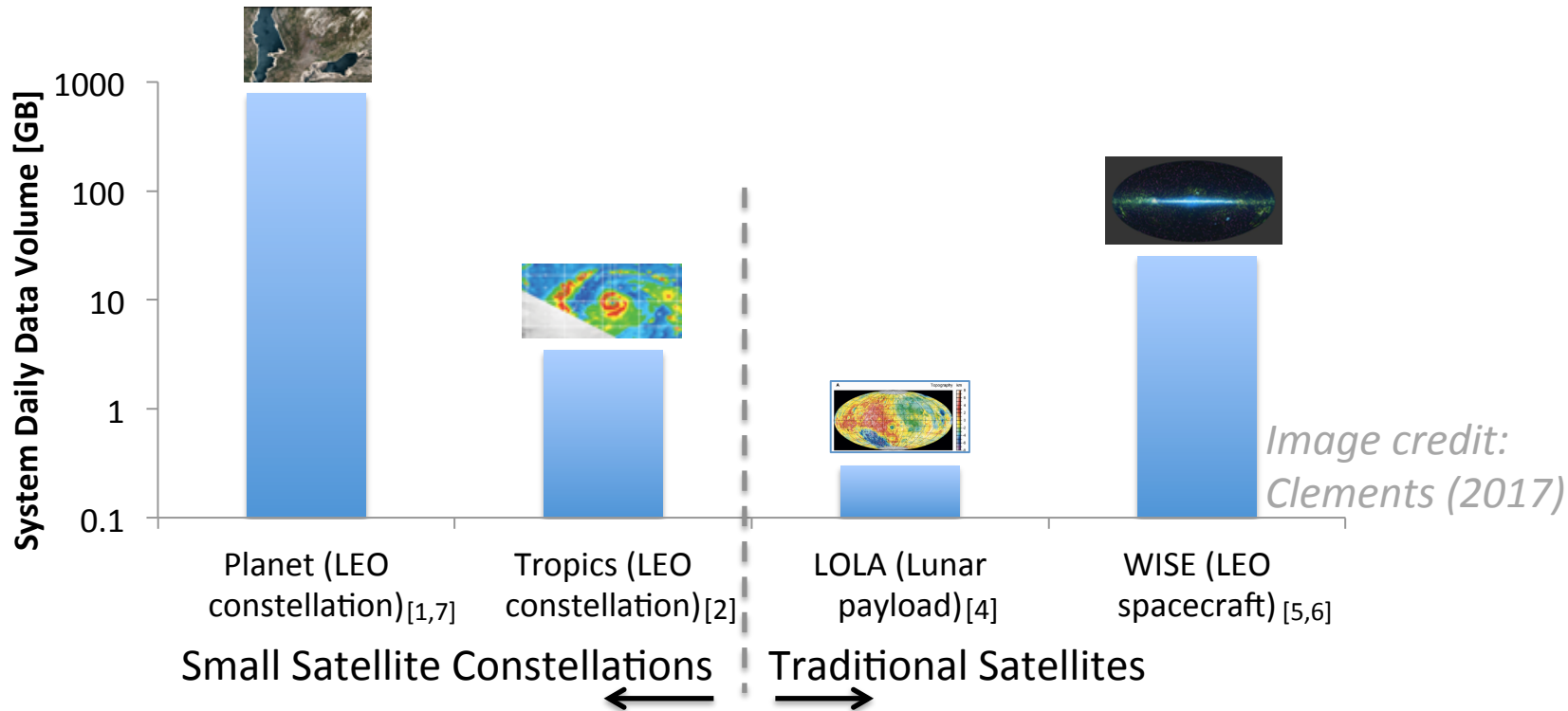
Satellite data are used to provide insight into many problems, such as...



**Can relaxed data constraints enable new capabilities?**

# Utility for Small Satellite Missions

- Small satellites offer a cost-effective solution to global coverage w/ improved temporal resolution
- Data need metrics are: Volume of data downlinked, Timeliness/latency



**Systems of small satellites can produce as much data as traditional satellites**

# RF and Lasercom Advantages & Challenges

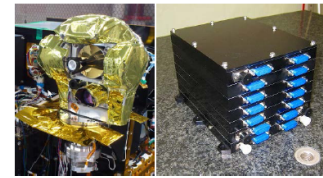
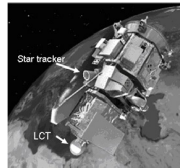
- Lasercom is more power-efficient than radio frequency (RF)

–  $P_{\text{received}} \propto \left(\frac{1}{\lambda}\right)^2$ , where  $P_{\text{received}}$  = received power,  $\lambda$  = wavelength

| Objective/Metric   | Radio Frequency  | Lasercom   |
|--|--|--|
| Data volume, $\mathbf{V}$  | Large transmit power and aperture size <sup>[8]</sup><br>(Selva, 2012) | <b>Higher downlink rates and lower SWAP<br/>(highly scalable for future needs)</b>   |
|  | Spectrum availability, large aperture ground station availability      | <b>Cloud cover</b> hinders access; Addressed by diversity techniques but large networks not available yet                            |
| Age of Information, $\mathbf{Aol}$<br>(latency)                                  | Depends on data volume   | Depends on ability to crosslink, depends on clear line of sight (e.g., cloud cover for downlinking, and ground-station diversity)    |
| Variance data vol. & Aol,<br>$\sigma^2(\mathbf{V})$ and $\sigma^2(\mathbf{Aol})$ | <b>Link losses are more predictable</b>                                | Dependent on atmospheric conditions, variable cloud cover, communication architecture (e.g., diversity techniques, crosslinks, etc.) |

# SmallSat\* Lasercom Missions

## SmallSat Lasercom Tech. Demos



NFIRE-TerraSAR-X<sup>[9]</sup>  
5.6 Gbps,  
LEO crosslink

NFIRE LCT<sup>[10]</sup>  
5.625 Gbps,  
LEO downlink

LLCD<sup>[11]</sup>  
622 Mbps  
Lunar downlink

SOTA<sup>[12]</sup>  
10 Mbps,  
LEO downlink

OCSD<sup>[15]</sup>  
NODE,<sup>[15]</sup> FLARE

2005

2010

2015

Future

## Missions that Advance Supporting Tech.

BRITE<sup>[13]</sup>  
0.0115° pointing



MINXSS<sup>[14]</sup>  
0.002° pointing, first  
flight of Blue Canyon  
wheels



### Related: UAV lasercom:

Facebook Aquila<sup>[17]</sup>  
Optical crosslinks between aircraft

Google Loon<sup>[18]</sup>  
155 Mbps crosslink,  
balloon lasercom system

\*Defined SmallSat as <500 kg<sup>[19]</sup>

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# System Block Diagram

- Communication system block diagram:



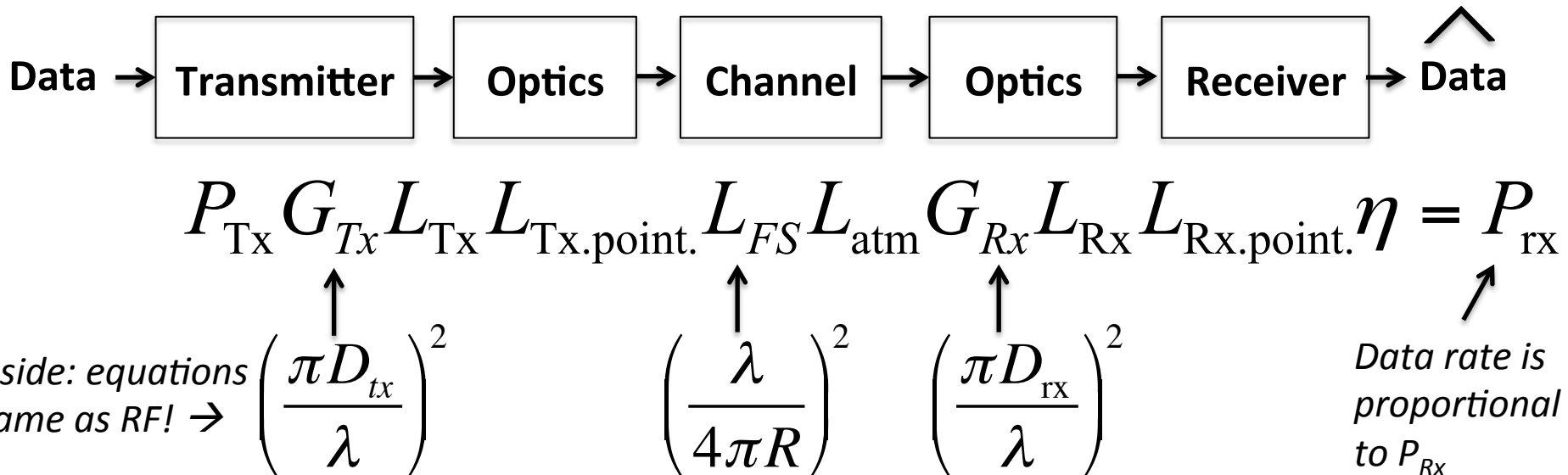
Adapted from Figure 2, Caplan, David O. "Laser communication transmitter and receiver design." *Journal of Optical and Fiber Communications Reports* 4.4-5 (2007): 225-362.<sup>[20]</sup>

- Additional system considerations
  - Pointing control
  - Onboard memory
  - Mechanical/thermal subsystems
  - System with multiple transmitters/receivers

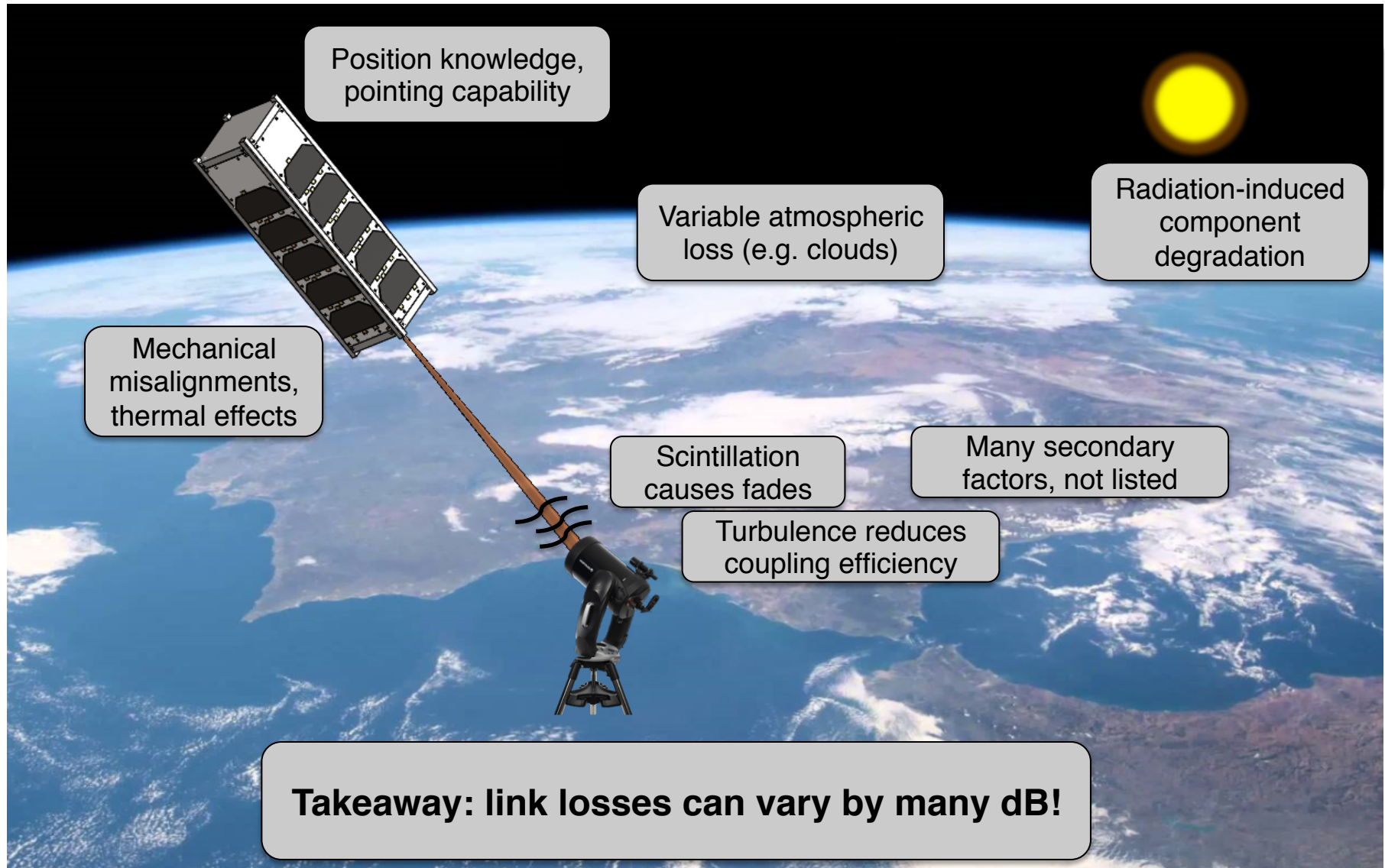


# Link Performance Modeling

- Received power is a function of gains and losses throughout the system:



# Performance Uncertainty Sources



# Link Performance Modeling

## Nominal Link Budget for NODE (LEO, CubeSat, downlink-only)

|                 | NODE       | Units     |
|-----------------|------------|-----------|
| Datarate        | 43         | Mbps      |
| $P_{tx}$        | -7.0       | dBW       |
| $G_{tx}$        | 69.6       | dB        |
| $L_{tx}$        | -1.5       | dB        |
| $L_{freespace}$ | -258.2     | dB        |
| $L_{atm}$       | -1.0       | dB        |
| $G_{rx}$        | 114.7      | dB        |
| $L_{rx}$        | -3.0       | dB        |
| $P_{rx}$        | -78.0      | dBW       |
| $P_{req}$       | -84.2      | dBW       |
| <b>Margin</b>   | <b>6.2</b> | <b>dB</b> |

Table from Clements et al. (2016)<sup>[15]</sup>

## Alternative modeling approach estimates input uncertainties and creates CDFs of link margin

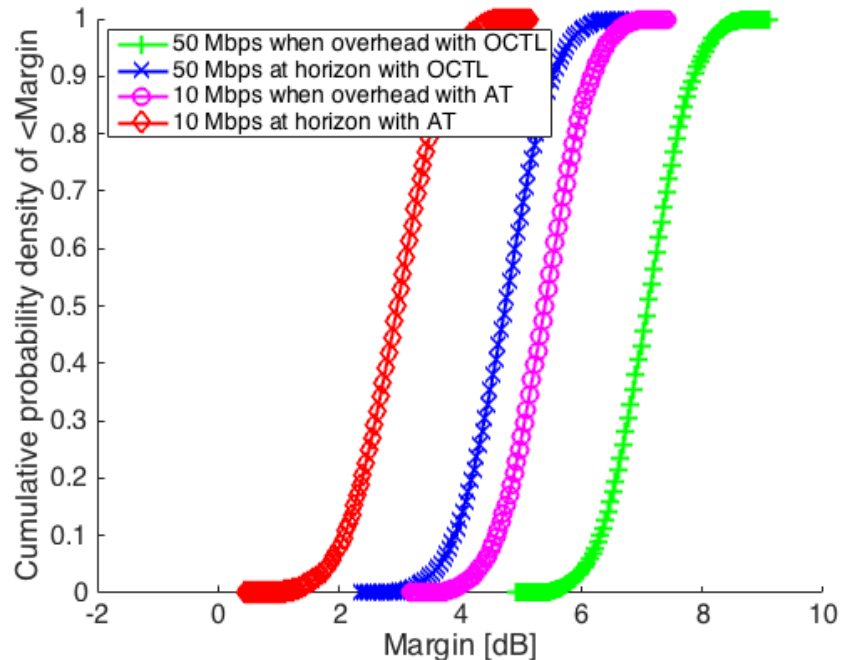


Figure from Clements, Cahoy (2017)<sup>[21]</sup>

## Can model deterministically or through Monte Carlo analysis

E.g., for NODE (MIT CubeSat lasercom downlink payload in development for resource-constrained systems)

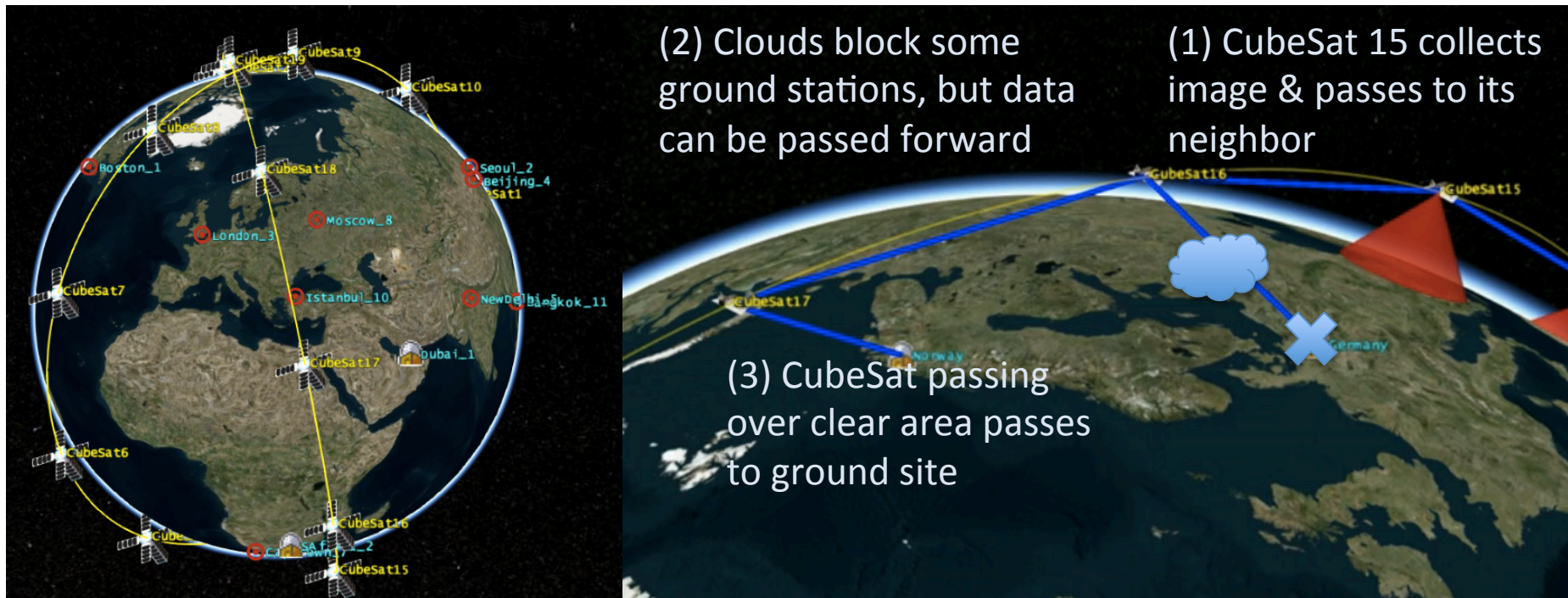
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# Constellation Opportunities

**Problem:** capacity saturation of ground stations for constellations of satellites with high data rate downlink needs

**Solutions:** (i) Many inexpensive ground terminals, (ii) Crosslinks



Visualization of Earth-observing small satellite mission using laser communication

Figure credit: A. Kennedy

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# Conclusion

- Small satellite communications depend on data volume, timeliness (latency), and reliability
- Lasercom can provide high data capabilities with power- and SWAP-efficient designs
- Primary challenge is that it is a relatively new technology in the space environment
  - Capabilities have been demonstrated (e.g., LLCD, TeSAT, etc.).
  - Potential for improvement is significant BUT experience is currently limited and operational uncertainties remain

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# Backup / from old talks

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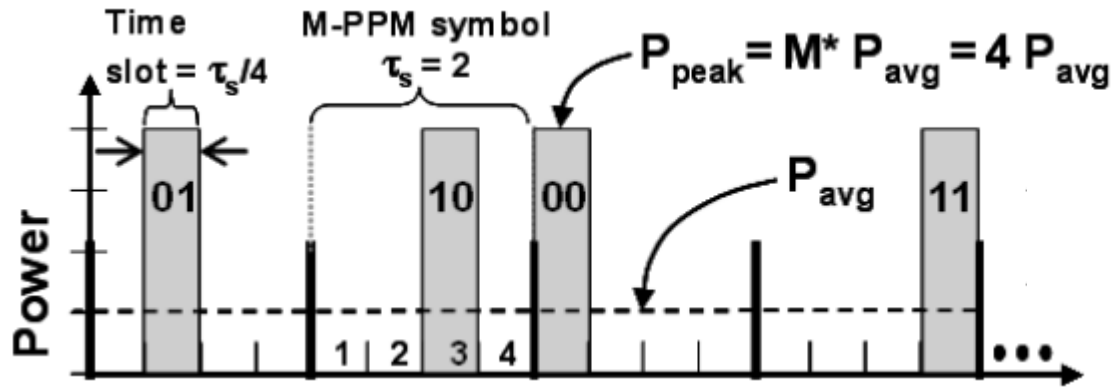
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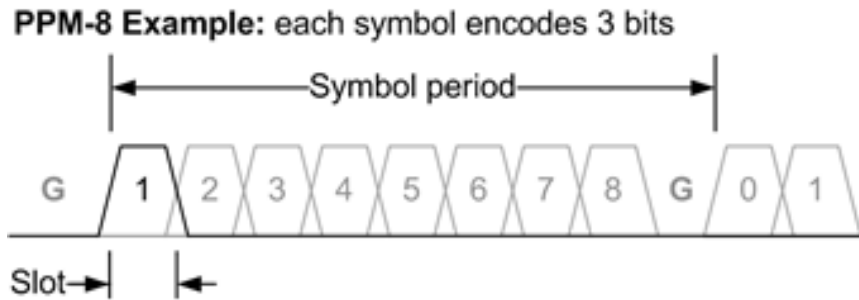
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# PPM Diagrams



Credit: *Laser Communication Transmitter and Receiver Design* by Dave Caplan



Credit: Ryan Kingsbury