

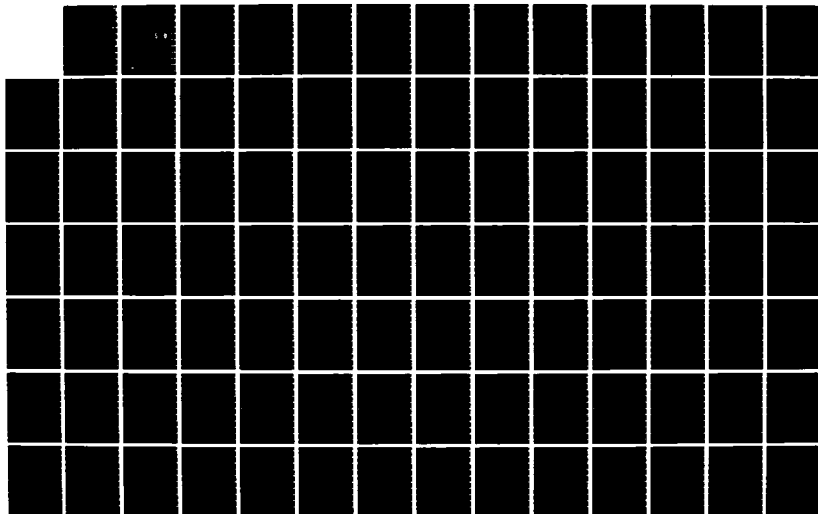
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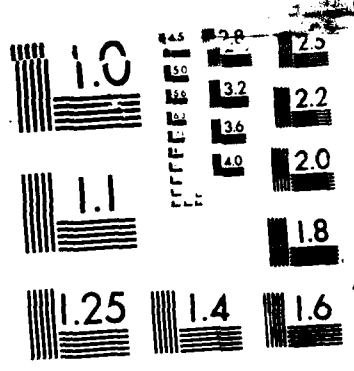
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FORMULATION OF NUMERICAL METHODS  
USED IN THE  
XYZ THREE-DIMENSIONAL POTENTIAL FLOW PROGRAM

An Engineering Report

by

WILLIAM JAMES BEARY JR.

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Submitted to the Faculty of  
the College of Engineering  
Texas A&M University

in partial fulfillment of the requirements for the degree of

MASTER OF ENGINEERING

May 1986

Major Subject: Ocean Engineering

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## ABSTRACT

The calculation of non-lifting potential flow about arbitrary three dimensional bodies is examined in detail with specific interest in the XYZ Potential Flow program developed by the David W. Taylor Naval Ship Research and Development Center. The program uses a surface singularity distribution to solve the Neumann boundary value problem by means of a source panel method assuming a flat element with a constant source density over the area of the element. Boundary conditions are applied at control points on the elements producing a system of linear equations for the source density. When the source density is known, velocities and pressure coefficients may be calculated.

The main purpose of this paper is to present the details of the approximation of an arbitrary three dimensional body using quadrilateral elements, and to provide a detailed derivation of the exact source panel integrations in order to gain insight for future research at Texas A&M University. A variation of the Hess method of surface discretization using quadrilateral source panels is described in detail as it is used in the XYZ Potential Flow program. The exact source panel integrations are derived in detail.

A general discussion of other aspects of the program is included. Velocities and pressure coefficients for flow about a triaxial ellipsoid are calculated using the XYZ Potential Flow Program solution, and the results are compared with the analytical solution and the Hess Program solution.



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## 1.0 INTRODUCTION

This paper examines two aspects of the development of the XYZ Potential Flow Program (hereafter referred to as the XYZPF Program), a FORTRAN program which uses a source panel method to approximate solutions to steady potential flow problems about arbitrary three dimensional bodies. The aspects examined in detail are (1) the description of the details of the approximation of an arbitrary three-dimensional body using quadrilateral elements, and (2) a detailed derivation of the exact source panel integrations.

The XYZPF Program was developed specifically for applications in numerical ship modelling and hydrodynamics studies at the David W. Taylor Naval Ship Research and Development Center (NSRDC) in Bethesda, Maryland. The format of the program is based on the work of Hess and Smith (1962) in the numerical calculation of non-lifting potential flow. A similar program is maintained by the Aerodynamics Division of the McDonnell-Douglas Corporation, referred to in this paper as the "Hess program." The XYZPF Program is a modification of what has come to be known generally as the Hess Method. The most significant modifications are improvements to the method of solving the matrix equation for the source density, and greater flexibility in the input options (Dawson and Dean 1972).

Though potential flow is a product of mathematics, and is never found in a real fluid, the results of potential flow calculations provide usable information for flow regions external to a thin boundary layer,



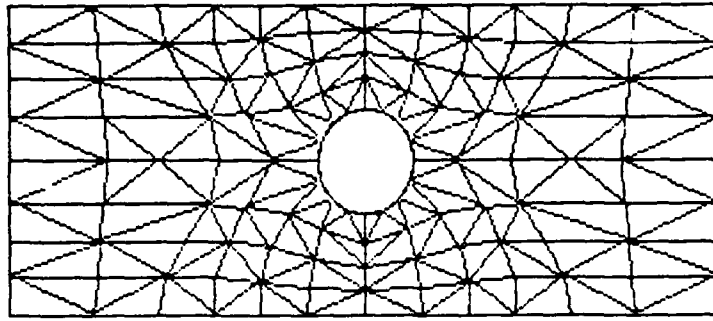
with little or no boundary layer separation. For such flow fields, the region outside the boundary layer may be considered to be effectively inviscid, and may be closely approximated by potential flow models. Small viscous effects can be accounted for by "thickening" the body by the appropriate displacement thickness. Displacement thickness accounts for the region of retarded fluid flow in the boundary layer inversely proportional to the square of the free stream velocity. Downstream of the point of boundary layer separation, the potential flow model no longer applies.

Prior to the development of numerical methods, analytical solutions were generally restricted to simple analytical shapes (Kellogg 1929). The need to solve boundary value problems for arbitrary boundaries in continuum mechanics has fostered the development of numerical approximations to the integral equation expressions. While the integration methods have been well known for quite some time, only since the advent of high speed computers have many of the problems been practical to solve by numerical methods. Among the numerical methods being used are finite differences, finite elements, and the boundary element method.

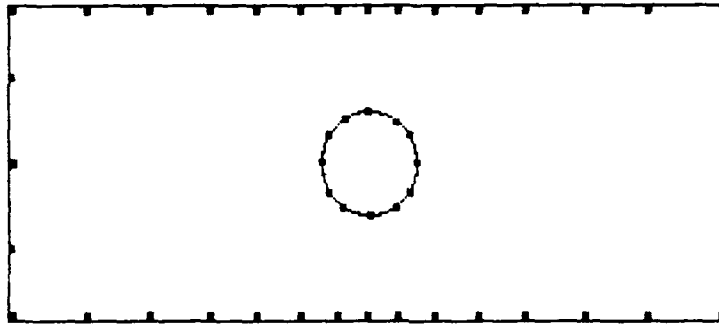
"Finite differences" and "finite elements" are numerical methods which satisfy the boundary conditions, and then approximate the solution to the governing equation in the fluid domain. These methods discretize the domain into a network of elements or cells.

Another approach is what is now known as the "Boundary Element Method," in which the governing equation is exactly satisfied in the domain, and the boundary conditions are applied through a boundary discretization method. The boundary value problem is reformulated as a boundary integral equation which is then discretized by subdividing the boundary into a finite number of surface elements. Each element is represented by an analytical function, and the source density function is integrated over the surface of each element. Two factors governing the accuracy of the boundary element method are the boundary discretization method and the source panel integration. These two factors are examined in detail in this report, as a detailed derivation of the exact source panel integration, including the development of the source panel geometry, has not previously appeared in literature.

The difference between the domain methods and the boundary methods is significant. The domain methods discretize the domain, while the boundary methods discretize the boundary. Thus, the boundary method reduces the dimension of the problem by one, as depicted in figure (1). In the application of the XYZPF Program, the problem is reduced from a three-dimensional problem in the domain to a two-dimensional problem on the boundaries. This method is well suited to problems in which the limits of the domain are infinite or difficult to define, in that the problem is applied to the boundary rather than the domain.



**FINITE ELEMENT DISCRETIZATION**



**BOUNDARY ELEMENT DISCRETIZATION**

**Figure 1. Discretization Methods**

Just as there are many variations of domain methods, there are also a variety of boundary methods. In general, they can be classified as "indirect" or "direct" formulations. The "indirect" method assumes a continuous source distribution over the surface of the body, and a solution which satisfies both the governing equation in the domain, and the boundary conditions on the body surface. The result is an integral equation on the boundary which has the surface source density function as its unknown. By enforcing the boundary conditions at control points on the surface, a system of equations is produced by which the source density may be determined.

The "direct" method solves the velocity potential function through

an application of Green's Second Identity requiring the solution of a source distribution and a dipole distribution on the boundary. The direct method has more physical significance to the general boundary value problem, and more versatility in its application as it can be applied to Neumann problems, Dirichlet problems, or mixed boundary value problems (Brebbia 1984).

The simplicity and accuracy of the indirect method has made it attractive for many applications. The source panel method is an application of the indirect formulation of the boundary element method to the Neumann type of potential flow problem, for which the normal derivative of the potential function is prescribed on the boundary.

## 1.1 OBJECTIVES

The purpose of this paper is (1) to describe the details of the approximation of an arbitrary three-dimensional body using quadrilateral elements, and (2) to provide a detailed derivation of the exact source panel integrations for use in future investigations at Texas A&M University using panels of higher order geometries and source density functions. This paper is not intended to be a user's manual, though a general discussion of other aspects of the program is also included. NSRDC Report 3892 (Dawson and Dean 1972) is a summary of the XYZPF Program for those strictly interested in its use.

## 2.0 HISTORICAL DEVELOPMENT

The foundations of the boundary element method were laid early in this century beginning with Fredholm in 1903 when he established the existence of solutions to the Neumann problem through a reconstruction of the problem using a discretized boundary (Kellogg 1929). The solution was determined to be the potential of a simple source distribution on a boundary with a continuous normal derivative for an infinite domain. Later works by Kellogg (1929) in potential theory demonstrated the application of the boundary integral equation method in electrostatics, heat transfer, flow in porous media, and fluid flow problems, but development was limited by the difficulty of obtaining analytical solutions. No significant advances were made until interest in boundary integral equation methods was revitalized with the advent of high speed electronic computers. Investigators were then able to discretize the boundaries and solve the integral equations numerically. This method of solution became known as the boundary element method. Early development of such numerical methods was pioneered by Hess and Smith (1962) and Jaswon and Symm (1963). Hess and Smith dealt primarily with the indirect formulation eventually leading to a solution for the three dimensional problem as described in this paper. In a parallel work, Jaswon and Symm developed a direct formulation approach to the two dimensional problem. The XYZPF Program is based primarily on the work of Hess and Smith. Hess has since developed a higher order panel method (Hess 1979) and Lefebvre modified the XYZPF Program for calculating velocity potentials for five degrees of freedom (Lefebvre 1982).

### 3.0 THEORETICAL DEVELOPMENT

#### 3.1 THE POTENTIAL FLOW PROBLEM IN THREE DIMENSIONS

The governing equation for ideal (incompressible, inviscid, irrotational) flow is Laplace's equation:

$$\nabla^2\phi = 0 \quad (1)$$

where  $\phi$  is the velocity potential, and  $\nabla^2$  is the Laplacian operator. The XYZPF Program deals with steady, uniform flow of an ideal fluid about an arbitrary three dimensional body. The velocity components at any point within the flow field may be obtained from the negative gradient of the velocity potential, i. e.

$$\mathbf{V} = -\nabla\phi = -\frac{\partial\phi}{\partial x}\mathbf{i} - \frac{\partial\phi}{\partial y}\mathbf{j} - \frac{\partial\phi}{\partial z}\mathbf{k} \quad (2)$$

The freestream flow  $\mathbf{V}_\infty$  is defined as a uniform stream of unit magnitude.

$$|\mathbf{V}_\infty| = \sqrt{V_{\infty x}^2 + V_{\infty y}^2 + V_{\infty z}^2} = 1 \quad (3)$$

The key to the boundary element method is the Divergence Theorem (Green's Theorem) which relates a volume integral to an equivalent surface integral reducing the three-dimensional problem to a

two-dimensional one. The expression for Green's second identity is (Lamb 1924):

$$\iiint (\phi \nabla^2 w - w \nabla^2 \phi) d\Omega = \iint (w \frac{\partial \phi}{\partial n} - \phi \frac{\partial w}{\partial n}) d\Gamma \quad (4)$$

in which  $\Omega$  represents the integration over the three dimensional domain, and  $\Gamma$  represents integration over the two dimensional boundary. The partial derivatives are taken with respect to the outward normal,  $n$ . The weighting function,  $w$ , is usually chosen to be the fundamental solution for three dimensions,  $w = 1/(4\pi r)$ , where  $r$  is the distance from the source to an arbitrary point on the boundary.

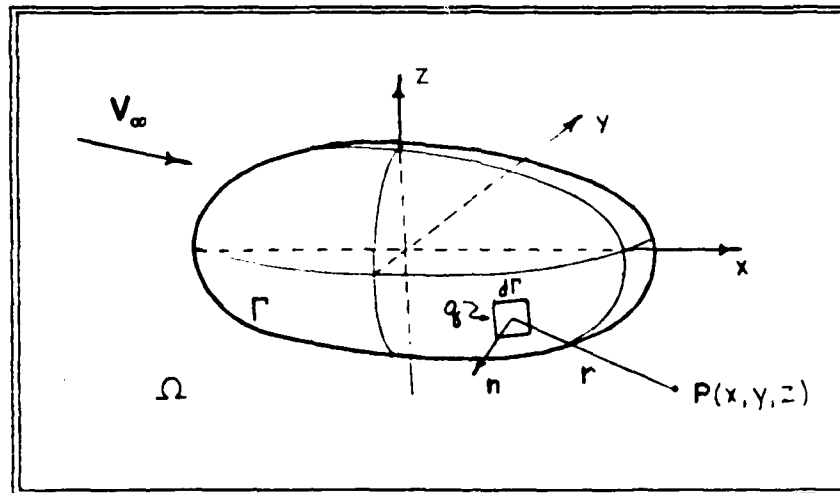


Figure 2. Potential Flow in Three Dimensions

Consider an arbitrary three-dimensional body with surface  $\Gamma$ , having an equation of the form  $F(x, y, z) = 0$  where  $x, y, z$  are Cartesian coordinates of the global reference system as shown in Figure (2). The unit outward normal,  $n$ , at any point on the surface is given by the gradient of the function describing the surface divided by the magnitude

of the gradient, i.e.

$$\mathbf{n} = \frac{\pm \nabla F}{|\nabla F|} \quad (5)$$

where the sign of the unit normal vector is chosen to ensure that the vector is an outward normal. The potential function  $\phi$  describing the flow field must meet the following boundary conditions:

a.  $\nabla^2 \phi = 0$  (Laplace's Equation) (6)

b. For an impermeable boundary, the velocity normal to the surface must be zero relative to the boundary (the Neumann boundary condition):

$$\left( \frac{\partial \phi}{\partial n} \right)_{\Gamma} = 0 \quad (7)$$

c. The velocity potential approaches the freestream velocity potential as the distance from the body goes to infinity:

$$\phi \rightarrow \phi_{\infty} \quad \text{as} \quad |\mathbf{r}| \rightarrow \infty \quad (8)$$

The total potential at any point in the domain is composed of the freestream potential and the disturbance potential due to the body.

$$\phi = \phi_{\infty} + \psi \quad (9)$$



The disturbance potential,  $\phi$ , satisfies the following boundary conditions:

a.  $\nabla^2\phi = 0$  (10)

b. From equation (7), the velocity normal to the boundary due to the disturbance and due to the onset flow must be of equal magnitude, but opposite sign. Then from equation (9)

$$\left(\frac{\partial\phi}{\partial n}\right)_{\Gamma} = \mathbf{n}(p) \cdot \mathbf{v}_{\infty} \quad (11)$$

Note that the normal vector is a function of position on the surface of the body.

c. The disturbance potential approaches zero as the distance from the body goes to infinity, i. e.

$$\phi \rightarrow 0 \quad \text{as} \quad |\mathbf{r}| \rightarrow \infty \quad (12)$$

### 3.2 MATHEMATICAL MODEL

The disturbance potential of the body may be represented by a distribution of a source density function  $\sigma$  over the body surface. The potential at an arbitrary point  $P(x, y, z)$  due to the surface potential is (Kellogg 1929):

$$\psi(x, y, z) = \iint \frac{\sigma(q)}{r(P,q)} d\Gamma \quad (13)$$

where  $q$  is the integration point on the surface, and

$$r(P,q) = \sqrt{(x - x_0)^2 + (y - y_0)^2 + (z - z_0)^2}$$

is the distance from the field point  $P$  to the integration point  $q$ .

The source density distribution function must satisfy the boundary conditions for the disturbance potential. Boundary conditions (10) and (12) are automatically satisfied by the form of the integrand. However, equation (11), the velocity normal to the boundary, combined with the Neumann boundary condition, equation (7), is the key to solving the boundary integral problem.

The integrand becomes singular as the surface of the body is approached, i. e.  $|r|$  goes to zero. The singularity represents the local fluid flux normal to the boundary due to the local source density. The principal value of the singularity is  $-2\pi\sigma(p)$ , determined through a limiting process of the Gauss Flux Theorem (Kellogg 1929). The point  $p$  represents a field point which lies on the boundary. The integral expression is now composed of the contribution of the local source density and the contribution of the source density function over the remainder of the body surface. Solving for the velocity normal to the surface yields the following expression:

$$\left(\frac{\partial\psi}{\partial n}\right)_{\Gamma} = -2\pi\sigma(p) + \iint \frac{\partial}{\partial n} \left[ \frac{\sigma(q)}{r(p,q)} \right] d\Gamma \quad (14)$$

From equation (11), this expression becomes:

$$2\pi\sigma(p) - \iint \frac{\partial}{\partial n} \left[ \frac{\sigma(q)}{r(p,q)} \right] d\Gamma = -\mathbf{n}(p) \cdot \mathbf{V}_{\infty} \quad (15)$$

This equation is a two dimensional Fredholm integral equation of the second kind, which ensures a unique solution, and that the diagonal elements of the system matrix will be dominant, each having a value of  $2\pi$  (Kellogg 1929). Once equation (15) has been solved for the source density  $\sigma$ , the velocity components at any point of the flow field may be obtained by differentiating the disturbance potential function (13) with respect to the coordinate directions and adding the components of the freestream flow,  $\mathbf{V}_{\infty}$ .

$$\mathbf{V}(x, y, z) = \mathbf{V}_{\infty} - \frac{\partial\psi}{\partial x} \mathbf{i} - \frac{\partial\psi}{\partial y} \mathbf{j} - \frac{\partial\psi}{\partial z} \mathbf{k} \quad (16)$$

The body shape does not have to be slender, axisymmetric, or simply connected, allowing for analysis of interior flow and a wide range of applications of the method. The only restriction imposed on the form of the body is that it must have a continuous normal vector. Discontinuities in the right hand side of equation (15) will produce unwanted singularities. Thus, in the process of approximating a body which has distinct corners, where there is clearly a discontinuity in the normal vector, the corner must be replaced by a surface with some finite

curvature. However, application of this method has shown that the flow calculations give correct results for convex corners, while unrounded concave corners may or may not produce significant error, depending on the angle produced by the corner (Hess and Smith 1962).

Because of the method of approximation, the calculation of flow velocities on the body surface are restricted to the points at which the boundary conditions were applied. Velocities at points other than those must be obtained by interpolation. Direct calculation of velocities at the edge of an element yields infinite velocities.

With the solution of the system of linear equations for the source densities, the flow velocities at any point in the domain may be obtained from equation (16), and pressure coefficients are then computed from the velocities using a form of the Bernoulli equation:

$$P(t) = \frac{p}{\rho} + \frac{1}{2} |\mathbf{v}|^2 + \frac{\partial \phi}{\partial t} \quad (17)$$

where  $P(t)$  is a constant independent of position. In the XYZPF Program, the flow is steady. Therefore, equation (17) can be reduced to

$$p + \frac{1}{2} \rho |\mathbf{v}|^2 = \text{constant} \quad (18)$$

and the pressure field can be expressed in terms of a dimensionless pressure coefficient  $C_p$  as:

$$C_p = \frac{p - p_\infty}{\frac{1}{2} \rho |\mathbf{v}_\infty|^2} = 1 - \frac{|\mathbf{v}|^2}{|\mathbf{v}_\infty|^2} \quad (19)$$

where  $p_{\infty}$  is the static pressure at infinity.

### 3.3 NUMERICAL MODEL

In order to represent the surface of a body in the domain mathematically, the body may be described by analytical expressions which may provide an exact representation of the surface. However, the types of bodies which can be adequately described by such methods are severely limited. Another way to represent the body is to use a large number of analytical expressions, each describing only a small portion of the body. Hess and Smith (1962) suggested the use of an assembly of flat quadrilateral elements to model the actual surface of the body, as shown in Figure (3). Each quadrilateral approximates a region of the surface described by points which lie on the actual surface of the body. As planar elements, these quadrilaterals are clearly analytical, and when carefully constructed, the elements can approximate arbitrary three dimensional body surfaces without restriction.

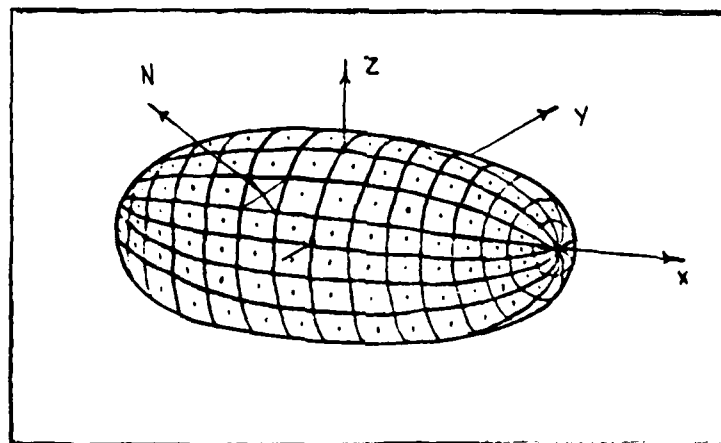


Figure 3. Approximation of the body by surface elements

The XYZPF Program uses the discretization procedure described by Hess and Smith (1962) with some minor modifications. The three dimensional body surface may be described using a large number of plane quadrilateral elements, each assumed to have a constant source density over the area of each element. Regions of the body requiring higher resolution for sharp curvature or anticipated velocity gradients will require a higher concentration of elements.

Because the plane quadrilateral elements cannot fit edge to edge on a rounded surface, small gaps in the panel approximation contribute to the error of the approximation. However, the error due to the gaps is negligible when compared with the error of the basic model, that is, using flat panels to approximate a curved surface. Triangular elements have been suggested in an attempt to eliminate the gaps (Levy 1959), but the increased accuracy is so small that it may not justify the additional work of organizing the triangular elements in lieu of the simpler quadrilaterals (Hess and Smith 1966). The method presented is valid for an polygonal element with any number of sides.

Equation (15) can now be decomposed into a summation of integrals, each representing the contribution of one element of the body surface. The unknown source density can be taken outside the integral, since it is assumed to be a constant over each element. The integration is performed over the area of the source element, and the boundary condition equation (11) is then enforced at a single point  $p$  in each remaining element. By performing this operation at each element of the

surface, a system of linear equations is generated which is equal in number to the number of surface elements and the number of unknown source densities. Equation (15) can now be approximated by the matrix equation (Dawson and Dean 1972):

$$\sigma_i = \sum_j \sigma_j C_{ij} + V_i \quad (20)$$

where

$$C_{ij} = \frac{1}{2\pi} \iint_j \frac{\partial}{\partial n_i} \left[ \frac{1}{r_{ij}} \right] dA$$

$$C_{ii} = 0$$

$$V_i = -\frac{1}{2\pi} \left[ n_i \cdot \mathbf{v}_\infty \right]$$

It is important to note that the influence coefficients  $C_{ij}$  and  $C_{ji}$  are functions of geometry only, and once computed, need not be recomputed for analysis of several different flows. From the solution of equation (20) on the discretized surface, equation (13) may be applied at any point in the domain. Then, the velocity at an arbitrary field point  $P(x, y, z)$  in the domain may be determined from equation (16). With the velocity known, the pressure coefficient is determined from equation (19).

#### 4.0 ORGANIZATION OF THE PROGRAM

The XYZPF Program is actually composed of seven independent programs, referred to as sections PF1 through PF7, each of which builds on data generated from a previous section. This type of organization allows the user the flexibility of rerunning portions of the program using different flow parameters without having to go through the time consuming process of recalculating the influence coefficient matrix, which is dependent only on the geometry of the body. While the NSRDC program is very similar to the Hess program, there are also some significant differences. The following list of differences is taken from NSRDC Report 3892 (Dawson and Dean 1972):

(1) The input to XYZ PF is arranged to facilitate the preparation of input for a series of problems in which only one part of the body is changed. Also, a number of checks are made on the input to help detect errors.

(2) An option was added for the recomputation of the source density and velocities for only part of the body when only small changes are made. This option also provides for the use of the solution of one problem as an initial guess for the solution of another problem.

(3) The matrix of influence coefficients is computed column by column instead of row by row. This column arrangement was used for the original LARC computer version because it required much less high speed memory. The computation is also about 10% faster this way than with the row-by-row arrangement.

(4) A simultaneous displacement iterative scheme is used to solve



the matrix coefficient for the source density. The scheme is slower than the successive displacement (Gauss-Seidel) scheme used in the Hess program, but it can be carried out using the matrix column by column instead of row by row.

(5) When possible, an extrapolation procedure is used to reduce the number of iterations required for convergence. One such method is the Richardson extrapolation.

The methods used in the XYZPF Program will be discussed in detail in the following sections.

## 5.0 DETAILS OF THE SURFACE APPROXIMATION

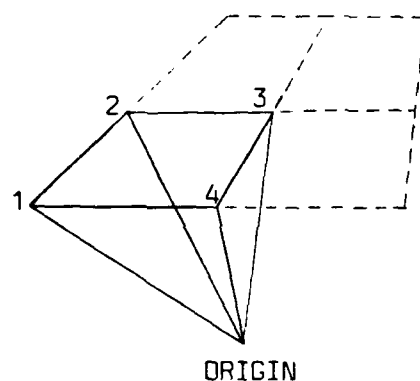
### 5.1 PREPARATION OF THE INPUT

Section PF1 is set up to read and store the input data, and to examine the cornerpoints of the quadrilaterals to detect obvious errors in the input. Because of the number of points which may have to be entered for a complex geometry, the user input is a major source of program error, and this first look for input errors will save lot of run time in the program as a whole. If Section PF1 detects major errors in the input, the program will not continue with the calculation of the coefficient matrix, but will stop and identify the grid location of the error. Minor errors may not cause the program to stop, but will be noted in the output.

One of the major advantages to this program is in the organization of the input data. The surface is input in sections so that small portions of the input geometry may be changed without having to recalculated the points for the entire surface. The program also takes advantage of symmetry to minimize the input effort. Only the portion of the body which has no redundancy needs to be entered point by point. The remainder of the body is reflected across the planes of symmetry by the program to complete the surface representation.

The surface is represented by a set of points in three-dimensional space which lie on the actual surface, and which will later be used to define the plane quadrilateral source elements. These points are defined in the global reference system. The points on the surface should be

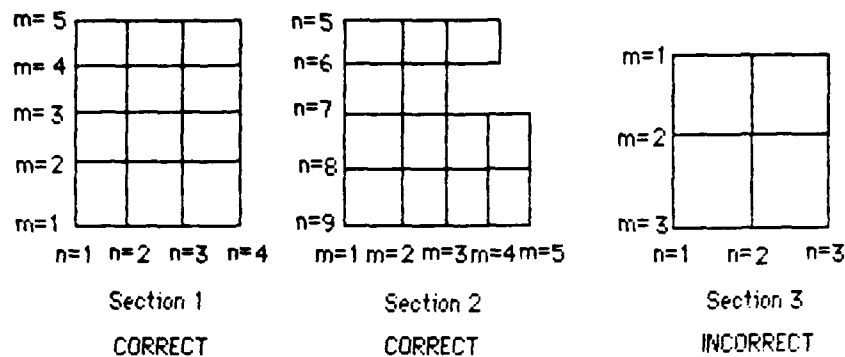
selected in such a way as to provide an accurate representation of the surface with the fewest number of points possible. Portions of the surface which are highly curved will require a larger number of points to provide adequate resolution. Additionally, portions of the surface in which the flow field is expected to change rapidly will require a large number of points to accurately determine the flow field in that region. Some familiarity with fluid dynamics will provide a somewhat intuitive approach to properly distributing the elements. Elements should change gradually in size from areas of high concentration to those of just a few large elements, changing no more than 50 percent in size between adjacent elements (Hess and Smith 1966). The accuracy is only as good as that provided by the largest element in a particular area. The use of quadrilateral elements facilitates the use of known analytical equations and body contours to determine the input points.



**Figure 4. The 3D quadrilateral element in global coordinates**

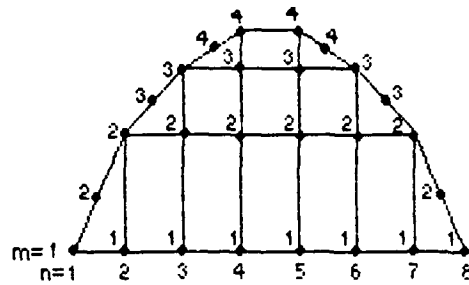
For the purposes of this program, the body surface is represented by a large number of plane quadrilateral elements as shown in figure (3), each of which is assumed to have a constant value of source density over the area of the element. Each element is defined by four input points

which lie on the actual surface as shown in figure (4). Since each input point can be used as a corner for up to four elements, there is no need to enter the same point four separate times. The input points are organized in groups of four to form the quadrilateral element, and each point may also be associated with adjacent quadrilaterals. This is accomplished through the use of a two dimensional coordinate system in which the user assigns a pair of integers,  $m$  and  $n$ , to each point which identifies the "row" and the "column" in which it lies. A column of points will be given a common value of  $n$ , and each point in that column will have a unique value of  $m$  corresponding to the row in which the point lies. The orientation of these "coordinate" integers determines the direction of the outward normal for each element. Looking from the flow field toward the section of elements, if the values of  $m$  are increasing upward, the values of  $n$  must increase to the right. Increasing  $m$  and  $n$  can point in any direction with respect to the global reference system. In fact, the orientation can change from one section to another. However, any other relationship between  $m$  and  $n$  will produce an incorrect normal vector. Once assigned, the values of  $m$  and  $n$  also serve to identify the element for which the corresponding point is a corner. The four points which form a quadrilateral element are two points of one column, or  $n$  line, with consecutive  $m$  numbers, and two points of the next higher  $n$  line with the same  $m$  numbers as the previous two points. Thus, the element  $m, n$  is composed of the points identified by  $(m, n)$ ,  $(m+1, n)$ ,  $(m, n+1)$ , and  $(m+1, n+1)$ .



**Figure 5. Quadrilateral index numbers**

Each section of the body surface is formed by specifying a set of corner points corresponding to the  $m, n$  pairs for all of the quadrilaterals of the section. The user will sequentially assign an  $m$  number to the points for each  $n$  line, and also number the  $n$  lines for the section points entered. The first point in each  $n$  line will always have  $m = 1$ . The  $n$  lines are also numbered sequentially, but the value of  $n$  is not reset for each new section. The sequence of  $n$  numbers runs through all the sections as shown in figure (5). Points on a particular row or column do not have to be strictly colinear. By forming nearly triangular elements, a rounded planform can be approximated without conflicting with the numbering convention, as shown in figure (6).



**Figure 6.** Approximating a thin region with rounded planform (Hess & Smith 1962)

By entering data in sections, small changes in geometry can be performed without having to reenter all the points associated with the body surface. This feature is unique to the XYZPF Program, and offers a great deal of flexibility in design work. However, with the added flexibility comes more restrictions on both the input of the original geometry and on any modifications. There are four important restrictions on the input which are required to provide quadrilateral elements in groups of four to facilitate geometry calculations (Dawson and Dean 1972):

(1) There must be an even number of elements in both the  $m$  and  $n$  directions in each section of the body.

(2) The common corner point of a group of four elements must not coincide with any other corner point. The sides between the elements serve to define the local coordinate system, and serve as the axis of rotation when the surface is flattened for numerical differentiation of

the velocity potential.

(3) Each set of four elements must have at least seven distinct corner points to allow the elements to more closely conform to a curved surface. This also allows for convergence of N-lines or M-lines as may occur, for example, at the leading edge of an ellipsoid. Thus, only two of the four quadrilateral may degenerate into triangles by having two of their corner points coincide. This does not necessarily eliminate the possibility of more than two "triangular" elements since the adjacent sides of a quadrilateral may be colinear as shown in figure (6).

(4) The normal vectors between two adjacent quadrilaterals in a group of four must be less than 90 degrees and preferably less than 45 degrees. If a sharp edge is required, it should be a concave corner with respect to the flow field, and the input should be arranged so that the edge is along an outside boundary of the groups of four, and not through the center.

When making small changes to the original geometry, the number of elements used in a new section must be the same as the number used in the original section unless the part being changed is at the end of the input data. Section configurations may be selected by natural divisions, as a matter of convenience to more easily handle large numbers of points, or as a tool to take advantage of symmetry.

In setting up input data to use planes of symmetry, it is important to note that the XYZPF Program has certain restrictions on the choice of

symmetry planes. The user only has the option to select the number of symmetry planes. The planes which will be used as symmetry planes are preselected by the program to optimize the calculation procedure. Therefore, knowing this, the preparation of input data must consider the following restrictions imposed by the program (Dean and Dawson 1972).

If only one plane of symmetry is used, the plane of symmetry is the  $y = 0$  plane of the global coordinate system. As such, all the  $y$  coordinates of the input points must be of the same sign, i. e., all positive or all negative. If the body is closed and intersects the plane of symmetry, the points touching the plane, i. e., corresponding to  $y = 0$ , must also be entered with the input points.

If two planes of symmetry are used, the planes of symmetry are the  $y = 0$  plane and the  $z = 0$  plane in the global coordinate system. Again, the  $y$  coordinates of all input points must have the same sign, positive or negative, and the  $z$  coordinates of all points must be of the same sign, positive or negative without regard to the sign of  $y$ . If the body surface intersects one or both of the planes of symmetry, the points which lie in the plane, i. e., those corresponding to  $y = 0$  or  $z = 0$ , must also be entered with the input points.

If three planes of symmetry are used, clearly the planes are the reference planes of the global coordinate system. As with the previous cases, all the  $x$  coordinates of the input points must be of the same sign, and similarly for the  $y$  and  $z$  coordinates. If any part of the body intersects any of the planes of symmetry, the points which lie in that



plane, i. e.,  $x = 0$ ,  $y = 0$  or  $z = 0$ , must also be entered with the input points.

## 5.2 SOURCE PANEL GEOMETRY

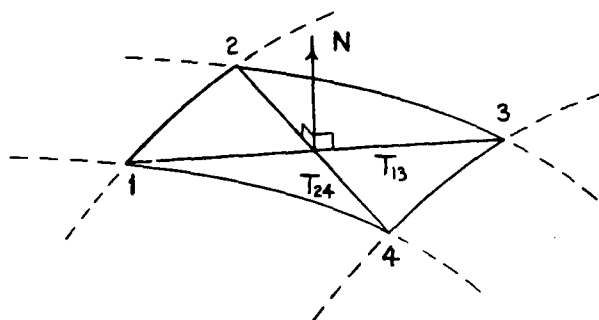


Figure 7. The outer normal to the quadrilateral element

With the surface points identified by the location numbers,  $m$  and  $n$ , and arranged in accordance with program requirements, calculation of various aspects of the source panel geometry and formation of the plane quadrilateral element is the next step in the numerical integration process. Formation of all of the planar elements is identical, so the following discussion of source panel geometry will deal with only one characteristic element. The four corner points forming the basic quadrilateral are numbered in a clockwise direction from 1 to 4 as shown in figure (7). It does not matter which corner point is identified with the number 1 subscript, but the remaining points must be numbered consecutively in a clockwise direction when observed from the flow field in order to ensure an outward directed normal vector. These subscripts will be used to identify the corner points for the remainder of this discussion. For this example, the points will be identified as follows:

<u>Position Numbers</u>	<u>Global Coordinates</u>
m, n	$X_1, Y_1, Z_1$
m+1, n	$X_2, Y_2, Z_2$
m+1, n+1	$X_3, Y_3, Z_3$
m, n+1	$X_4, Y_4, Z_4$

In forming the plane quadrilateral elements, the corner points, which are generally not coplanar, are used to form the local coordinate system, relative to the element. Recalling that the crossproduct of two vectors yields a vector solution which is perpendicular to both of the original vectors, the normal to the element may be obtained from the crossproduct of the diagonals of the element,

$$\mathbf{N} = \mathbf{T}_{24} \times \mathbf{T}_{13} \quad (21)$$

where  $\mathbf{T}_{13}$  is the vector from point 1 to point 3, and  $\mathbf{T}_{24}$  is the vector from corner point 2 to point 4. The unit normal is then:

$$\mathbf{n} = \frac{\mathbf{T}_{24} \times \mathbf{T}_{13}}{|\mathbf{T}_{24} \times \mathbf{T}_{13}|} \quad (22)$$

This unit normal now represents the first of the three local coordinate directions, this one in the  $\zeta$  direction. The side of the quadrilateral from point 2 to point 3 is then used to obtain the second coordinate vector.

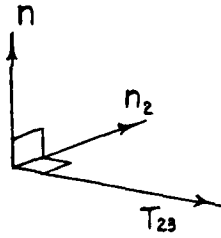


Figure 8. The second local coordinate vector

$$\mathbf{N}_2 = \mathbf{n} \times \mathbf{T}_{23} \quad (23)$$

and the unit vector

$$\mathbf{n}_2 = \frac{\mathbf{N}_2}{|\mathbf{N}_2|} \quad (24)$$

Similarly, the third local coordinate vector is obtained from the crossproduct of  $\mathbf{n}_2$  and  $\mathbf{n}$ , the result of which is a unit vector.

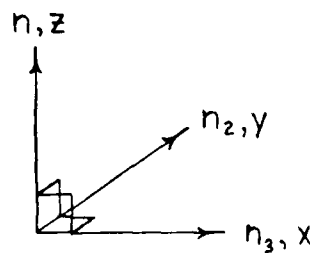


Figure 9. The third local coordinate vector

$$\mathbf{n}_3 = \mathbf{n}_2 \times \mathbf{n} \quad (25)$$

The unit vectors  $\mathbf{n}_3$ ,  $\mathbf{n}_2$ , and  $\mathbf{n}$  form an orthonormal basis and define the local coordinate system for the element in the  $\xi$ ,  $\eta$ , and  $\zeta$  directions

respectively. Other methods of obtaining an orthonormal basis could be used just as well, and would make no difference to the remaining computations. The origin of the local coordinate system would most correctly be located at the "null point," the point at which the velocity potential has no contribution to the tangential velocity component on the source element. The null point is the point in each quadrilateral element where the normal velocity boundary condition is applied. However, with the exception of long, thin quadrilaterals, the physical difference between the null point and the centroid of the quadrilateral is not significant. The XYZPF Program will print a warning in the output when a quadrilateral is long and thin enough to jeopardize the accuracy of the approximation in that region. By locating the origin of the local coordinate system at the centroid, rather than at the null point, the difficult process of locating the null point for each element can be eliminated, later calculations of the multipole expansion can be simplified, and the boundary conditions can be applied at the centroid without contributing significant error to the approximation (Hess and Smith 1966). Therefore, the origin for each local coordinate system is located at the centroid for the respective element.

### 5.3 LOCATING THE CENTROID

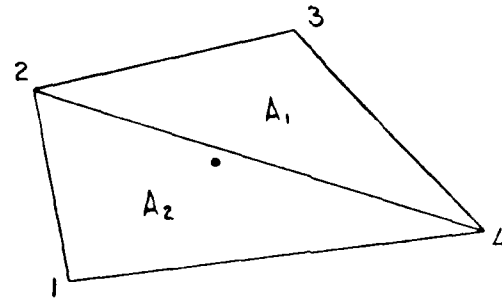


Figure 10. Locating the centroid of the quadrilateral

The centroid of the element may be calculated by first dividing the area of the quadrilateral into two triangular areas, the triangles being separated by the line from point 2 to point 4. The area  $A_1$  of the triangle defined by corner points 2, 3, and 4 is

$$A_1 = \frac{1}{2} |T_{24} \times T_{23}| \quad (26)$$

Similarly, the area  $A_2$  of the triangle defined by corner points 1, 2, and 4 is

$$A_2 = \frac{1}{2} |T_{12} \times T_{14}| \quad (27)$$

In the global coordinate system, the  $X$  component of the centroid is given by

$$\bar{X} = \frac{A_1 \bar{X}_1 + A_2 \bar{X}_2}{A_1 + A_2} \quad (28)$$

where  $X_1$  and  $X_2$  are the averages of the  $X$  components of the corner points of each triangle. Substituting the values for  $X_1$  and  $X_2$ :

$$\begin{aligned}
\bar{X} &= \frac{\frac{1}{3} A_1 (X_2 + X_3 + X_4) + \frac{1}{3} A_2 (X_1 + X_2 + X_4)}{A_1 + A_2} \\
&= \frac{1}{3} \left[ \frac{(A_1 + A_2) X_2 + (A_1 + A_2) X_4 + A_1 X_3 + A_2 X_1}{A_1 + A_2} \right] \\
&= \frac{1}{3} \left[ X_2 + X_4 + \frac{A_1 X_3 + A_2 X_1}{A_1 + A_2} \right] \quad (29)
\end{aligned}$$

Similarly

$$\bar{Y} = \frac{1}{3} \left[ Y_2 + Y_4 + \frac{A_1 Y_3 + A_2 Y_1}{A_1 + A_2} \right] \quad (30)$$

$$\bar{Z} = \frac{1}{3} \left[ Z_2 + Z_4 + \frac{A_1 Z_3 + A_2 Z_1}{A_1 + A_2} \right] \quad (31)$$

#### 5.4 COORDINATE TRANSFORMATION

Now that the local coordinate system is formed and properly located at the centroid of the element, the global coordinates of the corner points (X, Y, Z) are transformed to local coordinates ( $\xi, \eta, \zeta$ ) through the components of the reference vectors of the local coordinate system as follows:

$$\begin{bmatrix} n_{3x} & n_{3y} & n_{3z} \\ n_{2x} & n_{2y} & n_{2z} \\ n_x & n_y & n_z \end{bmatrix} \begin{bmatrix} X - \bar{X} \\ Y - \bar{Y} \\ Z - \bar{Z} \end{bmatrix} = \begin{bmatrix} \xi \\ \eta \\ \zeta \end{bmatrix} \quad (32)$$

The corner points are projected into the plane of the quadrilateral element by setting the  $\zeta$  components to zero. The original diagonal vectors,  $T_{13}$  and  $T_{24}$ , will be on opposite sides of the resulting plane. The plane quadrilateral element is now completely defined. The program will sweep through all of the input elements using the assigned location numbers, and repeat this process for each element.

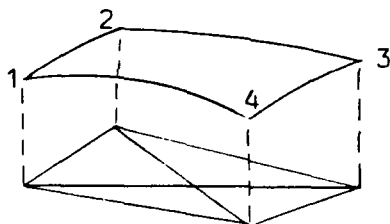


Figure 11. Forming the plane quadrilateral element

## 5.5 MOMENTS OF INERTIA

The calculation of the moments of inertia for each element are performed for use in the computation of the velocity coefficients using the quadrupole method. Any calculus text will give the moments of inertia of a planar section with a constant unit density about the origin to be:

$$I_{xx} = \iint_A \xi^2 d\xi d\eta \quad (33)$$

$$I_{yy} = \iint_A \eta^2 d\xi d\eta \quad (34)$$

$$I_{xy} = \iint_A \xi \eta d\xi d\eta \quad (35)$$

For the triangular region defined by the corner points 2, 3, and 4,

$$I_{xx} = \frac{A}{12} [(\xi_2 + \xi_3)^2 + (\xi_3 + \xi_4)^2 + (\xi_4 + \xi_2)^2] \quad (36)$$

$$I_{yy} = \frac{A}{12} [(\eta_2 + \eta_3)^2 + (\eta_3 + \eta_4)^2 + (\eta_4 + \eta_2)^2] \quad (37)$$

$$I_{xy} = \frac{A}{12} [(\xi_2 + \xi_3)(\eta_2 + \eta_3) + (\xi_3 + \xi_4)(\eta_3 + \eta_4) + (\xi_4 + \xi_2)(\eta_4 + \eta_2)] \quad (38)$$

Similar equations can be generated for the triangular region defined by the corner points 1, 2, and 4. The moment of inertia for the entire quadrilateral is the sum of the corresponding expressions for each of the triangles. The resulting equations are:

$$I_{xx} = \frac{A}{12} [(\xi_2 + \xi_3)^2 + (\xi_3 + \xi_4)^2 + (\xi_4 + \xi_2)^2] + \frac{A}{12} [(\xi_1 + \xi_2)^2 + (\xi_2 + \xi_4)^2 + (\xi_4 + \xi_1)^2] \quad (39)$$

$$I_{yy} = \frac{A}{12} [(\eta_2 + \eta_3)^2 + (\eta_3 + \eta_4)^2 + (\eta_4 + \eta_2)^2] + \frac{A}{12} [(\eta_1 + \eta_2)^2 + (\eta_2 + \eta_4)^2 + (\eta_4 + \eta_1)^2] \quad (40)$$

$$I_{xy} = \frac{A}{12} [(\xi_2 + \xi_3)(\eta_2 + \eta_3) + (\xi_3 + \xi_4)(\eta_3 + \eta_4) + (\xi_4 + \xi_2)(\eta_4 + \eta_2)] + \frac{A}{12} [(\xi_1 + \xi_2)(\eta_1 + \eta_2) + (\xi_2 + \xi_4)(\eta_2 + \eta_4) + (\xi_4 + \xi_1)(\eta_4 + \eta_1)] \quad (41)$$



## 6.0 THE MATRIX OF INFLUENCE COEFFICIENTS

With the quadrilaterals completely formed, the next step is to calculate the velocities induced by the elements at the centroids of all the other elements. The total number of elements forming the surface will be represented by  $N$ . Let the source element be the  $(j)$ th element, and the element for which the velocity components are to be calculated at the centroid is the  $(i)$ th element. It does not matter how the  $(i)$ th elements are arranged in relation to each other as the sequence progresses. However, the sequence must be consistent as the calculations proceed from one source element to another. This program sweeps through the  $(i)$ th elements in the order of their location numbers,  $m$  and  $n$ . For each consecutive  $n$  line, the elements are swept in order of increasing  $m$  numbers.

The result of the induced velocity calculations for the elements with unit source densities is an  $N$  by  $N$  square matrix of the values of induced velocities at each element, known also as the "matrix of influence coefficients." The XYZ potential flow program calculates the coefficients column by column, while the Hess program calculates them row by row. The advantage of one over the other depends on the method of later solving the matrix for the source densities. In calculating the influence coefficients, twenty-five quantities which describe the geometry of the source element are required to adequately calculate the induced velocity at the centroid of the  $(i)$ th element. These quantities include the coordinates of the centroid in the global coordinate system, the elements of the coordinate transformation matrix, the local

coordinates of the corner points, the maximum diagonal, the area, and the second moments of the quadrilateral element. Additionally, the Hess program uses the coordinates of the null point, making a total of twenty-eight quantities for that method (Hess and Smith 1962).

When calculating row by row, the first (i)th element is selected, containing the "null" point, and the influence coefficients are computed for all of the (j)th elements in sequence before proceeding to the (i+1)th element. This procedure requires the twenty-five quantities for each (j)th element to be available for calculation of the influence coefficient. Because each of the N (j)th elements is used N times with this procedure, calculating the geometric quantities or retrieving the values from low speed memory would be very time consuming, since the calculations or memory access would need to be performed  $N^2$  times. Therefore, it is more practical to have the values available in high speed memory, although large matrices may exceed the storage capacity of high speed memory, imposing a limit on the number of elements which can be used. The advantage to the row-by-row calculation is that solution of the resulting matrix by the Gauss-Seidel reduction method does not require transposing the matrix, which would be another time consuming process (Hess and Smith 1962).

Another alternative is calculation of the influence coefficients column by column. This method calculates the influence coefficients by sweeping all the (i)th elements for each (j)th element before proceeding to the (j+1)th element. Therefore, the twenty-five geometric quantities are retrieved from low speed storage only once for each (j)th element.

for a total of N times. This procedure is not limited by the capacity of high speed memory, and calculation of the coefficient matrix is approximately 10% faster than the row-by-row method (Dawson and Dean 1972). This is the calculation method used by the XYZ Potential Flow Program.

An influence coefficient represents the combined effects on one element of the velocity potentials of all the other elements comprising the body surface. For the quadrilateral element with a unit source density in the xy-plane, from equation (13), the potential at point P (x, y, z) due to the element is

$$\psi = \iint_A \frac{1}{r} dA = \iint_A \frac{d\xi d\eta}{\sqrt{(x - \xi)^2 + (y - \eta)^2 + z^2}} \quad (42)$$

The integrand,  $1/r$ , can be expanded in a series about the origin in powers of  $\xi$  and  $\eta$ . Each term of the series will contain the product of some powers of  $\xi$  and  $\eta$  with a corresponding derivative of  $1/r_0$ , where  $r_0$  is the distance of the field point P from the quadrilateral origin.

$$r_0 = \sqrt{x^2 + y^2 + z^2}$$

and let

$$w = \frac{1}{r_0}$$

Then the series expansion through the second order term is (Hess and Smith 1962):

$$\phi = Aw - (M_x w_x + M_y w_y) + 1/2(I_{xx} w_{xx} + 2 I_{xy} w_{xy} + I_{yy} w_{yy}) + \dots \quad (43)$$

The subscripts, x and y, used with w represent the respective partial derivatives. This series represents the multipole expansion of the velocity potential, since each term can be interpreted as a point singularity of a particular order. The first term is the potential at point P due to a point source of strength A located at the origin. The second term is the sum of two dipoles of strengths  $M_x$  and  $M_y$  located at the origin, oriented along the x and y axis respectively. The choice of the centroid of the quadrilateral as the origin of the local coordinate system causes the first moments,  $M_x$  and  $M_y$ , to be zero. Therefore, the dipole terms disappear, and are not dealt with anywhere in the program. The third term is the sum of three quadrupoles of strengths  $I_{xx}$ ,  $I_{xy}$ , and  $I_{yy}$  located at the origin. Kellogg (1929) shows that this second order approximation is absolutely and uniformly convergent, and Hess and Smith (1962) show that convergence is rapid enough with an increase in  $r_0$  that certain further approximations may be made without significant error at large distances  $r_0$  from the source quadrilateral.

Hess and Smith (1962) presented a comparison of velocities calculated using the exact formulas, a simple point source, and a second order approximation. The comparisons were based on the ratio of the distance  $r_0$ , between the centroid of the source quadrilateral and the field point P, to the length of the maximum dimension t, of the source quadrilateral, typically the maximum diagonal. The non-dimensional ratio is then  $r_0/t$ . The results show that sufficient accuracy is maintained

while using a simple point source at ratios of  $(r_0/t) \geq 4$ , using the second order source and quadrupole solution for the range  $2.45 \leq (r_0/t) < 4$ , and using the exact solution for ratios of  $(r_0/t) < 2.45$ . In any case, the error goes to infinity as the field point approaches the edge of the quadrilateral where calculations indicate an infinite velocity. The XYZ Potential Flow Program uses a monopole source for  $(r_0/t) > 4$ , the source - quadrupole formulae for  $2 < (r_0/t) \leq 4$ , and the exact formulae for  $(r_0/t) \leq 2$ . Hess and Smith (1962) reported a maximum error of 0.001 in approximating any velocity component using the above criteria.

## 7.0 DERIVATION OF THE EXACT SOURCE PANEL INTEGRATION

From equations (2) and (42), the components of the velocity at the field point  $P(x, y, z)$  due to the source quadrilateral are:

$$V_x = - \frac{\partial \phi}{\partial x} = \iint_A \frac{(x - \xi) d\xi d\eta}{r^3} \quad (44)$$

$$V_y = - \frac{\partial \phi}{\partial y} = \iint_A \frac{(y - \eta) d\xi d\eta}{r^3} \quad (45)$$

$$V_z = - \frac{\partial \phi}{\partial z} = \iint_A \frac{z d\xi d\eta}{r^3} \quad (46)$$

Equations (44), (45) and (46) are evaluated by expressing each of the integrals as the sum of four terms, each term representing the effect of one side of the quadrilateral (Hess and Smith 1962). This method can also be generalized for polygonal elements with any number of sides. The potential function for each side of the quadrilateral is the combined potentials of semi-infinite strips whose boundaries are the side of the quadrilateral and two semi-infinite lines parallel to either the  $x$  or  $y$  axis. When observed from the domain, and the sides are traversed in a clockwise direction, the source strip on the right will have a source density of  $\sigma = +1/2$  and the source strip on the left will have a source density of  $\sigma = -1/2$  as shown in figure (12). When the sides are recombined to form the quadrilateral, the source densities outside the quadrilateral cancel each other, and the source densities within the quadrilateral combine to form a source density of  $\sigma = +1$ . This will be

true for a planar element with any number of sides and in any relative orientation within the plane.

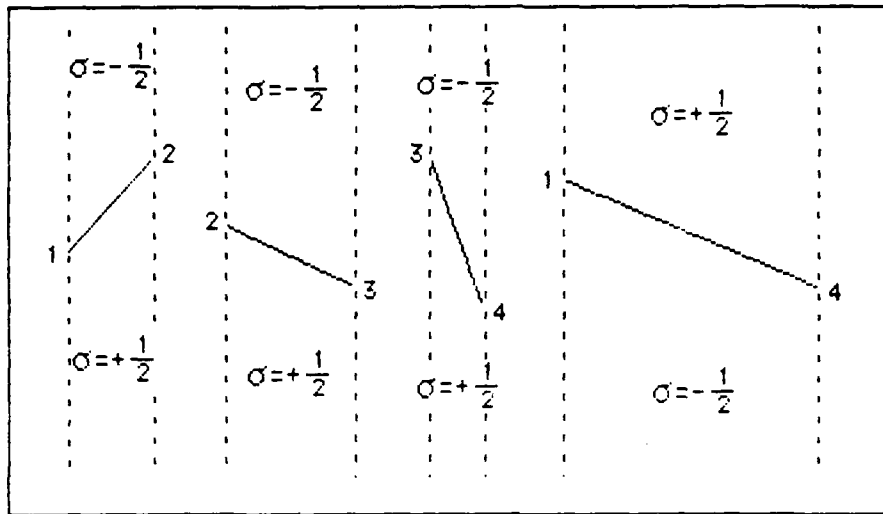


Figure 12. Fundamental potential functions for sides of a quadrilateral (Hess & Smith 1962)

## 7.1 THE Y VELOCITY COMPONENT

From equation (45), the velocity component  $V_y$  is found by summing the four terms representing the contributions of the sides of the quadrilateral. For the side from point  $(\xi_1, \eta_1)$  to point  $(\xi_2, \eta_2)$ , the contribution is expressed as the integral over the area of the semi-infinite strips with the source densities of  $\sigma = +1/2$  and  $\sigma = -1/2$  rather than the unit source density of equation (45).

$$V_{y_{12}} = \int_{\xi_1}^{\xi_2} d\xi \left[ \frac{1}{2} \int_{-\infty}^{\eta_{12}} - \frac{1}{2} \int_{\eta_{12}}^{\infty} \right] \frac{(y - \eta) d\eta}{r^3} \quad (47)$$

$$V_{y_{12}} = \frac{1}{2} \int_{\xi_1}^{\xi_2} d\xi \left[ \int_{-\infty}^{\eta_{12}} - \int_{\eta_{12}}^{\infty} \right] \frac{(y - \eta) d\eta}{[(x - \xi)^2 + (y - \eta)^2 + z^2]^{3/2}}$$

Integrating with respect to  $\eta$ :

$$V_{y_{12}} = \frac{1}{2} \int_{\xi_1}^{\xi_2} d\xi \left\{ \frac{1}{[(x - \xi)^2 + (y - \eta)^2 + z^2]^{1/2}} \Big|_{\eta_{12}} \right. \\ \left. - \frac{1}{[(x - \xi)^2 + (y - \eta)^2 + z^2]^{1/2}} \Big|_{-\infty} - \frac{1}{[(x - \xi)^2 + (y - \eta)^2 + z^2]^{1/2}} \Big|_{\infty} \right. \\ \left. + \frac{1}{[(x - \xi)^2 + (y - \eta)^2 + z^2]^{1/2}} \Big|_{\eta_{12}} \right\}$$

The terms evaluated at  $\eta = +\infty$  and  $\eta = -\infty$  cancel, and the terms evaluated at  $\eta = \eta_{12}$  add to obtain the following expression:

$$V_{y_{12}} = \frac{1}{2} \int_{\xi_1}^{\xi_2} \frac{2 d\xi}{[(x - \xi)^2 + (y - \eta)^2 + z^2]^{1/2}} \\ V_{y_{12}} = \int_{\xi_1}^{\xi_2} \frac{d\xi}{r} \quad (48)$$

Equation (48) is changed to a function of arclength  $s$  by the relation

$$\frac{d\xi}{ds} = \frac{\xi_2 - \xi_1}{\sqrt{(\xi_2 - \xi_1)^2 + (\eta_2 - \eta_1)^2}} = \frac{\xi_2 - \xi_1}{d_{12}} \quad (49)$$

where  $d_{12}$  is the length of the side of the quadrilateral from  $(\xi_1, \eta_1)$  to  $(\xi_2, \eta_2)$  as shown in figure (13).



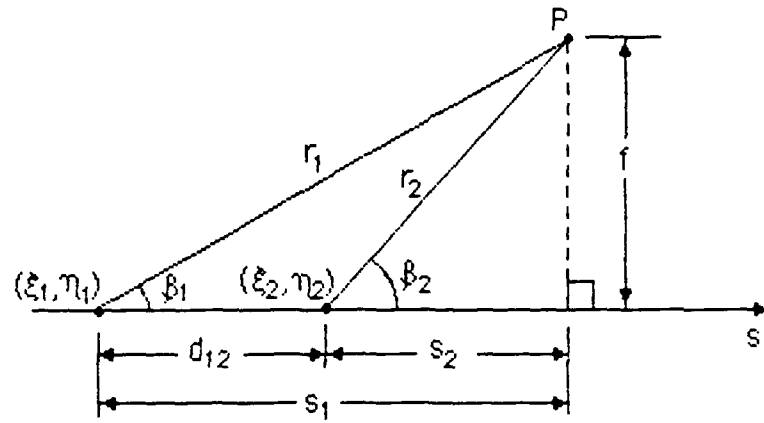


Figure 13. The potential due to a finite line source (Hess & Smith 1962)

Substituting equation (49) into equation (48)

$$V_{y_{12}} = \frac{\xi_2 - \xi_1}{d_{12}} \int_0^{d_{12}} \frac{ds}{r} \quad (50)$$

From figure (13), it can be seen that, in terms of arclength  $s$ , the distance  $r$  from point  $P$  to any point on the line from point 1 to point 2 is given by

$$r = \sqrt{f^2 + (s_1 - s)^2}$$

Substituting into equation (50) yields

$$\begin{aligned} V_{y_{12}} &= \frac{\xi_2 - \xi_1}{d_{12}} \int_0^{d_{12}} \frac{ds}{\sqrt{f^2 + (s_1 - s)^2}} \\ &= \frac{\xi_2 - \xi_1}{d_{12}} \int_0^{d_{12}} \frac{ds}{\sqrt{f^2 + (s - s_1)^2}} \quad (51) \end{aligned}$$

## Evaluating the integral

$$\begin{aligned}
 V_{y_{12}} &= \frac{\xi_2 - \xi_1}{d_{12}} \log \left[ (s - s_1) + \sqrt{f^2 + (s - s_1)^2} \right] \Big|_0^{d_{12}} \\
 &= \frac{\xi_2 - \xi_1}{d_{12}} \left\{ \log \left[ (d_{12} - s_1) + \sqrt{f^2 + (d_{12} - s_1)^2} \right] \right. \\
 &\quad \left. - \log \left[ (-s_1) + \sqrt{f^2 + (-s_1)^2} \right] \right\} \\
 &= \frac{\xi_2 - \xi_1}{d_{12}} \left\{ \log (r_2 - s_2) - \log (r_1 - s_1) \right\} \\
 &= \frac{\xi_2 - \xi_1}{d_{12}} \log \frac{(r_2 - s_2)}{(r_1 - s_1)} \quad (52)
 \end{aligned}$$

The quantities  $r_1$ ,  $r_2$ ,  $s_1$ , and  $s_2$  used in equation (52) are as shown in figure (13). Equation (52) is singular when  $r_1 = s_1$ , which occurs when the field point P is located anywhere along the line defined by the side of the quadrilateral. This singularity may be removed by using the law of cosines (Hess and Smith 1962).

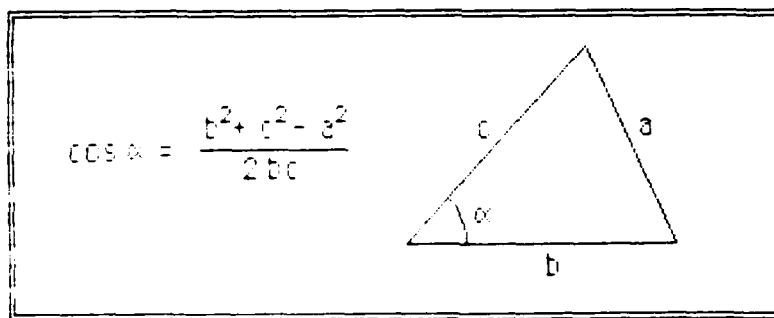


Figure 14. The law of cosines

From equation (52)

$$V_{y_{12}} = \frac{\xi_2 - \xi_1}{d_{12}} \log \frac{r_2(1 - \cos \beta_2)}{r_1(1 - \cos \beta_1)} \quad (53)$$

where  $\beta_1$  and  $\beta_2$  are the interior angles shown in figure (13). Applying the law of cosines to figure (13)

$$\cos \beta_1 = \frac{r_1^2 + d_{12}^2 - r_2^2}{2r_1d_{12}} \quad (54)$$

$$\cos \beta_2 = -\frac{r_2^2 + d_{12}^2 - r_1^2}{2r_2d_{12}} = \frac{r_1^2 - d_{12}^2 - r_2^2}{2r_2d_{12}} \quad (55)$$

From equations (54) and (55):

$$\begin{aligned} \frac{r_2(1 - \cos \beta_2)}{r_1(1 - \cos \beta_1)} &= \frac{r_2 \left[ 1 - \frac{r_1^2 - d_{12}^2 - r_2^2}{2r_2d_{12}} \right]}{r_1 \left[ 1 - \frac{r_1^2 + d_{12}^2 - r_2^2}{2r_1d_{12}} \right]} \\ &= \frac{r_2 \left[ \frac{2r_2d_{12} - r_1^2 + d_{12}^2 + r_2^2}{2r_2d_{12}} \right]}{r_1 \left[ \frac{2r_1d_{12} - r_1^2 - d_{12}^2 + r_2^2}{2r_1d_{12}} \right]} \\ &= \frac{2r_2d_{12} - r_1^2 + d_{12}^2 + r_2^2}{2r_1d_{12} - r_1^2 - d_{12}^2 + r_2^2} = \frac{(r_2 + d_{12})^2 - r_1^2}{-(r_1 - d_{12})^2 + r_2^2} \\ &= \frac{[(r_2 + d_{12}) + r_1][(r_2 + d_{12}) - r_1]}{[r_2 + (r_1 - d_{12})][r_2 - (r_1 - d_{12})]} = \frac{r_1 + r_2 + d_{12}}{r_1 + r_2 - d_{12}} \quad (56) \end{aligned}$$

Substituting equation (56) into equation (53) yields the final form of the exact equation of the y component of the velocity induced by the side of the quadrilateral from point 1 to point 2:

$$V_{y_{12}} = \frac{\xi_2 - \xi_1}{d_{12}} \log \frac{r_1 + r_2 + d_{12}}{r_1 + r_2 - d_{12}} \quad (57)$$

Equation (57) is applied to the remaining sides of the quadrilateral simply by substituting the appropriate point numbers for the corner points of each side. The total contribution of the quadrilateral to the y component of the velocity is the sum of the four terms representing the contributions of each of the sides. The y component of the velocity at the field point P is now given by:

$$\begin{aligned} V_y = & \frac{\xi_2 - \xi_1}{d_{12}} \log \frac{r_1 + r_2 + d_{12}}{r_1 + r_2 - d_{12}} + \frac{\xi_3 - \xi_2}{d_{23}} \log \frac{r_2 + r_3 + d_{23}}{r_2 + r_3 - d_{23}} \\ & + \frac{\xi_4 - \xi_3}{d_{34}} \log \frac{r_3 + r_4 + d_{34}}{r_3 + r_4 - d_{34}} + \frac{\xi_1 - \xi_4}{d_{41}} \log \frac{r_4 + r_1 + d_{41}}{r_4 + r_1 - d_{41}} \end{aligned} \quad (58)$$

## 7.2 THE X VELOCITY COMPONENT

A similar derivation process is used to produce the equation for the x component of the velocity induced by the side of the quadrilateral from point 1 to point 2. The semi-infinite source strips are constructed parallel to the x axis, and the order of integration is reversed.

The x component of the velocity at the field point P due to the quadrilateral is given by:

$$\begin{aligned}
 V_x = & \frac{\eta_2 - \eta_1}{d_{12}} \log \frac{r_1 + r_2 + d_{12}}{r_1 + r_2 - d_{12}} + \frac{\eta_3 - \eta_2}{d_{23}} \log \frac{r_2 + r_3 + d_{23}}{r_2 + r_3 - d_{23}} \\
 & + \frac{\eta_4 - \eta_3}{d_{34}} \log \frac{r_3 + r_4 + d_{34}}{r_3 + r_4 - d_{34}} + \frac{\eta_1 - \eta_4}{d_{41}} \log \frac{r_4 + r_1 + d_{41}}{r_4 + r_1 - d_{41}}
 \end{aligned} \quad (59)$$

### 7.3 THE Z VELOCITY COMPONENT

The z component of the velocity at the field point P due to the quadrilateral is obtained in a similar fashion, using semi-infinite source strips, this time parallel to the y axis. From equation (46), the fundamental velocity potential of the semi infinite source strips is integrated in a manner similar to that used to obtain equation (47), and the z component of the velocity due to the side from  $(\xi_1, \eta_1)$  to  $(\xi_2, \eta_2)$  is given by

$$V_{z12} = \int_{\xi_1}^{\xi_2} d\xi \left[ \frac{1}{z} \int_{-\infty}^{\eta_{12}} - \frac{1}{z} \int_{\eta_{12}}^{\infty} \right] \frac{z d\eta}{r^3} \quad (60)$$

$$V_{z12} = \frac{z}{2} \int_{\xi_1}^{\xi_2} d\xi \left[ \int_{-\infty}^{\eta_{12}} - \int_{\eta_{12}}^{\infty} \right] \frac{d\eta}{[(x - \xi)^2 + (y - \eta)^2 + z^2]^{\frac{3}{2}}}$$

Performing the integration with respect to  $\eta$ , the integral

$$\int \frac{d\eta}{[(x - \xi)^2 + (y - \eta)^2 + z^2]^{\frac{3}{2}}}$$

fits the integral form

$$-\int \frac{dF}{[C^2 + F^2]^n} = \frac{-1}{2C^2(n-1)} \left[ \frac{F}{[C^2 + F^2]^{n-1}} + (2n-3) \int \frac{dF}{[C^2 + F^2]^{n-1}} \right]$$

where

$$C^2 = (x - \xi)^2 + z^2$$

$$F = (y - \eta)$$

$$dF = -d\eta$$

$$n = 3/2$$

Then, from equation (60)

$$\begin{aligned} V_{z12} = & -\frac{z}{2} \int_{\xi_1}^{\xi_2} d\xi \left\{ \frac{(y - \eta)}{[(x - \xi)^2 + z^2][(x - \xi)^2 + (y - \eta)^2 + z^2]^{1/2}} \Big|_{\eta_{12}} \right. \\ & - \frac{(y - \eta)}{[(x - \xi)^2 + z^2][(x - \xi)^2 + (y - \eta)^2 + z^2]^{1/2}} \Big|_{-\infty} \\ & - \frac{(y - \eta)}{[(x - \xi)^2 + z^2][(x - \xi)^2 + (y - \eta)^2 + z^2]^{1/2}} \Big|_{\infty} \\ & \left. - \frac{(y - \eta)}{[(x - \xi)^2 + z^2][(x - \xi)^2 + (y - \eta)^2 + z^2]^{1/2}} \Big|_{\eta_{12}} \right\} \quad (61) \end{aligned}$$

Again, the terms evaluated at  $+\infty$  and  $-\infty$  cancel and the terms evaluated at  $\eta_{12}$  add to obtain the following expression:

$$V_{z12} = -z \int_{\xi_1}^{\xi_2} \frac{(y - \eta_{12}) d\xi}{[(x - \xi)^2 + z^2][(x - \xi)^2 + (y - \eta_{12})^2 + z^2]^{1/2}} \quad (62)$$

Without a convenient substitution with which to integrate equation (62), the integration is performed directly. Recognizing that along the line defined by the side of the quadrilateral from  $(\xi_1, \eta_1)$  to  $(\xi_2, \eta_2)$ ,  $\eta_{12}$  may be expressed as a function of  $\xi$ :

$$\eta_{12} = m_{12} \xi + b_{12} \quad (63)$$

where the slope of the side,  $m_{12}$  is given by

$$m_{12} = \frac{\eta_2 - \eta_1}{\xi_2 - \xi_1} \quad (64)$$

and  $b_{12}$  may be determined knowing that  $\eta_{12} = \eta_1$  when  $\xi = \xi_1$ .

$$b_{12} = \frac{\xi_2 \eta_1 - \xi_1 \eta_2}{\xi_2 - \xi_1} \quad (65)$$

Substituting equation (63) into equation (62) yields

$$V_{z_{12}} = -z \int_{\xi_1}^{\xi_2} \frac{(y - m_{12}\xi - b_{12}) d\xi}{[(x - \xi)^2 + z^2][(x - \xi)^2 + (y - m_{12}\xi - b_{12})^2 + z^2]^{1/2}} \quad (66)$$

Define the quantities

$$q_{12} = y - b_{12} - m_{12}x \quad (67)$$

$$u = x - \xi \quad (68)$$

Then  $du = -d\xi \quad (69)$

$$y - b_{12} - m_{12}\xi = m_{12}u + q_{12} \quad (70)$$

By a change of variable, equation (66) is now expressed as a function of  $u$ :

$$V_{z,12} = z \int_{x-\xi_1}^{x-\xi_2} \frac{(m_{12}u + q_{12}) du}{[u^2 + z^2][u^2 + (m_{12}u + q_{12})^2 + z^2]^{\frac{1}{2}}} \quad (71)$$

$$= z \int_{x-\xi_1}^{x-\xi_2} \frac{(m_{12}u + q_{12}) du}{[u^2 + z^2][(m_{12}^2 + 1)u^2 + 2m_{12}q_{12}u + q_{12}^2 + z^2]^{\frac{1}{2}}} \quad (72)$$

which fits the form of

$$\int \frac{(Lu + M) du}{(Au^2 + 2Bu + C)\sqrt{(au^2 + 2bu + c)}} \quad (73)$$

where

$$L = m_{12}$$

$$M = q_{12}$$

$$A = 1$$

$$a = (m_{12}^2 + 1)$$

$$B = 0$$

$$b = m_{12}q_{12}$$

$$C = z^2$$

$$c = q_{12}^2 + z^2$$

From Hardy (1944), this integral form may be integrated by the substitution

$$u = \frac{\mu t + v}{t + 1} \quad (74)$$

where  $\mu$  and  $v$  satisfy

$$a\mu v + b(\mu + v) + c = 0 \quad (75)$$

$$A\mu v + B(\mu + v) + C = 0 \quad (76)$$

and are the roots of the equation

$$(aB - bA)\xi^2 - (cA - aC)\xi + (bC - cB) = 0 \quad (77)$$



Substituting the appropriate values into equation (75), the roots of the quadratic equation are

$$\mu = -\frac{q_{12}}{m_{12}} \quad (78)$$

$$v = \frac{m_{12}z^2}{q_{12}} \quad (79)$$

It can be verified that these values satisfy equations (75) and (76).

Substituting equations (78) and (79) into equation (74)

$$u = \frac{\frac{m_{12}z^2}{q_{12}} - \frac{q_{12}t}{m_{12}}}{t + 1} \quad (80)$$

$$du = \left[ \frac{-\frac{q_{12}t}{m_{12}} - \frac{m_{12}z^2}{q_{12}}}{(t + 1)^2} \right] dt \quad (81)$$

By substitution and a change of variable, equation (72) becomes a function of the parameter  $t$ . After simplification, the integral now fits the form of

$$K \int \frac{dt}{(\alpha t^2 + \beta)\sqrt{(\delta t^2 + \epsilon)}} \quad (82)$$

where

$$K = -(m_{12}^6 q_{12} z^4 + 2m_{12}^4 q_{12}^3 z^2 + q_{12}^5 m_{12}^2)$$

$$\alpha = q_{12}^4 + m_{12}^2 q_{12}^2 z^2$$

$$\beta = m_{12}^4 z^4 + m_{12}^2 q_{12}^2 z^2$$

$$\gamma = q_{12}^4 + m_{12}^2 q_{12}^2 z^2$$

$$\delta = m_{12}^6 z^4 + m_{12}^4 z^4 + 2m_{12}^4 q_{12}^2 z^2 + m_{12}^2 q_{12}^4 + m_{12}^2 q_{12}^2 z^2$$

Equation (82) can be rationalized by the substitution

$$v = \frac{t}{\sqrt{\gamma t^2 + \delta}} \quad (83)$$

from which it can be shown that

$$t^2 = \frac{v^2 \delta}{1 - v^2 \gamma} \quad (84)$$

$$dt = \left[ \frac{\delta}{(1 - v^2 \gamma)^3} \right]^{1/2} dv \quad (85)$$

Substituting equations (84) and (85) into equation (82) and simplifying yields the integral in terms of the parameter  $v$ :

$$K \int \frac{dt}{(\alpha t^2 + \beta) \sqrt{(\gamma t^2 + \delta)}} = K \int \frac{dv}{\beta + (\alpha \delta - \beta \gamma) v^2} \quad (86)$$

which fits the form

$$\int \frac{dv}{a^2 + b^2 v^2} = \frac{1}{ab} \tan^{-1} \frac{bv}{a} \quad (87)$$

where  $a^2 = \beta$

$$b^2 = (\alpha \delta - \beta \gamma)$$

Performing the integration

$$K \int \frac{dv}{\beta + (\alpha\delta - \beta\gamma)v^2} = \frac{K}{\sqrt{\beta(\alpha\delta - \beta\gamma)}} \tan^{-1} \left[ v \sqrt{\frac{\alpha\delta - \beta\gamma}{\beta}} \right] \quad (86)$$

From equations (80), (83), and the expressions for  $\alpha$ ,  $\beta$ ,  $\delta$ , and  $\gamma$  from equation (82), and after a considerable amount of algebraic manipulation and simplification, equation (88) becomes

$$\begin{aligned} & \frac{K}{\sqrt{\beta(\alpha\delta - \beta\gamma)}} \tan^{-1} \left[ v \sqrt{\frac{\alpha\delta - \beta\gamma}{\beta}} \right] \\ &= -\frac{1}{z} \tan^{-1} \left[ \frac{m_{12}z^2 - q_{12}u}{z \sqrt{z^2 + u^2 + (m_{12}u + q_{12})^2}} \right] \quad (89) \end{aligned}$$

From equations (63) and (67), equation (89) becomes

$$\begin{aligned} & -\frac{1}{z} \tan^{-1} \left[ \frac{m_{12}z^2 - q_{12}u}{z \sqrt{z^2 + u^2 + (m_{12}u + q_{12})^2}} \right] \\ &= -\frac{1}{z} \tan^{-1} \left[ \frac{m_{12}(u^2 + z^2) - (y - \eta_{12})u}{z \sqrt{u^2 + (y - \eta_{12})^2 + z^2}} \right] \quad (90) \end{aligned}$$

Finally, applying these results to equation (71)

$$\begin{aligned}
 V_{z_{12}} &= z \int_{x-\xi_1}^{x-\xi_2} \frac{(m_{12}u + q_{12}) du}{[u^2 + z^2][u^2 + (m_{12}u + q_{12})^2 + z^2]}^{1/2} \\
 &= -\tan^{-1} \left[ \frac{m_{12}(u^2 + z^2) - (y - \eta_{12})u}{z \sqrt{u^2 + (y - \eta_{12})^2 + z^2}} \right] \Big|_{x-\xi_1}^{x-\xi_2} \\
 &= \tan^{-1} \left[ \frac{m_{12}((x - \xi_1)^2 + z^2) - (y - \eta_{12})(x - \xi_1)}{z \sqrt{(x - \xi_1)^2 + (y - \eta_{12})^2 + z^2}} \right] \\
 &\quad - \tan^{-1} \left[ \frac{m_{12}((x - \xi_2)^2 + z^2) - (y - \eta_{12})(x - \xi_2)}{z \sqrt{(x - \xi_2)^2 + (y - \eta_{12})^2 + z^2}} \right] \tag{91}
 \end{aligned}$$

Recall that when  $x = \xi_1$ ,  $y = \eta_1$  and when  $x = \xi_2$ ,  $y = \eta_2$ . Then, for the sake of a more compact equation, define the following quantities:

$$\begin{aligned}
 e_1 &= (x - \xi_1)^2 + z^2 & e_2 &= (x - \xi_2)^2 + z^2 \\
 h_1 &= (y - \eta_1)(x - \xi_1) & h_2 &= (y - \eta_2)(x - \xi_2)
 \end{aligned}$$

The quantities  $r_1$  and  $r_2$  are as shown in figure (12), where

$$r_1 = \sqrt{(x - \xi_1)^2 + (y - \eta_1)^2 + z^2} \quad r_2 = \sqrt{(x - \xi_2)^2 + (y - \eta_2)^2 + z^2}$$

Substituting these quantities into equation (91) yields the exact  $z$  component of velocity due to the side from point  $(\xi_1, \eta_1)$  to  $(\xi_2, \eta_2)$  in the form used by the XYZ Potential Flow Program:

$$V_{z_{12}} = \tan^{-1} \left[ \frac{m_{12}e_1 - h_1}{z r_1} \right] - \tan^{-1} \left[ \frac{m_{12}e_2 - h_2}{z r_2} \right] \tag{92}$$

The total z component of the velocity at the field point P(x, y, z) due to the quadrilateral element is the sum of the four sides:

$$\begin{aligned}
 V_z = & \tan^{-1} \left[ \frac{m_{12}e_1 - h_1}{z r_1} \right] - \tan^{-1} \left[ \frac{m_{12}e_2 - h_2}{z r_2} \right] \\
 & + \tan^{-1} \left[ \frac{m_{23}e_2 - h_2}{z r_2} \right] - \tan^{-1} \left[ \frac{m_{23}e_3 - h_3}{z r_3} \right] \\
 & + \tan^{-1} \left[ \frac{m_{34}e_3 - h_3}{z r_3} \right] - \tan^{-1} \left[ \frac{m_{34}e_4 - h_4}{z r_4} \right] \\
 & + \tan^{-1} \left[ \frac{m_{41}e_4 - h_4}{z r_4} \right] - \tan^{-1} \left[ \frac{m_{41}e_1 - h_1}{z r_1} \right]
 \end{aligned} \tag{93}$$

## 8.0 APPROXIMATIONS OF THE INDUCED VELOCITY

### 8.1 QUADRUPOLE METHOD

As previously mentioned, as the ratio of  $(r_0/t)$  exceeds the value of 2, then certain approximations may be made which greatly reduce the calculation effort otherwise required by the exact method. In the range of  $2 < (r_0/t) \leq 4$ , the XYZ Potential Flow Program uses the second order approximation of the potential described by equation (43). With the origin at the centroid of the quadrilateral, the first moments are zero, and the second order approximation is

$$\psi = Aw + (1/2)(I_{xx}w_{xx} + 2I_{xy}w_{xy} + I_{yy}w_{yy}) \quad (94)$$

where the first term is a point source of strength A, the second term is composed of three quadrupoles of strengths  $I_{xx}$ ,  $I_{xy}$ , and  $I_{yy}$  located at the local origin, and the subscripts on w indicate the partial derivatives of w with respect to those variables as before. The quantity A is the area of the element, and the terms  $I_{xx}$ ,  $I_{xy}$ , and  $I_{yy}$  are the respective moments of inertia of the source element given by equations (39), (40), and (41). To obtain the velocity components at the field point, equation (94) is differentiated with respect to the coordinate directions giving:

$$V_x = -\frac{\partial \phi}{\partial x} = -\left[ A W_x + \frac{1}{2} I_{xx} W_{xxx} + I_{xy} W_{xxy} + \frac{1}{2} I_{yy} W_{xyy} \right] \quad (95)$$

$$V_y = -\frac{\partial \phi}{\partial y} = -\left[ A W_y + \frac{1}{2} I_{xx} W_{xyx} + I_{xy} W_{xyy} + \frac{1}{2} I_{yy} W_{yyy} \right] \quad (96)$$

$$V_z = -\frac{\partial \phi}{\partial z} = -\left[ A W_z + \frac{1}{2} I_{xx} W_{xxz} + I_{xy} W_{xyz} + \frac{1}{2} I_{yy} W_{yyz} \right] \quad (97)$$

Recalling that  $w = \frac{1}{\sqrt{x^2 + y^2 + z^2}} = \frac{1}{r_0}$

the derivatives of  $w$ , as expressed by Hess and Smith (1962) and as used in the XYZPF program, are

$$\left. \begin{aligned} W_x &= -x r_0^{-3} \\ W_y &= -y r_0^{-3} \\ W_z &= -z r_0^{-3} \end{aligned} \right\} \quad (99)$$

$$\left. \begin{aligned} W_{xxx} &= 3x(3p + 10x^2) r_0^{-7} \\ W_{xxy} &= 3y p r_0^{-7} \\ W_{xyy} &= 3x q r_0^{-7} \\ W_{yyy} &= 3y(3q + 10y^2) r_0^{-7} \\ W_{xxz} &= 3z p r_0^{-7} \\ W_{xyz} &= -15xyz r_0^{-7} \\ W_{yyz} &= 3z q r_0^{-7} \end{aligned} \right\} \quad (100)$$

where

$$p = y^2 + z^2 - 4x^2$$

$$q = x^2 + z^2 - 4y^2$$

## 8.2 MONOPOLE METHOD

When the ratio of  $(r_0/t)$  is greater than 4, then the quadrilateral may be approximated by a simple source corresponding to the first term of equation (43). Then the velocity components at the field point due to the quadrilateral are given by

$$V_x = - \frac{\partial \psi}{\partial x} = - Aw_x \quad (101)$$

$$V_y = - \frac{\partial \psi}{\partial y} = - Aw_y \quad (102)$$

$$V_z = - \frac{\partial \psi}{\partial z} = - Aw_z \quad (103)$$

where the partial derivatives of  $w$  are those given in equation (99).



## 9.0 SOLVING THE MATRIX EQUATION FOR SOURCE DENSITY

### 9.1 JACOBI'S ITERATIVE METHOD

From equation (20), the matrix equation may be solved for the constant source density  $\sigma_i$  for each element which satisfies the boundary condition equation (11). Equation (20) suggests the use of Jacobi's iterative method of matrix solution in the form

$$\sigma_i^{(m+1)} = V_i + \sum_{\substack{j=1 \\ j \neq i}}^N C_{ij} \sigma_j^{(m)}, \quad i = 1, 2, \dots, N \quad (104)$$

where  $N$  is the number of elements composing the body surface, and  $m$  is the number of iterations completed. A partial sum of equation (104) is computed for each of the  $i$ th elements before proceeding to the next  $j$ th element. The iteration is complete when the summation of equation (104) includes all of the  $j$ th elements. Because the values of the source densities at all of the elements are recomputed before any of them are used in the iteration, this method is also called the simultaneous displacement method (Ralston 1965). This is contrasted with the Gauss-Seidel iterative method used in the Douglas program. In the Gauss-Seidel method, as each new  $\sigma_i$  is computed, it is used immediately in the iteration process for calculation of  $\sigma_{(i+1)}$ . This is also known as the successive displacement method and is expressed as

$$\sigma_i^{(m+1)} = V_i + \sum_{j=1}^{i-1} C_{ij} \sigma_j^{(m+1)} + \sum_{j=i+1}^N C_{ij} \sigma_j^{(m)} \quad (105)$$

$$i = 1, 2, \dots, N$$

Though the Gauss-Seidel iterative method is faster, the Jacobi iteration method was selected for use in the XYZPF program in order to be able to perform the iterations column by column, since the coefficient matrix is also computed column by column, and the matrix does not have to be transposed for solution.

When the (m+1)th iteration is complete, the values of the source densities are compared with those of the (m)th iteration and the differences summed for all of the elements. The total difference between successive iterations is then compared to a convergence criteria input by the user. If the difference is less than the convergence criteria, then the matrix solution is complete and the values of the source densities are stored for later use in computing velocities and pressure coefficients. If the convergence criteria is not met, then the iteration process is repeated. After every five iterations, if the convergence criteria is still not met, then an extrapolation is attempted in order to accelerate the convergence. The XYZ Potential Flow Program uses a Richardson extrapolation method, a numerical procedure which uses two approximate results to obtain a third approximation which is closer to the exact solution (Ralston 1965).

## 9.2 RICHARDSON EXTRAPOLATION

The Richardson extrapolation assumes that the iterative process is convergent. For the iterative solutions  $S_0$ ,  $S_1$ , and  $S_2$ , where  $S_0$  is the most recent approximation and  $S_2$  the oldest, the solution is convergent if

$$\frac{S_0 - S_1}{S_1 - S_2} = \lambda < 1 \quad (106)$$

While a Richardson-type extrapolation can take many forms, the XYZPF program uses a procedure developed from the following approximations (Dawson and Dean 1972). If there is only one dominant eigenvalue and a sufficient number of iterations have been completed, the iterative solutions may be approximated by

$$\begin{aligned} S_0 &\approx S_f + E \lambda^m \\ S_1 &\approx S_f + E \lambda^{m-1} \\ S_2 &\approx S_f + E \lambda^{m-2} \\ S_i &\approx S_f + E \lambda^{m-i} \end{aligned} \quad (107)$$

where  $S_f$  is the true solution  
 $\lambda$  is the eigenvalue  
 $E$  is the eigenfunction  
 $m$  is the number of completed iterations

Define the linear combination which, from equation (107), may be

approximated as

$$A S_0 + (1 - A) S_1 \approx S_f + E \lambda^{n-1} (A \lambda + 1 - A) \quad (108)$$

The value of A may be chosen such that

$$(A \lambda + 1 - A) = 0 \quad (109)$$

Then, from equations (108) and (109)

$$A S_0 + (1 - A) S_1 \approx S_f \quad (110)$$

where the expression on the left converges to the exact solution.

From equations (106) and (109)

$$\lambda = \frac{S_0 - S_1}{S_1 - S_2} = 1 - \frac{1}{A} \quad (111)$$

Solving for A,

$$A = \frac{S_2 - S_1}{S_0 - 2 S_1 + S_2} = \frac{S_2 - S_1}{D} \quad (112)$$

Since the value of A generally changes from element to element, a weighted average of A is used in the extrapolation, where

$$\bar{A} = \frac{\sum_{i=1}^N (S_2(i) - S_1(i)) (\text{sign of } D(i))}{\sum_{i=1}^N D(i)} \quad (113)$$

Equation (113) is recomputed after every fifth iteration. If the difference between the new value and the old value is less than 0.02, then the solution is extrapolated. From equation (110), the extrapolated solution is

$$S^* = \bar{A} S_0 + (1 - \bar{A}) S_1 \quad (114)$$

When there are two dominant eigenvalues, then the iterative solutions may be approximated by

$$S_i \approx S_f + E_1 \lambda_1^{m-i} + E_2 \lambda_2^{m-i} \quad (115)$$

where  $S_f$  is the true solution

$\lambda_1$  and  $\lambda_2$  are the eigenvalues

$E_1$  and  $E_2$  are the eigenfunctions

$m$  is the number of completed iterations

Define the linear combination which, from equation (115), may be approximated as

$$\begin{aligned} & B_2 S_0 + B_1 S_1 + (1 - B_1 - B_2) S_2 \\ & \approx S_f + E_1 \lambda_1^{m-2} [B_2 \lambda_1^2 + B_1 \lambda_1 + (1 - B_1 - B_2)] \\ & \quad + E_2 \lambda_2^{m-2} [B_2 \lambda_2^2 + B_1 \lambda_2 + (1 - B_1 - B_2)] \end{aligned} \quad (116)$$

The values of  $B_1$  and  $B_2$  may be determined for which the eigenvalues  $\lambda_1$  and  $\lambda_2$  are roots of the quadratic equation

$$B_2 \lambda^2 + B_1 \lambda + (1 - B_1 - B_2) = 0 \quad (117)$$

Then, from equation (116)

$$B_2 S_0 + B_1 S_1 + (1 - B_1 - B_2) S_2 \approx S_f \quad (118)$$

where the left side of the equation (118) converges to the exact solution.

Using equation (115) and eliminating terms containing  $E_2$ :

$$\begin{aligned}(S_0 - S_1) - \lambda_2 (S_1 - S_2) &= E_1 \lambda_1^{n-2} (\lambda_1 - \lambda_2) (\lambda_1 - 1) \\(S_1 - S_2) - \lambda_2 (S_2 - S_3) &= E_1 \lambda_1^{n-3} (\lambda_1 - \lambda_2) (\lambda_1 - 1) \\(S_2 - S_3) - \lambda_2 (S_3 - S_4) &= E_1 \lambda_1^{n-4} (\lambda_1 - \lambda_2) (\lambda_1 - 1)\end{aligned}\quad (119)$$

Solving for  $\lambda_1$

$$\lambda_1 = \frac{(S_0 - S_1) - \lambda_2 (S_1 - S_2)}{(S_1 - S_2) - \lambda_2 (S_2 - S_3)} = \frac{(S_1 - S_2) - \lambda_2 (S_2 - S_3)}{(S_2 - S_3) - \lambda_2 (S_3 - S_4)} \quad (120)$$

From equations (117) and (120)

$$B_1 = \frac{(S_4 - S_3)(S_0 - 2S_2 + S_4) - (S_4 - S_2)[(S_1 - S_2) - (S_3 - S_4)]}{D} \quad (121)$$

$$B_2 = \frac{(S_4 - S_2)(S_4 - 2S_3 + S_2) - (S_4 - S_3)[(S_1 - S_2) - (S_3 - S_4)]}{D} \quad (122)$$

where  $D = (S_4 - 2S_3 - S_2)(S_0 - 2S_2 + S_4) - (S_1 - S_2 - S_3 + S_4)^2$

The weighted averages of  $B_1$  and  $B_2$  are used for the extrapolation as done with  $A$  in equation (113). If the sum of the absolute values of the weighted averages of  $B_1$  and  $B_2$  changes by less than 2%, then the extrapolation is performed. Then from equation (118), the extrapolated solution is

$$S^* = \bar{B}_2 S_0 + \bar{B}_1 S_1 + (1 - \bar{B}_1 - \bar{B}_2) S_2 \quad (123)$$

## 10.0 CALCULATION OF VELOCITIES AND PRESSURE COEFFICIENTS

With the influence coefficients and the source densities determined, the calculation of velocities is a relatively simple matter. From equation (9), the total velocity is the sum of the freestream velocity and the disturbance velocity due to the body. The product of the source densities and the influence coefficients are summed for all of the elements, and then added to the freestream velocity to determine the total velocity at any point in the domain. Velocities on the surface of the body are calculated at the null points only, as the boundary conditions are enforced only at the null point of each element, and velocities at other points in the element would produce significant error due to the method of approximation. The components of the velocity at the centroid of the  $i$ th element are

$$\begin{aligned}V_{i_x} &= V_{\infty_x} + \sum_{j=1}^N C_{ij_x} \sigma_j \\V_{i_y} &= V_{\infty_y} + \sum_{j=1}^N C_{ij_y} \sigma_j \\V_{i_z} &= V_{\infty_z} + \sum_{j=1}^N C_{ij_z} \sigma_j\end{aligned}\tag{124}$$

From equation (15), the velocity induced by an element at its own null point has a magnitude of  $2\pi$  directed along the outward normal vector of the element.

At a point off the surface of the body, the components of the velocity are determined just as if the point of interest was a null point of a single element. The total velocity at the field point is the sum of the freestream velocity and the contributions of each of the elements of the body surface. The contribution of each of the elements is determined by calculating the influence coefficient based on the element geometry, and multiplying the result by the source density for the element. The total velocity at the field point may be expressed as

$$V_p = V_{\infty} + \sum_{q=1}^N C_{pq} \sigma_q \quad (125)$$

where p and q represent the field point and the source element respectively and the influence coefficient,

$$C_{pq} = \iint \frac{\partial}{\partial n} \left[ \frac{1}{r(p,q)} \right] d\Gamma \quad (126)$$

As discussed in section 6.0, the influence coefficient may be calculated by the exact method, or it may be approximated by the quadrupole or monopole method depending on the ratio of the distance,  $r_0$ , between the field point and the centroid of the source element to the maximum dimension,  $t$ , of the source element.



The magnitude of the velocity at either the on-body or off-body points is given by

$$|\mathbf{v}| = \sqrt{V_x^2 + V_y^2 + V_z^2} \quad (127)$$

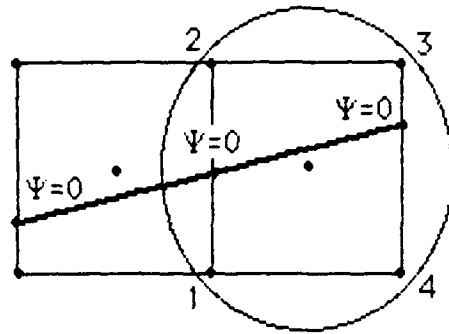
The pressure coefficient is calculated by using the result of equation (127) in equation (19), renumbered here for clarity.

$$C_p = \frac{p - p_\infty}{\frac{1}{2} \rho |\mathbf{v}_\infty|^2} = 1 - \frac{|\mathbf{v}|^2}{|\mathbf{v}_\infty|^2} \quad (128)$$

## 11.0 STREAMLINE CALCULATIONS

For the calculation of streamlines off the surface of the body, a timestep procedure is performed by calculating the velocity at the starting point of the streamline from equation (125), and advancing the streamline one time increment by a fourth order Runge-Kutta integration to a new point (Ralston 1965). The timestep procedure is repeated, thus creating a streamline composed of finite segments.

For the calculation of streamlines on the surface of the body, the streamline is started at a specified point and quadrilateral number. The local velocity is calculated from equation (124), and the values of a stream function are computed for each corner point. The stream function is chosen so that it has a value of zero at the last point on the streamline in the quadrilateral. The side of the quadrilateral through which the streamline exits is determined, and coordinates of the point on the side which has a stream function value of zero are computed. The direction of the streamline is verified by comparing it with the known direction of positive velocity. The next quadrilateral through which the streamline passes is determined by calculating the proximity of the new quadrilateral to the most recent point on the streamline. A circular area is computed which encloses the new quadrilateral with an additional 10% margin. If the last point of the streamline falls outside the circle, then the quadrilateral is discarded and a new one selected until the streamline is adjacent to the new quadrilateral.



**Figure 15. Calculation of on-body streamlines**

This procedure is repeated along the surface of the body until all of the surface elements have been tested. The result is a streamline composed of segments from one side of an element to another.

## 12.0 DEVELOPMENT OF HIGHER ORDER PANEL METHODS

The XYZ Potential Flow program assumes a constant element source panel as described in Section 3.3. Extensive use of the constant element source panel method has shown that the primary disadvantage of the method is that, in order to obtain a highly accurate solution, a large number of surface elements must be used to discretize the body surface. The method has been applied to problems of increasingly complex configurations (Hess 1977). By doing so, the size of the coefficient matrix is increased resulting in increased computer time and cost. Additional cost is accrued due to the manhours required to prepare the input. Therefore, while the constant element methods have proven to be very successful, the cost has motivated the development of higher order methods.

The higher order surface singularity methods discretize the body surface with curved elements having a variable source density, as compared to the flat elements of constant source strength used in the basic method. Hess (1973) showed that the effect of a curved surface and the effect of a variable source density are of the same order of magnitude. Therefore, the two effects must be used together to provide a "consistent" solution. The consistent higher order panel method provides the increased accuracy and speed desired for three dimensional Neumann problems (Hess 1979).

According to Hess (1979), the evolution of the higher order panel

method from the constant element method involved the derivation of new influence coefficients based on the integration of a variable source density over a curved element. Other portions of the method were unchanged. However, the development of the higher order velocity equations also required different programming logic.

In examining the potential for development of the higher order methods, Hess (1979) noted that "a consistent approach always uses a source polynomial one degree less than the panel polynomial." Through an independent effort, Brebbia (1984) presented a higher order approach using the direct method to solve for a surface potential polynomial stating that the potential function must be of a degree at least equal to the degree of the polynomial describing the element. Knowing that the velocity function is the derivative of the potential function, these two observations agree. As a result of his derivations, Hess showed that the solution of a flat element with a constant source requires one integral, a paraboloidal panel with a linearly varying source density requires six integrals, and a cubic element with a quadratic source density requires twenty-three integrals. Development of higher order methods has focused on the paraboloidal element with the linearly varying surface, as solutions of higher order than that offered little benefit for the amount of effort required to produce a working program (Hess 1979). Hess (1979) and Eriksson (1983) have independently developed programs for three dimensional higher order panel methods. The higher order Hess program evolved from the constant element program which he developed in the early 1960s, while Eriksson developed a new program based on the

work of Johnson and Rubbert (1975). Continued work in the near future is expected to deal primarily with refinement of the paraboloidal element with a linearly varying source (Eriksson 1983).

In order to alleviate the burden of preparing the input, a geometry package for input data generation has been developed which is incorporated into the Hess higher order panel program. This allows the user to enter relatively few points to describe the body. The geometry package enhances the surface representation by distributing additional points on the surface based on one of many algorithms or recurring geometries (Halsey 1978).

As the state of the art in fluid dynamics has progressed, the XYZ Potential Flow program has seen increasingly complex applications requiring a great deal of effort in preparing the input, and requiring long computer run times. Hess (1979) reported the use of the Hess constant element program for a configuration utilizing 7000 effective elements. Realizing that the computation time increases as the square of the number of elements, it is easy to see the motivation for developing the higher order panel methods. Though modern computers offer storage capacities which can handle most applications of the constant element method, the higher order panel methods can provide equal accuracy for much less user effort. While the constant element method is still a versatile tool, future generations of the surface singularity methods will be able to handle the more complex applications being demanded in fluid dynamics.

### 13.0 VELOCITY CALCULATIONS FOR A TRIAXIAL ELLIPSOID

As the only true body for which an analytical solution exists, a triaxial ellipsoid was selected for the sample calculations in order to compare calculated results with the analytical solution. Hess has made use of the triaxial ellipsoid throughout his works in developing both the constant element method and the higher order panel method. Therefore, the XYZPF program will be compared with existing results of the Hess method (Hess 1979).

The triaxial ellipsoid utilized for the calculations has semiaxes dimensions of 1, 2, and 0.5 in the x, y, and z directions respectively. The surface was discretized by selecting fixed intervals of 0.1 in the y direction, and fourteen equal divisions of the 90° sector in the x-z plane. The values of x and z were then solved in terms of y and an angle  $\theta$ . This method yielded 280 elements in the first octant for a total of 2240 effective elements after employing symmetry. A FORTRAN program was used to generate the corner points and to prepare the input file for later use by the XYZPF program.

Figures (16) and (17) show excellent correlation with the analytical solution and little difference from the Hess solution using 4320 effective elements. The use of the centroid as the control point is an approximation used to simplify the multipole expansion of the potential about the origin of the local coordinate system. This approximation is valid for most elements. However, for elements which are long and thin,

the physical difference between the centroid location and the null point location is significant, and use of the centroid can produce significant error as may be observed in figure (16) when the value of  $y$  approaches 2.0.

Recent calculations on the same body (Hess 1979) showed that results of at least equal accuracy could be obtained using only 480 effective elements using the higher order panel method. These results are a significant demonstration of the value of the higher order panel method. Using the higher order panel method rather than the constant element method, the user has the option of obtaining equal accuracy with cruder discretization or higher accuracy for the same discretization effort. While the results of the triaxial ellipsoid show relatively little improvement in accuracy, the most significant advantages of the higher order panel method are evident for a body with concave regions (Hess 1977).



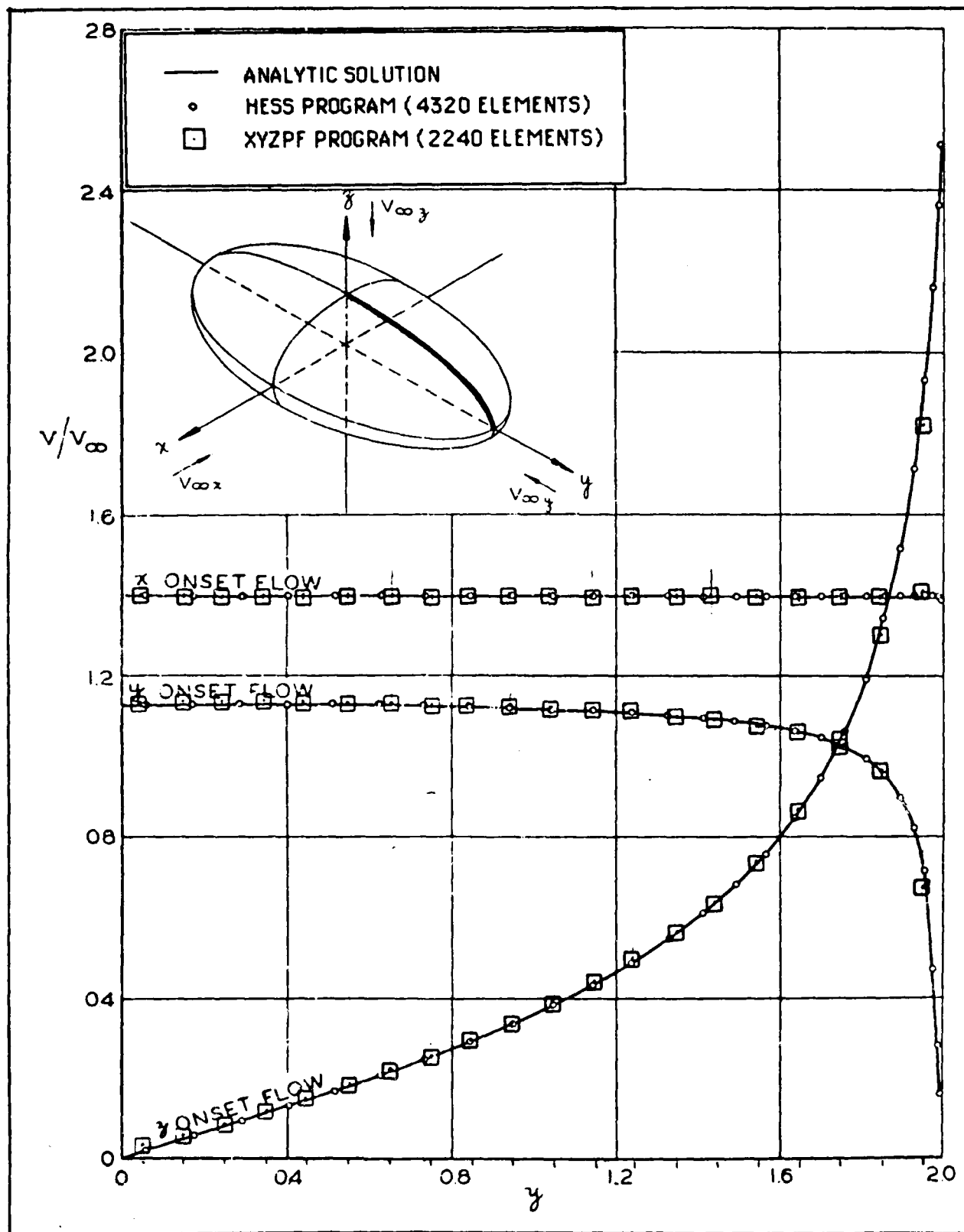


Figure 16. Comparison of analytic and calculated velocity distributions on an ellipsoid with axes ratios 1:2:0.5. Velocities in the  $xz$ -plane. (from Hess and Smith 1962)

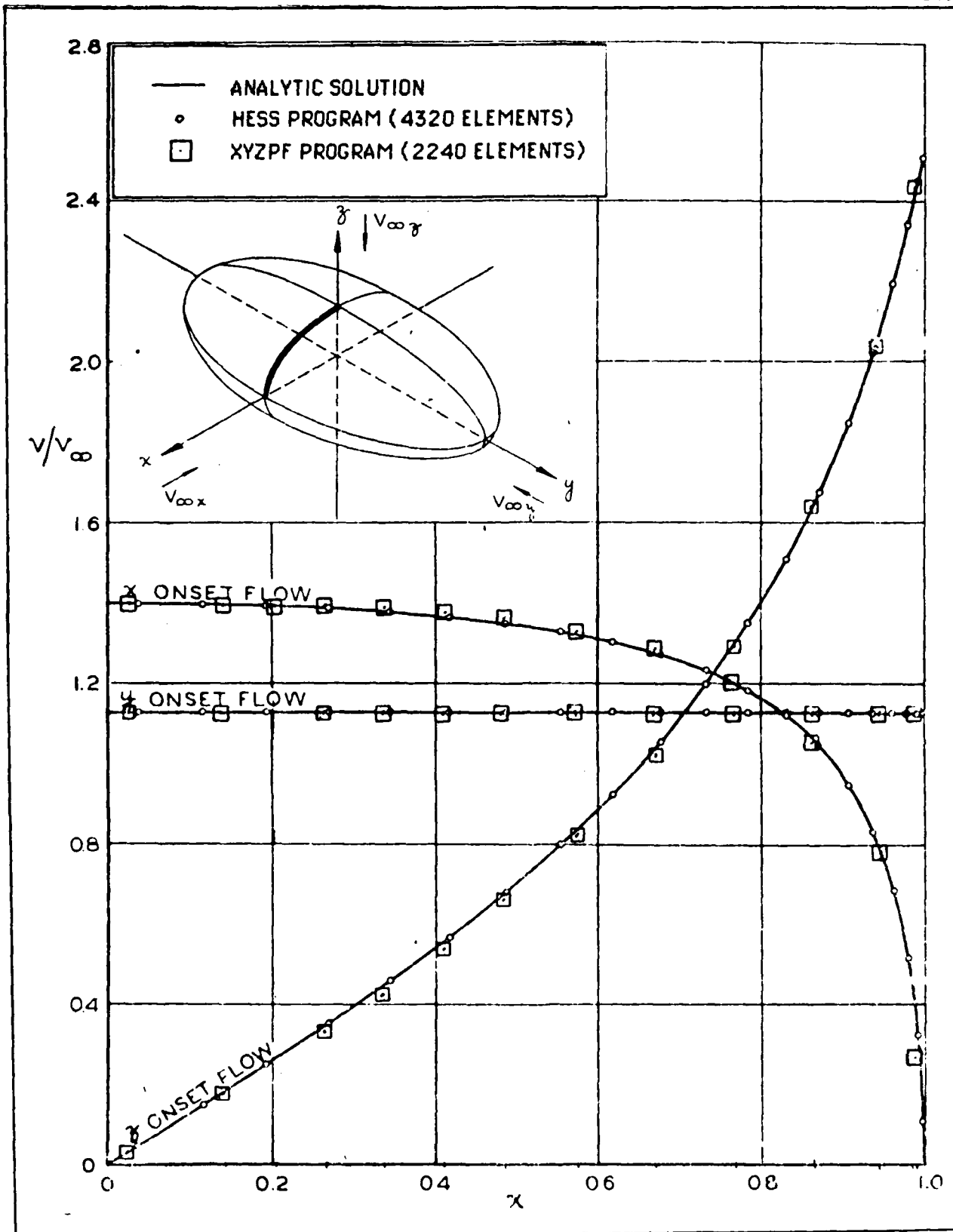


Figure 17. Comparison of analytic and calculated velocity distributions on an ellipsoid with axes ratios 1:2:0.5. Velocities in the yz-plane. (from Hess and Smith 1962)

## 14.0 CONCLUSION AND REMARKS

The objectives of this paper were (1) to describe the details of the approximation of an arbitrary three-dimensional body using quadrilateral elements, and (2) to provide a detailed derivation of the exact source panel integrations. Both of these objectives were met.

The method of surface discretization and source panel geometry is easily described using basic principles of geometry and vector algebra. By using quadrilateral surface elements, many surfaces can be discretized in a very straight forward logical fashion. The user can frequently visualize the contour lines of the surface which may be used to form the quadrilaterals, with some help from an intuitive approach to the fluid dynamics problem. The method of forming the planar quadrilateral element in the XYZPF Program differs slightly from the method presented by Hess and Smith (1962). The differences lie in the formation of the local coordinate system and the use of the centroid rather than the null point as the control point for applying the boundary conditions. The method of forming the local coordinate system has no effect on the potential flow calculations as long as one of the coordinate vectors is the outer normal to the planar element. The use of the centroid as the control point is an approximation used to simplify the multipole expansion of the potential about the origin of the local coordinate system. This approximation is valid for most elements. However, for elements which are long and thin, the physical difference between the centroid location and the null point location is significant, and use of the

centroid can produce significant error as may be observed in figure (16) when the value of  $y$  approaches 2.0.

A detailed derivation of the exact source panel integration has not previously appeared in literature, though the results are summarized by Hess and Smith (1962). The derivations presented in this paper verify the equations presented by Hess and Smith (1962), and the equations used in the XYZPF Program. The integral expressions for the velocity components were evaluated exactly with no assumptions or approximations used in the course of the integrations. Since the method of integration reduces the surface integral to a line integral around each of the sides of the element, the integration method can be generalized for a planar element with any number of sides, though the surface discretization used in the XYZPF Program uses only quadrilateral elements.

The calculation of potential flow about arbitrary three dimensional bodies is an engineering tool which is basic to design involving fluid dynamics. The XYZPF Program is a useful tool which has proven its value over the past 14 years. However, the increasing demands placed on this method are exposing the errors of the approximation as evident in the sample calculations presented in this paper. The requirement for increased accuracy has motivated the development of the higher order panel methods. Some of the limitations imposed on the XYZPF Program were due to computer memory and speed limitations. Advances in computer performance may allow future investigators to eliminate some

of the simplifying approximations used in the XYZPF Program, allowing increased accuracy without violating computer limitations. Some modifications might include the use of the null point as the control point rather than the panel centroid (as is used in the Hess program), or extending the range in which the exact velocity calculations are performed. The gains in accuracy by modifying the "constant element method" are limited by the basic approximations of the planar element and the constant source density for each element. Significant gains are most evident in the higher-order panel methods. This author concurs with Eriksson (1983) in expecting advances in the surface singularity methods to focus on the "development and refinement" of the higher order panel methods.

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**APPENDIX I - XYZPF SECTION PF1**

```

PROGRAM PPF1(INPUT=128,OUTPUT=128,TAPE5=INPUT,TAPE6=OUTPUT,TAPE03,
1      TAPE3=TAPE03,TAPE04,TAPE4=TAPE04,TAPE50=128)
C
C XYZ POTENTIAL FLOW PROGRAM VERSION 4 SECTION 1
C READS INPUT AND COMPUTES QUADRILATERAL PARAMETERS
C
C FOR INFORMATION CONTACT
C BILL CHENG OR JANET DEAN
C NUMERICAL FLUID DYNAMICS BRANCH CODE 1843
C NAVAL SHIP RESEARCH AND DEVELOPMENT CENTER
C BETHESDA, MARYLAND 20084
C
C DIMENSION INDEX(9,3),G(9,6),F(9),CZ(9),IP(9),XP(9),YP(9),ZF(9)
1      ,MSK(100),WS(240),PROB(15),DM(650)
COMMON X(800),Y(800),Z(800),ID(41,71),B(250),T(4600),KP(100)
EQUIVALENCE (CZ(1),F(1))
EQUIVALENCE(WS(1),KP(1)),(WS(101),MSK(1)),(WS(201),NF ),(WS(202),
2 NSP),(WS(203),NEP),(WS(204),NSE),(WS(205),MIX),(WS(206),MIY),
3 (WS(207),MIZ),(WS(209),IUCT),(WS(210),ISM),(WS(211),K),(WS(213),
4 EPS),(WS(208),IPS ),(WS(212),IPF ),(WS(217),XI),(WS(218),YI),
5(WS(219),ZI)
EQUIVALENCE (Y12,Y23),(Y34,Y41)
INTEGER P,P1,P2,P3,P4,PC,PS,P6,P7,P8 ,P9
WRITE (6,5)
5 FORMAT(49H1XYZ POTENTIAL FLOW PROGRAM SECTION 1, VERSION 4 )
10 FORMAT (11,15A4)
20 FORMAT (1X,15I4)
50 FORMAT (1X,2I7,6E12.5)
30 FORMAT (1X,5F12.9)
C A. READ IN CONTROL PARAMETERS
WS(220)=4.
K1=0
ID1=0
ID2=0
ID3=0
ID4=0
ID5=0
ID6=0
ID7=0
MAXN=70
MAXM=40
MAXNQE=650
MAXPC=800
ICNTAL=1
EOF50=0.
READ (5,10)J,(PROB(1),I=1,15)
IF (EOF(5) .EQ. 0.) GO TO 9
WRITE(6,8)
8 FORMAT(39HNO TITLE CARD FOUND - PROGRAM ABORTED )
STOP
9 CONTINUE
J=0
WRITE (6,10) J,(PROB(1),I=1,15)
SA=.0
SB=0.
21 FORMAT (17HNO. OF QUADS. =,14 /17H NO. OF SECTIONS=,
214/31H MAX. NO. OF ITERATIONS X FLOW ,13,9H Y FLOW ,
313,9H Z FLOW ,13)
READ (5,11)NQE,NSE,MIX,MIY,MIZ,ISM,EPS,IUCT,IPS,IPF,ISP

```

```

1      , IEDIT1, IEDIT3, IEDIT4, ITAPE, XCENTER, YCENTER, ZCENTER
11  FORMAT(6I4, F8.5, 8I4, 1X, 3F5.3)
    IF (EOF(5) .EQ. 0.) GO TO 19
    WRITE(6, 18)
18  FORMAT(43HONO PARAMETER CARD FOUND - PROGRAM ABORTED  )
    STOP
19  CONTINUE
    IF (IEDIT1.EQ.1) ICNTRL=0
    WRITE (6, 21) NQE, NSE, MIX, MIV, MIZ
31  FORMAT ("CONVERGENCE CRITERIA", F8.5)
41  FORMAT (4H0  M, 7X, 2HX1, 12X, 2HX2, 12X, 2HX3, 12X, 2HX4, 12X, 2HXP, 12X,
1  2HXN, 12X, 1HA, 13X, 3HCZ4/4H  N, 7X, 2HY1, 12X, 2HY2, 12X, 2HY3, 12X, 2HY4,
2  12X, 2HYP, 12X, 2HYN, 12X, 2HFL, 12X, 3HCZ5/4H  P, 7X, 2HZ1, 12X, 2HZ2, 12X,
3  2HZ3, 12X, 2HZ4, 12X, 2HZP, 12X, 2HZN, 12X, 4HCZ1  , 10X, 3HCZ6/)
42  FORMAT (4H1  M, 7X, 2HX1, 12X, 2HX2, 12X, 2HX3, 12X, 2HX4, 12X, 2HXP, 12X,
1  2HXN, 12X, 1HA, 13X, 3HCZ4/4H  N, 7X, 2HY1, 12X, 2HY2, 12X, 2HY3, 12X, 2HY4,
2  12X, 2HYP, 12X, 2HYN, 12X, 2HFL, 12X, 3HCZ5/4H  P, 7X, 2HZ1, 12X, 2HZ2, 12X,
3  2HZ3, 12X, 2HZ4, 12X, 2HZP, 12X, 2HZN, 12X, 4HCZ1  , 10X, 3HCZ6/)
24  FORMAT (1H , 11, 19H PLANES OF SYMMETRY)
240 WRITE (6, 24) ISM
270 WRITE (6, 31) EPS
280 IF (IFS.LE.0) GO TO 290
285 WRITE (6, 36) IPS, IPF
36  FORMAT (45HNEW SOURCE DENSITY TO BE COMPUTED FOR QUADS., 14, 3H - ,
114)
290 K=0
    WRITE(6, 39)ISP
39  FORMAT (9H0  ISP = , 13 )
    WRITE(6, 37) IEDIT1, IEDIT3, IEDIT4, ITAPE
37  FORMAT (9H0IEDIT1 = , 13/9H IEDIT3 = , 13/9H IEDIT4 = , 13/9H ITAPE = ,
1  13)
    WRITE(6, 38) XCENTER, YCENTER, ZCENTER
38  FORMAT(10HXCENTER = , F5.2/10H YCENTER = , F5.2/10H ZCENTER = , F5.2)
    MM=0
    MN=0
    P=1
    Q=1.0
    DO 291 I=1, 41
    DO 291 J=1, 71
291  ID(I, J)=0
    J=0
C      B. READ FIRST PT.
    IERR=0
    IF (ITAPE.EQ.1) GO TO 292
2000 READ (5, 40) X1, Y1, Z1, N1, M1, NS, NE, UN
    IF (EOF(5).NE.0. OR. NS.LE.0) GO TO 2050
    WRITE(6, 45) ICNTRL, NS
45  FORMAT(11, 9H SECTION , 14)
    LINE=0
    GO TO 293
2050 IF(IERR.EQ.0) GO TO 2200
    WRITE(6, 2100)
2100 FORMAT(39HONO POINT CARDS FOUND - PROGRAM ABORTED  )
    STOP
2200 IERR = 1
    ITAPE=1
    WRITE(6, 2300)
2300 FORMAT(47HERROR IN INPUT - POINT CARDS NOT ON INPUT FILE , 10X,
1  53HPROGRAM WILL CHANGE ITAPE TO 1 AND TRY TO READ TAPE50 )
    IF ( EOF(5).EQ.0 ) WRITE(6, 2400) X1, Y1, Z1
2400 FORMAT(11H0EXTRA FLOW, 3F12.5, 5X, 20HWILL NOT BE COMPUTED )

```

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292 READ(50,40) XI,YI,ZI,NI,MI,NS,NE,UN
40  FORMAT (3F12.9,4I4,F12.9)
    EOF50=EOF(50)
    IF (EOF50.NE.0 .OR. NS.LE.0) GO TO 2450
    WRITE(6,45) ICNTRL,NS
    LINE=0
    GO TO 293
2450 IF (IERR.EQ.0) GO TO 2500
    WRITE(6,2100)
    STOP
2500 IERR=1
    ITAPE=0
    WRITE(6,2600)
2600 FORMAT("ERROR IN INPUT - POINT CARDS NOT ON TAPE50",10X,
1"PROGRAM WILL CHANGE ITAPE TO 0 AND TRY TO READ INPUT FILE")
    GO TO 2000
293  UNR=UN
    NSS=NS
    PC=1
    IF (NE.EQ.0) GO TO 2700
    MMIN=NI
    MMAX=NI
    NMIN=NI
    NMAX=NI
    GO TO 300
2700 MMIN=MI
    MMAX=MI
    NMIN=NI
    NMAX=NI
    GO TO 300
295  IF (ITAPE.EQ.1) GO TO 297
    READ (5,40) XI,YI,ZI,NI,MI,NS,ME,UN
    IF (EOF(5).EQ.0.) GO TO 299
    NS=0
    XI=0.
    YI=0.
    ZI=0.
    GO TO 299
297  READ(50,40) XI,YI,ZI,NI,MI,NS,ME,UN
    EOF50=EOF(50)
    IF (EOF50.NE.0) NS=0
299  PC=PC+1
    IF (NS.NE.NSS) GO TO 330
300  IF (NE.EQ.0) GO TO 304
301  IW=NI
    NI=MI
    MI=IW
C    C. STORE PT. IN PT. ARRAY
304  IF (MAXPC+1-PC) 295,305,310
305  WRITE(6,306) NS,MI,NI
306  FORMAT(60H ERROR IN INPUT - THERE ARE TOO MANY DATA POINTS IN SEC
1TION ,14,30H - POINTS BEGINNING WITH M = ,14,5H N = ,14,
2 17H WILL BE IGNORED )
    LINE=LINE+1
    ID4=ID4+1
    GO TO 295
310  X(PC)=XI
    Y(PC)=YI
    Z(PC)=ZI
    IF (MI.LE.MAXM .AND. NI.LE.MAXN) GO TO 315
    WRITE(6,311) MI,NI

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LINE=LINE+1
311 FORMAT(38H ERROR IN INPUT - INVALID M,N INDICES ,10X,
1 14HPPOINT WITH M = ,14,5H N = ,14,17H WILL BE IGNORED )
ID5=ID5+1
PC=PC-1
GO TO 295
315 ID(M1,N1)=PC
MMAX=MAXO(MMAX,M1)
MMIN=MINO(MMIN,M1)
NMAX=MAXO(NMAX,N1)
NMIN=MINO(NMIN,N1)
GO TO 295
330 IF (IEDIT1.EQ.1) GO TO 294
IF (LINE.LT.40) GO TO 333
WRITE(6,42)
LINE=0
GO TO 294
333 WRITE(6,41)
294 CONTINUE
C E. DO LOOPS TO SWEEP PT. ARRAY
N1=NMIN
MM2=MMAX-MMIN
NN2=NMAX-NMIN
IF ( MOD(MM2,2).EQ.0 .AND. MOD(NN2,2).EQ.0) GO TO 332
WRITE(6,331) NSS,MMIN,MMAX,NMIN,NMAX
LINE=LINE+1
331 FORMAT(16HERROR - SECTION ,15,45H DOES NOT HAVE QUADS ARRANGED IN
1 BLOCKS OF 4 ,9H MMIN= ,12,6H MMAX= ,12,6H NMIN= ,12 ,
26H NMAX= ,12)
ID6=ID6+1
332 MM2=MM2/2
NN2=NN2/2
DO 404 NN=1,NN2
M1=MMIN
DO 402 MM=1,MM2
N0=1
C F. HAVE 9 CORNER PTS. BEEN GIVEN
IT=ID(M1,N1)*ID(M1+1,N1)*ID(M1+2,N1)*ID(M1,N1+1)*ID(M1+1,N1+1)*
1 ID(M1+1,N1+2)*ID(M1,N1+2)*ID(M1+1,N1+2)*ID(M1+2,N1+2)
IF( IT.EQ.0 ) GO TO 402
IERR=0
M2=M1+1
N2=N1+1 =M1,M2
DO 400 N=N1,N2
GO TO (334,335,336,337) N0
334 P1=ID(M ,N )
P2=ID(M+1,N )
P3=ID(M+1,N+1)
P4=ID(M ,N+1)
P5=ID(M+2,N )
P6=ID(M+2,N+1)
P7=ID(M+1,N+2)
P8=ID(M ,N+2)
P9=P1
IF((X(P1).NE.X(P2).OR.Y(P1).NE.Y(P2).OR.Z(P1).NE.Z(P2)) .AND.
1 (X(P1).NE.X(P4).OR.Y(P1).NE.Y(P4).OR.Z(P1).NE.Z(P4))) GO TO 340
P9=ID(M+2,N+2)
GO TO 340
335 P1=ID(M ,N+1)
P2=ID(M ,N )
P3=ID(M+1,N )

```

```

P4=ID(M+1,N+1)
P5=ID(M,N-1)
P6=ID(M+1,N-1)
P7=ID(M+2,N)
P8=ID(M+2,N+1)
P9=P1
IF((X(P1).NE.X(P2)).OR.Y(P1).NE.Y(P2)).OR.Z(P1).NE.Z(P2)) .AND.
1 (X(P1).NE.X(P4)).OR.Y(P1).NE.Y(P4)).OR.Z(P1).NE.Z(P4))) GO TO 340
P9=ID(M+2,N-1)
GO TO 340
336 P1=ID(M+1,N)
P2=ID(M+1,N+1)
P3=ID(M,N+1)
P4=ID(M,N)
P5=ID(M+1,N+2)
P6=ID(M,N+2)
P7=ID(M-1,N+1)
P8=ID(M-1,N)
P9=P1
IF((X(P1).NE.X(P2)).OR.Y(P1).NE.Y(P2)).OR.Z(P1).NE.Z(P2)) .AND.
1 (X(P1).NE.X(P4)).OR.Y(P1).NE.Y(P4)).OR.Z(P1).NE.Z(P4))) GO TO 340
P9=ID(M-1,N+2)
GO TO 340
337 P1=ID(M+1,N+1)
P2=ID(M,N+1)
P3=ID(M,N)
P4=ID(M+1,N)
P5=ID(M-1,N+1)
P6=ID(M-1,N)
P7=ID(M,N-1)
P8=ID(M+1,N-1)
P9=P1
IF((X(P1).NE.X(P2)).OR.Y(P1).NE.Y(P2)).OR.Z(P1).NE.Z(P2)) .AND.
1 (X(P1).NE.X(P4)).OR.Y(P1).NE.Y(P4)).OR.Z(P1).NE.Z(P4))) GO TO 340
P9=ID(M-1,N-1)
340 IP(1)=P1
IP(2)=P2
IP(3)=P3
IP(4)=P4
IP(5)=P5
IP(6)=P6
IP(7)=P7
IP(8)=P8
IP(9)=P9
C G2 COMPUTE NORMAL VECTOR (XN,YN,ZN)
X1=X(P3)-X(P1)
X2=X(P4)-X(P2)
Y1=Y(P3)-Y(P1)
Y2=Y(P4)-Y(P2)
Z1=Z(P3)-Z(P1)
Z2=Z(P4)-Z(P2)
XN=Y2*Z1-Y1*Z2
YN=X1*Z2-X2*Z1
ZN=X2*Y1-X1*Y2
R=SQ2(XN,YN,ZN)
IF (R.GT. .00000000001) GO TO 345
WRITE(6,343)
343 FORMAT(33H ERROR IN INPUT - ZERO AREA QUAD )
LINE=LINE+1
ID1=ID1+1
AQ=0.

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AD-A168 167

FORMULATION OF NUMERICAL METHODS USED IN THE XYZ  
THREE-DIMENSIONAL POTENT. (U) TEXAS A AND M UNIV  
COLLEGE STATION COLL OF ENGINEERING W J BEARY MAY 86

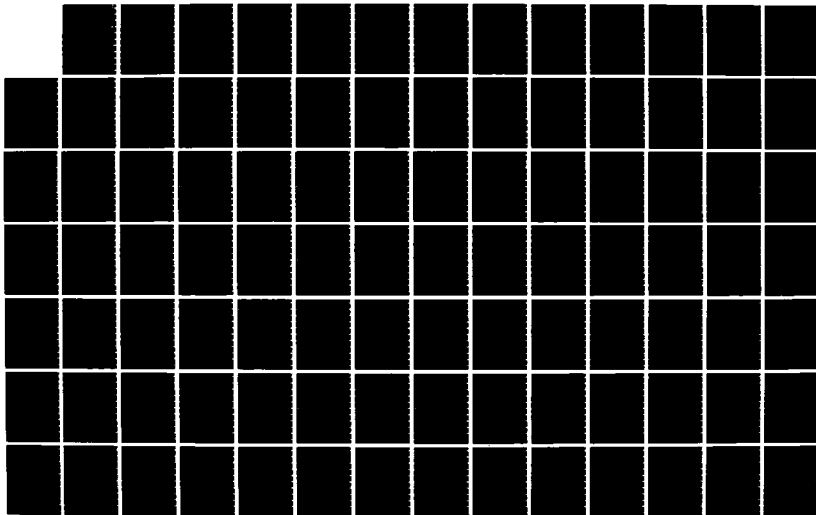
2/8

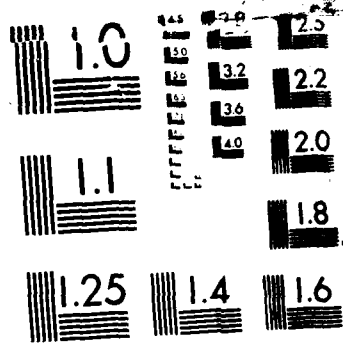
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F/G 20/4

NL





MICROCOPY RESOLUTION TEST CHART  
1010A



```

XC=0.
YC=0.
ZC=0.
FL=0.
CZ(1)=0.
CZ(4)=0.
CZ(5)=0.
CZ(6)=0.
IERR=1
GO TO 351
345 CONTINUE
XN=XN/R
YN=YN/R
ZN=ZN/R
A0=.5*R
C   COMPUTE CENTROID
X1=X(P3)-X(P2)
Y1=Y(P3)-Y(P2)
Z1=Z(P3)-Z(P2)
X5=Y1*Z2-Y2*Z1
Y5=Z1*X2-Z2*X1
Z5=X1*Y2-X2*Y1
A1=SQ2(X5,Y5,Z5)
A2=R-A1
IT=1
XC=(X(P2)+X(P4)+(A1*X(P3)+A2*X(P1))/R)/3.
YC=(Y(P2)+Y(P4)+(A1*Y(P3)+A2*Y(P1))/R)/3.
ZC=(Z(P2)+Z(P4)+(A1*Z(P3)+A2*Z(P1))/R)/3.
C   COMPUTE SECOND AND THIRD VECTORS
945 X4=YN*Z1-Y1*ZN
Y4=ZN*X1-Z1*XN
Z4=XN*Y1-X1*YN
A=1./SQ2(X4,Y4,Z4)
X4=X4*A
Y4=Y4*A
Z4=Z4*A
X3=ZN*Y4-Z4*YN
Y3=XN*Z4-X4*ZN
Z3=YN*X4-Y4*XN
C   COMPUTE POINTS IN QUAD SYSTEM
DO 947 I=1,9
L=I*(I)
XP(I)=X3*(X(L)-XC)+Y3*(Y(L)-YC)+Z3*(Z(L)-ZC)
YP(I)=X4*(X(L)-XC)+Y4*(Y(L)-YC)+Z4*(Z(L)-ZC)
947 ZP(I)=XN*(X(L)-XC)+YN*(Y(L)-YC)+ZN*(Z(L)-ZC)
C   COMPUTE MATRIX COEF. TO FIND SURFACE EQ.
DO 949 I=2,9
G(1,1)=1.
G(1,2)=XP(I)
G(1,3)=YP(I)
G(1,4)=XP(I)**2
G(1,5)=YP(I)**2
G(1,6)=YP(I)*XP(I)
949 F(I)=ZP(I)
DO 953 I=1,6
G(1,1)=G(9,1)
G(5,1)=G(5,1)+G(6,1)
953 G(6,1)=G(7,1)+G(8,1)
F(1)=F(9)
F(5)=F(5)+F(6)
F(6)=F(7)+F(8)

```

```

C      SOLVE MATRIX EQ.  G*CZ=F  FOR CZ
      CALL MATINS(G,9,6,F,6,1,DETERM,IDM,INDEX)
      IF (IDM.EQ. 1)      GO TO (955,960) IT
      IERR=1
      WRITE(6,954)
954  FORMAT (33H ERROR IN INPUT - SINGULAR MATRIX  )
      LINE=LINE+1
      ID2=ID2+1
      GO TO 960
955  IT=2
C      FIND NEW NORMAL VECTOR
      XN=XN-CZ(2)*X3-CZ(3)*X4
      YN=YN-CZ(2)*Y3-CZ(3)*Y4
      ZN=ZN-CZ(2)*Z3-CZ(3)*Z4
      A=1./SQ2(XN,YN,ZN)
      XN=XN*A
      YN=YN*A
      ZN=ZN*A
      GO TO 945
C      STORE DATA
960  B(J+1)=XP(1)
      B(J+2)=YP(1)
      B(J+3)=XP(2)
      B(J+4)=YP(2)
      B(J+5)=XP(3)
      B(J+6)=XP(4)
      B(J+7)=YP(4)
      B(J+8)=X3
      B(J+9)=Y3
      B(J+10)=Z3
      B(J+11)=X4
      B(J+12)=Y4
      B(J+13)=Z4
      B(J+14)=CZ(1)
      B(J+15)=CZ(4)
      B(J+16)=CZ(5)
      B(J+17)=CZ(6)
      IF (K.LT. 7*MAXNDE) GO TO 965
      ID7=ID7+1
      K1=K+K1
      K=0
965  CONTINUE
      T(K+1)=XC
      T(K+2)=YC
      T(K+3)=ZC
      T(K+4)=XN
      T(K+5)=YN
      T(K+6)=ZN
      T(K+7)=AQ
C      COMPUTE QUADRUPOLE MOMENTS
      X11=XP(1)+XP(2)
      X12=XP(1)+XP(4)
      X13=XP(3)+XP(2)
      X14=XP(3)+XP(4)
      X15=XP(2)+XP(4)
      Y11=YP(1)+YP(2)
      Y12=YP(1)+YP(4)
      Y13=YP(3)+YP(2)
      Y14=YP(3)+YP(4)
      Y15=YP(2)+YP(4)
      R1=R1/24.

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R2=R2/24.
R3=AQ/12.
RXX=(X11**2+X12**2)*R1+(X13**2+X14**2)*R2+X15**2*R3
RXY=(X11*Y11+X12*Y12)*R1+(X13*Y13+X14*Y14)*R2+X15*Y15*R3
RYY=(Y11**2+Y12**2)*R1+(Y13**2+Y14**2)*R2+Y15**2*R3
C   COMPUTE SOLID ANGLE
XX=XC-XCENTER
YY=YC-YCENTER
ZZ=ZC-ZCENTER
X1=XX*X3+YY*Y3+ZZ*Z3
Y1=XX*X4+YY*Y4+ZZ*Z4
Z1=XX*XN+YY*YN+ZZ*ZN
RD=1./SQ2(X1,Y1,Z1)
RCU=RD**3
RSU=RCU**2*RD
SA=SA+Z1*(AQ*RCU-((RXX*(Y1**2+Z1**2-4.*X1**2)
1   +RYY*(X1**2+Z1**2-4.*Y1**2))*1.5-15.*X1*Y1*AXY)*RSU)
B(J+18)=RXX
B(J+19)=RXY
B(J+20)=RYY
C   ERROR TESTS
D1=SQ2((XP(3)-XP(1)),(YP(3)-YP(1)),0.)
D2=SQ2((XP(4)-XP(2)),(YP(4)-YP(2)),0.)
FL=.5*AMAX1(D1,D2)
CZ23=ABS(CZ(2)) + ABS(CZ(3))
IF(ABS(CZ(2))+ABS(CZ(3)) .GT. FL*.001) GO TO 970
IF(ABS(CZ(1)) .LT. FL*.3) GO TO 977
970 WRITE(6,975) CZ23
975 FORMAT(29H QUESTIONABLE POINT -POOR FIT ,6E14.3)
IERR=1
LINE=LINE+1
977 IF (YP(1)*XP(2)-YP(2)*XP(1) .GE. 0. .AND.
1   YP(2)*XP(3)-YP(3)*XP(2) .GE. 0. .AND.
2   YP(3)*XP(4)-YP(4)*XP(3) .GE. 0. .AND.
3   YP(4)*XP(1)-YP(1)*XP(4) .GE. 0.) GO TO 984
980 WRITE(6,1000) (XP(1),YP(1), I=1,4)
1000 FORMAT(41H ERROR IN INPUT - CROSSED OR CONCAVE QUAD
1   4(2F10.5,3X) )
IERR=1
LINE=LINE+1
ID3=ID3+1
984 CRCF=SQ2((XP(2)-XP(1)),(YP(2)-YP(1)),0)+XP(3)-XP(2)+ SQ2((XP(1)-
1   XP(4)),(YP(1)-YP(4)),0.)+SQ2((XP(4)-XP(3)),(YP(4)-YP(3)),0.)
IF(35.*AQ .GT. CRCF**2) GO TO 986
LINE=LINE+1
WRITE(6,951)
961 FORMAT(24H WARNING LONG THIN QUAD.)
986 IF (Z1 .GE. 0.) GO TO 351
350 WRITE (6,35)
85 FORMAT(35H QUESTIONABLE POINT - INWARD NORMAL)
IERR=1
LINE=LINE+1
C   J. EDITE QUAD INFORMATION
351 IF (IEDIT1.EQ.2.AND. IERR.EQ.0) GO TO 354
IF (IEDIT1.EQ.1) GO TO 354
GO TO (356,357,358,359) NO
356 WRITE(6,51) M,X(P1),X(P2),X(P3),X(P4),XC,XN,AQ ,CZ(4) ,
1   N,Y(P1),Y(P2),Y(P3),Y(P4),YC,YN,FL ,CZ(5) ,
2   P,Z(P1),Z(P2),Z(P3),Z(P4),ZC,ZN,CZ(1),CZ(6)
GO TO 360
357 WRITE(6,51) M,X(P2),X(P3),X(P4),X(P1),XC,XN,AQ ,CZ(4) ,

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1          N,Y(P2),Y(P3),Y(P4),Y(P1),YC,YN,FL ,CZ(5) ,
2          P,Z(P2),Z(P3),Z(P4),Z(P1),ZC,ZN,CZ(1),CZ(6)
GO TO 360
358 WRITE(6,51) M,X(P4),X(P1),X(P2),X(P3),XC,XN,AQ ,CZ(4) ,
1          N,Y(P4),Y(P1),Y(P2),Y(P3),YC,YN,FL ,CZ(5) ,
2          P,Z(P4),Z(P1),Z(P2),Z(P3),ZC,ZN,CZ(1),CZ(6)
GO TO 360
359 WRITE(6,51) M,X(P3),X(P4),X(P1),X(P2),XC,XN,AQ ,CZ(4) ,
1          N,Y(P3),Y(P4),Y(P1),Y(P2),YC,YN,FL ,CZ(5) ,
2          P,Z(P3),Z(P4),Z(P1),Z(P2),ZC,ZN,CZ(1),CZ(6)
360 CONTINUE
51 FORMAT (1H ,13,8E14.5/1X,13,8E14.5/1X,13,8E14.5/)
LINE=LINE+4
IF (LINE.LT.50) GO TO 354
352 WRITE (6,42)
LINE=0
354 CONTINUE
J=J+20
I=P
DM(I)=UNR
P=P+1
NQ=NQ+1
IERR=0
349 K=K+7
C      K. WRITE OUT BLOCK OF B ARRAY IF FULL
IF (J.LT.240) GO TO 400
355 WRITE (04) Q,(B(I),I=1,240)
Q=P
J=0
C      L. END OF DO LOOP OVER PT. ARRAY
400 CONTINUE
402 M1=M1+2
404 N1=N1+2
C      M. SET FOR NEXT SECTION
NSS=NS
DO 405 M=MMIN,MMAX
DO 405 N=NM1N,NMAX
405 ID(M,N)=0
PC=1
IF (ME .EQ.0) GO TO 410
MMAX=NI
MMIN=NI
NMIN=MI
NMAX=MI
GO TO 420
410 MMAX=MI
MMIN=MI
NMIN=NI
NMAX=NI
420 NE=ME
UNR=UN
IF (NS.LE.0) GO TO 500
WRITE(6,45) ICNTRL,NS
LINE=0
GO TO 300
500 WRITE (04) Q,(B(I),I=1,240)
550 NP=(K+K1)/7
ISMP = ISM + 1
GO TO (595,590,580,570),ISMP
570 SA=SA+SA
580 SA=SA+SA

```

```

590 SA=SA+SA
C   01 WRITE PARAMETERS AND T ARRAY ON TAPE
595 J = 1
   IF (ITAPE.EQ.1 .AND. EOF50 .NE.0 )   GO TO 601
597 WS(J) = XI
   WS(J+20) = YI
   WS(J+40) = ZI
   J = J+1
   IF(XI**2 + YI**2 + ZI**2) 599,599,598
598 WRITE(6,600)  XI,YI,ZI
600 FORMAT(11H0EXTRA FLOW,10X,3F12.5)
601 READ(5,40)  XI,YI,ZI
   IF (EOF(5) .EQ. 0) GO TO 597
   XI=0.
   YI=0.
   ZI=0.
   GO TO 597
599 IF (ISP.LT.0) GO TO 605
   WRITE (03) (PROB(I),I=1,15)
   WRITE (03) (WS(I),I=1,220),IEDIT3,IEDIT4
   WRITE (03) (T(I),I=1,K)
   WRITE (03) (DM(I),I=1,NP)
605 CONTINUE
C   N1 CHECK SOLID ANGLE
610 WRITE (6,83) SA
   80 FORMAT(14HSOLID ANGLE = ,F8.3)
620 REWIND 04
   REWIND 03
622 IDS=ID1+ID2+ID3+ID4+ID5+ID6+ID7
   IF (IDS.EQ.0 .AND. NP.EQ.NQE) GO TO 638
   WRITE(6,625)
   IF (ID1 .GT.0) WRITE(6,628) ID1
   IF (ID2 .GT.0) WRITE(6,629) ID2
   IF (ID3 .GT.0) WRITE(6,630) ID3
   IF (ID4 .GT.0) WRITE(6,631) ID4
   IF (ID5 .GT.0) WRITE(6,632) ID5
   IF (ID6 .GT.0) WRITE(6,633) ID6
   IF (ID7 .GT.0) WRITE(6,634) NP,MAXNQE
   IF (NP.NE.NQE) WRITE(6,637) NP,NQE
   STOP
625 FORMAT(3SHOFATAL ERROR IN DATA - PROGRAM ABORTED)
628 FORMAT(1H0,15,31H  QUADRILATERALS WITH ZERO AREA          )
629 FORMAT(1H0,15,44H  QUADRILATERALS GENERATE A SINGULAR MATRIX  )
630 FORMAT(1H0,15,26H  CROSSED QUADRILATERALS                )
631 FORMAT(1H0,15,32H  SECTIONS HAVE TOO MANY POINTS        )
632 FORMAT(1H0,15,34H  POINTS HAVE INVALID M,N INDICES      )
633 FORMAT(1H0,15,52H  SECTIONS DO NOT HAVE QUADS ARRANGED IN GROUPS 0
   IF 4  )
634 FORMAT(1H0,15,48H  QUADRILATERALS GIVEN, EXCEEDING THE LIMIT OF ,
   1 14)
637 FORMAT(1H0,15,28H  QUADRILATERALS GIVEN, NOT ,14)
638 IF (ISP.LE.0) GO TO 640
   WRITE(6,639) ISP
639 FORMAT(7H0  ISP= ,14,19H - PROGRAM ABORTED  )
   STOP
640 CONTINUE
C   02 READ PEPS2 AND TRANSFER TO IT
   STOP 1
   END
C
   FUNCTION SQ2(X,Y,Z)

```

```

C COMPUTE SQUAR ROOT OF R**2
  R= ABS(X)+ABS(Y) +ABS(Z) +.0000000000001
700 RS=X**2+Y**2 +Z**2
  R=R+RS/R
  R=.25*R+RS/R
  R=R+RS/R
  SQ2=.25*R+RS/R
  RETURN
  END

C
C SUBROUTINE MATINS(A, NR, N1, B, NC, M1, DETERM, ID, INDEX)
C   MATRIX INVERSION WITH ACCOMPANYING SOLUTION OF SIMUL. EQ.
C   PIVOT METHOD
C   FORTRAN IV SINGLE PRECISION WITH ADJUSTABLE DIMENSION
C   NOVEMBER 1971 S GOOD NAVAL SHIP R & D CENTER
C   WHERE CALLING PROGRAM MUST INCLUDE
C     DIMENSION A(NR, NR), B(NR, NC), INDEX(NR, 3)
C     WHERE NR, NC ARE DIMENSIONS OF A, B, INDEX
C     N1 IS THE ORDER OF A
C     M1 IS THE NUMBER OF COLUMN VECTORS IN B (MAY BE 0)
C     DETERM WILL CONTAIN DETERMINANT ON EXIT
C     ID WILL BE SET BY ROUTINE TO 2 IF MATRIX A IS
C     SINGULAR, 1 IF INVERSION WAS SUCCESSFUL
C     MATRIX A (INPUT MATRIX) WILL BE REPLACED BY A INV
C     MATRIX B: THE COLUMN VECTORS WILL BE REPLACED
C     BY CORRESPONDING SOLUTION VECTORS
C     INDEX: WORKING STORAGE ARRAY
C   IF IT IS DESIRED TO SCALE, THE DETERMINANT CARD 29 MAY BE
C   DELETED AND DETERM PRESET BEFORE ENTERING THE ROUTINE
C
C   DIMENSION A(NR, NR), B(NR, NC), INDEX(NR, 3)
C   EQUIVALENCE (IROW, JROW), (ICOLUMN, JCOLUMN), (AMAX, T, SWAP)
C
C   INITIALIZATION
C
C     N=N1
C     M=M1
C     DETERM=1.0
C     DO 20 J=1, N
C
C 20 INDEX(J, 3)=0
C     DO 550 I=1, N
C
C   SEARCH FOR PIVOT ELEMENT
C
C     AMAX = 0.0
C     DO 105 J=1, N
C       IF<INDEX(J, 3)-1> 60, 105, 60
C 60 DO 100 K=1, N
C       IF<INDEX(K, 3)-1> 80, 100, 715
C 80 IF ( AMAX -ABS (A(J, K))) 85, 100, 100
C 85 IROW = J
C       ICOLUMN = K
C       AMAX = ABS (A(J, K))
C 100 CONTINUE
C 105 CONTINUE
C     INDEX(ICOLUMN, 3) = INDEX(ICOLUMN, 3) + 1
C     INDEX(1, 1) = IROW
C     INDEX(1, 2) = ICOLUMN
C
C   INTERCHANGE ROWS TO PUT PIVOT ELEMENT ON DIAGONAL

```

```

C      IF (IROW-ICOLUM) 140, 310, 140
140  DETERM= -DETERM
      DO 200 L=1,N
          SWAP= A(IROW,L)
          A(IROW,L)=A(ICOLUM,L)
200  A(ICOLUM,L)=SWAP
      IF (M) 310, 310, 210
210  DO 250 L=1,M
          SWAP=B(IROW,L)
          B(IROW,L)=B(ICOLUM,L)
250  B(ICOLUM,L)=SWAP
C
C      DIVIDE PIVOT ROW BY PIVOT ELEMENT
C
310  PIVOT = A(ICOLUM,ICOLUM)
      DETERM=DETERM*PIVOT
330  A(ICOLUM,ICOLUM) = 1.0
      DO 350 L=1,N
350  A(ICOLUM,L)=A(ICOLUM,L)/PIVOT
      IF (M) 380, 380, 360
360  DO 370 L=1,M
370  B(ICOLUM,L)=B(ICOLUM,L)/PIVOT
C
C      REDUCE NON-PIVOT ROWS
C
380  DO 550 L1=1,N
      IF (L1-ICOLUM) 400, 550, 400
400  T=A(L1,ICOLUM)
      A(L1,ICOLUM)=0.0
      DO 450 L=1,N
450  A(L1,L)=A(L1,L)-A(ICOLUM,L)*T
      IF (M) 550, 550, 460
460  DO 500 L=1,M
460  B(L1,L)=B(L1,L)-B(ICOLUM,L)*T
500  B(L1,L)=B(L1,L)-B(ICOLUM,L)*T
550  CONTINUE
C
C      INTERCHANGE COLUMNS
C
      DO 710 I=1,N
          L=N+1-I
          IF (INDEX(L,1)-INDEX(L,2)) 630, 710, 630
630  JROW= INDEX(L,1)
          JCOLUM=INDEX(L,2)
          DO 705 K=1,N
              SWAP = A(K,JROW)
              A(K,JROW)=A(K,JCOLUM)
              A(K,JCOLUM)=SWAP
705  CONTINUE
710  CONTINUE
          DO 730 K=1,N
              IF (INDEX(K,3)-1) 715, 720, 715
720  CONTINUE
730  CONTINUE
          ID=1
810  RETURN
715  ID=2
          GO TO 810
          END
/

```

**APPENDIX II - XYZPF SECTION PF2**

```

PROGRAM PF2(OUTPUT=128, TAPE6=OUTPUT, TAPE03, TAPE02, TAPE08,
1          TAPE09, TAPE01, TAPE11, TAPE04, TAPE4=TAPE04,
3          TAPE1=TAPE01,
2          TAPE3=TAPE03, TAPE2=TAPE02, TAPE8=TAPE08, TAPE9=TAPE09)
C
C   XYZ POTENTIAL FLOW PROGRAM VERSION 4 SECTION 2
C   COMPUTES MATRIX COEFFICIENTS
C
COMMON  B(241), T(4600), U1(1000), U2(1000), U3(1000), C1( 900), C2( 9
200), C3( 900),      WS(300), PROB(15),
3UX(8), UY(8), UZ(8)
EQUIVALENCE(Y3, Y2)      , (WS(201), NP), (WS(210), SYM), (WS(211), KM)
1      , (WS(212), IPF)  , (WS(208), IPS) , (KM, MK)
INTEGER SYM
BLK=1.0
IDN=0
READ(03)  (PROB(I), I=1, 15)
WRITE(6, 5)
5 FORMAT(49HXYZ POTENTIAL FLOW PROGRAM SECTION 2, VERSION 4 )
90 FORMAT(1H0, 15A4)
WRITE (6 ,90)      (PROB(I), I=1, 15)
C   A. READ PARAMETERS, T ARRAY, FIRST BLOCK OF B ARRAY
READ(03)  (WS(I), I=1, 220)
READ(03)  (T(I), I=1, MK)
READ(04)  (B(I), I=1, 241)
C   B. START LOOP OVER QUADRILATERALS
K=1
J=1
P=1
JC=1
JU=2
NPN=5*((NP+4)/5)
KMM=7*NPN+1
290 IF(B(1)-P)595, 295, 595
98 FORMAT("POINTS OUT OF ORDER B(1)=", 1F4.0, " P=", 1F4.0)
295 J=2
296 X1=B(J)
Y1=B(J+1)
X2=B(J+2)
Y2=B(J+3)
X3=B(J+4)
C   Y3=Y2
X4=B(J+5)
Y4=B(J+6)
XN=T(K+3)
YN=T(K+4)
ZN=T(K+5)
XC=T(K)
YC=T(K+1)
ZC=T(K+2)
A=T(K+6)
XX=B(J+7)
YX=B(J+8)
ZX=B(J+9)
XY=B(J+10)
YY=B(J+11)
ZY=B(J+12)
C   C1 COMPUTE LENGTH OF SIDES OF QUAD
D12=SQ2F(X1, X2, Y1, Y2, 0., 0.)
D23=SQ2F(X2, X3, Y2, Y3, 0., 0.)

```



```

      D34=SQ2F(X3,X4,Y3,Y4,.0,.0)
      D41=SQ2F(X4,X1,Y4,Y1,.0,.0)
C     C2 COMPUTE SLOPE OF SIDES
      IF(X2-X3)305,300,305
300  C123=1.
      GO TO 310
305  CM23=(Y2-Y3)/(X2-X3)
      C123=0.
310  IF(X3-X4)315,311,315
311  C134=1.
      GO TO 320
315  CM34=(Y4-Y3)/(X4-X3)
      C134=0.
320  IF(X4-X1)325,321,325
321  C141=1.
      GO TO 330
325  CM41=(Y1-Y4)/(X1-X4)
      C141=0.
330  IF(X1-X2)335,331,335
331  C112=1.
      GO TO 340
335  CM12=(Y2-Y1)/(X2-X1)
      C112=0.
C     C3 COMPUTE QUADRAPOLE MOMENTS
340  C1XX=B(J+17)
      C1XY=B(J+18)
      C1YV=B(J+19)
      CV12=0.0
      CX12=0.0
      CV23=0.0
      CX23=0.0
      CV34=0.0
      CX34=0.0
      CV41=0.0
      CX41=0.0
C     C4 COMPUTE SIN AND COS OF SLOPE ANGLE FOR EACH SIDE
      IF (D12) 9341,9342,9341
9341  CY12=(Y2-Y1)/D12
      CX12=(X1-X2)/D12
9342  IF(D23)9343,9344,9343
9343  CV23=(Y3-Y2)/D23
      CX23=(X2-X3)/D23
9344  IF(D34)9345,9346,9345
9345  CV34=(Y4-Y3)/D34
      CX34=(X3-X4)/D34
9346  IF(D41)9347,9348,9347
9347  CV41=(Y1-Y4)/D41
      CX41=(X4-X1)/D41
C     C5 COMPUTE MAX LENGTH OF QUAD
9348  ST=ABS(X1-X3)
      ST2=SQ2F(X2,X4,Y2,Y4,.0,.0)
      ST=AMAX1(ST,ST2,D12,D23,D34,D41)
C     D. START LOOP OVER NULL PTS
342  K0=1
      L=1
343  I=K0
      IF(I=PS) 9360,9360,9350
9350  IF(L=IPS) 9355,9352,9352
9352  IF(L=IPF) 9360,9360,9355
9355  C1(JC)=.0
      C2(JC)=.0
      C3(JC)=.0

```

```

GO TO 541
9360 IS=1
      XCQ=T(1)
      YCQ=T(1+1)
      ZCQ=T(1+2)
      XNQ=T(1+3)
      YNQ=T(1+4)
344 ZNQ=T(1+5)
C     E. COMPUTE DISTANCE BETWEEN QUAD AND NULL PT.
C     DETERMIN METHOD
345 RPQ=SQ2F(XC,XCQ,YC,YCQ,ZC,ZCQ)
      IF(RPQ-ST*4)350,350,460
350 X=(XCQ-XC)*XX+(YCQ-YC)*YX+(ZCQ-ZC)*ZX
      Y=(XCQ-XC)*XY+(YCQ-YC)*YY+(ZCQ-ZC)*ZY
      Z=(XCQ-XC)*XN+(YCQ-YC)*YN+(ZCQ-ZC)*ZN
      IF(RPQ-ST*2.0)355,355,400
C     F. COMPUTE VELOCITY COEF. BY EXACT METHOD
355 R1=SQ2F(X,X1,Y,Y1,Z,0.)
      R2=SQ2F(X,X2,Y,Y2,Z,0.)
      R3=SQ2F(X,X3,Y,Y3,Z,0.)
      R4=SQ2F(X,X4,Y,Y4,Z,0.)
      IF((R1+R2).LE.D12)GO TO 1000
      IF((R2+R3).LE.D23)GO TO 1000
      IF((R3+R4).LE.D34)GO TO 1000
      IF((R4+R1).LE.D41)GO TO 1000
      CLA1=ALOG((R1+R2-D12)/(R1+R2+D12))
      CLA2=ALOG((R2+R3-D23)/(R2+R3+D23))
      CLA3=ALOG((R3+R4-D34)/(R3+R4+D34))
      CLA4=ALOG((R4+R1-D41)/(R4+R1+D41))
      TUX=CX12*CLA1+CY23*CLA2+CY34*CLA3+CY41*CLA4
      TUY=CX12*CLA1+CX23*CLA2+CX34*CLA3+CX41*CLA4
      TUZ=0.
      IF(ABS(Z/ST)-.010)375,361,361
361 ZSQ=Z**2
      E1=ZSQ+(X-X1)**2
      E2=ZSQ+(X-X2)**2
      E3=ZSQ+(X-X3)**2
      E4=ZSQ+(X-X4)**2
      H1=(Y-Y1)*(X-X1)
      H2=(Y-Y2)*(X-X2)
      H3=(Y-Y3)*(X-X3)
      H4=(Y-Y4)*(X-X4)
      IF(C112)363,363,364
363 WS1=(C112*E1-H1)/(Z*R1)
      WS2=(C112*E2-H2)/(Z*R2)
      AT1=ATAN(WS1)
      AT2=ATAN(WS2)
      TUZ=AT1-AT2
364 IF(C123)365,365,367
365 AT3=ATAN((C123*E2-H2)/(Z*R2))
      AT4=ATAN((C123*E3-H3)/(Z*R3))
      TUZ=TUZ+AT3-AT4
367 IF(C134)368,368,369
368 AT5=ATAN((C134*E3-H3)/(Z*R3))
      AT6=ATAN((C134*E4-H4)/(Z*R4))
      TUZ=TUZ+AT5-AT6
369 IF(C141)370,370,375
370 AT7=ATAN((C141*E4-H4)/(Z*R4))
      AT8=ATAN((C141*E1-H1)/(Z*R1))
      TUZ=TUZ+AT7-AT8
375 GO TO 450
C     G. COMPUTE VELOCITY COEF. BY QUADRAPOLE METHOD

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```

400 RPQ3=RPQ**3
   RPQ7=(RPQ3**2)*RPQ
   WS1=X/RPQ3
   XSQ=X**2
   YSQ=Y**2
   ZSQ=Z**2
   PS=YSQ+ZSQ-4.*XSQ
   QS=XSQ+ZSQ-4.*YSQ
   WS2=X*(9.*PS+30.*XSQ)/RPQ7
   WS3=3.*Y*PS/RPQ7
   WS4=3.*X*QS/RPQ7
   TUX=A*WS1-C1XX*WS3-C1XY*WS2-C1YY*WS4
   WS1=Y/RPQ3
   WS2=Y*(9.*QS+30.*YSQ)/RPQ7
   TUY=A*WS1-C1XX*WS3-C1XY*WS4-C1YY*WS2
   TUZ=Z*(A/RPQ3-3.*(C1XX*PS-5.*C1XY*X*Y+C1YY*QS)/RPQ7)
450 UX(1S)=TUX*XX+TUY*XY+TUZ *XN
   UY(1S)=TUX*YX+TUY*YY+TUZ*YH
   UZ(1S)=TUX*ZX+TUY*ZY+TUZ*ZH
   GO TO 470
C     H. COMPUTE VELOCITY COEF. BY MONOPOLE METHOD
460 ARPQ3=A/(RPQ**3)
   UX(1S)= (XCQ-XC)*ARPQ3
   UY(1S)= (YCQ-YC)*ARPQ3
   UZ(1S)= (ZCQ-ZC)*ARPQ3
C     I. REFLECT NULL PT. IN PLANE OF SYMETRY
470 GO TO(480,485,490,495,500,505,510,515),1S
C     DO LOOPS SET UP TO FORCE USE OF INDEX REGISTERS
480 J1=JU
   J2=JC
   UDY=UX(1)
   UDY=UY(1)
   UOZ=UZ(1)
   U1(J1)=UX(1)
   U2(J1)=UX(1)
   U3(J1)=UX(1)
   U1(J1+1)=UY(1)
   U2(J1+1)=UY(1)
   U3(J1+1)=UY(1)
   U1(J1+2)=UZ(1)
   U2(J1+2)=UZ(1)
   U3(J1+2)=UZ(1)
   IF(SYM) 530,530,481
481 1S=2
C     XZ SYMETRY
   YCQ=-YCQ
   GO TO 345
485 IF(SYM-1)517,517,486
C     XY SYMETRY
486 1S=3
   ZCQ=-ZCQ
   GO TO 345
490 1S=4
   YCQ=-YCQ
   GO TO 345
495 IF(SYM-2)516,516,496
C     YZ SYMETRY
496 1S=5
   XCQ=-XCQ
   GO TO 345
500 1S=6
   YCQ=-YCQ

```

```

GO TO 345
505 IS=7
ZCQ=-ZCQ
GO TO 345
510 IS=8
YCQ=-YCQ
GO TO 345
C      J. ADD CONTRIBUTIONS OF ALL REFLECTIONS
515 U1(J1)=U1(J1)+UX(8)+UX(7)+UX(6)+UX(5)
      U2(J1)=U2(J1)-UX(8)+UX(7)+UX(6)-UX(5)
      U3(J1)=U3(J1)-UX(8)-UX(7)+UX(6)+UX(5)
      U1(J1+1)=U1(J1+1)-UY(8)+UY(7)+UY(6)-UY(5)
      U2(J1+1)=U2(J1+1)+UY(8)+UY(7)+UY(6)+UY(5)
      U3(J1+1)=U3(J1+1)+UY(8)-UY(7)+UY(6)-UY(5)
      U1(J1+2)=U1(J1+2)-UZ(8)-UZ(7)+UZ(6)+UZ(5)
      U2(J1+2)=U2(J1+2)+UZ(8)-UZ(7)+UZ(6)-UZ(5)
      U3(J1+2)=U3(J1+2)+UZ(8)+UZ(7)+UZ(6)+UZ(5)
516 U1(J1)=U1(J1)+UX(4)+UX(3)
      U2(J1)=U2(J1)+UX(4)-UX(3)
      U3(J1)=U3(J1)-UX(4)-UX(3)
      U1(J1+1)=U1(J1+1)+UY(4)-UY(3)
      U2(J1+1)=U2(J1+1)+UY(4)+UY(3)
      U3(J1+1)=U3(J1+1)-UY(4)+UY(3)
      U1(J1+2)=U1(J1+2)-UZ(4)-UZ(3)
      U2(J1+2)=U2(J1+2)-UZ(4)+UZ(3)
      U3(J1+2)=U3(J1+2)+UZ(4)+UZ(3)
517 U1(J1)=U1(J1)+UX(2)
      U2(J1)=U2(J1)-UX(2)
      U3(J1)=U3(J1)+UX(2)
      U1(J1+1)=U1(J1+1)-UY(2)
      U2(J1+1)=U2(J1+1)+UY(2)
      U3(J1+1)=U3(J1+1)-UY(2)
      U1(J1+2)=U1(J1+2)+UZ(2)
      U2(J1+2)=U2(J1+2)-UZ(2)
      U3(J1+2)=U3(J1+2)+UZ(2)
530 C1(J2)=XNQ*U1(J1)+YNO*U1(J1+1)+ZNO*U1(J1+2)
      C2(J2)=XNQ*U2(J1)+YNO*U2(J1+1)+ZNO*U2(J1+2)
      C3(J2)=XNQ*U3(J1)+YNO*U3(J1+1)+ZNO*U3(J1+2)
540 JW=JW+3
541 JC=JC+1
C      D. WRITE COEFFICIENTS
C      K. WRITE COEF. ON TAPE OR DRUM IF STORAGE AREA IS FULL
545 IF(JW-100)570,555,555
555 JW=2
      U1(1)=BLK
      U2(1)=BLK
      U3(1)=BLK
      IF(BLK-636.0) 560,553,566
553 WRITE (01) BLK,U1,U2,U3
      GO TO 568
563 REWIND 01
566 WRITE(11) BLK,U1,U2,U3
568 BLK=BLK+1.
570 IF(JC-90)580,571,571
571 IDW=IDW+1
      WRITE (02) IDW,C1
      WRITE (08) IDW,C2
      WRITE (09) IDW,C3
576 JC=1
580 KO=KO+7
      L=L+1
C      L. END OF LOOP OVER NULL PTS.

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```

      IF(KQ-KM )343,343,581
581 C1(JC)=0
      C2(JC)=0
      C3(JC)=0
      IF(KQ-KMM)541,585,585
585 P=P+1
      K=K+7
      J=J+20
      IF(K-KM )586,586,600
C      N. END OF LOOP OVER QUADS.
C      M. READ NEXT BLOCK OF B ARRAY IF NEEDED
586 IF(J-241)296,590,590
590 READ(04)(B(I),I=1,241)
      J=2
      IF(B(1)-P)595,296,595
595 WRITE (6,98) B(1),P
      STOP
600 IF(BLK-636.0) 610,620,630
C      O. WRITE REMAINING COEF. ON TAPE OR DRUM
610 WRITE(01)BLK,U1,U2,U3
      REWIND 01
      GO TO 640
620 REWIND 01
630 WRITE(11) BLK,U1,U2,U3
      REWIND 11
640 WRITE (02) IDW,C1
      WRITE (08) IDW,C2
      WRITE (09) IDW,C3
      REWIND 02
      REWIND 03
      REWIND 04
      REWIND 08
      REWIND 09
C      P. TRANSFER TO PFP53
      GO TO 5000
1000 WRITE(6,2000) L,P
2000 FORMAT(3H L= ,15,20X,3H P= ,F5.1)
5000 CONTINUE
      STOP 2
      END
      FUNCTION SQZF(X1,X2,Y1,Y2,Z1,Z2)
      X=X1-X2
      Y=Y1-Y2
      Z=Z1-Z2
      RS=Z**2+Y**2+X**2
      R=ABS(X)+ABS(Y)+ABS(Z)+ 1.0E-20
      R=R+RS/R
      R=.25*R+RS/R
      R= R+RS/R
      SQZF= .25*R+RS/R
      RETURN
      END

```

**APPENDIX III - XYZPF SECTION PF3**

```

PROGRAM PFP3(OUTPUT=128, TAPE02, TAPE08, TAPE09,
1      TAPE12, TAPE03, TAPE6=OUTPUT, TAPE2=TAPE02,
2      TAPES=TAPE08, TAPE9=TAPE09, TAPE3=TAPE03)
C
C      XYZ POTENTIAL FLOW PROGRAM VERSION 1 SECTION 3
C      SOLVES MATRIX EQUATION FOR SOURCE DENSITY
C
COMMON  SN(654),VIP(650),S(5,650),PROB(15),WS(220),DM(650),
1      B(220),COEF(900),XN(650),YN(650),ZN(650)
EQUIVALENCE (WS(213),EPS), (KK,B(201))
EQUIVALENCE (MIX,WS(205)), (MIY,WS(206)), (MIZ,WS(207))
EQUIVALENCE (WS(201),NP), (WS(208),IPS), (WS(212),IPF), (WS(211),KM),
1(KM,MK)
5  FORMAT(49H1XYZ POTENTIAL FLOW PROGRAM SECTION 3, VERSION 4 )
WRITE (6,5)
READ   (03)(PROB(1),I=1,15)
WRITE (6,1001)(PROB(1),I=1,15)
1001  FORMAT(1H0,15A4)
READ (03) (WS(I),I=1,220),IEDIT3,IEDIT4
READ   (03)(SKIP,SKIP,SKIP,XN(1),YN(1),ZN(1),SKIP,I=1,NP)
D=-.5/3.14159265
READ   (03)(DM(I),I=1,NP)
K1=1
K2=NP
240  FORMAT (18HOCHANGES IN PROB -,15A4)
IF(IPS)1220,1220,1231
1231  READ   (12)(B(K),K=1,15)
WRITE (6,240)(B(K),K=1,15)
READ   (12)(B(K),K=1,220)
READ   (12)SKIP
READ   (12)SKIP
K1=IPS
K2=IPF
C      A. SET CONDITIONS FOR FLOW OF -1 IN X DIRECTION
1220  FX=-1
      FY=0
      FZ=0
      NF=-1
C      B. COMPUTE INITIAL APPROXIMATION TO THE SOURCE
1240  DO 1250 K=1,NP
      VIP(K)=XN(K)*FX+YN(K)*FY+ZN(K)*FZ-DM(K)
      S(5,K)=-VIP(K)*.11936
C      C. SET PARTIAL SUM VECTOR TO ZERO
1250  SN(K)=0.
      SN(NP+1)=0.
      SN(NP+2)=0.
      SN(NP+3)=0.
      SN(NP+4)=0.
      WRITE(6,997) FX,FY,FZ
      WRITE (6,998)
998  FORMAT(27H0 ITERATION SUM OF CHANGES ,9X,1HR,10X,2HB1,10X,2HB2)
      IT=1
      IC=5
      IF (IPS) 1260,1260,1255
1255  READ(12) (S(5,K), K=1,KK)
      DO 1256 K=1,KK
      DO 1256 I=1,4
1256  S(I,K)=S(5,K)
C      D. START ITERATION
1260  BAND=0

```

```

      IF (NF)1261, 1262, 1263
1261 READ (02)IDW,COEF
      GO TO 1264
1262 READ (08)IDW,COEF
      GO TO 1264
1263 READ (09)IDW,COEF
1264 J=0
C      D. READ FIRST BLOCK OF COEF
C      E. START LOOP OVER QUADS.
      DO 1290 K=1,NP
C      F. PICK UP SOURCE DENSITY
      SP=S(IC,K)
C      G. START LOOP OVER NULL PTS.
      DO 1290 KP=1,NP,5
      IF(J-900)80,65,65
65 IF (NF)67,68,69
67 READ (02)IDW,COEF
      GO TO 70
68 READ (08)IDW,COEF
      GO TO 70
69 READ (09)IDW,COEF
70 J=0
C      H. COMPUTE PARTIAL SUM FOR NEXT 5 PTS.
80 SN(KP)=SN(KP)+COEF(J+1)*SP
      SN(KP+1)=SN(KP+1)+COEF(J+2)*SP
      SN(KP+2)=SN(KP+2)+COEF(J+3)*SP
      SN(KP+3)=SN(KP+3)+COEF(J+4)*SP
      SN(KP+4)=SN(KP+4)+COEF(J+5)*SP
      J=J+5
C      J. END OF LOOP OVER NULL PTS.
C      K. END OF LOOP OVER QUADS.
1290 CONTINUE
C      L. COMPUTE NEW SOURCE
      IF (NF) 91,92,93
91 REWIND 02
      GO TO 94
92 REWIND 08
      GO TO 94
93 REWIND 09
94 PASS=1.0
      SUM=0.
      DO 100 K=K1,K2
      SN(K)=( SN(K)+VIP(K) ) *D
      TEST=ABS(SN(K)-S(IC,K))
      SUM=SUM+TEST
      IF (TEST .GT. EPS) PASS=-1.0
100 CONTINUE
      IF (PASS .EQ. 1.0) GO TO 180
      IF (IT.GE.MIX) GO TO 180
      IF (IDIT3 .EQ. 0) WRITE(6,99) IT,SUM
      IT=IT+1
      IC=IC-1
      IF (IC .EQ. 0) GO TO 120
      DO 110 K=K1,K2
      S(IC,K)=SN(K)
110 SN(K)=0.
      GO TO 1260
120 A=0.
      B1=0.
      B2=0.
      DA=0.
      D1=0.

```

```

D2=0.
DO 140 K= K1,K2
DS9=2*S(1,K)-SN(K)-S(2,K)
IF(DS9 .GT. 0.) GO TO 122
A=A+S(2,K)-S(1,K)
DA=DA-DS9
GO TO 125
122 A =A +S(1,K)-S(2,K)
DA=DA+DS9
125 DS1=S(4,K)-S(3,K)
DS2=S(3,K)-S(2,K)
DS3=DS1-DS2
DSS=S(2,K)-S(1,K)
DS5=DS2-DSS
DS6=DS1-DSS
DS4=DS2-S(1,K)+SN(K)
DS7=DS3*DS4-DS5*DS6
DS8=DS6*DS5-DS4*DS3
IF(DS7 .GT. 0.) GO TO 128
B1=B1-DS1*DS4+DS2*DS6
D1=D1-DS7
GO TO 130
128 B1=B1+DS1*DS4-DS2*DS6
D1=D1+DS7
130 IF (DS9 .GT. 0.) GO TO 132
B2=B2-DS1*DS5+DS2*DS3
D2=D2-DS8
GO TO 140
132 B2=B2+DS1*DS5-DS2*DS3
D2=D2+DS8
140 CONTINUE
A=A/DA
B1=B1/D1
B2=B2/D2
IF(IT .EQ. 6) GO TO 155
AA=A-AS
AA=ABS(AA)
IF (AA .GT. .02) GO TO 148
DO 145 K=K1,K2
S(5,K)=A*(SN(K)-S(1,K))+S(1,K)
145 SN(K)=0.
WRITE(6,6000)
6000 FORMAT(29X,17H8 EXTRAPOLATION )
GO TO 160
148 BB1=B1-BB1
BB1=ABS(BB1)
BB1=50.*BB1
BB2=B2-BB2
BB2=ABS(BB2)
BB2=50.*BB2
BBB=ABS(BB1) + ABS(BB2)
IF ( (BB1 .GT. BBB) .OR. (BB2 .GT. BBB) ) GO TO 155
DO 150 K=K1,K2
S(5,K)=S(2,K)+B1*(S(1,K)-S(2,K))+B2*(SN(K)-S(2,K))
150 SN(K)=0.
WRITE(6,7000)
7000 FORMAT(29X,17H8 EXTRAPOLATION )
GO TO 160
155 DO 158 K=K1,K2
S(5,K)=SN(K)
158 SN(K)=0.
160 IC=5

```



```

WRITE(6,161) A, B1, B2
161 FORMAT(29X,3E12.3)
AS=A
BS1=B1
BS2=B2
GO TO 1260
180 WRITE(6,99) IT,SUM
DO 182 K=K1,K2
182 S(1,K)=SN(K)
WRITE(03) (S(1,K), K=1,NP)
99 FORMAT(4X,13,E18.5)
997 FORMAT (13H0 X VELOCITY=,F4.1,15H Y VELOCITY=,F4.1,
115H Z VELOCITY=,F4.1)
IF(FZ)1400,1390,1400
C P1 IF THIS WAS NOT LAST FLOW, SET FOR NEXT FLOW
1390 FZ=FY
FV=FX
FX=0
MIX=M1Y
M1Y=M1Z
NF=NF+1
GO TO 1240
1400 REWIND 03
C P2 READ IN PFFS4 AND TRANSFER TO IT
STOP 3
END

```

**APPENDIX IU - XYZPF SECTION PF4**

```

PROGRAM PFP4(OUTPUT,TAPE6=OUTPUT,TAPE03,TAPE01,TAPE11,
1      TAPE3=TAPE03,TAPE1=TAPE01)
C
C XYZ POTENTIAL FLOW PROGRAM VERSION 1 SECTION 4
C COMPUTES VELOCITIES AND PRESSURE COEFFICIENTS FOR
C POINTS ON THE BODY
C
COMMON   UX1(650),UY1(650),UZ1(650),   UX2(650),UY2(650),UZ2(650)
1      ,UX3(650),UY3(650),UZ3(650),   S1(650),S2(650),S3(650)
2      ,X(650),Y(650),Z(650),   T4(650),T5(650),T6(650)
3      ,DM(650)
DIMENSION PROB(15),WS(220),CV1(1000),CV2(1000),CV3(1000)
EQUIVALENCE (WS(201),NP),(WS(211),KM),(WS(217),UX1),(WS(218),UY1),
1(WS(219),UZ1),(WS(208),IPS),(WS(212),IPF)
EQUIVALENCE (MIX,WS(205)),(MIY,WS(206)),(MIZ,WS(207))
5 FORMAT(49HOXYZ POTENTIAL FLOW PROGRAM SECTION 4, VERSION 4 )
WRITE (6,5)
READ (03)(PROB(I),I=1,15)
100 FORMAT(1H ,15R4)
WRITE (6,100)(PROB(I),I=1,15)
C
A. READ PARAMETERS AND SOURCE
READ (03)(WS(I),I=1,220),IEDIT3,IEDIT4
READ (03)(X(I),Y(I),Z(I),T4(I),T5(I),T6(I),SKIP,I=1,NP)
READ (03)(DM(I),I=1,NP)
D=-.5/3.14159265
READ (03)(S1(I),I=1,NP)
READ (03)(S2(I),I=1,NP)
READ (03)(S3(I),I=1,NP)
J=1
K1=1
K2=NP
IF(IPS)108,108,102
102 K1=1PS
K2=1PF
108 BBR=1.
C
B. READ FIRST BLOCK OF COEF.
ITAPE=01
READ (01)BB,CV1,CV2,CV3
IF (BB-BBR) 300,120,300
120 DO 125I=K1,K2
UX1(I)=-1.0      -S1(I)*T4(I) /D
UY1(I)=          -S1(I)*T5(I) /D
UZ1(I)=          -S1(I)*T6(I) /D
UX2(I)=          -S2(I)*T4(I) /D
UY2(I)=-1.0      -S2(I)*T5(I) /D
UZ2(I)=          -S2(I)*T6(I) /D
UX3(I)=          -S3(I)*T4(I) /D
UY3(I)=          -S3(I)*T5(I) /D
125 UZ3(I)=-1.0   -S3(I)*T6(I) /D
C
C. SET UP LOOP OVER QUADS.
JC=2
C
D. PICK UP SOURCE
130 S1J=S1(J)
S2J=S2(J)
S3J=S3(J)
C
E. SET UP LOOP OVER NULL PTS.
DO 180 JP=K1,K2
C
F. COMPUTE PARTIAL SUM FOR 3 COMPONENTS OF 3 VELOCITIES
UX1(JP)=UX1(JP)+S1J*CV1(JC)
UY1(JP)=UY1(JP)+S1J*CV1(JC+1)

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UZ1(JP)=UZ1(JP)+S1J*CV1(JC+2)
UX2(JP)=UX2(JP)+S2J*CV2(JC)
UY2(JP)=UY2(JP)+S2J*CV2(JC+1)
UZ2(JP)=UZ2(JP)+S2J*CV2(JC+2)
UX3(JP)=UX3(JP)+S3J*CV3(JC)
UY3(JP)=UY3(JP)+S3J*CV3(JC+1)
UZ3(JP)=UZ3(JP)+S3J*CV3(JC+2)
JC=JC+3
C   G. READ MORE COEF. IF NEEDED.
   IF (JC-1000)180, 135, 135
135 JC=2
   IF (BBR-635.0) 140, 150, 155
140 READ (01) BB, CV1, CV2, CV3
   GO TO 160
150 REWIND 01
155 READ (11) BB, CV1, CV2, CV3
160 BBR=BBR+1.
   IF (BBR-BB) 300, 180, 300
C   H. END OF LOOP OVER NULL PTS.
180 CONTINUE
   J=J+1
C   I. END OF LOOP OVER QUADS.
   IF (J-NP)130, 130, 200
200 IF (BBR-635.0) 231, 231, 232
231 REWIND 01
   GO TO 233
232 REWIND 11
C   K. EDIT THE VELOCITIES ETC. AND WRITE THEM ON TAPE
233 WRITE (03)(UX1(I), UY1(I), UZ1(I) , I=1, NP)
   WRITE (03)(UX2(I), UY2(I), UZ2(I) , I=1, NP)
   WRITE (03)(UX3(I), UY3(I), UZ3(I) , I=1, NP)
235 FORMAT(1H1, 15A4, 8H PAGE =, 115)
   REWIND 03
   IP=K1+49
   IS=K1
   IPAGE=1
   IF (EDIT4 .EQ. 1) GO TO 293
   IF (MIX .LE. 0) GO TO 265
242 FORMAT(8H0 X FLOW)
240 FORMAT(4H PT., 10X, 1HX, 9X, 1HY, 9X, 1HZ, 13X, 2HUX, 8X, 2HUY, 8X, 2HUZ, 10X,
1 5HABS.U, 7X, 2HCP, 6X, 6HSOURCE, 4X, 8HU NORMAL)
245 FORMAT (1X, 13, 4X, 3F10.5, 4X, 3F10.5, 1F13.5, 2F11.5, E12.2)
250 IF (IP-K2)255, 255, 260
C   J. COMPUTE PRESSURE AND ABS. VALUE OF VELOCITY
255 WRITE (6, 235)(PROB(I), I=1, 15), IPAGE
   WRITE (6, 242)
   WRITE (6, 240)
   DO 257 I=15, IP
   USQ=UX1(I)**2+UY1(I)**2+UZ1(I)**2
   UM=(ABS(UX1(I))+ABS(UY1(I))+ABS(UZ1(I)))*.79
   UM=UM+USQ/UM
   UM=.25*UM+USQ/UM
   UM=.5*(UM+USQ/UM)
   CP =1.-USQ
   UNR =UX1(I)*T4(I) +UY1(I)*T5(I) +UZ1(I)*T6(I)
257 WRITE (6, 245) I, X(I), Y(I), Z(I), UX1(I), UY1(I), UZ1(I), UM
1 , CP , S1(I), UNR
   IS=IS+50
   IP=IP+50
   IPAGE=IPAGE+1
   IF (K2-IS) 265, 260, 250
260 IP=K2

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```

GO TO 255
265 IP=K1+49
   IS=K1
   IF (M1V .LE. 0 ) GO TO 280
267 IF (IP-K2)275,275,270
270 IP=K2
275 WRITE          (6,235)(PROB(1), I=1, 15), IPAGE
   WRITE          (6,277)
277 FORMAT(8H0 Y FLOW)
   WRITE          (6,240)
   DO 278 I=IS, IP
   USQ=UX2(1)**2+UY2(1)**2 +UZ2(1)**2
   UM=(ABS(UX2(1))+ABS(UY2(1))+ABS(UZ2(1)))*.79
   UM=UM+USQ/UM
   UM=.25*UM+USQ/UM
   UM  =.5*(UM+USQ/UM)
   UNR  =UX2(1)*T4(1) +UY2(1)*T5(1) +UZ2(1)*T6(1)
   CP   =1.-USQ
278 WRITE (6,245) I,X(1),Y(1),Z(1),UX2(1),UY2(1),UZ2(1),UM
   1 ,CP ,S2(1),UNR
   IS=IS+50
   IP=IP+50
   IPAGE=IPAGE+1
   IF (IS-K2)267,267,280
280 IF=K1+49
   IS=K1
   IF (M1Z .LE. 0 ) GO TO 293
282 IF (IP-K2)290,290,285
285 IP=K2
290 WRITE          (6,235)(PROB(1), I=1, 15), IPAGE
   WRITE          (6,292)
292 FORMAT(8H0 Z FLOW)
   WRITE          (6,240)
   DO 291 I=IS, IP
   USQ=UX3(1)**2+UY3(1)**2+UZ3(1)**2
   UM=(ABS(UX3(1))+ABS(UY3(1))+ABS(UZ3(1)))*.79
   UM=UM+USQ/UM
   UM=.25*UM+USQ/UM
   UM  =.5*(UM+USQ/UM)
   CP   =1.-USQ
   UNR  =UX3(1)*T4(1) +UY3(1)*T5(1) +UZ3(1)*T6(1)
291 WRITE (6,245) I,X(1),Y(1),Z(1),UX3(1),UY3(1),UZ3(1),UM
   1 ,CP ,S3(1),UNR
   IS=IS+50
   IP=IP+50
   IPAGE=IPAGE+1
   IF (IS-K2)282,282,293
C   L. CHECK FOR A FOURTH FLOW
293 J = 1
294 UX1 = WS(J)
   UY1 = WS(J+20)
   UZ1 = WS(J+40)
295 IF (UX1**2+UY1**2+UZ1**2) 400,400,301
301 IS=K1
   IP=K1+49
320 IF (IP-K2)330,325,325
C   N. EDIT THE VELOCITY AND PRESSURE FOR FOURTH FLOW
325 IP=K2
330 WRITE          (6,235)(PROB(1), I=1, 15), IPAGE
   WRITE          (6,315)UX1,UY1,UZ1
   WRITE          (6,340)
C   M. COMPUTE VELOCITY AND PRESSURE FOR FOURTH FLOW

```

```

DO 333 I=IS,IP
UX4  =- (UX1*UX1<I>+UY1*UX2<I>+UZ1*UX3<I>)
UY4  =- (UX1*UY1<I>+UY1*UY2<I>+UZ1*UY3<I>)
UZ4  =- (UX1*UZ1<I>+UY1*UZ2<I>+UZ1*UZ3<I>)
USQ=UX4  **2+UY4  **2+UZ4  **2
UM=(ABS<UX4  >+ABS<UY4  >+ABS<UZ4  >)*.79
UM=UM+USQ/UM
UM=.25*UM+USQ/UM
UM  =.5*(UM+USQ/UM)
CP  = 1.-<USQ>/< UY1**2+UZ1**2+UX1**2>
333 WRITE (6,345) I,X<I>,Y<I>,Z<I>,UX4  ,UY4  ,UZ4  ,UM
      I ,CP
      IS=IS+50
      IP=IP+50
      IPAGE=IPAGE+1
      IF<K2-IS>350,325,320
350 J = J+1
      GO TO 294
315 FORMAT<19H0  ONSET FLOW, UX1=,F6.3,2X,4HUY1=,F6.3,2X,4HUZ1=,F6.3>
340 FORMAT<4H FT., 10X, 1HX, 9X, 1HY, 9X, 1HZ, 13X, 2HUX, 8X, 2HUY, 8X, 2HUZ, 10X,
      1 5HABS.U, 7X, 2HCP>
345 FORMAT <1X, 13, 4X, 3F10.5, 4X, 3F10.5, 1F13.5, 1F11.5>
400 CONTINUE
      GO TO 5000
300 CONTINUE
      WRITE (6,310) ITAPE, BBR, BB
310 FORMAT (6H1TAPE ,12, 17H OUT OF POSITION/114, 6F6.1)
5000 CONTINUE
      STOP 4
      END

```

**APPENDIX U - XYZPF SECTION PFS**

```

PROGRAM PFP5(INPUT=128,OUTPUT=128,TAPE03,TAPE04,
1TAPES=INPUT,TAPE6=OUTPUT,TAPE3=TAPE03,TAPE4=TAPE04)
C
C XYZ POTENTIAL FLOW PROGRAM VERSION 4 SECTION 5
C COMPUTES VELOCITIES AND PRESSURE COEFFICIENTS FOR
C OFF BODY POINTS
C
COMMON B(241), XP(500),YP(500),ZP(500),UX(500),WS(220)
1,UY(500),UZ(500),UX2(500),UY2(500),UZ2(500),UX3(500),UY3(500)
2,UZ3(500),UX(8),UY(8),UZ(8),S1(650),S2(650),S3(650),U1(3),U2(3),U3
3(3),PROB(15),XN(650),YN(650),ZN(650),TA(650),TX(650)
4,TY(650),TZ(650)
EQUIVALENCE (KM,WS(211)),(KM,MK),(NF,WS(201)),(SYM,WS(210))
1,(Y2,Y3),(MIX,WS(205)),(M1Y,WS(206)),(M1Z,WS(207))
INTEGER PAGE
INTEGER SYM
5 FORMAT(49H0XYZ POTENTIAL FLOW PROGRAM SECTION 5, VERSION 4 )
WRITE (6,5)
C A. READ INPUT
READ(5,25) NOBP,IEDITS,IREAD
C A. READ THE OFF BODY POINTS
NOBP=NOBP
DO 10 I=1,NOBP
READ(5,26) XP(I),YP(I),ZP(I)
IF (EOF(5) .EQ. 0.) GO TO 10
NOBP=I-1
WRITE(6,9) NOBP,NOB
9 FORMAT(1H0,15,31H OFF BODY POINTS SPECIFIED NOT ,13)
GO TO 11
10 CONTINUE
11 CONTINUE
25 FORMAT(314)
26 FORMAT(3F12.5)
P=1.
READ (03) (PROB(I),I=1,15)
WRITE (6,90) (PROB(I),I=1,15)
90 FORMAT(1H0,18A4)
WRITE(6,91) NOBP,IEDITS,IREAD
91 FORMAT(8H0NOBP =,14 /8H IEDITS=,14/8H IREAD =,14)
WRITE(6,92)
92 FORMAT(17H0 OFF BODY POINTS / 4H PT.,11X,1HX,12X,1HY,12X,1HZ)
WRITE(6,93) (I,XP(I),YP(I),ZP(I),I=1,NOBP)
93 FORMAT(1X,13,2X,3F13.5)
C B. READ THE PARAMETERS, T ARRAY AND SOURCE FROM TAPE 31
READ (03) (WS(I),I=1,220)
READ (03) (TX(I),TY(I),TZ(I),XN(I),YN(I),ZN(I),TA(I),I=1,NP)
READ (03) SKIP
C
C FORMERLY: WS(220) .EQ. 2.
C
IF(WS(220) .EQ. 5.) READ(03) SKIP
READ (03) (S1(I),I=1,NP)
READ (03) (S2(I),I=1,NP)
READ (03) (S3(I),I=1,NP)
C C. READ THE FIRST BLOCK OF THE B ARRAY
READ (04) (B(I),I=1,241)
K=1
J=1
DO 100 I=1,NOBP
C D. SET THE PARTIAL VELOCITY TO THE FREE STREAM VELOCITY

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```

    UX1(I)=-1.0
    UY1(I)=0.
    UZ1(I)=0.
    UX2(I)=0.
    UY2(I)=-1.0
    UX3(I)=0.0
    UY3(I)=0.0
    UZ3(I)=-1.0
100 UZ2(I)=0.
290 IF(B(I)-P)291,295,291
291 WRITE (6,98) B(I),P
    98 FORMAT(28H0 POINTS OUT OF ORDER B(I)=,1F4.0,4H P=,1F4.0)
    STOP
C      E.  START LOOP OVER THE QUADS.
295 J=2
C      F1  PICK UP QUAD. INFORMATION
296 X1=B(J)
    Y1=B(J+1)
    X2=B(J+2)
    Y2=B(J+3)
    X3=B(J+4)
    X4=B(J+5)
    Y4=B(J+6)
    XC=TX(K)
    YC=TY(K)
    ZC=TZ(K)
    A =TA(K)
    XX=B(J+7)
    YX=B(J+8)
    ZX=B(J+9)
    XY=B(J+10)
    YY=B(J+11)
    ZY=B(J+12)
C      F2  COMPUTE LENGTH OF SIDES OF QUAD.
    D12=SQ2F(X1,X2,Y1,Y2,0.,0.)
    D23=SQ2F(X2,X3,Y2,Y3,.0,.0)
    D34=SQ2F(X3,X4,Y3,Y4,.0,.0)
    D41=SQ2F(X4,X1,Y4,Y1,.0,.0)
C      F3  COMPUTE SLOPE OF SIDES
    IF(X2-X3)305,300,305
300 C123=1.
    GO TO 310
305 CM23=(Y2-Y3)/(X2-X3)
    C123=0.
310 IF(X3-X4)315,311,315
311 C134=1.
    GO TO 320
315 CM34=(Y4-Y3)/(X4-X3)
    C134=0.
320 IF(X4-X1)325,321,325
321 C141=1.
    GO TO 330
325 CM41=(Y1-Y4)/(X1-X4)
    C141=0.
330 IF(X1-X2)335,331,335
331 C112=1.
    GO TO 340
335 CM12=(Y2-Y1)/(X2-X1)
    C112=0.
C      F4  COMPUTE QUADRAPOLE MOMENTS
340 C1XX=B(J+17)
    C1XY=B(J+18)

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```

C1YY=B(J+1Q)
C   F5 COMPUTE SIN AND COS OF SLOPE ANGLE FOR EACH SIDE
CY12=(Y2-Y1)/D12
CY23=(Y3-Y2)/D23
CY34=(Y4-Y3)/D34
CY41=(Y1-Y4)/D41
CX12=(X1-X2)/D12
CX23=(X2-X3)/D23
CX34=(X3-X4)/D34
CX41=(X4-X1)/D41
C   F6 COMPUTE MAX DIAGONAL
ST=SQ2F(X1,X3,Y1,Y3,0.,0.)
ST2=SQ2F(X2,X4,Y2,Y4,0.,0.)
IF(ST-ST2)341,342,342
341 ST=ST2
C   G. START LOOP OVER THE OFF BODY POINTS
342 DO 530 JQ=1,NDBP
    IS=1
    XCQ=XP(JQ)
    YCQ=YP(JQ)
    ZCQ=ZP(JQ)
    J1=1
345 RPQ=SQ2F(XC,XCQ,YC,YCQ,ZC,ZCQ)
C   H. DETERMIN METHOD
    IF(RPQ-ST*4)350,350,460
350 X=(XCQ-XC)*XX+(YCQ-YC)*YX+(ZCQ-ZC)*ZX
    Y=(XCQ-XC)*XY+(YCQ-YC)*YY+(ZCQ-ZC)*ZY
    Z=(XCQ-XC)*XN(K)+(YCQ-YC)*YN(K)+(ZCQ-ZC)*ZN(K)
    IF(RPQ-ST*2.5)355,355,400
C   I. COMPUTE INDUCED VELOCITY BY EXACT METHOD
355 R1=SQ2F(X,X1,Y,Y1,Z,0.)
    R2=SQ2F(X,X2,Y,Y2,Z,0.)
    R3=SQ2F(X,X3,Y,Y3,Z,0.)
    R4=SQ2F(X,X4,Y,Y4,Z,0.)
    IF((R1+R2).LE.D12) GO TO 1000
    IF((R3+R2).LE.D23) GO TO 1000
    IF((R3+R4).LE.D34) GO TO 1000
    IF((R1+R4).LE.D41) GO TO 1000
    CLA1=ALOG((R1+R2-D12)/(R1+R2+D12))
    CLA2=ALOG((R2+R3-D23)/(R2+R3+D23))
    CLA3=ALOG((R3+R4-D34)/(R3+R4+D34))
    CLA4=ALOG((R4+R1-D41)/(R4+R1+D41))
    TUX=CY12*CLA1+CY23*CLA2+CY34*CLA3+CY41*CLA4
    TUY=CX12*CLA1+CX23*CLA2+CX34*CLA3+CX41*CLA4
    TUZ=0.
    IF(ABS(Z)-.001*ST)375,361,361
361 ZSQ=Z**2
    E1=ZSQ+(X-X1)**2
    E2=ZSQ+(X-X2)**2
    E3=ZSQ+(X-X3)**2
    E4=ZSQ+(X-X4)**2
    H1=(Y-Y1)*(X-X1)
    H2=(Y-Y2)*(X-X2)
    H3=(Y-Y3)*(X-X3)
    H4=(Y-Y4)*(X-X4)
    IF(D12)363,363,354
363 WS1=(D112*E1-H1)/(Z*R1)
    WS2=(D112*E2-H2)/(Z*R2)
    AT1=ATAN(WS1)
    AT2=ATAN(WS2)
    TUZ=AT1-AT2
364 IF(D123)366,366,367

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366 AT3=ATAN((CM23*E2-H2)/(Z*R2))
AT4=ATAN((CM23*E3-H3)/(Z*R3))
TUZ=TU2+AT3-AT4
367 IF(C134)368,368,369
368 AT5=ATAN((CM34*E3-H3)/(Z*R3))
AT6=ATAN((CM34*E4-H4)/(Z*R4))
TUZ=TU2+AT5-AT6
369 IF(C141)370,370,375
370 AT7=ATAN((CM41*E4-H4)/(Z*R4))
AT8=ATAN((CM41*E1-H1)/(Z*R1))
TUZ=TU2+AT7-AT8
375 GO TO 450
C J. COMPUTE INDUCED VELOCITY BY QUADRAPOLE METHOD
400 RPQ3=RPQ**3
RPQ7=(RPQ**2)*RPQ
WS1=X/RPQ3
XSQ=X**2
YSQ=Y**2
ZSQ=Z**2
PS=YSQ+ZSQ-4.*XSQ
QS=XSQ+ZSQ-4.*YSQ
WS2=X*(9.*PS+30.*XSQ)/RPQ7
WS3=3.*Y*PS/RPQ7
WS4=3.*X*QS/RPQ7
TUX=A*WS1-C1XY*WS3-C1XX*WS2-C1YY*WS4
US1=Y/RPQ3
US2=Y*(9.*QS+30.*YSQ)/RPQ7
TUY=A*WS1-C1XX*WS3-C1XY*WS4-C1YY*WS2
TUZ=Z*(A/RPQ3-3.*(C1XX*PS-5.*(C1XY*X*Y+C1YY*QS))/RPQ7)
450 UX(1S)=TUX*XX+TUY*XY+TUZ*XN(K)
UY(1S)=TUX*YX+TUY*YY+TUZ*YN(K)
UZ(1S)=TUX*ZX+TUY*ZY+TUZ*ZN(K)
GO TO 470
C K. COMPUTE INDUCED VELOCITY BY MONOPOLE METHOD
460 ARPQ3=A/(RPQ**3)
UX(1S)=(XCQ-XC)*ARPQ3
UY(1S)=(YCQ-YC)*ARPQ3
UZ(1S)=(ZCQ-ZC)*ARPQ3
C L. REFLECT OFF BODY POINT IN PLANE OF SYMMETRY
470 GO TO(480,485,490,495,500,505,510,515),IS
480 U1(J1)=UX(1)
U2(J1)=UX(1)
U3(J1)=UX(1)
U1(J1+1)=UY(1)
U2(J1+1)=UY(1)
U3(J1+1)=UY(1)
U1(J1+2)=UZ(1)
U2(J1+2)=UZ(1)
U3(J1+2)=UZ(1)
IF(SYM) 525,525,481
481 IS=2
C XZ SYMMETRY
YCQ=-YCQ
GO TO 345
485 IF(SYM) 517,517,486
C XY SYMMETRY
486 IS=3
ZCQ=-ZCQ
GO TO 345
490 IS=4
YCQ=-YCQ
GO TO 345

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495 IF(SYM-2)516,516,496
C      YZ SYMETRY
496 IS=5
      XCQ=-XCQ
      GO TO 345
500 IS=6
      YCQ=-YCQ
      GO TO 345
505 IS=7
      ZCQ=-ZCQ
      GO TO 345
510 IS=8
      YCQ=-YCQ
      GO TO 345
C      M. ADD CONTRIBUTIONS OF ALL REFLECTIONS
515 U1(J1)=U1(J1)+UX(8)+UX(7)+UX(6)+UX(5)
      U2(J1)=U2(J1)-UX(8)+UX(7)+UX(6)-UX(5)
      U3(J1)=U3(J1)-UX(8)-UX(7)+UX(6)+UX(5)
      U1(J1+1)=U1(J1+1)-UY(8)+UY(7)+UY(6)-UY(5)
      U2(J1+1)=U2(J1+1)+UY(8)+UY(7)+UY(6)+UY(5)
      U3(J1+1)=U3(J1+1)+UY(8)-UY(7)+UY(6)-UY(5)
      U1(J1+2)=U1(J1+2)-UZ(8)-UZ(7)+UZ(6)+UZ(5)
      U2(J1+2)=U2(J1+2)+UZ(8)-UZ(7)+UZ(6)-UZ(5)
      U3(J1+2)=U3(J1+2)+UZ(8)+UZ(7)+UZ(6)+UZ(5)
516 U1(J1)=U1(J1)+UX(4)+UX(3)
      U2(J1)=U2(J1)+UX(4)-UX(3)
      U3(J1)=U3(J1)-UX(4)-UX(3)
      U1(J1+1)=U1(J1+1)+UY(4)-UY(3)
      U2(J1+1)=U2(J1+1)+UY(4)+UY(3)
      U3(J1+1)=U3(J1+1)-UY(4)+UY(3)
      U1(J1+2)=U1(J1+2)-UZ(4)-UZ(3)
      U2(J1+2)=U2(J1+2)-UZ(4)+UZ(3)
      U3(J1+2)=U3(J1+2)+UZ(4)+UZ(3)
517 U1(J1)=U1(J1)+UX(2)
      U2(J1)=U2(J1)-UX(2)
      U3(J1)=U3(J1)+UX(2)
      U1(J1+1)=U1(J1+1)-UY(2)
      U2(J1+1)=U2(J1+1)+UY(2)
      U3(J1+1)=U3(J1+1)-UY(2)
      U1(J1+2)=U1(J1+2)+UZ(2)
      U2(J1+2)=U2(J1+2)-UZ(2)
      U3(J1+2)=U3(J1+2)+UZ(2)
525 L=P
      UX1(JQ)=UX1(JQ)+U1(1)*S1(L)
      UY1(JQ)=UY1(JQ)+U1(2)*S1(L)
      UZ1(JQ)=UZ1(JQ)+U1(3)*S1(L)
      UX2(JQ)=UX2(JQ)+U2(1)*S2(L)
      UY2(JQ)=UY2(JQ)+U2(2)*S2(L)
      UZ2(JQ)=UZ2(JQ)+U2(3)*S2(L)
      UX3(JQ)=UX3(JQ)+U3(1)*S3(L)
      UY3(JQ)=UY3(JQ)+U3(2)*S3(L)
      UZ3(JQ)=UZ3(JQ)+U3(3)*S3(L)
530 CONTINUE
C      N. END OF LOOP OVER OFF BODY POINTS
585 P=P+1
      K=K+1
      J=J+20
      IF(K-NP)586,586,599
586 IF(J-241)296,590,590
C      O. READ NEXT BLOCK OF B ARRAY IF NEEDED
590 READ (04) (B(I), I=1,241)
      J=2

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      IF(B(1)-P)291,296,291
C     P.  END OF LOOP OVER QUADS
599 CONTINUE
      PAGE = 1
      IF (IEDITS .EQ. 1) GO TO 825
601 FORMAT(4H PT., 11X, 1HX, 12X, 1HY, 12X, 1HZ, 14X, 2HUX, 11X, 2HUY, 11X, 2HUZ
      1, 14X, 2HCP)
602 FORMAT(7H X FLOW)
603 FORMAT(7H Y FLOW)
604 FORMAT(7H Z FLOW)
605 FORMAT(1H1, 15A4, 10X, 15HOFF BODY POINTS , 10X, 5HPAGE , 13)
      IF (MIX.EQ.0) GO TO 700
      WRITE(6,605) PROB,PAGE
      WRITE (6,602)
      WRITE (6,601)
      LINE=1
      LAST=53
606 IF(NOBP-LAST)607,610,610
607 LAST=NOBP
610 DO 615 I=LINE, LAST
611 FORMAT(1X, 113, 2X, 3F13.5, 2X, 3F13.5, 3X, F13.5)
C     Q.  COMPUTE PRESSURE AND EDIT 3 BASIC FLOWS
      CP1=1.-(UX1(I)**2+UY1(I)**2+UZ1(I)**2)
615 WRITE (6,611) I,XP(I),YP(I),ZP(I),UX1(I),UY1(I),UZ1(I),CP1
      LINE=LAST+1
      LAST=LINE+54
      PAGE=PAGE+1
      IF(LINE=NOBP)620,620,700
620 WRITE(6,605) PROB,PAGE
      WRITE (6,601)
      GO TO 606
700 IF (M1Y.EQ.0) GO TO 800
      WRITE(6,605) PROB,PAGE
      WRITE (6,603)
      WRITE (6,601)
      LINE=1
      LAST=55
706 IF(NOBP-LAST)707,710,710
707 LAST=NOBP
710 DO 715 I=LINE, LAST
      CP2=1.-(UX2(I)**2+UY2(I)**2+UZ2(I)**2)
715 WRITE (6,611) I,XP(I),YP(I),ZP(I),UX2(I),UY2(I),UZ2(I),CP2
      LINE=LAST+1
      LAST=LINE+54
      PAGE=PAGE+1
      IF(LINE=NOBP)720,720,800
720 WRITE(6,605) PROB,PAGE
      WRITE (6,601)
      GO TO 706
800 IF (M1Z.EQ.0) GO TO 825
      WRITE(6,605) PROB,PAGE
      WRITE (6,604)
      WRITE (6,601)
      LINE=1
      LAST=55
806 IF(NOBP-LAST)807,810,810
807 LAST=NOBP
810 DO 815 I=LINE, LAST
      CP3=1.-(UX3(I)**2+UY3(I)**2+UZ3(I)**2)
815 WRITE (6,611) I,XP(I),YP(I),ZP(I),UX3(I),UY3(I),UZ3(I),CP3
      LINE=LAST+1
      LAST=LINE+54

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```

PAGE=PAGE+1
IF(LINE-NOBP)820,820,825
820 WRITE(6,605)PROB,PAGE
WRITE(6,601)
GO TO 806
825 J = 1
826 IF (IREAD.EQ.0) GO TO 827
READ(5,26) UX4,UY4,UZ4
IF (EOF(5).NE.0.) GO TO 900
GO TO 828
827 UX4=WS(J)
UY4 = WS(J+20)
UZ4 = WS(J+40)
828 CP=UX4**2+UY4**2+UZ4**2
IF(CP)900,900,830
C R. COMPUTE FOURTH FLOW AND EDIT IT
830 LINE=1
LAST=51
WRITE(6,605)PROB,PAGE
831 FORMAT(19HCONSET FLOW UX =,F7.3/15X,4HVY =,
IF7.3/15X,4HVZ =,F7.3)
WRITE(6,831)UX4,UY4,UZ4
WRITE(6,601)
835 IF(NOBP-LAST)837,840,840
837 LAST=NOBP
840 DO 845 I=LINE, LAST
UXP=-UX4*UX1(I)-UY4*UX2(I)-UZ4*UX3(I)
UYP=-UX4*UY1(I)-UY4*UY2(I)-UZ4*UY3(I)
UZP=-UX4*UZ1(I)-UY4*UZ2(I)-UZ4*UZ3(I)
CP4= 1.-(UXP**2+UYP**2+UZP**2)/CP
845 WRITE(6,611)I,XP(I),YP(I),ZP(I),UXP,UYP,UZP,CP4
LINE=LAST+1
LAST=LINE+54
PAGE=PAGE+1
IF(LINE-NOBP)850,850,860
850 WRITE(6,605)PROB,PAGE
WRITE(6,601)
GO TO 835
860 J = J+1
GO TO 826
1000 WRITE(6,1001)JQ,L,XP(JQ),YP(JQ),ZP(JQ)
1001 FORMAT(16HDOFF BODY POINT ,13,23H ON BOUNDARY OF QUAD ,13/
1 3H X=,F12.5,5X,2HVY=,F12.5,5X,2HZ=,F12.5)
GO TO 530
900 CONTINUE
C S. REWIND TAPES AND STOP
REWIND 03
REWIND 04
STOP 5
END
FUNCTION SQ2F(X1,X2,Y1,Y2,Z1,Z2)
X=X1-X2
Y=Y1-Y2
Z=Z1-Z2
RS=Z**2+Y**2+X**2
R=ABS(X)+ABS(Y)+ABS(Z)+ 1.0E-20
R=R+RS/R
R= .25*R+RS/R
R= R+RS/R
SQ2F= .25*R+RS/R
RETURN
END

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**APPENDIX UI - XYZPF SECTION PF6**

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PROGRAM PPF6(INPUT=128,TAPE16,OUTPUT=128,TAPE03,TAPE04,
1          TAPE5=INPUT,TAPE6=OUTPUT,TAPE3=TAPE03,TAPE4=TAPE04)
C
C XYZ POTENTIAL FLOW PROGRAM VERSION 4 SECTION 6
C COMPUTES VELOCITIES AND PRESSURE COEFFICIENTS FOR
C OFF BODY STREAMLINES
C
COMMON  XP(100),YP(100),ZP(100),UX1(100),WS(220)
1,UY1(100),UZ1(100),UX2(100),UY2(100),UZ2(100),UX3(100),UY3(100)
2,UZ3(100),UX(8),UY(8),UZ(8),S1(650),S2(650),S3(650),U1(3),U2(3),U3
3(3),PROB(15),XN(650),YN(650),ZN(650),TX(650),TY(650),TZ(650)
1,B(13000),          XT(100),YT(100),ZT(100),AP(5),GM(4),SKY(100),
1,SKZ(100),SKX(100),DX(100),DY(100),DZ(100),CP(100),TA(650)
EQUIVALENCE  (KM,WS(211)),(KM,MK),(NP,WS(201)),(SYM,WS(210))
1,(Y2,Y3)
INTEGER SYM ,P
C      A. READ INPUT
WRITE (6,5)
6 FORMAT(3I4,4F12.5)
8 FORMAT(3F12.5)
5 FORMAT(49H0XYZ POTENTIAL FLOW PROGRAM SECTION 6, VERSION 4 )
7 FORMAT (1X,9F12.5)
20 FORMAT(1H1,14,31H STREAMLINES TO BE COMPUTED AT ,14,10H STEPS OF
1 ,F8.4,28H T FOR AN ONSET VELOCITY OF ,3F8.4)
21 FORMAT(1X,15HSTARTING POINTS/,3X,2HPT,5X,1HX,11X,1HV,11X,1HZ)
22 FORMAT(1X,14,3F12.5)
READ (03) (PROB(I),I=1,15)
WRITE (6,90) (PROB(I),I=1,15)
90 FORMAT(1H0,15A4)
C      B. READ THE PARAMETERS, T ARRAY AND SOURCE FROM TAPE 31
READ (03) (WS(I),I=1,220)
READ(03) (TX(I),TY(I),TZ(I),XN(I),YN(I),ZN(I),TA(I),I=1,NP)
READ (03) SKIP
IF ( WS(220) .EQ. 2. ) READ(03) SKIP
READ (03) (S1(I),I=1,NP)
READ (03) (S2(I),I=1,NP)
READ (03) (S3(I),I=1,NP)
C      C. READ THE B ARRAY
WZ = NP
WZ=(WZ+11.0)/12.0
NB = WZ
IS = 2
IF=241
DO 12 IP = 1,NB
READ (04) P, (B(I),I=IS,IF)
IS=IS+240
12 IF=IF+240
AP(1)= .5
AP(2) = .5
AP(3) = 1.
AP(4)=0.
AP(5)=0.
GM(1) = 1./6.
GM(2) = 1./3.
GM(3) = 1./3.
83 READ (5,6) NOBP,NST,IEND,DT,UX1,UY1,UZ1
USQ=UX1**2+UY1**2+UZ1**2
NOB=NOBP
DO 10 I=1,NOB
READ(5, 8) XP(I),YP(I),ZP(I)

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      IF (EOF(5) .EQ. 0.) GO TO 10
      NOBP=I-1
      WRITE(6,9) NOBP,NOB
9     FORMAT(1H0,15, 28H STREAMLINES SPECIFIED NOT ,13)
      GO TO 11
10    CONTINUE
11    CONTINUE
      WRITE(16) NOBP,NST,IEND,UX1,UY1,UZ1
      WRITE(6,20) NOBP,NST,DT,UX1,UY1,UZ1
      WRITE(6,21)
      WRITE(6,22) (I,XP(I),YP(I),ZP(I), I=1,NOBP)
C     NOBP - NUMBER OF STREAMLINES TO BE TRACED.
C     NST - NUMBER OF STATIONS AT WHICH STREAMLINES SHOULD BE COMPUTED.
      DO 15 I=1,NOBP
      XT(I) = XP(I)
      YT(I) = YP(I)
      ZT(I) = ZP(I)
      SKX(I)=0.
      SKY(I) = 0.
15    SKZ(I) = 0.
      ITC=0
      IRK=5
98    K=1
      P = 1
      J=1
      DO 100 I=1,NOBP
C     D. SET THE PARTIAL VELOCITY TO THE FREE STREAM VELOCITY
      UX1(I)=-1.0
      UY1(I)=0.
      UZ1(I)=0.
      UX2(I)=0.
      UY2(I)=-1.0
      UX3(I)=0.0
      UY3(I)=0.0
      UZ3(I)=-1.0
100   UZ2(I)=0.
C     E. START LOOP OVER THE QUADS.
295   J=2
C     F1 PICK UP QUAD. INFORMATION
296   X1=B(J)
      Y1=B(J+1)
      X2=B(J+2)
      Y2=B(J+3)
      X3=B(J+4)
      X4=B(J+5)
      Y4=B(J+6)
      XC=TX(K)
      YC=TY(K)
      ZC=TZ(K)
      A =TR(K)
      XX=B(J+7)
      YX=B(J+8)
      ZX=B(J+9)
      XY=B(J+10)
      YY=B(J+11)
      ZY=B(J+12)
C     F2 COMPUTE LENGTH OF SIDES OF QUAD.
      D12=SQ2F(X1,X2,Y1,Y2,0.,0.)
      D23=SQ2F(X2,X3,Y2,Y3,0.,0.)
      D34=SQ2F(X3,X4,Y3,Y4,0.,0.)
      D41=SQ2F(X4,X1,Y4,Y1,0.,0.)
C     F3 COMPUTE SLOPE OF SIDES

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      IF(X2-X3)305,300,305
300 C123=1.
      GO TO 310
305 CM23=(Y2-Y3)/(X2-X3)
      C123=0.
310 IF(X3-X4)315,311,315
311 C134=1.
      GO TO 320
315 CM34=(Y4-Y3)/(X4-X3)
      C134=0.
320 IF(X4-X1)325,321,325
321 C141=1.
      GO TO 330
325 CM41=(Y1-Y4)/(X1-X4)
      C141=0.
330 IF(X1-X2)335,331,335
331 C112=1.
      GO TO 340
335 CM12=(Y2-Y1)/(X2-X1)
      C112=0.
C      F4 COMPUTE QUADRAPOLE MOMENTS
340 C1XX=B(J+17)
      C1XY=B(J+18)
      C1VY=B(J+19)
C      F5 COMPUTE SIN AND COS OF SLOPE ANGLE FOR EACH SIDE
      CY12=(Y2-Y1)/D12
      CY23=(Y3-Y2)/D23
      CY34=(Y4-Y3)/D34
      CY41=(Y1-Y4)/D41
      CX12=(X1-X2)/D12
      CX23=(X2-X3)/D23
      CX34=(X3-X4)/D34
      CX41=(X4-X1)/D41
C      F6 COMPUTE MAX DIAGONAL
      ST=SQ2F(X1,X3,Y1,Y3,0.,0.)
      ST2=SQ2F(X2,X4,Y2,Y4,0.,0.)
      IF(ST-ST2)341,342,342
341 ST=ST2
C      G. START LOOP OVER THE OFF BODY POINTS
342 DO 530 JQ=1,NOBP
      IS=1
      XCQ=XP(JQ)
      YCQ=YP(JQ)
      ZCQ=ZP(JQ)
      J1=1
345 RPQ=SQ2F(XC,XCQ,YC,YCQ,ZC,ZCQ)
C      H. DETERMIN METHOD
      IF(RPQ-ST*4)350,350,460
350 Y=(XCQ-XC)*XX+(YCQ-YC)*YY+(ZCQ-ZC)*ZY
      Y=(XCQ-XC)*XY+(YCQ-YC)*YY+(ZCQ-ZC)*ZY
      Z=(XCQ-XC)*XN(K)+(YCQ-YC)*YN(K)+(ZCQ-ZC)*ZN(K)
      IF(RPQ-ST*2.5)355,355,400
C      I. COMPUTE INDUCED VELOCITY BY EXACT METHOD
355 R1=SQ2F(X,X1,Y,Y1,Z,0.)
      R2=SQ2F(X,X2,Y,Y2,Z,0.)
      R3=SQ2F(X,X3,Y,Y3,Z,0.)
      R4=SQ2F(X,X4,Y,Y4,Z,0.)
      CLA1=ALOG((R1+R2-D12)/(R1+R2+D12))
      CLA2=ALOG((R2+R3-D23)/(R2+R3+D23))
      CLA3=ALOG((R3+R4-D34)/(R3+R4+D34))
      CLA4=ALOG((R4+R1-D41)/(R4+R1+D41))
      TVX=CY12*CLA1+CY23*CLA2+CY34*CLA3+CY41*CLA4

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```

TUV=CX12*CLA1+CX23*CLA2+CX34*CLA3+CX41*CLA4
TUZ=0.
IF (ABS(Z)-.001*ST)375,361,361
361 ZSQ=Z**2
E1=ZSQ+(X-X1)**2
E2=ZSQ+(X-X2)**2
E3=ZSQ+(X-X3)**2
E4=ZSQ+(X-X4)**2
H1=(Y-Y1)*(X-X1)
H2=(Y-Y2)*(X-X2)
H3=(Y-Y3)*(X-X3)
H4=(Y-Y4)*(X-X4)
IF (C112)363,363,364
363 WS1=(CM12*E1-H1)/(Z*R1)
WS2=(CM12*E2-H2)/(Z*R2)
AT1=ATAN(WS1)
AT2=ATAN(WS2)
TUZ=AT1-AT2
364 IF (C123)366,366,367
366 AT3=ATAN((CM23*E2-H2)/(Z*R2))
AT4=ATAN((CM23*E3-H3)/(Z*R3))
TUZ=TUZ+AT3-AT4
367 IF (C134)368,368,369
368 AT5=ATAN((CM34*E3-H3)/(Z*R3))
AT6=ATAN((CM34*E4-H4)/(Z*R4))
TUZ=TUZ+AT5-AT6
369 IF (C141)370,370,375
370 AT7=ATAN((CM41*E4-H4)/(Z*R4))
AT8=ATAN((CM41*E1-H1)/(Z*R1))
TUZ=TUZ+AT7-AT8
375 GO TO 450
C J. COMPUTE INDUSED VELOCITY BY QUADRAPOLE METHOD
400 RPQ3=RPQ**3
RPQ7=(RPQ3**2)*RPQ
WS1=X/RPQ3
XSQ=X**2
YSQ=Y**2
ZSQ=Z**2
PS=YSQ+ZSQ-4.*XSQ
QS=XSQ+ZSQ-4.*YSQ
WS2=X*(9.*PS+30.*XSQ)/RPQ7
WS3=3.*Y*PS/RPQ7
WS4=3.*Y*QS/RPQ7
TUX=A*WS1-C1XY*WS3-C1XX*WS2-C1YY*WS4
WS1=Y/RPQ3
WS2=Y*(9.*QS+30.*YSQ)/RPQ7
TUY=A*WS1-C1XX*WS3-C1XY*WS4-C1YY*WS2
TUZ=2*(A/RPQ3-3.*(C1XX*PS-5.*C1XY*X*Y+C1YY*QS)/RPQ7)
450 UX(I5)=TUX*XX+TUY*XY+TUZ*XN(K)
UY(I5)=TUX*YX+TUY*YY+TUZ*YN(K)
UZ(I5)=TUX*ZX+TUY*ZY+TUZ*ZN(K)
GO TO 470
C K. COMPUTE INDUSED VELOCITY BY MONOPOLE METHOD
460 ARPQ3=A/(RPQ**3)
UX(I5)=(XC0-XC)*ARPQ3
UY(I5)=(YC0-YC)*ARPQ3
UZ(I5)=(ZC0-ZC)*ARPQ3
C L. REFLECT OFF BODY POINT IN PLANE OF SYMETRY
470 GO TO(480,485,490,495,500,505,510,515),IS
480 U1(J1)=U(I5)
U2(J1)=UX(I5)
U3(J1)=UY(I5)

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U1(J1+1)=UY(1)
U2(J1+1)=UY(1)
U3(J1+1)=UY(1)
U1(J1+2)=UZ(1)
U2(J1+2)=UZ(1)
U3(J1+2)=UZ(1)
IF(SYM) 525,525,481
481 IS=2
C      XZ SYMETRY
      YCQ=-YCQ
      GO TO 345
485 IF(SYM-1)517,517,486
C      XY SYMETRY
486 IS=3
      ZCQ=-ZCQ
      GO TO 345
490 IS=4
      YCQ=-YCQ
      GO TO 345
495 IF(SYM-2)516,516,496
C      YZ SYMETRY
496 IS=5
      XCQ=-XCQ
      GO TO 345
500 IS=6
      YCQ=-YCQ
      GO TO 345
505 IS=7
      ZCQ=-ZCQ
      GO TO 345
510 IS=8
      YCQ=-YCQ
      GO TO 345
C      M. ADD CONTRIBUTIONS OF ALL REFLECTIONS
515 U1(J1)=U1(J1)+UX(8)+UX(7)+UX(6)+UX(5)
      U2(J1)=U2(J1)-UX(8)+UX(7)+UX(6)-UX(5)
      U3(J1)=U3(J1)-UX(8)-UX(7)+UX(6)+UX(5)
      U1(J1+1)=U1(J1+1)-UY(8)+UY(7)+UY(6)-UY(5)
      U2(J1+1)=U2(J1+1)+UY(8)+UY(7)+UY(6)+UY(5)
      U3(J1+1)=U3(J1+1)+UY(8)-UY(7)+UY(6)-UY(5)
      U1(J1+2)=U1(J1+2)-UZ(8)-UZ(7)+UZ(6)+UZ(5)
      U2(J1+2)=U2(J1+2)+UZ(8)-UZ(7)+UZ(6)-UZ(5)
      U3(J1+2)=U3(J1+2)+UZ(8)+UZ(7)+UZ(6)+UZ(5)
516 U1(J1)=U1(J1)+UX(4)+UX(3)
      U2(J1)=U2(J1)+UX(4)-UX(3)
      U3(J1)=U3(J1)-UX(4)-UX(3)
      U1(J1+1)=U1(J1+1)+UY(4)-UY(3)
      U2(J1+1)=U2(J1+1)+UY(4)+UY(3)
      U3(J1+1)=U3(J1+1)-UY(4)+UY(3)
      U1(J1+2)=U1(J1+2)-UZ(4)-UZ(3)
      U2(J1+2)=U2(J1+2)-UZ(4)+UZ(3)
      U3(J1+2)=U3(J1+2)+UZ(4)+UZ(3)
517 U1(J1)=U1(J1)+UX(2)
      U2(J1)=U2(J1)-UX(2)
      U3(J1)=U3(J1)+UX(2)
      U1(J1+1)=U1(J1+1)-UY(2)
      U2(J1+1)=U2(J1+1)+UY(2)
      U3(J1+1)=U3(J1+1)-UY(2)
      U1(J1+2)=U1(J1+2)+UZ(2)
      U2(J1+2)=U2(J1+2)-UZ(2)
      U3(J1+2)=U3(J1+2)+UZ(2)
525 UX1(J0)=UX1(J0)+U1(1)*S1(P)

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UY1(JQ)=UY1(JQ)+U1(2)*S1(P)
UZ1(JQ)=UZ1(JQ)+U1(3)*S1(P)
UX2(JQ)=UX2(JQ)+U2(1)*S2(P)
UY2(JQ)=UY2(JQ)+U2(2)*S2(P)
UZ2(JQ)=UZ2(JQ)+U2(3)*S2(P)
UX3(JQ)=UX3(JQ)+U3(1)*S3(P)
UY3(JQ)=UY3(JQ)+U3(2)*S3(P)
530 UZ3(JQ)=UZ3(JQ)+U3(3)*S3(P)
C   N.  END OF LOOP OVER OFF BODY POINTS
585 P=P+1
    K=K+1
    J=J+20
    IF(K-NP >296,296,599)
C   P.  END OF LOOP OVER QUADS
599 H=AP(IJK)*DT
    DO 730 I = 1,NOBP
63  FORMAT(2X,13,3F12.5,9X,4F12.5)
    DX(I)=- (UX1*UX1(I)+UY1*UY2(I)+UZ1*UX3(I))
    DY(I)=- (UX1*UY1(I)+UY1*UY2(I)+UZ1*UY3(I))
730  DZ(I)=- (UX1*UZ1(I)+UY1*UZ2(I)+UZ1*UZ3(I))
    IF(IJK.EQ.5) GO TO 900
    IF(IJK.EQ.4) GO TO 800
    DO 750 I=1,NOBP
    XP(I)=XT(I)+DX(I)*H
    YP(I)=YT(I)+DY(I)*H
    ZP(I)=ZT(I)+DZ(I)*H
    SKX(I)=SKX(I)+GM(IJK)*DX(I)
    SKY(I)=SKY(I)+GM(IJK)*DY(I)
750  SKZ(I)=SKZ(I)+GM(IJK)*DZ(I)
    IJK = IJK + 1
    GO TO 98
800 H=DT
    DO 830 I=1,NOBP
    DX(I)=- (UX1*UX1(I)+UY1*UY2(I)+UZ1*UX3(I))
    DY(I)=- (UX1*UY1(I)+UY1*UY2(I)+UZ1*UY3(I))
    DZ(I)=- (UX1*UZ1(I)+UY1*UZ2(I)+UZ1*UZ3(I))
    XP(I)=XT(I)+H*DX(I)/6.+SKX(I)*H
    XT(I)=XP(I)
    YP(I)=YT(I)+H*DY(I)/6.+SKY(I)*H
    YT(I)=YP(I)
    ZP(I)=ZT(I)+H*DZ(I)/6.+SKZ(I)*H
    ZT(I)=ZP(I)
    SKX(I)=0.
    SKY(I) = 0.
830  SKZ(I) = 0.
    IJK = 5
    GO TO 98
900 IJK = 1
    DO 905 I=1,NOBP
    DSQ=DX(I)**2+DY(I)**2+DZ(I)**2
    CP(I)=1.-DSQ/USQ
905  CONTINUE
    WRITE(6,61) ITC
61  FORMAT(6H0 STEP,14/)
    WRITE(6,62)
62  FORMAT(3X,4HLINE,5X,1HX,11X,1HY,11X,1HZ,20X,2HUX,10X,2HUY,
1    10X,2HUZ,10X,2HCF)
    WRITE(6,63) (I,XP(I),YP(I),ZP(I),DX(I),DY(I),DZ(I),CP(I),I=1,NOBP)
    WRITE(16) (XP(I),YP(I),ZP(I),I=1,NOBP)
    IF(ITC.EQ.NST) GO TO 910
    ITC=ITC+1
    GO TO 599

```

```
910 IF( IEND.EQ.0 ) GO TO 93
REWIND 03
REWIND 04
ENDFILE 16
REWIND 16
STOP 6
END
FUNCTION SQ2F(X1,X2,Y1,Y2,Z1,Z2)
X=X1-X2
Y=Y1-Y2
Z=Z1-Z2
RS=Z**2+Y**2+X**2
R=ABS(X)+ABS(Y)+ABS(Z)+ 1.0E-20
R= .3422*(R+(RS+RS)/R)
R= R+RS/R
SQ2F= .25*R+RS/R
RETURN
END
```

**APPENDIX VII - XYZPF SECTION PF7**

PROGRAM PFP7(TAPE7, INPUT=128, OUTPUT=128, TAPE5=INPUT, TAPE17,  
1TAPE6=OUTPUT, TAPE03, TAPE04, TAPE3=TAPE03, TAPE4=TAPE04)

C  
C  
C  
C  
C

XYZ POTENTIAL FLOW PROGRAM VERSION 4 AND VERSION 5 SECTION 7  
COMPUTES VELOCITIES AND PRESSURE COEFFICIENTS FOR  
ON BODY STREAMLINES

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COMMON X(658),Y(658),Z(658),XN(650),YN(650)
1,ZN(650),UX1(650),UY1(650),UZ1(650),UX2(650),UY2(650)
2,UZ2(650),UX3(650),UY3(650),UZ3(650),XC1(658),YC1(658)
3,XC2(658),YC2(658),XC3(658),XC4(658),YC4(658),
4X3(658),Y3(658),Z3(658),X4(658),Y4(658),Z4(658)
DIMENSION XL(150),YL(150),ZL(150),UX(150),UY(150),UZ(150),
1CP(150),GK1(150),GK2(150),H2(150),STML(150),URBS(150),NQUAD(150)
5,DMX(650),PROB(15),YOC3(658),SF(5),XCR(5),YCR(5),
7NSP(50),WS(220),XST(50),YST(50),ZST(50)
EQUIVALENCE (WS(201),NP),(YOC3,YOC2)
READ(03)(PROB(I), I=1, 15)
READ(03)(WS(I), I=1, 220)
READ(03)(X(I),Y(I),Z(I),XN(I),YN(I),ZN(I),
1SKIP, I=1, NP)
READ(03)SKIP
1UER =WS(220)
WRITE(6,5) 1UER
5 FORMAT(46HXYZ POTENTIAL FLOW PROGRAM SECTION 7, VERSION ,12)
IF (1UER.EQ.5) READ(03) SKIP
READ(03)SKIP
READ(03)SKIP
READ(03)SKIP
READ(03)(UX1(I),UY1(I),UZ1(I), I=1, NP)
READ(03)(UX2(I),UY2(I),UZ2(I), I=1, NP)
READ(03)(UX3(I),UY3(I),UZ3(I), I=1, NP)
REWIND 03
NB=(NP+11)/12
DO 80 I=1,NB
1FN=I*12
1S=1FN-11
READ(04) 0, (XC1(J),YC1(J),XC2(J),
1YC2(J),XC3(J),XC4(J),YC4(J),X3(J),Y3(J),
2Z3(J),X4(J),Y4(J),Z4(J), (SKIP,K=1,7), J=1S, 1FN)
NO=0
IF(NO.NE.1S) GO TO 450
80 CONTINUE
REWIND 04
DO 90 I=1,NP
D1=(XC1(I)**2+YC1(I)**2)*1.01
D2=(XC2(I)**2+YC2(I)**2)*1.01
D3=(XC3(I)**2+YC3(I)**2)*1.01
D4=(XC4(I)**2+YC4(I)**2)*1.01
90 DMX(I)=AMAX1(D1, D2, D3, D4)
11 FORMAT(3F12.4, 3I4, F12.4)
12 FORMAT(3F12.4, 14)
MID=75
100 READ(5, 11) UX1,UY1,UZ1,NLIN,MAXJ,1WRITE,AMACH
IF (EOF(5).NE.0.) NLIN=0.
MAXJ=MAXJ
IF (MAXJ.LE.0 .OR. MAXJ.GT. NP/2) MAXJ = NP/2
MINJ=MID-MAXJ
MAXJ=MID+MAXJ
IF (MAXJ.GT.MID*2) MAXJ=MID*2

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      IF (MINJ .LT. 1) MINJ = 1
      WRITE(7) NLIN
      WRITE(17) NLIN,UXI,UYI,UZI
      IF(NLIN.LE.0) GO TO 550
      WRITE(6,30) (PROB(I),I=1,15)
      WRITE(6,34) UXI,UYI,UZI,NLIN,MXJ,IWRITE,AMACH
34  FORMAT(34H00N BODY STREAMLINES - INPUT DATA /6H UXI =,F10.5/
16H UYI =,F10.5/6H UZI =,F10.5/6H NLIN=,I10/
2 6H JMAX=,I10,/,8H IWRITE=,I10,/,9H MACH NO=,F10.5)
      WRITE(6,38)
38  FORMAT(27H08STREAMLINE STARTING POINTS/5H LINE,11X,1HX,12X,1HY,
1 12X,1HZ,10X,3HNSP)
      LIN=NLIN
      DO 45 I=1,LIN
      READ(5,12) XST(I),YST(I),ZST(I),NSP(I)
      IF (EOF(5).EQ.0.) GO TO 45
      NLIN=I-1
      WRITE(6,42) NLIN,LIN
42  FORMAT(1H0,15,28H STREAMLINES SPECIFIED NOT ,13)
      IF(NLIN.LE.0) GO TO 550
      GO TO 48
45  WRITE(6,46) I,XST(I),YST(I),ZST(I),NSP(I)
46  FORMAT(1X,13,2X,3F13.5,19)
48  CONTINUE
      USQ=UXI**2+UYI**2+UZI**2
      IF (AMACH .EQ. 0.) GO TO 1130
C *** COMPUTE CRITICAL MACH NO.
      USD = 0
      DO 1100 I=1,NP
      US = (UXI*UX1(I)+UYI*UY2(I)+UZI*UY3(I))**2 +
1 (UXI*UY1(I)+UYI*UY2(I)+UZI*UY3(I))**2 +
2 (UXI*UZ1(I)+UYI*UZ2(I)+UZI*UZ3(I))**2
      IF (US .GT. USD) USD = US
1100 CONTINUE
      U = SQRT(USD/USQ)
      CMNA = 1./U
      DO 1110 I=1,3
      CMNE = (((CMNA**2+5.)/6.))**1.75/U
      CMNC = (((CMNA**2+5.)/6.))**1.75/U
1110 CMNA = (CMNA*CMNC-CMNE**2)/(CMNA+CMNC-2.*CMNE)
      WRITE(6,1120) CMNA
1120 FORMAT(21H CRITICAL MACH NO. =,F5.3)
1130 CONTINUE
C START LOOP OVER STREAMLINES
      DO 400 LL=1,NLIN
      DIAT=1.
101  JI=1
      AF=1.
      U(MID)=0.
      VY(MID)=0.
      UZ(MID)=0.
      CP(MID)=0.
      H2(MID)=1.
      GK1(MID)=0.
      GK2(MID)=0.
      STAL(MID)=0
102  NQ=NSP(LL)
      LND=NQ
      XL(MID)=XST(LL)
      VL(MID)=YST(LL)
      ZL(MID)=ZST(LL)
      J=MID

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      JL=J
C     SEPARATE CALCULATION OF SECOND
C     POINT FROM MAIN LOOP
      XLT=(XL(J)-X(NQ))*X3(NQ)+(VL(J)-Y(NQ))*Y3(NQ)
      1      + (ZL(J)-Z(NQ))*Z3(NQ)
      YLT=(XL(J)-X(NQ))*X4(NQ)+(VL(J)-Y(NQ))*Y4(NQ)
      1      + (ZL(J)-Z(NQ))*Z4(NQ)
      XL(J)=XLT*X3(NQ)+YLT*X4(NQ)+X(NQ)
      VL(J)=XLT*Y3(NQ)+YLT*Y4(NQ)+Y(NQ)
      ZL(J)=XLT*Z3(NQ)+YLT*Z4(NQ)+Z(NQ)
105  IQT=MOD(NQ,4) + 1
      GO TO (630,600,610,620) IQT
600  NR=NQ+1
      NU=NQ+2
      GO TO 107
610  NR=NQ+2
      NU=NQ-1
      GO TO 107
620  NR=NQ-2
      NU=NQ+1
      GO TO 107
630  NR=NQ-1
      NU=NQ-2
107  UXQ=-(UX1*UX1(NQ)+UY1*UX2(NQ)+UZ1*UX3(NQ))
      UYQ=-(UX1*UY1(NQ)+UY1*UY2(NQ)+UZ1*UY3(NQ))
      UZQ=-(UX1*UZ1(NQ)+UY1*UZ2(NQ)+UZ1*UZ3(NQ))
      UXR=-(UX1*UX1(NR)+UY1*UX2(NR)+UZ1*UX3(NR))
      UYR=-(UX1*UY1(NR)+UY1*UY2(NR)+UZ1*UY3(NR))
      UZR=-(UX1*UZ1(NR)+UY1*UZ2(NR)+UZ1*UZ3(NR))
      UXU=-(UX1*UX1(NU)+UY1*UX2(NU)+UZ1*UX3(NU))
      UYU=-(UX1*UY1(NU)+UY1*UY2(NU)+UZ1*UY3(NU))
      UZU=-(UX1*UZ1(NU)+UY1*UZ2(NU)+UZ1*UZ3(NU))
C     TRANSFORM VELOCITIES TO QUAD SYSTEM
      UQ=UXQ*X3(NQ)+UYQ*Y3(NQ)+UZQ*Z3(NQ)
      VQ=UXQ*X4(NQ)+UYQ*Y4(NQ)+UZQ*Z4(NQ)
      CSR=1./CXN(NQ)*XN(NR)+YN(NQ)*YN(NR)+ZN(NQ)*ZN(NR)
      UT=UXR*X3(NR)+UYR*Y3(NR)+UZR*Z3(NR)
      VT=(UXR*X4(NR)+UYR*Y4(NR)+UZR*Z4(NR))*CSR
      XXR= (X3(NR)*X3(NQ)+Y3(NR)*Y3(NQ)+Z3(NR)*Z3(NQ))
      XYR= (X4(NR)*X3(NQ)+Y4(NR)*Y3(NQ)+Z4(NR)*Z3(NQ))
      UR=UT*XXR+VT*XYR
      YXR= (X3(NR)*X4(NQ)+Y3(NR)*Y4(NQ)+Z3(NR)*Z4(NQ))
      YVR= (X4(NR)*X4(NQ)+Y4(NR)*Y4(NQ)+Z4(NR)*Z4(NQ))
      UR=UT*YXR+VT*YVR
      UU=UXU*X3(NQ)+UYU*Y3(NQ)+UZU*Z3(NQ)
      VU= (XN(NQ)*XN(NU)+YN(NQ)*YN(NU)+ZN(NQ)*ZN(NU))
      UQ=(UXU*X4(NQ)+UYU*Y4(NQ)+UZU*Z4(NQ))/VU
C     FIND RELATIVE COORDINATES OF NEIGHBORING QUADS
      XD=X(NR)-X(NQ)
      YD=Y(NR)-Y(NQ)
      ZD=Z(NR)-Z(NQ)
      XT =XD*X3(NR)+YD*Y3(NR)+ZD*Z3(NR)
      YTT=XD*X4(NR)+YD*Y4(NR)+ZD*Z4(NR)
      ZT =XD*XN(NR)+YD*YN(NR)+ZD*ZN(NR)
      YT=(-4*SQRT(YTT**2+ZT**2)+YTT*CSR+YTT)*CSR+.16666667
      XR=XT*XXR+YT*XYR
      YR=XT*YXR+YT*YVR
      XD=X(NU)-X(NQ)
      YD=Y(NU)-Y(NQ)
      ZD=Z(NU)-Z(NQ)
      YU=XD*X3(NQ)+YD*Y3(NQ)+ZD*Z3(NQ)
      VT=XD*X4(NQ)+YD*Y4(NQ)+ZD*Z4(NQ)

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      ZT=XD*YN(NQ)+YD*YV(NQ)+ZD*ZN(NQ)
      YU=(4.*SQRT(YT**2+ZT**2)+YT/CSU+YT)*.16666667
C     FIND COEFFICIENTS OF VELOCITY FUNCTIONS
      DEN=1./<XR*YU-XU*YR>
      U1=<<UR-UQ>*YU-<UJ-UQ>*YR> *DEN
      U2=-<<UR-UQ>*XU-<UJ-UQ>*XR >*DEN
      V1=<<UR-UQ>*YU-<UJ-UQ>*YR> *DEN
      V2=-<<UR-UQ>*XU-<UJ-UQ>*XR >*DEN
C     FIND VELOCITY AT STREAMLINE POINT
      USL=UQ+U1*XLT+U2*YLT
      USL=UQ+U1*XLT+U2*YLT
      UXP=USL*X3(NQ)+USL*X4(NQ)
      UYP=USL*Y3(NQ)+USL*Y4(NQ)
      UZP=USL*Z3(NQ)+USL*Z4(NQ)
C     FIND GEODESIC CURVATURES GK1,GK2
      USQD=USL**2+USL**2
      DEN=USQD*SQRT(USQD)
      GK1P=<USL*(USL*U2-USL*U2)-USL*(USL*U1-USL*U1)>/DEN
      GK2P=<USL*(USL*U1+USL*U2)-USL*(USL*U1+USL*U2)>/DEN
C     FIND LOCAL STREAM FUNCTION
      CXV=<U1*U2**2+U2*U2**2>/USQD
      CVY=<U2-UQ*X3*(U1+U2)>/USQD
      CXX=<U2-CVY-U1>
      CG=<XLT*UQ-YLT*UQ-CXV*XLT*YLT-CVY*YLT**2-CXX*XLT**2>
C     FIND STREAM FUNCTION AT CORNER POINTS
      XCR(1)=XC1(NQ)
      XCR(2)=XC2(NQ)
      XCR(3)=XC3(NQ)
      XCR(4)=XC4(NQ)
      XCR(5)=XCR(1)
      YCR(1)=YC1(NQ)
      YCR(2)=YC2(NQ)
      YCR(3)=YC3(NQ)
      YCR(4)=YC4(NQ)
      YCR(5)=YCR(1)
      DO 110 N=1,4
110 SF(N)=CG-UQ*XCR(N)+UQ*YCR(N)+CXV*XCR(N)*YCR(N)+CVY*YCR(N)**2
      +CXX*XCR(N)**2
      SF(5)=SF(1)
      TEST=C
      DO 120 N=1,4
      IF <SF(N)*SF(N+1) GE. 0. > GO TO 120
      XM=<XCR(N)+XCR(N+1)>*.5
      VM=<YCR(N)+YCR(N+1)>*.5
C     FIND INTERSECTION WITH SIDE OF QUAD.
      SFM=<CG-UQ*XM+UQ*VM+CXV*XM*VM+CVY*VM**2+CXX*XM**2>
      AC=2.*<SF(N)-2.*SFM+SF(N+1)>
      BC=2.*<SF(N)-4.*SFM+SF(N+1)>
      IF <AC .EQ. 0 > GO TO 113
      SR=SQRT<BC**2-4.*AC*SF(N)>
      TP=<BC+SR>/<2.*AC>
      IF <TP .LE. 1 .AND. TP .GE. 0 > GO TO 115
      TP=<BC-SR>/<2.*AC>
      GO TO 115
113 IF <BC .EQ. 0 > GO TO 120
      TP=SF(N)/BC
115 XNP=<(1.-TP)*XCR(N)+TP*XCR(N+1)>
      YNP=<(1.-TP)*YCR(N)+TP*YCR(N+1)>
      TESTP=<<XNP-XLT>*UQ+<YNP-YLT>*UQ>*DIRT
      IF <TESTP .LE. TEST > GO TO 120
      TEST=TESTP
      XNT=XNP

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VNT=VNF
120 CONTINUE
IF( TEST .EQ. 0) GO TO 280
C AVERAGE LAST VELOCITY AND CURVATURE
UX(J)=(UX(J)+UXP)*AF
UY(J)=(UY(J)+UYP)*AF
UZ(J)=(UZ(J)+UZP)*AF
GK1(J)=(GK1(J)+GK1P)*AF
GK2(J)=(GK2(J)+GK2P)*AF
H2(J)=H2(JL)*(2.-GK1(JL))*(STML(J)-STML(JL))/(2.+GK1(J))*
1 STML(J)-STML(JL))
CP(J)=1.-(UX(J)**2+UY(J)**2+UZ(J)**2)/US0
UABS(J)=SQRT(1.-CP(J))
C COMPUTE VELOCITY AT NEXT POINT
NQUAD(J)=NO
JL=J
J=J+1
USL=UQ+XNT*U1+VNT*U2
USL=UQ+XNT*U1+VNT*U2
UX(J)=USL*X3(NQ)+USL*X4(NQ)
UY(J)=USL*Y3(NQ)+USL*Y4(NQ)
UZ(J)=USL*Z3(NQ)+USL*Z4(NQ)
C COMPUTE GEODESIC CURVATURES
USQ0=USL**2+USL**2
DEN=USQ0*SQRT(USQ0)
GK1(J)=(USL*(USL*U2-USL*U2)-USL*(USL*U1-USL*U1))/DEN
GK2(J)=(USL*(USL*U1+USL*U2)-USL*(USL*U1+USL*U2))/DEN
CORD=SQRT((XNT-XLT)**2+(VNT-VLT)**2)*DIRT
STML(J)=STML(JL)+CORD
C COMPUTE H2
H2(J)=H2(JL)*(2.-CORD*GK1(JL))/(2.+CORD*GK1(J))
CP(J)=1.-USQ0/US0
UABS(J)=SQRT(1.-CP(J))
AF=.5
LNQ=NO
XL(J)=XNT*X3(NQ)+VNT*X4(NQ)+X(NQ)
YL(J)=XNT*Y3(NQ)+VNT*Y4(NQ)+Y(NQ)
ZL(J)=XNT*Z3(NQ)+VNT*Z4(NQ)+Z(NQ)
IF (J.LE. MINJ .OR. J.GE. MAXJ) GO TO 280
C FIND NEXT QUAD.
I=1
250 NO=I
IF(1.EQ.LNQ) GO TO 280
TEST=(XL(J)-X(I))**2+(YL(J)-Y(I))**2+
1(ZL(J)-Z(I))**2-DMY(I)
IF( TEST .GT. 0.) GO TO 280
DS1=(XC1(I)-XC2(I))**2+(YC1(I)-YC2(I))**2
DS2=(XC2(I)-XC3(I))**2+(YC2(I)-YC3(I))**2
DS3=(XC3(I)-XC4(I))**2+(YC3(I)-YC4(I))**2
DS4=(XC4(I)-XC1(I))**2+(YC4(I)-YC1(I))**2
XLT=(XL(J)-X(I))*X3(I)+(YL(J)-Y(I))*Y3(I)+
1(ZL(J)-Z(I))*Z3(I)
VLT=(XL(J)-X(I))*Y4(I)+(YL(J)-Y(I))*Y4(I)+
1(ZL(J)-Z(I))*Z4(I)
ZLT=(XL(J)-X(I))*XN(I)+(YL(J)-Y(I))*YN(I)+
1(ZL(J)-Z(I))*ZN(I)
ZS0=ZLT**2
TEST=ZS0-.1*DMY(I)
IF( TEST .GT. 0.) GO TO 280
RC1=SQRT(ZS0+(XLT-XC1(I))**2+(VLT-YC1(I))
1**2)
RC2=SQRT(ZS0+(XLT-XC2(I))**2+(VLT-YC2(I))**2)

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RC3=SQRT(ZSQ+(XLT-XC3(I))**2+(YLT-YC3(I))**2)
RC4=SQRT(ZSQ+(XLT-XC4(I))**2+(YLT-YC4(I))**2)
TEST= ((RC1+RC2)**2)-DS1 *1.21
IF(TEST.LT.0.) GO TO 105
TEST= ((RC2+RC3)**2)-DS2 *1.21
IF(TEST.LT.0.) GO TO 105
TEST= ((RC3+RC4)**2)-DS3 *1.21
IF(TEST.LT.0.) GO TO 105
TEST= ((RC4+RC1)**2)-DS4 *1.21
IF(TEST.LT.0.) GO TO 105
280 I=I+1
IF(I.LE.NP) GO TO 250
282 IF (DIAT .LT. 0.) GO TO 285
DIAT=-1.
JI=-1
JMAX=J
GO TO 102
285 JMIN=J
SS=STML(JMIN)
DO 290 J=JMIN,JMAX
290 STML(J)=STML(J)-SS
JMN=JMIN+1
JMX=JMAX-2
AF=1.
L=JMN
WRITE(6,30)(PROB(I),I=1,15)
30 FORMAT(1H1,15A4)
WRITE(6,20)UX1,UY1,UZ1
20 FORMAT(18H0 ONSET FLOW, UX1=,F6.3,2X,4HUY1=,F6.3,2X,4HUZ1=,F6.3)
WRITE(6,50) LL,NSP(LL),XST(LL),YST(LL),ZST(LL)
50 FORMAT(11H0 LINE NO. ,12,31H PASSING THROUGH QUADRILATERAL ,13,
1 28H WITH STARTING POINT, X=,F12.5,2X,2HY=,F12.5,2X,
2 2HZ=,F12.5 //)
IF (JMIN.LE.MINU .OR. JMAX.GE.MAXJ) WRITE(6,55)
65 FORMAT(35H PROBABLE ERROR - LINE IS VERY LONG)
DO 330 J=JMN,JMX
IF ((STML(J+2)-STML(L-1)).LT.8.*(STML(J+1)-STML(L))) GO TO 320
WRITE(6,310) XL(L),YL(L),ZL(L),XL(J+1),YL(J+1),ZL(J+1)
310 FORMAT(14H POINT DELETED ,10X,3F12.5,10X,3F12.5)
STML(L)=(AF*STML(L)+STML(J+1))/(AF+1.)
XL(L)=(AF*XL(L)+XL(J+1))/(AF+1.)
YL(L)=(AF*YL(L)+YL(J+1))/(AF+1.)
ZL(L)=(AF*ZL(L)+ZL(J+1))/(AF+1.)
UX(L)=(AF*UX(L)+UX(J+1))/(AF+1.)
UY(L)=(AF*UY(L)+UY(J+1))/(AF+1.)
UZ(L)=(AF*UZ(L)+UZ(J+1))/(AF+1.)
GK1(L)=(AF*GK1(L)+GK1(J+1))/(AF+1.)
GK2(L)=(AF*GK2(L)+GK2(J+1))/(AF+1.)
H2(L)=(AF*H2(L)+H2(J+1))/(AF+1.)
CP(L)=1.-(UX(L)**2+UY(L)**2+UZ(L)**2)/USQ
UABS(L)=SQRT(1.-CP(L))
AF=AF+1.
GO TO 330
320 AF=1.
L=L+1
K=J+1
STML(L)=STML(K)
XL(L)=XL(K)
YL(L)=YL(K)
ZL(L)=ZL(K)
UX(L)=UX(K)
UY(L)=UY(K)

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UZ(L)=UZ(K)
GK1(L)=GK1(K)
GK2(L)=GK2(K)
H2(L)=H2(K)
CP(L)=CP(K)
UABS(L)=UABS(K)
NQAD(L)=NQAD(K)
330 CONTINUE
L=L+1
STML(L)=STML(JMAX)
XL(L)=XL(JMAX)
YL(L)=YL(JMAX)
ZL(L)=ZL(JMAX)
UX(L)=UX(JMAX)
UY(L)=UY(JMAX)
UZ(L)=UZ(JMAX)
GK1(L)=GK1(JMAX)
GK2(L)=GK2(JMAX)
H2(L)=H2(JMAX)
CP(L)=CP(JMAX)
UABS(L)=UABS(JMAX)
JMAX=L
NQAD(JMAX)=NQAD(JMAX-1)
NQAD(JMIN)=NQAD(JMIN+1)
WRITE(6,51)
51 FORMAT(4H0 1,6X,1HX,9X,1HY,
19X,1HZ,09X,2HUX,8X,2HUY,8X,2HUZ,09X,
22HCP, 8X,2HK1, 8X,2HK2, 8X,2HHZ,8X,2HSL,8X,1HV,9X,1HP)
IF (AMACH.EQ.0.) GOTO 1160
C *** COMPUTE COMPRESSIBILITY CORRECTION
DO 1150 J=JMIN,JMAX
USD = (UX(J)**2+UY(J)**2+UZ(J)**2)/USQ
USDA = USD
SM = AMACH**2
DO 1140 I=1,3
R = (1.+2*SM*(1.-USDA))
IF (R.LT..000001) R = .000001
USDB = USD/R**2.5
R = (1.+2*SM*(1.-USDB))
IF (R.LT..000001) R = .000001
USDC = USD/R**2.5
1140 USDA = (USDC*USDA-USDB**2)/(USDC+USDA-2.*USDB)
R = (1.+2*SM*(1.-USDA))
IF (R.LT..000001) R = .000001
R = R**1.25
UX(J) = UX(J)/R
UY(J) = UY(J)/R
UZ(J) = UZ(J)/R
UABS(J) = SQRT(USDA)
1150 CP(J) = (R**2.8-1.)/(1.7*SM)
1150 CONTINUE
K=0
DO 53 I=JMIN,JMAX
K=K+1
53 WRITE(6,60) K,XL(I),YL(I),ZL(I),UX(I),UY(I),UZ(I),CP(I),
1 GK1(I),GK2(I),H2(I),STML(I),UABS(I),NQAD(I)
60 FORMAT(1X,13,3F10.5,1X,3F10.5,1X,6F10.5,1E)
8 FORMAT(3F12.5)
WRITE(17) K, (XL(I),YL(I),ZL(I),NQAD(I), I=JMIN,JMAX)
C WRITE .LE. 0 -- WRITE SL,U,H2,K2
C WRITE .GE. 2 -- WRITE X,Y,Z,CP
C WRITE .EQ. 1 -- WRITE SL,U,H2,K2 AND X,Y,Z,CP

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      IF (IWRITE.GT.1) GO TO 340
      WRITE(7) K, (STML(I),UABS(I),H2(I),GK2(I),I=JMIN,JMAX)
340  IF (IWRITE.LT.1) GO TO 400
      WRITE(7) K, (XL(I),YL(I),ZL(I),CP(I), I=JMIN,JMAX)
      GO TO 400
300  WRITE(6,50) NSP(LL)
      WRITE(6,65)
      GO TO 282
400  CONTINUE
      GO TO 100
C    READ NEXT SET OF STREAMLINES
450  WRITE(6,451)IS,NQ
451  FORMAT(14H TAPE 04 ERROR,2I4)
550  ENDFILE 7
      REWIND 7
      ENDFILE 17
      REWIND 17
      REWIND 04
      STOP 7
      END

```

APPENDIX VIII - TRIAXIAL ELLIPSOID INPUT FILE

SAMPLE	PROBLEM	TRIAIAL	ELLIPSOID												
280	5	150	150	150	3	.00001	0	0	0	0	0	0	0	0	.000 .000 .0
	1.00000		.00000			.00000	1	1	1	0					.00000
	.97861		.00000			.10286	1	2	1	0					.00000
	.90789		.00000			.20960	1	3	1	0					.00000
	.82360		.00000			.28359	1	4	1	0					.00000
	.71583		.00000			.34914	1	5	1	0					.00000
	.62478		.00000			.39040	1	6	1	0					.00000
	.52537		.00000			.42544	1	7	1	0					.00000
	.44721		.00000			.44721	1	8	1	0					.00000
	.37530		.00000			.46345	1	9	1	0					.00000
	.29822		.00000			.47725	1	10	1	0					.00000
	.23692		.00000			.48576	1	11	1	0					.00000
	.16967		.00000			.49275	1	12	1	0					.00000
	.11467		.00000			.49670	1	13	1	0					.00000
	.05248		.00000			.49931	1	14	1	0					.00000
	.00000		.00000			.50000	1	15	1	0					.00000
	.99875		.10000			.00000	2	1	1	0					.00000
	.97739		.10000			.10273	2	2	1	0					.00000
	.90676		.10000			.20934	2	3	1	0					.00000
	.82257		.10000			.28323	2	4	1	0					.00000
	.71494		.10000			.34870	2	5	1	0					.00000
	.62399		.10000			.38991	2	6	1	0					.00000
	.52471		.10000			.42490	2	7	1	0					.00000
	.44565		.10000			.44665	2	8	1	0					.00000
	.37463		.10000			.46287	2	9	1	0					.00000
	.29785		.10000			.47665	2	10	1	0					.00000
	.23663		.10000			.48516	2	11	1	0					.00000
	.16946		.10000			.49213	2	12	1	0					.00000
	.11453		.10000			.49608	2	13	1	0					.00000
	.05241		.10000			.49859	2	14	1	0					.00000
	.00000		.10000			.49937	2	15	1	0					.00000
	.99499		.20000			.00000	3	1	1	0					.00000
	.97371		.20000			.10234	3	2	1	0					.00000
	.90334		.20000			.20855	3	3	1	0					.00000
	.81947		.20000			.28217	3	4	1	0					.00000
	.71225		.20000			.34739	3	5	1	0					.00000
	.62164		.20000			.38845	3	6	1	0					.00000
	.52274		.20000			.42330	3	7	1	0					.00000
	.44497		.20000			.44497	3	8	1	0					.00000
	.37342		.20000			.45113	3	9	1	0					.00000
	.29672		.20000			.47486	3	10	1	0					.00000
	.23574		.20000			.48333	3	11	1	0					.00000
	.16952		.20000			.49028	3	12	1	0					.00000
	.11410		.20000			.49421	3	13	1	0					.00000
	.05222		.20000			.49581	3	14	1	0					.00000
	.00000		.20000			.49749	3	15	1	0					.00000
	.98869		.30000			.00000	4	1	1	0					.00000
	.96754		.30000			.10169	4	2	1	0					.00000
	.89762		.30000			.20723	4	3	1	0					.00000
	.81428		.30000			.28038	4	4	1	0					.00000
	.70774		.30000			.34519	4	5	1	0					.00000
	.61771		.30000			.38599	4	6	1	0					.00000
	.51943		.30000			.42052	4	7	1	0					.00000
	.44215		.30000			.44215	4	8	1	0					.00000
	.37105		.30000			.45821	4	9	1	0					.00000
	.29484		.30000			.47185	4	10	1	0					.00000
	.23424		.30000			.48027	4	11	1	0					.00000
	.16775		.30000			.48718	4	12	1	0					.00000

.11338	.30000	.49108	4	13	1	0	.00000
.05189	.30000	.49366	4	14	1	0	.00000
.00000	.30000	.49434	4	15	1	0	.00000
.97980	.40000	.00000	5	1	1	0	.00000
.95884	.40000	.10078	5	2	1	0	.00000
.88955	.40000	.20537	5	3	1	0	.00000
.80696	.40000	.27786	5	4	1	0	.00000
.70137	.40000	.34208	5	5	1	0	.00000
.61215	.40000	.38251	5	6	1	0	.00000
.51476	.40000	.41684	5	7	1	0	.00000
.43818	.40000	.43818	5	8	1	0	.00000
.36771	.40000	.45409	5	9	1	0	.00000
.29219	.40000	.46761	5	10	1	0	.00000
.23214	.40000	.47595	5	11	1	0	.00000
.16624	.40000	.48280	5	12	1	0	.00000
.11236	.40000	.48667	5	13	1	0	.00000
.05142	.40000	.48922	5	14	1	0	.00000
.00000	.40000	.48990	5	15	1	0	.00000
.97980	.40000	.00000	6	1	2	0	.00000
.95884	.40000	.10078	6	2	2	0	.00000
.88955	.40000	.20537	6	3	2	0	.00000
.80696	.40000	.27786	6	4	2	0	.00000
.70137	.40000	.34208	6	5	2	0	.00000
.61215	.40000	.38251	6	6	2	0	.00000
.51476	.40000	.41684	6	7	2	0	.00000
.43818	.40000	.43818	6	8	2	0	.00000
.36771	.40000	.45409	6	9	2	0	.00000
.29219	.40000	.46761	6	10	2	0	.00000
.23214	.40000	.47595	6	11	2	0	.00000
.16624	.40000	.48280	6	12	2	0	.00000
.11236	.40000	.48667	6	13	2	0	.00000
.05142	.40000	.48922	6	14	2	0	.00000
.00000	.40000	.48990	6	15	2	0	.00000
.95825	.50000	.00000	7	1	2	0	.00000
.94754	.50000	.09959	7	2	2	0	.00000
.87906	.50000	.20295	7	3	2	0	.00000
.79745	.50000	.27458	7	4	2	0	.00000
.69310	.50000	.33805	7	5	2	0	.00000
.60494	.50000	.37801	7	6	2	0	.00000
.50869	.50000	.41193	7	7	2	0	.00000
.43301	.50000	.43301	7	8	2	0	.00000
.36338	.50000	.44874	7	9	2	0	.00000
.28875	.50000	.46209	7	10	2	0	.00000
.22940	.50000	.47034	7	11	2	0	.00000
.16428	.50000	.47710	7	12	2	0	.00000
.11103	.50000	.48093	7	13	2	0	.00000
.05081	.50000	.48346	7	14	2	0	.00000
.00000	.50000	.48412	7	15	2	0	.00000
.95334	.60000	.00000	8	1	2	0	.00000
.93354	.60000	.09212	8	2	2	0	.00000
.85507	.60000	.19995	8	3	2	0	.00000
.78566	.60000	.27052	8	4	2	0	.00000
.68286	.60000	.33305	8	5	2	0	.00000
.59500	.60000	.37242	8	6	2	0	.00000
.50117	.60000	.40564	8	7	2	0	.00000
.42662	.60000	.42661	8	8	2	0	.00000
.35801	.60000	.44111	8	9	2	0	.00000
.28448	.60000	.45527	8	10	2	0	.00000
.22601	.60000	.46339	8	11	2	0	.00000
.16185	.60000	.47005	8	12	2	0	.00000
.10939	.60000	.47382	8	13	2	0	.00000

.05006	.60000	.47531	8	14	2	0	.00000
.00000	.60000	.47697	8	15	2	0	.00000
.93675	.70000	.00000	9	1	2	0	.00000
.91671	.70000	.09635	9	2	2	0	.00000
.85047	.70000	.19635	9	3	2	0	.00000
.77150	.70000	.26565	9	4	2	0	.00000
.67056	.70000	.32705	9	5	2	0	.00000
.58526	.70000	.36571	9	6	2	0	.00000
.49214	.70000	.39853	9	7	2	0	.00000
.41893	.70000	.41893	9	8	2	0	.00000
.35156	.70000	.43414	9	9	2	0	.00000
.27936	.70000	.44706	9	10	2	0	.00000
.22194	.70000	.45504	9	11	2	0	.00000
.15894	.70000	.45158	9	12	2	0	.00000
.10742	.70000	.46529	9	13	2	0	.00000
.04916	.70000	.46773	9	14	2	0	.00000
.00000	.70000	.46837	9	15	2	0	.00000
.91652	.80000	.00000	10	1	2	0	.00000
.89691	.80000	.09427	10	2	2	0	.00000
.83210	.80000	.19210	10	3	2	0	.00000
.75484	.80000	.25991	10	4	2	0	.00000
.65607	.80000	.31999	10	5	2	0	.00000
.57262	.80000	.35781	10	6	2	0	.00000
.48151	.80000	.38992	10	7	2	0	.00000
.40988	.80000	.40988	10	8	2	0	.00000
.34397	.80000	.42476	10	9	2	0	.00000
.27332	.80000	.43741	10	10	2	0	.00000
.21714	.80000	.44521	10	11	2	0	.00000
.15550	.80000	.45161	10	12	2	0	.00000
.10510	.80000	.45523	10	13	2	0	.00000
.04810	.80000	.45763	10	14	2	0	.00000
.00000	.80000	.45826	10	15	2	0	.00000
.91652	.80000	.00000	11	1	3	0	.00000
.89691	.80000	.09427	11	2	3	0	.00000
.83210	.80000	.19210	11	3	3	0	.00000
.75484	.80000	.25991	11	4	3	0	.00000
.65607	.80000	.31999	11	5	3	0	.00000
.57262	.80000	.35781	11	6	3	0	.00000
.48151	.80000	.38992	11	7	3	0	.00000
.40988	.80000	.40988	11	8	3	0	.00000
.34397	.80000	.42476	11	9	3	0	.00000
.27332	.80000	.43741	11	10	3	0	.00000
.21714	.80000	.44521	11	11	3	0	.00000
.15550	.80000	.45161	11	12	3	0	.00000
.10510	.80000	.45523	11	13	3	0	.00000
.04810	.80000	.45763	11	14	3	0	.00000
.00000	.80000	.45826	11	15	3	0	.00000
.89303	.90000	.00000	12	1	3	0	.00000
.87393	.90000	.09185	12	2	3	0	.00000
.81077	.90000	.18718	12	3	3	0	.00000
.73550	.90000	.25325	12	4	3	0	.00000
.63926	.90000	.31179	12	5	3	0	.00000
.55794	.90000	.34854	12	6	3	0	.00000
.46917	.90000	.37993	12	7	3	0	.00000
.39937	.90000	.39937	12	8	3	0	.00000
.33515	.90000	.41368	12	9	3	0	.00000
.26632	.90000	.42620	12	10	3	0	.00000
.21158	.90000	.43380	12	11	3	0	.00000
.15152	.90000	.44004	12	12	3	0	.00000
.10241	.90000	.44357	12	13	3	0	.00000
.04687	.90000	.44590	12	14	3	0	.00000

.00000	.90000	.44651	12	15	3	0	.00000
.86603	1.00000	.00000	13	1	3	0	.00000
.84750	1.00000	.08908	13	2	3	0	.00000
.78626	1.00000	.18152	13	3	3	0	.00000
.71326	1.00000	.24559	13	4	3	0	.00000
.61993	1.00000	.30236	13	5	3	0	.00000
.54107	1.00000	.33810	13	6	3	0	.00000
.45498	1.00000	.36844	13	7	3	0	.00000
.38730	1.00000	.38730	13	8	3	0	.00000
.32502	1.00000	.40136	13	9	3	0	.00000
.25826	1.00000	.41331	13	10	3	0	.00000
.20518	1.00000	.42068	13	11	3	0	.00000
.14594	1.00000	.42573	13	12	3	0	.00000
.09931	1.00000	.43016	13	13	3	0	.00000
.04545	1.00000	.43242	13	14	3	0	.00000
.00000	1.00000	.43301	13	15	3	0	.00000
.83516	1.10000	.00000	14	1	3	0	.00000
.81730	1.10000	.08590	14	2	3	0	.00000
.75824	1.10000	.17505	14	3	3	0	.00000
.68784	1.10000	.23684	14	4	3	0	.00000
.59784	1.10000	.29159	14	5	3	0	.00000
.52179	1.10000	.32605	14	6	3	0	.00000
.43877	1.10000	.35531	14	7	3	0	.00000
.37350	1.10000	.37350	14	8	3	0	.00000
.31343	1.10000	.38706	14	9	3	0	.00000
.24905	1.10000	.39858	14	10	3	0	.00000
.19787	1.10000	.40559	14	11	3	0	.00000
.14170	1.10000	.41153	14	12	3	0	.00000
.09577	1.10000	.41483	14	13	3	0	.00000
.04383	1.10000	.41701	14	14	3	0	.00000
.00000	1.10000	.41758	14	15	3	0	.00000
.80000	1.20000	.00000	15	1	3	0	.00000
.78289	1.20000	.08228	15	2	3	0	.00000
.72631	1.20000	.16768	15	3	3	0	.00000
.65888	1.20000	.22687	15	4	3	0	.00000
.57267	1.20000	.27931	15	5	3	0	.00000
.49982	1.20000	.31232	15	6	3	0	.00000
.42030	1.20000	.34035	15	7	3	0	.00000
.35777	1.20000	.35777	15	8	3	0	.00000
.30024	1.20000	.37076	15	9	3	0	.00000
.23957	1.20000	.38180	15	10	3	0	.00000
.18954	1.20000	.38861	15	11	3	0	.00000
.13573	1.20000	.39420	15	12	3	0	.00000
.09174	1.20000	.39736	15	13	3	0	.00000
.04198	1.20000	.39945	15	14	3	0	.00000
.00000	1.20000	.40000	15	15	3	0	.00000
.80000	1.20000	.00000	16	1	4	0	.00000
.78289	1.20000	.08228	16	2	4	0	.00000
.72631	1.20000	.16768	16	3	4	0	.00000
.65888	1.20000	.22687	16	4	4	0	.00000
.57267	1.20000	.27931	16	5	4	0	.00000
.49982	1.20000	.31232	16	6	4	0	.00000
.42030	1.20000	.34035	16	7	4	0	.00000
.35777	1.20000	.35777	16	8	4	0	.00000
.30024	1.20000	.37076	16	9	4	0	.00000
.23957	1.20000	.38180	16	10	4	0	.00000
.18954	1.20000	.38861	16	11	4	0	.00000
.13573	1.20000	.39420	16	12	4	0	.00000
.09174	1.20000	.39736	16	13	4	0	.00000
.04198	1.20000	.39945	16	14	4	0	.00000
.00000	1.20000	.40000	16	15	4	0	.00000

.75993	1.30000	.00000	17	1	4	0	.00000
.74368	1.30000	.07816	17	2	4	0	.00000
.68994	1.30000	.15928	17	3	4	0	.00000
.62588	1.30000	.21551	17	4	4	0	.00000
.54299	1.30000	.26532	17	5	4	0	.00000
.47479	1.30000	.29668	17	6	4	0	.00000
.39925	1.30000	.32330	17	7	4	0	.00000
.32985	1.30000	.33985	17	8	4	0	.00000
.28520	1.30000	.35219	17	9	4	0	.00000
.22663	1.30000	.36268	17	10	4	0	.00000
.18005	1.30000	.36915	17	11	4	0	.00000
.12894	1.30000	.37446	17	12	4	0	.00000
.08714	1.30000	.37746	17	13	4	0	.00000
.03988	1.30000	.37944	17	14	4	0	.00000
.00000	1.30000	.37997	17	15	4	0	.00000
.71414	1.40000	.00000	18	1	4	0	.00000
.69887	1.40000	.07345	18	2	4	0	.00000
.64836	1.40000	.14969	18	3	4	0	.00000
.58817	1.40000	.20252	18	4	4	0	.00000
.51121	1.40000	.24933	18	5	4	0	.00000
.44518	1.40000	.27880	18	6	4	0	.00000
.37519	1.40000	.30382	18	7	4	0	.00000
.31937	1.40000	.31937	18	8	4	0	.00000
.26902	1.40000	.33097	18	9	4	0	.00000
.21297	1.40000	.34082	18	10	4	0	.00000
.16920	1.40000	.34690	18	11	4	0	.00000
.12117	1.40000	.35189	18	12	4	0	.00000
.08189	1.40000	.35472	18	13	4	0	.00000
.03748	1.40000	.35558	18	14	4	0	.00000
.00000	1.40000	.35707	18	15	4	0	.00000
.66144	1.50000	.00000	19	1	4	0	.00000
.64729	1.50000	.06903	19	2	4	0	.00000
.60051	1.50000	.13854	19	3	4	0	.00000
.54476	1.50000	.18758	19	4	4	0	.00000
.47348	1.50000	.23093	19	5	4	0	.00000
.41325	1.50000	.25823	19	6	4	0	.00000
.34750	1.50000	.28140	19	7	4	0	.00000
.29580	1.50000	.29580	19	8	4	0	.00000
.24824	1.50000	.30654	19	9	4	0	.00000
.19725	1.50000	.31567	19	10	4	0	.00000
.15671	1.50000	.32130	19	11	4	0	.00000
.11222	1.50000	.32592	19	12	4	0	.00000
.07585	1.50000	.32854	19	13	4	0	.00000
.03471	1.50000	.33026	19	14	4	0	.00000
.00000	1.50000	.33072	19	15	4	0	.00000
.60000	1.60000	.00000	20	1	4	0	.00000
.58717	1.60000	.06171	20	2	4	0	.00000
.54473	1.60000	.12576	20	3	4	0	.00000
.49416	1.60000	.17015	20	4	4	0	.00000
.42950	1.60000	.20948	20	5	4	0	.00000
.37487	1.60000	.23424	20	6	4	0	.00000
.31522	1.60000	.25526	20	7	4	0	.00000
.25233	1.60000	.26833	20	8	4	0	.00000
.22518	1.60000	.27807	20	9	4	0	.00000
.17893	1.60000	.28635	20	10	4	0	.00000
.14215	1.60000	.29145	20	11	4	0	.00000
.10180	1.60000	.29565	20	12	4	0	.00000
.06890	1.60000	.29802	20	13	4	0	.00000
.03143	1.60000	.29959	20	14	4	0	.00000
.00000	1.60000	.30000	20	15	4	0	.00000
.60000	1.60000	.00000	21	1	5	0	.00000



.58717	1.60000	.06171	21	2	5	0	.00000
.54473	1.60000	.12576	21	3	5	0	.00000
.49416	1.60000	.17015	21	4	5	0	.00000
.42950	1.60000	.20948	21	5	5	0	.00000
.37487	1.60000	.23424	21	6	5	0	.00000
.31522	1.60000	.25526	21	7	5	0	.00000
.26833	1.60000	.26833	21	8	5	0	.00000
.22518	1.60000	.27807	21	9	5	0	.00000
.17893	1.60000	.28635	21	10	5	0	.00000
.14215	1.60000	.29146	21	11	5	0	.00000
.10180	1.60000	.29565	21	12	5	0	.00000
.06880	1.60000	.29802	21	13	5	0	.00000
.03149	1.60000	.29959	21	14	5	0	.00000
.00000	1.60000	.30000	21	15	5	0	.00000
.52678	1.70000	.00000	22	1	5	0	.00000
.51552	1.70000	.05418	22	2	5	0	.00000
.47826	1.70000	.11042	22	3	5	0	.00000
.43386	1.70000	.14939	22	4	5	0	.00000
.37709	1.70000	.18392	22	5	5	0	.00000
.32912	1.70000	.20566	22	6	5	0	.00000
.27676	1.70000	.22411	22	7	5	0	.00000
.23553	1.70000	.23558	22	8	5	0	.00000
.19770	1.70000	.24414	22	9	5	0	.00000
.15710	1.70000	.25141	22	10	5	0	.00000
.12481	1.70000	.25589	22	11	5	0	.00000
.08938	1.70000	.25957	22	12	5	0	.00000
.05041	1.70000	.26165	22	13	5	0	.00000
.02765	1.70000	.26303	22	14	5	0	.00000
.00000	1.70000	.26339	22	15	5	0	.00000
.43539	1.80000	.00000	23	1	5	0	.00000
.42657	1.80000	.04483	23	2	5	0	.00000
.39574	1.80000	.09136	23	3	5	0	.00000
.35900	1.80000	.12361	23	4	5	0	.00000
.31202	1.80000	.15218	23	5	5	0	.00000
.27233	1.80000	.17017	23	6	5	0	.00000
.22900	1.80000	.18544	23	7	5	0	.00000
.19494	1.80000	.19494	23	8	5	0	.00000
.16359	1.80000	.20201	23	9	5	0	.00000
.12999	1.80000	.20803	23	10	5	0	.00000
.10327	1.80000	.21174	23	11	5	0	.00000
.07396	1.80000	.21479	23	12	5	0	.00000
.04996	1.80000	.21651	23	13	5	0	.00000
.02288	1.80000	.21764	23	14	5	0	.00000
.00000	1.80000	.21794	23	15	5	0	.00000
.31225	1.90000	.00000	24	1	5	0	.00000
.30557	1.90000	.03212	24	2	5	0	.00000
.28349	1.90000	.06545	24	3	5	0	.00000
.25717	1.90000	.08855	24	4	5	0	.00000
.22352	1.90000	.10902	24	5	5	0	.00000
.19509	1.90000	.12190	24	6	5	0	.00000
.16405	1.90000	.13234	24	7	5	0	.00000
.13964	1.90000	.13954	24	8	5	0	.00000
.11719	1.90000	.14471	24	9	5	0	.00000
.09312	1.90000	.14902	24	10	5	0	.00000
.07338	1.90000	.15168	24	11	5	0	.00000
.05292	1.90000	.15385	24	12	5	0	.00000
.03581	1.90000	.15519	24	13	5	0	.00000
.01639	1.90000	.15591	24	14	5	0	.00000
.00000	1.90000	.15612	24	15	5	0	.00000
.00000	2.00000	.00000	25	1	5	0	.00000
.00000	2.00000	.00000	25	2	5	0	.00000

.00000	2.00000	.00000	25	3	5	0	.00000
.00000	2.00000	.00000	25	4	5	0	.00000
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.00000	2.00000	.00000	25	11	5	0	.00000
.00000	2.00000	.00000	25	12	5	0	.00000
.00000	2.00000	.00000	25	13	5	0	.00000
.00000	2.00000	.00000	25	14	5	0	.00000
.00000	2.00000	.00000	25	15	5	0	.00000

(EOR)

3	0	0					
2.00000	.00000	.00000					
.00000	.00000	1.50000					
.00000	3.00000	.00000					

(EOR)

2	20	1	-1.00000	.10000	.00000	.00000	
1.00000	1.00000	.00000					
1.50000	.00000	.00000					

(EOR)

-1.0000	.0000	.0000	1	0	1		
1.0000	.0500	.0000	1				

(EOR)

(EOR)

APPENDIX IX TRIAXIAL ELLIPSOID OUTPUT FILE  
XYZ POTENTIAL FLOW PROGRAM SECTION 4, VERSION 4

SAMPLE PROBLEM TRIAXIAL ELLIPSOID

NU. OF QUADS. = 290  
NU. OF SECTIONS = 3  
MAX. NO. OF ITERATIONS X FLOW 100 Y FLOW 100 Z FLOW 100  
3 PLANES OF SYMMETRY  
CONVERGENCE CRITERIA .00001

ISP = 0  
IEDIT1 = 0  
IEDIT3 = 0  
IEDIT4 = 0  
ITAPE = 0  
XCENTER = .00  
YCENTER = .00  
ZCENTER = .00



N P Y1 Y2 Y3 Y4 Y5 Y6 Y7 Y8 Y9 Y0  
 7 2 14 8 1 15 8 2 16 9 1 17 9 2 18 10 1 19 10 2 20 11 1 21 11 2 22 12 1 23 12 2 24 13 1 25 13 2 26

N	P	Y1	Y2	Y3	Y4	Y5	Y6	Y7	Y8	Y9	Y0	FL	CL	CL6
7		52471E+00	44605E+00	44997E+00	52279E+00	46477E+00	26835E+00	80900E-02	-32183E+00					
2		10000E+00	10000E+00	20000E+00	20000E+00	14997E+00	20727E-01	64733E-01	-68965E-01					
14		44450E+00	44005E+00	44997E+00	42330E+00	43996E+00	90310E+00	69804E-03	-85764E-02					
8		44721E+00	37530E+00	37463E+00	44605E+00	41100E+00	22024E+00	73677E-02	-32082E+00					
1		00000E+00	00000E+00	10000E+00	10000E+00	49990E-01	67038E-02	62218E-01	-67036E-01					
15		44721E+00	46345E+00	45287E+00	44605E+00	43505E+00	97541E+00	60291E-03	-24607E-02					
8		44665E+00	37463E+00	37342E+00	44497E+00	40997E+00	22026E+00	73505E-02	-67020E-01					
2		10000E+00	10000E+00	20000E+00	20000E+00	14997E+00	20108E-01	62395E-01	-32189E+00					
16		44665E+00	46287E+00	45113E+00	44497E+00	43539E+00	97523E+00	60228E-03	94007E-03					
9		37530E+00	24922E+00	24762E+00	37463E+00	33655E+00	17622E+00	78256E-02	-65399E-01					
1		00000E+00	00000E+00	10000E+00	10000E+00	49990E-01	65574E-02	63586E-01	-27940E+00					
17		46345E+00	47725E+00	47667E+00	46287E+00	47666E+00	98433E+00	59108E-03	35907E-03					
9		37483E+00	24762E+00	24762E+00	37342E+00	33571E+00	17617E+00	78076E-02	-27971E+00					
2		10000E+00	10000E+00	20000E+00	20000E+00	14997E+00	13635E-01	63728E-01	-65426E-01					
18		46287E+00	47665E+00	47466E+00	46113E+00	46868E+00	98416E+00	58967E-03	-52681E-02					
10		29822E+00	23663E+00	23663E+00	24762E+00	26741E+00	13759E+00	61850E-02	-27942E+00					
1		00000E+00	00000E+00	10000E+00	10000E+00	49990E-01	64023E-02	58856E-01	-64671E-01					
19		47252E+00	48516E+00	48516E+00	47667E+00	48121E+00	99047E+00	42796E-03	-31447E-02					
10		29785E+00	23663E+00	23574E+00	24672E+00	26674E+00	13760E+00	61699E-02	-64660E-01					
2		10000E+00	10000E+00	20000E+00	20000E+00	14997E+00	14331E-01	58954E-01	-27975E+00					
20		47665E+00	48516E+00	48333E+00	47466E+00	48000E+00	99030E+00	42729E-03	-61924E-03					
11		23663E+00	16967E+00	16746E+00	23663E+00	20317E+00	10330E+00	67573E-02	-63885E-01					
1		00000E+00	00000E+00	10000E+00	10000E+00	49990E-01	63313E-02	60397E-01	-25888E+00					
21		48516E+00	49275E+00	49213E+00	48516E+00	48895E+00	99463E+00	45426E-03	-13246E-02					
11		23663E+00	16967E+00	16862E+00	23574E+00	26266E+00	10324E+00	67418E-02	-25913E+00					
2		10000E+00	10000E+00	20000E+00	20000E+00	14997E+00	19106E-01	60466E-01	-63901E-01					
22		48113E+00	49275E+00	49275E+00	48333E+00	48773E+00	99447E+00	45363E-03	-36368E-02					
12		16967E+00	11457E+00	11457E+00	16967E+00	14208E+00	71678E-01	59108E-02	-25890E+00					
1		00000E+00	00000E+00	10000E+00	10000E+00	49990E-01	63130E-02	57122E-01	-63172E-01					
23		49275E+00	49976E+00	49976E+00	48516E+00	49442E+00	99741E+00	35406E-03	-15840E-02					
12		16946E+00	11457E+00	11410E+00	16862E+00	14173E+00	71667E-01	54976E-02	-63170E-01					
2		10000E+00	10000E+00	20000E+00	20000E+00	14997E+00	16343E-01	57160E-01	-25915E+00					
24		49213E+00	49976E+00	49421E+00	49028E+00	49318E+00	99725E+00	35352E-03	-79794E-03					
13		11467E+00	52471E-01	52471E-01	11457E+00	85231E-01	41954E-01	62211E-02	-63181E-01					
1		00000E+00	00000E+00	10000E+00	10000E+00	49990E-01	62428E-02	56907E-01	-25222E+00					
25		49670E+00	49931E+00	49697E+00	49028E+00	49769E+00	99910E+00	40223E-03	70397E-03					
13		11453E+00	52471E-01	52471E-01	11410E+00	85231E-01	41972E-01	62066E-02	-25358E+00					
2		10000E+00	10000E+00	20000E+00	20000E+00	14997E+00	16875E-01	56913E-01	-63182E-01					
26		49931E+00	49976E+00	49976E+00	49213E+00	49645E+00	99894E+00	40212E-03	-11849E-02					



N P Y1 Y2 Y3 Y4 Y5 Y6 Y7 Y8 Y9 YN YN C25 C26  
 6 4 40 7 3 41 7 4 42 8 3 43 8 4 44 9 3 45 9 4 46 10 3 47 10 4 48 11 3 49 11 4 50 12 3 51 12 4 52

6	61771+00	51943+00	51475+00	61215+00	50602+00	33195+00	10307+01	-74777E-01
4	30000+00	40000+00	40000+00	40000+00	34922+00	51368+01	73400+01	-43096E+00
40	30599+00	42002+00	41604+00	36221+00	40149+00	94170+00	13452+02	33122E-02
7	52274E+00	44471+00	44215+00	51943+00	46232+00	26823+00	80526+02	-70941E-01
3	20000+00	30000+00	44215+00	30000+00	24995+00	34752+01	64904+01	-32329E+00
41	42330+00	44471+00	44215+00	42002+00	43276+00	96273+00	70054+03	40087E-02
7	51943E+00	44215+00	43818E+00	51475+00	47803E+00	26808E+00	74956E-02	-32575E+00
4	30000E+00	40000E+00	43818E+00	40000E+00	34922E+00	46948E-01	65019E-01	-71303E-01
42	42002E+00	44215E+00	43818E+00	41604E+00	42445E+00	96215E+00	64684E-03	-20941E-01
8	44471E+00	37342E+00	37105E+00	44215E+00	40790E+00	22019E+00	74163E-02	-32311E+00
3	20000E+00	30000E+00	30000E+00	30000E+00	24995E+00	33740E-01	62529E-01	-69037E-01
43	44471E+00	40113E+00	45821E+00	44215E+00	45162E+00	97487E+00	60426E-03	-12483E-01
8	44215E+00	37105E+00	36771E+00	43818E+00	40478E+00	22003E+00	72649E-02	-68865E-01
4	30000E+00	40000E+00	40000E+00	40000E+00	34943E+00	47520E-01	62618E-01	-32599E+00
44	44215E+00	45821E+00	43818E+00	43818E+00	44816E+00	97434E+00	60090E-03	98076E-03
9	37342E+00	24672E+00	24672E+00	37105E+00	33401E+00	17610E+00	77712E-02	-67399E-01
3	20000E+00	20000E+00	30000E+00	30000E+00	24995E+00	32960E-01	63816E-01	-28124E+00
45	44113E+00	47408E+00	47165E+00	45821E+00	46651E+00	98382E+00	54256E-03	28884E-02
9	37105E+00	24672E+00	24219E+00	36771E+00	33145E+00	17601E+00	77154E-02	-28330E+00
4	30000E+00	30000E+00	40000E+00	40000E+00	34942E+00	46439E-01	63850E-01	-67558E-01
46	45821E+00	47165E+00	45409E+00	45409E+00	46294E+00	98329E+00	58930E-03	-12965E-01
10	24672E+00	23574E+00	23424E+00	24672E+00	26539E+00	13753E+00	61406E-02	-28118E+00
3	20000E+00	20000E+00	30000E+00	30000E+00	24995E+00	32346E-01	54019E-01	-66344E-01
47	47486E+00	45333E+00	48027E+00	47165E+00	47758E+00	98997E+00	43046E-03	-75106E-02
10	24484E+00	23424E+00	23214E+00	24219E+00	26335E+00	13745E+00	60968E-02	-66271E-01
4	30000E+00	30000E+00	40000E+00	40000E+00	34942E+00	45621E-01	59051E-01	-28340E+00
48	47165E+00	45927E+00	47595E+00	46701E+00	47342E+00	98946E+00	42851E-03	54621E-03
11	23574E+00	16775E+00	16775E+00	23424E+00	20164E+00	10328E+00	67098E-02	-65772E-01
3	20000E+00	20000E+00	30000E+00	30000E+00	24995E+00	31977E-01	60493E-01	-26176E+00
49	48333E+00	49027E+00	48271E+00	48027E+00	48527E+00	99414E+00	45931E-03	28002E-02
11	23424E+00	16775E+00	16624E+00	23214E+00	20009E+00	10327E+00	66619E-02	-26399E+00
4	30000E+00	30000E+00	40000E+00	40000E+00	34943E+00	45124E-01	60478E-01	-65822E-01
50	48027E+00	48718E+00	48202E+00	47595E+00	48155E+00	99363E+00	45647E-03	-72947E-02
12	16862E+00	11410E+00	11338E+00	16775E+00	14101E+00	71555E-01	54713E-02	-26176E+00
3	20000E+00	20000E+00	30000E+00	30000E+00	24995E+00	31713E-01	57171E-01	-65073E-01
51	44028E+00	44421E+00	44104E+00	46718E+00	44064E+00	94693E+00	35844E-03	-26159E-02
12	16775E+00	11338E+00	11236E+00	16624E+00	13943E+00	71525E-01	54319E-02	-65058E-01
4	30000E+00	30000E+00	40000E+00	40000E+00	34942E+00	44725E-01	57158E-01	-26444E+00
52	48718E+00	47165E+00	46667E+00	46200E+00	46644E+00	94644E+00	35777E-03	12042E-02





M	X1	X2	X3	X4	XP	XN	FL	CZ4
N	Y1	Y2	Y3	Y4	YP	YN	FL	CZ5
P	Z1	Z2	Z3	Z4	ZP	ZN	CZ1	CZ6
1	97960E+00	95854E+00	94754E+00	96825E+00	96362E+00	97273E+00	10298E-01	-0.13315E+00
6	40000E+00	40000E+00	50000E+00	50000E+00	44990E+00	11344E+00	72366E-01	-0.18220E+01
57	00000E+00	10078E+00	99590E-01	00000E+00	50093E-01	20224E+00	51058E-02	0.25252E-01
1	96825E+00	94754E+00	93354E+00	95354E+00	95084E+00	96941E+00	10196E-01	-0.18411E+01
7	50000E+00	50000E+00	60000E+00	60000E+00	54988E+00	14006E+00	72167E-01	-0.13398E+00
58	00000E+00	99590E-01	96120E-01	00000E+00	49428E-01	20157E+00	50318E-02	-0.79263E-01
2	95854E+00	88955E+00	87906E+00	94754E+00	91876E+00	82950E+00	12536E-01	-0.18037E+01
6	40000E+00	40000E+00	50000E+00	50000E+00	44990E+00	10190E+00	81859E-01	-0.12337E+00
59	10078E+00	20297E+00	20295E+00	99590E-01	15217E+00	54943E+00	73177E-02	-0.16165E+00
2	94754E+00	87906E+00	86607E+00	93354E+00	90657E+00	82700E+00	12403E-01	-0.11943E+00
7	50000E+00	50000E+00	60000E+00	60000E+00	54988E+00	12507E+00	81715E-01	-0.18266E+01
60	99590E-01	20295E+00	19995E+00	96120E-01	15016E+00	54793E+00	71985E-02	-0.17945E-01
3	82955E+00	80696E+00	79745E+00	87906E+00	84326E+00	65712E+00	10965E-01	-0.10319E+00
6	40000E+00	40000E+00	50000E+00	50000E+00	44990E+00	87571E-01	76277E-01	-0.76373E+00
61	20297E+00	27786E+00	27458E+00	20295E+00	24019E+00	74868E+00	25312E-02	0.20143E-01
3	87906E+00	79745E+00	76506E+00	86607E+00	83208E+00	65576E+00	10842E-01	-0.77007E+00
7	50000E+00	50000E+00	60000E+00	60000E+00	54988E+00	10826E+00	76304E-01	-0.10710E+00
62	20295E+00	27458E+00	27022E+00	19995E+00	25700E+00	74716E+00	24969E-02	-0.10646E+00
4	80696E+00	70157E+00	69310E+00	79745E+00	74973E+00	51807E+00	12323E-01	-0.75918E+00
6	40000E+00	40000E+00	50000E+00	50000E+00	44990E+00	77847E-01	81527E-01	-0.94748E-01
63	27786E+00	34208E+00	33805E+00	27458E+00	30815E+00	85179E+00	30969E-02	-0.86518E-01
4	79745E+00	69310E+00	66206E+00	79745E+00	73978E+00	51726E+00	12179E-01	-0.91774E-01
7	50000E+00	50000E+00	60000E+00	60000E+00	54988E+00	96319E-01	81470E-01	-0.71177E+00
64	27458E+00	33305E+00	33305E+00	27022E+00	30406E+00	85039E+00	30551E-02	0.89605E-03
5	70157E+00	61245E+00	60494E+00	69310E+00	65290E+00	41176E+00	97617E-02	-0.84372E-01
6	40000E+00	40000E+00	50000E+00	50000E+00	44990E+00	70847E-01	71746E-01	-0.43567E+00
65	34208E+00	36291E+00	37801E+00	33805E+00	36017E+00	90854E+00	12426E-02	0.13917E-01
5	69310E+00	60494E+00	59600E+00	68266E+00	64424E+00	41124E+00	96449E-02	-0.44000E+00
7	50000E+00	50000E+00	60000E+00	60000E+00	54988E+00	87760E-01	71780E-01	-0.86404E-01
66	33805E+00	37801E+00	37242E+00	33305E+00	35539E+00	90729E+00	12286E-02	-0.57336E-01
6	61215E+00	51476E+00	50807E+00	60494E+00	56014E+00	33168E+00	10288E-01	-0.43394E+00
6	40000E+00	40000E+00	50000E+00	50000E+00	44990E+00	66554E-01	73433E-01	-0.80464E-01
67	31251E+00	41624E+00	41193E+00	37801E+00	39733E+00	94104E+00	13438E-02	-0.39939E-01
6	60494E+00	50809E+00	50171E+00	59600E+00	55271E+00	33155E+00	10164E-01	-0.79258E-01
7	50000E+00	50000E+00	60000E+00	60000E+00	54988E+00	82469E-01	73367E-01	-0.44135E+00
68	37801E+00	41193E+00	40254E+00	37242E+00	39206E+00	93993E+00	13281E-02	0.38303E-02
7	51476E+00	43153E+00	43061E+00	50809E+00	47367E+00	26764E+00	79189E-02	-0.75409E-01
6	40000E+00	40000E+00	50000E+00	50000E+00	44990E+00	66612E-01	65066E-01	-0.32925E+00
69	41624E+00	46322E+00	45301E+00	41193E+00	42500E+00	96136E+00	70198E-03	0.73503E-02

N	P	Y1	Y2	Y3	Y4	Y5	Z1	Z2	Z3	Z4	Z5	Z6	Z7	Z8	Z9	FL	CL5	CL6
7		50869E+00	4330E+00	42562E+00	50117E+00	46738E+00	26752E+00	76217E-02	-0.33211E+00									
7		50000E+00	50000E+00	50000E+00	50000E+00	54937E+00	78742E-01	65078E-01	-0.73671E-01									
70		41193E+00	4330E+00	42661E+00	40769E+00	41936E+00	49033E+00	64443E-03	-0.33856E-01									
8		43818E+00	3671E+00	36350E+00	4330E+00	40077E+00	21987E+00	71951E-02	-0.32860E+00									
6		40000E+00	40000E+00	50000E+00	50000E+00	44990E+00	61741E-01	62663E-01	-0.73674E-01									
71		43818E+00	4540E+00	44474E+00	4330E+00	44351E+00	47577E+00	60609E-03	-0.24103E-01									
8		4330E+00	3635E+00	3560E+00	4262E+00	37526E+00	21971E+00	71069E-02	-0.73069E-01									
7		50000E+00	50000E+00	60000E+00	60000E+00	54938E+00	76367E-01	62665E-01	-0.33278E+00									
72		4330E+00	4487E+00	4421E+00	4266E+00	43763E+00	47277E+00	60101E-03	0.36245E-02									
9		36771E+00	2921E+00	2875E+00	3635E+00	32801E+00	17563E+00	76406E-02	-0.71294E-01									
6		40000E+00	4070E+00	4070E+00	5000E+00	44990E+00	60318E-01	63833E-01	-0.28574E+00									
73		4540E+00	4676E+00	4620E+00	4487E+00	45814E+00	49827E+00	54931E-03	0.46377E-02									
9		36338E+00	2687E+00	2644E+00	3580E+00	32366E+00	17564E+00	75466E-02	-0.28891E+00									
7		50000E+00	5000E+00	6000E+00	6000E+00	54988E+00	74584E-01	63773E-01	-0.71735E-01									
74		44874E+00	4620E+00	4552E+00	4421E+00	45206E+00	49816E+00	58789E-03	-0.21403E-01									
10		29219E+00	2321E+00	2294E+00	2887E+00	2660E+00	13738E+00	60380E-02	-0.28550E+00									
6		40000E+00	4000E+00	5000E+00	5000E+00	44990E+00	54913E-01	59055E-01	-0.70256E-01									
75		4676E+00	4759E+00	4703E+00	4633E+00	46900E+00	49887E+00	43453E-03	-0.14267E-01									
10		28875E+00	2244E+00	2260E+00	2644E+00	25716E+00	13725E+00	54963E-02	-0.70017E-01									
7		50000E+00	5000E+00	6000E+00	6000E+00	54988E+00	73327E-01	54030E-01	-0.28915E+00									
76		4620E+00	4703E+00	4633E+00	4552E+00	46278E+00	49876E+00	43101E-03	0.75220E-03									
11		23214E+00	1662E+00	1642E+00	2260E+00	1980E+00	10314E+00	65976E-02	-0.69391E-01									
6		40000E+00	4000E+00	5000E+00	5000E+00	44990E+00	58629E-01	60428E-01	-0.26535E+00									
77		4759E+00	4820E+00	4770E+00	4703E+00	47655E+00	49924E+00	46106E-03	0.26133E-02									
11		22940E+00	1642E+00	1615E+00	2260E+00	1953E+00	10298E+00	65159E-02	-0.26783E+00									
7		50000E+00	5000E+00	6000E+00	6000E+00	54988E+00	72506E-01	60339E-01	-0.69533E-01									
78		4703E+00	4770E+00	4705E+00	4633E+00	47023E+00	49924E+00	45728E-03	-0.11550E-01									
12		16624E+00	1123E+00	1103E+00	1642E+00	13848E+00	71569E-01	53744E-02	-0.26527E+00									
6		40000E+00	4000E+00	5000E+00	5000E+00	44990E+00	58165E-01	57122E-01	-0.68896E-01									
79		46280E+00	4866E+00	4809E+00	4809E+00	48188E+00	49574E+00	36353E-03	-0.74949E-02									
12		16428E+00	1116E+00	1093E+00	1615E+00	13664E+00	71524E-01	53129E-02	-0.68825E-01									
7		50000E+00	5000E+00	6000E+00	6000E+00	54988E+00	71929E-01	57061E-01	-0.26790E+00									
80		4770E+00	4809E+00	4736E+00	4700E+00	47548E+00	49484E+00	36077E-03	-0.82666E-04									
13		11236E+00	5142E+00	5061E+00	1110E+00	8140E+00	41821E-01	60735E-02	-0.69052E-01									
6		40000E+00	4000E+00	5000E+00	5000E+00	44990E+00	57862E-01	58734E-01	-0.25738E+00									
81		4866E+00	4872E+00	4834E+00	4809E+00	48507E+00	49474E+00	41059E-03	0.40314E-02									
13		11103E+00	5041E+00	5009E+00	1073E+00	8324E-01	41846E-01	54981E-02	-0.26136E+00									
7		50000E+00	5000E+00	6000E+00	6000E+00	54988E+00	71667E-01	56666E-01	-0.69061E-01									
82		4809E+00	4834E+00	4736E+00	4736E+00	47664E+00	49635E+00	40761E-03	-0.42261E-02									





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13	.10939E+00	.50060E-01	.41600E-01	.10742E+00	.74010E-01	.41734E-01	.59064E-02	-.75785E-01
8	.60000E+00	.60000E+00	.70000E+00	.70000E+00	.64985E+00	.85819E-01	.56448E-01	-.26468E+00
109	.47382E+00	.47631E+00	.46773E+00	.46529E+00	.47600E+00	.99544E+00	.41959E-03	.27733E-02
13	.10742E+00	.49160E-01	.46100E-01	.10510E+00	.77448E-01	.41741E-01	.57977E-02	-.27066E+00
9	.70000E+00	.70000E+00	.80000E+00	.80000E+00	.74982E+00	.10097E+00	.56261E-01	-.75856E-01
110	.46529E+00	.46773E+00	.46773E+00	.45523E+00	.46149E+00	.99401E+00	.41551E-03	-.90339E-02
14	.50060E-01	.00000E+00	.00000E+00	.49100E-01	.24806E-01	.13052E-01	.49797E-02	-.26466E+00
8	.60000E+00	.60000E+00	.70000E+00	.70000E+00	.64985E+00	.85672E-01	.56056E-01	-.74860E-01
111	.47631E+00	.47697E+00	.46937E+00	.46773E+00	.47236E+00	.99624E+00	.35211E-03	-.14001E-03
14	.49160E-01	.00000E+00	.00000E+00	.48100E-01	.24316E-01	.12991E-01	.46882E-02	-.74861E-01
9	.70000E+00	.70000E+00	.80000E+00	.80000E+00	.74982E+00	.10064E+00	.59916E-01	-.27066E+00
112	.46773E+00	.46837E+00	.46837E+00	.45763E+00	.46302E+00	.99489E+00	.34946E-03	-.69016E-03





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14	.48100E+01	.00000E+00	.46870E-01	.23744E-01	.12966E+01	.47816E-02	-.27705E+00
11	.80000E+00	.00000E+00	.00000E+00	.64978E+00	.11671E+00	.55761E-01	-.85562E-01
139	.45763E+00	.44651E+00	.44651E+00	.45210E+00	.94930E+00	.37342E-03	-.17241E-02
14	.46870E-01	.00000E+00	.45450E-01	.23062E-01	.12660E-01	.40563E-02	-.85565E-01
12	.90000E+00	.00000E+01	.10000E+01	.94974E+00	.13362E+00	.55594E-01	-.28422E+00
140	.44590E+00	.44651E+00	.43242E+00	.43949E+00	.94902E+00	.36915E-03	-.22173E-02
1	.86603E+00	.81730E+00	.63516E+00	.84159E+00	.93643E+00	.93353E-02	-.18265E+00
13	.10000E+01	.11000E+01	.11000E+01	.10497E+01	.24165E+00	.70273E-01	-.20237E+01
141	.00000E+00	.89900E-01	.00000E-01	.43750E-01	.19474E+00	.45329E-02	.74322E-01
1	.83516E+00	.76209E+00	.80000E+00	.80876E+00	.92479E+00	.90835E-02	-.20795E+01
14	.11000E+01	.12000E+01	.12000E+01	.11496E+01	.32852E+00	.69825E-01	-.18759E+00
142	.00000E+00	.82269E-01	.00000E-01	.42052E-01	.19229E+00	.43481E-02	-.20849E+00
2	.84750E+00	.75824E+00	.81730E+00	.80241E+00	.60410E+00	.11279E-01	-.19718E+01
13	.10000E+01	.11000E+01	.11000E+01	.10497E+01	.26391E+00	.74621E-01	-.19478E+00
143	.84960E-01	.18152E+00	.17505E+00	.15290E+00	.53270E+00	.63010E-02	-.45905E+00
2	.81730E+00	.72631E+00	.76269E+00	.77150E+00	.74588E+00	.10951E-01	-.16524E+00
14	.11000E+01	.12000E+01	.12000E+01	.11496E+01	.29759E+00	.74005E-01	-.20565E+01
144	.85900E-01	.17505E+00	.82269E-01	.12775E+00	.52727E+00	.60008E-02	-.36669E-01
3	.78626E+00	.60764E+00	.75824E+00	.73648E+00	.64218E+00	.97965E-02	-.14684E+00
13	.10000E+01	.11000E+01	.11000E+01	.10497E+01	.22859E+00	.75410E-01	-.87831E+00
145	.18152E+00	.24559E+00	.23634E+00	.20977E+00	.73168E+00	.23454E-02	.67158E-01
3	.75824E+00	.68764E+00	.72631E+00	.70793E+00	.63726E+00	.94855E-02	-.86764E+00
14	.11000E+01	.12000E+01	.12000E+01	.11496E+01	.25845E+00	.75096E-01	-.17359E+00
146	.17505E+00	.23634E+00	.17505E+00	.20164E+00	.72602E+00	.22600E-02	-.29497E+00
4	.71326E+00	.59784E+00	.68764E+00	.65479E+00	.50875E+00	.10956E-01	-.84725E+00
13	.10000E+01	.11000E+01	.11000E+01	.10497E+01	.20417E+00	.74746E-01	-.15385E+00
147	.24559E+00	.30256E+00	.29199E+00	.26912E+00	.83635E+00	.27911E-02	-.25303E+00
4	.68764E+00	.59764E+00	.65858E+00	.62940E+00	.50562E+00	.10595E-01	-.13067E+00
14	.11000E+01	.12000E+01	.12000E+01	.11496E+01	.23122E+00	.74164E-01	-.90333E+00
148	.23634E+00	.27931E+00	.22637E+00	.25869E+00	.83120E+00	.26819E-02	.11636E-01
5	.61993E+00	.54107E+00	.59764E+00	.57022E+00	.40552E+00	.86547E-02	-.12288E+00
13	.10000E+01	.11000E+01	.11000E+01	.10497E+01	.18655E+00	.71050E-01	-.56209E+00
149	.30236E+00	.33810E+00	.32655E+00	.31456E+00	.89465E+00	.12103E-02	.44606E-01
5	.59784E+00	.49962E+00	.57267E+00	.54911E+00	.40359E+00	.85617E-02	-.50818E+00
14	.11000E+01	.12000E+01	.12000E+01	.11496E+01	.21149E+00	.70777E-01	-.13726E+00
150	.24159E+00	.32665E+00	.27931E+00	.30236E+00	.89025E+00	.11756E-02	-.15997E+00
6	.54107E+00	.45498E+00	.52179E+00	.48921E+00	.32723E+00	.91057E-02	-.48886E+00
13	.10000E+01	.10900E+01	.11000E+01	.10497E+01	.17566E+00	.72044E-01	-.12523E+00
151	.33810E+00	.35944E+00	.32655E+00	.34701E+00	.92849E+00	.12863E-02	-.12102E+00



6	52179E+00	4377E+00	42030E+00	4742E+00	47024E+00	32576E+00	87420E-02	-11502E+00
14	11000E+01	11000E+01	12000E+01	12000E+01	11496E+01	14947E+00	71547E-01	-51813E+00
152	32605E+00	35531E+00	34035E+00	31232E+00	32356E+00	92423E+00	12457E-02	18433E-01
7	45498E+00	3730L+00	3730L+00	4387E+00	41308E+00	20464E+00	6998E-02	-11220E+00
13	10000E+01	10000E+01	10000E+01	11000E+01	1047E+01	16783E+00	64546E-01	-37851E+00
153	36844E+00	38730L+00	3730L+00	32531E+00	37116L+00	94903E+00	73579E-03	30637E-01
7	43877E+00	3730L+00	3730L+00	42030E+00	39765E+00	26330E+00	67567E-02	-38639E+00
14	11000E+01	11000E+01	12000E+01	12000E+01	11496E+01	19045E+00	64357E-01	-11872E+00
154	35531E+00	3730L+00	3730L+00	34035E+00	35679L+00	94507L+00	71938E-03	-92806E-01
8	36730L+00	32502E+00	31343E+00	3730L+00	34985E+00	21726E+00	63562E-02	-37270E+00
13	10000E+01	10000E+01	10000E+01	11000E+01	1047E+01	16300E+00	62163E-01	-11332E+00
155	38730E+00	40156E+00	30706E+00	3730L+00	38735E+00	92424E+00	64975E-03	-68153E-01
8	37350L+00	31343E+00	30024E+00	3577E+00	33629E+00	21642E+00	61337E-02	-10878E+00
14	11000E+01	11000E+01	12000E+01	12000E+01	11496E+01	18504E+00	61947E-01	-39126E+00
156	37350L+00	38706L+00	3707E+00	3577E+00	37233E+00	95801E+00	63609E-03	16074E-01
9	32502E+00	29425E+00	29066E+00	31343E+00	28647E+00	17393E+00	67467E-02	-10727E+00
13	10000E+01	10000E+01	10000E+01	11000E+01	1047E+01	15933E+00	62805E-01	-32650E+00
157	40136E+00	41331E+00	39828E+00	30706E+00	40012L+00	97178E+00	63582E-03	21138E-01
9	31343E+00	24906E+00	23857E+00	30024E+00	27537E+00	17329L+00	65093E-02	-33599E+00
14	11000E+01	11000E+01	12000E+01	12000E+01	11496E+01	18086E+00	62513E-01	-11019E+00
158	38706E+00	34628E+00	3010L+00	3707E+00	38401E+00	96812E+00	62252E-03	-58258E-01
10	25826E+00	20514E+00	1787E+00	24906E+00	22762E+00	13585E+00	53243E-02	-32403E+00
13	10000E+01	10000E+01	11000E+01	11000E+01	1047E+01	15668E+00	58534E-01	-10716E+00
159	41331E+00	4208E+00	40509E+00	39828E+00	40961E+00	97826E+00	45586E-03	-39703E-01
10	24906E+00	14767E+00	10954E+00	23857E+00	21879E+00	13538E+00	51410E-02	-10532E+00
14	11000E+01	11000E+01	12000E+01	12000E+01	11496E+01	17866E+00	58400E-01	-33787E+00
160	39858E+00	40509E+00	3861E+00	30100E+00	39373E+00	97470E+00	46844E-03	95512E-02
11	20518E+00	14674E+00	14170L+00	14767E+00	17244E+00	10212E+00	58214E-02	-10501E+00
13	10000E+01	10000E+01	11000E+01	11000E+01	1047E+01	15485E+00	54400E-01	-30131E+00
161	4208E+00	42073E+00	41153E+00	40509E+00	41620E+00	98265E+00	51826E-03	14596E-01
11	19767E+00	14170E+00	10954E+00	10954E+00	16624E+00	10176E+00	56160E-02	-31245E+00
14	11000E+01	11000E+01	12000E+01	12000E+01	11496E+01	17566E+00	59147E-01	-10592E+00
162	40509E+00	41153E+00	3420E+00	38601L+00	40007E+00	97914E+00	51045E-03	-31981E-01
12	14694E+00	9310E+01	9370L+01	14170L+01	12094E+00	70895E-01	47464E-02	-30043E+00
13	10000E+01	10000E+01	11000E+01	11000E+01	1047E+01	15362E+00	56400E-01	-10464E+00
163	42673E+00	43016E+00	41403E+00	41153L+00	42006L+00	98558E+00	43316E-03	-17898E-01
12	14170E+00	9570E+01	91740E-01	13573E+00	11625E+00	7058E-01	45778E-02	-10418E+00
14	11000E+01	11000E+01	12000E+01	12000E+01	11496E+01	14448E+00	56340E-01	-31243E+00
164	41153E+00	41403E+00	3730E+00	34420E+00	40454E+00	98213E+00	42784E-03	35531E-02

Y1 Z1 Y2 Z2 Y3 Z3 Y4 Z4 Y5 Z5 Y6 Z6 Y7 Z7 Y8 Z8 Y9 Z9 Y10 Z10  
 C75 C76 C71 C72 C73 C74 C75 C76 C71 C72 C73 C74 C75 C76 C71 C72 C73 C74 C75 C76

13	.99310E-01	.45450E-01	.43530E-01	.9770E-01	.71098E-01	.41856E-01	.53576E-02	-.10373E+00
13	.10000E+01	.10000E+01	.10000E+01	.11000E+01	.10497E+01	.15291E+00	.57556E-01	-.29333E+00
165	.43016E+00	.43242E+00	.41701E+00	.41403E+00	.42365E+00	.98757E+00	.47113E-03	.76684E-02
13	.95770E-01	.43850E-01	.41900E-01	.91740E-01	.68341E-01	.41312E-01	.51679E-02	-.30358E+00
14	.11000E+01	.11000E+01	.10000E+01	.12000E+01	.11496E+01	.17304E+00	.57293E-01	-.10386E+00
166	.41483E+00	.41701E+00	.39945E+00	.39756E+00	.40723E+00	.98394E+00	.46392E-03	-.13463E-01
14	.45450E-01	.00000E+00	.00000E+00	.43850E-01	.22322E-01	.12840E-01	.45172E-02	-.29321E+00
13	.10000E+01	.10000E+01	.10000E+01	.11000E+01	.10497E+01	.15291E+00	.55421E-01	-.10339E+00
167	.45242E+00	.43301E+00	.41758E+00	.41701E+00	.42505E+00	.98822E+00	.41069E-03	-.37618E-02
14	.43830E-01	.00000E+00	.00000E+00	.41900E-01	.21456E-01	.12854E-01	.45567E-02	-.10337E+00
14	.11000E+01	.11000E+01	.12000E+01	.12000E+01	.11496E+01	.17319E+00	.55250E-01	-.30357E+00
168	.41701E+00	.41758E+00	.40000E+00	.39945E+00	.40857E+00	.98460E+00	.40660E-03	-.12933E-02

M	X1	X2	X3	X4	X5	X6	X7	X8	A	C74
N	Y1	Y2	Y3	Y4	Y5	Y6	Y7	Y8	FL	C75
P	Z1	Z2	Z3	Z4	Z5	Z6	Z7	Z8	C71	C76
1	.80000E+00	.75289E+00	.74368E+00	.75993E+00	.77177E+00	.81027E+00	.86015E-02	.86015E-02		-.22103E+00
16	.12000E+01	.12000E+01	.13000E+01	.13000E+01	.13000E+01	.3E022E+00	.69428E-01	.69428E-01		-.21565E+01
169	.00000E+00	.32260E-01	.70100E-01	.00000E+00	.40119E-01	.16927E+00	.42490E-02	.42490E-02		.10084E+00
1	.75993E+00	.74368E+00	.75289E+00	.74368E+00	.75993E+00	.84147E+00	.84851E-02	.84851E-02		-.22359E+01
17	.13000E+01	.13000E+01	.13000E+01	.13000E+01	.13495E+01	.41232E+00	.69143E-01	.69143E-01		-.22951E+00
170	.00000E+00	.78160E-01	.16768E+00	.00000E+00	.37915E-01	.1E544E+00	.40165E-02	.40165E-02		-.28226E+00
2	.78289E+00	.72631E+00	.66994E+00	.74368E+00	.73587E+00	.7E560E+00	.10560E-01	.10560E-01		-.20778E+01
16	.12000E+01	.12000E+01	.13000E+01	.13000E+01	.12496E+01	.33459E+00	.78372E-01	.78372E-01		-.25177E+00
171	.82280E-01	.16768E+00	.13928E+00	.7E160E-01	.12100E+00	.52040E+00	.57658E-02	.57658E-02		-.61434E+00
2	.74368E+00	.68994E+00	.64567E+00	.69594E+00	.69594E+00	.77249E+00	.10164E-01	.10164E-01		-.20059E+00
17	.13000E+01	.13000E+01	.14000E+01	.14000E+01	.13495E+01	.37595E+00	.77785E-01	.77785E-01		-.22108E+01
172	.78160E-01	.15928E+00	.14969E+00	.7E340E-01	.11518E+00	.51177E+00	.53842E-02	.53842E-02		-.39262E-01
3	.72631E+00	.65888E+00	.62588E+00	.68994E+00	.67540E+00	.63103E+00	.91410E-02	.91410E-02		-.18219E+00
16	.12000E+01	.12000E+01	.13000E+01	.13000E+01	.12496E+01	.29154E+00	.74789E-01	.74789E-01		-.96117E+00
173	.16768E+00	.22667E+00	.24551E+00	.15928E+00	.19236E+00	.71809E+00	.22799E-02	.22799E-02		.99881E-01
3	.68994E+00	.62588E+00	.58617E+00	.64836E+00	.63829E+00	.62246E+00	.87481E-02	.87481E-02		-.96919E+00
17	.13000E+01	.13000E+01	.14000E+01	.14000E+01	.13495E+01	.32895E+00	.74543E-01	.74543E-01		-.22948E+00
174	.15928E+00	.21551E+00	.20252E+00	.14969E+00	.16181E+00	.70973E+00	.21815E-02	.21815E-02		-.40906E+00
4	.65888E+00	.57267E+00	.54399E+00	.62588E+00	.60049E+00	.50162E+00	.10186E-01	.10186E-01		-.90666E+00
16	.12000E+01	.12000E+01	.13000E+01	.13000E+01	.12496E+01	.26133E+00	.78546E-01	.78546E-01		-.20387E+00
175	.22667E+00	.27931E+00	.26532E+00	.24551E+00	.24681E+00	.82467E+00	.26543E-02	.26543E-02		-.34540E+00
4	.62588E+00	.54399E+00	.51121E+00	.58617E+00	.56749E+00	.49645E+00	.97244E-02	.97244E-02		-.16206E+00
17	.13000E+01	.13000E+01	.14000E+01	.14000E+01	.13495E+01	.29558E+00	.77931E-01	.77931E-01		-.99884E+00
176	.21551E+00	.26532E+00	.24933E+00	.24933E+00	.23325E+00	.81619E+00	.25144E-02	.25144E-02		.26612E-01
5	.57267E+00	.49822E+00	.47499E+00	.54399E+00	.52293E+00	.40076E+00	.80298E-02	.80298E-02		-.15546E+00
16	.12000E+01	.12000E+01	.13000E+01	.13000E+01	.12496E+01	.25953E+00	.70502E-01	.70502E-01		-.55061E+00
177	.27931E+00	.31232E+00	.27608E+00	.26532E+00	.26847E+00	.88437E+00	.12288E-02	.12288E-02		.70403E-01
5	.54399E+00	.47499E+00	.44618E+00	.51121E+00	.49420E+00	.39730E+00	.76538E-02	.76538E-02		-.55702E+00
17	.13000E+01	.13000E+01	.14000E+01	.14000E+01	.13495E+01	.27118E+00	.70265E-01	.70265E-01		-.18187E+00
178	.26532E+00	.29568E+00	.27800E+00	.24933E+00	.27262E+00	.87670E+00	.11866E-02	.11866E-02		-.22485E+00
6	.49822E+00	.42030E+00	.39225E+00	.47499E+00	.44804E+00	.32384E+00	.84364E-02	.84364E-02		-.52634E+00
16	.12000E+01	.12000E+01	.13000E+01	.13000E+01	.12496E+01	.22557E+00	.71125E-01	.71125E-01		-.16421E+00
179	.31232E+00	.34035E+00	.32330E+00	.29088E+00	.31823E+00	.91863E+00	.12885E-02	.12885E-02		-.16638E+00
6	.47499E+00	.39925E+00	.37519E+00	.44618E+00	.42399E+00	.32132E+00	.80337E-02	.80337E-02		-.14536E+00
17	.13000E+01	.13000E+01	.14000E+01	.14000E+01	.13495E+01	.25581E+00	.70660E-01	.70660E-01		-.57560E+00
180	.24668E+00	.32330E+00	.30362E+00	.27608E+00	.30079E+00	.91176E+00	.12384E-02	.12384E-02		.33748E-01
7	.42030E+00	.35777E+00	.33969E+00	.41121E+00	.37930E+00	.26266E+00	.64810E-02	.64810E-02		-.14331E+00
16	.12000E+01	.12000E+01	.13000E+01	.13000E+01	.12496E+01	.21561E+00	.64173E-01	.64173E-01		-.41212E+00
181	.34035E+00	.35777E+00	.32330E+00	.29088E+00	.34039E+00	.94061E+00	.76457E-03	.76457E-03		.47916E-01

M P Y1 Y2 Y3 Y4 Y5 Y6 Y7 Y8 Y9 Y0  
 A FL C75 C76  
 C71 C72 C73 C74 C75 C76

7	39025E+00	33965E+00	34517E+00	39123E+00	26021E+00	61677E-02	-64229E+00
17	13000E+01	12000E+01	14000E+01	13975E+01	24907E+00	64024E-01	-13556E+00
182	32330E+00	35965E+00	34378E+00	32167E+00	93379E+00	70507E-03	-21314E+00
8	35777E+00	30066E+00	25220E+00	33905E+00	24535E+00	56807E-02	-640140E+00
16	12000E+01	12000E+01	13000E+01	12490E+01	20972E+00	61542E-01	-14729E+00
183	35777E+00	37076E+00	32216E+00	33965E+00	93379E+00	70200E-03	-694092E-01
8	33965E+00	26320E+00	26320E+00	31937E+00	21397E+00	55940E-02	-13868E+00
17	13000E+01	13000E+01	14000E+01	13495E+01	23005E+00	61722E-01	-64104E+00
184	33965E+00	35219E+00	32097E+00	31937E+00	93379E+00	66723E-03	-26727E-01
9	30024E+00	23007E+00	26320E+00	26320E+00	17251E+00	62399E-02	-13832E+00
16	12000E+01	12000E+01	13000E+01	13900E+01	20504E+00	62217E-01	-35532E+00
185	37076E+00	30180E+00	32200E+00	35219E+00	96343E+00	69128E-03	-34304E-01
9	28520E+00	22663E+00	21297E+00	24828E+00	17157E+00	59341E-02	-36873E+00
17	13000E+01	13000E+01	14000E+01	13495E+01	23206E+00	61940E-01	-14351E+00
186	35219E+00	36218E+00	34092E+00	34675E+00	93729E+00	67611E-03	-80040E-01
10	23957E+00	18934E+00	15005E+00	20874E+00	13475E+00	49275E-02	-35062E+00
16	12000E+01	12000E+01	13000E+01	12490E+01	20107E+00	50277E-01	-13902E+00
187	38180E+00	30801E+00	30915E+00	37504E+00	97014E+00	50262E-03	-54726E-01
10	22663E+00	16005E+00	16920E+00	19726E+00	13392E+00	40854E-02	-13613E+00
17	13000E+01	13000E+01	14000E+01	13495E+01	22914E+00	58196E-01	-37236E+00
188	36208E+00	34945E+00	34062E+00	35500E+00	96414E+00	55447E-03	-17787E-01
11	18954E+00	13573E+00	12894E+00	15860E+00	10126E+00	55822E-02	-13587E+00
16	12000E+01	12000E+01	13000E+01	12490E+01	19939E+00	50891E-01	-32739E+00
189	38612E+00	34940E+00	37496E+00	30169E+00	97467E+00	58130E-03	-21314E-01
11	18005E+00	12894E+00	12117E+00	14969E+00	10065E+00	51166E-02	-34181E+00
17	13000E+01	13000E+01	14000E+01	13495E+01	22659E+00	56602E-01	-13767E+00
190	36515E+00	37446E+00	35189E+00	36072E+00	96878E+00	57120E-03	-46236E-01
12	13573E+00	91740E-01	87140E-01	11091E+00	70203E-01	43872E-02	-32581E+00
16	12000E+01	12000E+01	13000E+01	12490E+01	19762E+00	50217E-01	-13538E+00
191	39440E+00	39736E+00	37792E+00	30595E+00	97772E+00	50182E-03	-26605E-01
12	12894E+00	87140E-01	81870E-01	10982E+00	69886E-01	41712E-02	-13437E+00
17	13000E+01	13000E+01	14000E+01	13495E+01	22476E+00	50132E-01	-34289E+00
192	37446E+00	37746E+00	35189E+00	30475E+00	97140E+00	49717E-03	-10263E-01
13	91740E-01	41800E-01	37900E-01	65199E-01	41092E-01	49522E-02	-13305E+00
16	12000E+01	12000E+01	13000E+01	12490E+01	19692E+00	57032E-01	-31497E+00
193	3736E+00	39445E+00	37746E+00	30651E+00	97950E+00	55357E-03	-10015E-01
13	87140E-01	39800E-01	37490E-01	61617E-01	40792E-01	47067E-02	-33052E+00
17	13000E+01	13000E+01	14000E+01	13495E+01	22306E+00	50794E-01	-13409E+00
194	37746E+00	37944E+00	35620E+00	30717E+00	97531E+00	52639E-03	-16347E-01





M	N	P	S1			S2			S3			S4			S5			S6			S7		
			Y1	Y2	Y3	Y4	Y5	Y6	Y7	Y8	Y9	Y10	Y11	Y12	Y13	Y14	Y15	Y16	Y17	Y18	Y19	Y20	
13			.84000E+01	.37000E+01	.37000E+01	.75000E+01	.25011E+01	.40424E+01	.44200E+02														
18			.14000E+01	.14000E+01	.14000E+01	.15000E+01	.14444E+01	.25534E+00	.50022E+01														
221			.35022E+00	.35022E+00	.35022E+00	.34054E+00	.34209E+00	.96001E+00	.60047E+03														
13			.75000E+01	.34700E+01	.34700E+01	.00000E+01	.52754E+01	.40050E+01	.44078E+02														
19			.15000E+01	.15000E+01	.15000E+01	.16000E+01	.15442E+01	.24415E+00	.50582E+01														
222			.33000E+00	.27000E+00	.27000E+00	.29000E+00	.31435E+00	.95472E+00	.67806E+03														
14			.37400E+01	.00000E+00	.00000E+00	.34700E+01	.16050E+01	.12720E+01	.37330E+02														
18			.14000E+01	.14000E+01	.14000E+01	.15000E+01	.14444E+01	.25473E+00	.54940E+01														
223			.35022E+00	.35022E+00	.35022E+00	.33072E+00	.34383E+00	.96043E+00	.63305E+03														
14			.34710E+01	.00000E+00	.00000E+00	.34700E+01	.16563E+01	.12559E+01	.34629E+02														
19			.15000E+01	.15000E+01	.15000E+01	.16000E+01	.15442E+01	.24347E+00	.53046E+01														
224			.33026E+00	.33026E+00	.33026E+00	.30000E+00	.31539E+00	.95504E+00	.63472E+03														

M	N	P	X			Y			Z			A			C74	C75	C76
			X1	X2	X3	Y1	Y2	Y3	Z1	Z2	Z3	A1	A2	A3			
1	1	1	60000E+00	53717E+00	52575E+00	13000E+01	17600E+01	17000E+01	10400E+01	10400E+01	10400E+01	74330E+00	74330E+00	74330E+00	72619E-02	-41274E+00	
21	21	21	13000E+01	15000E+01	17600E+01	17600E+01	17600E+01	17600E+01	17600E+01	17600E+01	17600E+01	70238E-01	70238E-01	70238E-01	70238E-01	-20529E+01	
225	225	225	00000E+00	61710E-01	54100E-01	54100E-01	54100E-01	54100E-01	54100E-01	54100E-01	54100E-01	30709E-02	30709E-02	30709E-02	30709E-02	213+8E+00	
1	1	1	51678E+00	51552E+00	42607E+00	17000E+01	13000E+01	10000E+01	47761E+00	17404E+01	00732E+00	72918E+00	72918E+00	72918E+00	67685E-02	-28476E+01	
22	22	22	17000E+01	17000E+01	13000E+01	13000E+01	13000E+01	13000E+01	17404E+01	17404E+01	00732E+00	00732E+00	00732E+00	00732E+00	74249E-01	-44100E+00	
226	226	226	00000E+00	54160E-01	44630E-01	44630E-01	44630E-01	44630E-01	24426E-01	24426E-01	15157E+00	15157E+00	15157E+00	15157E+00	30899E-02	-58335E+00	
2	2	2	51522E+00	54473E+00	47826E+00	17000E+01	17000E+01	17000E+01	53217E+00	10469E+01	09905E+00	09905E+00	09905E+00	09905E+00	85615E-02	-24360E+01	
21	21	21	10000E+01	10000E+01	17000E+01	17000E+01	17000E+01	17000E+01	10469E+01	10469E+01	54352E+00	54352E+00	54352E+00	54352E+00	77836E-01	-52630E+00	
227	227	227	01710E-01	12576E-01	14942E-01	14942E-01	14942E-01	14942E-01	00141E-01	00141E-01	40369E+00	40369E+00	40369E+00	40369E+00	46531E-02	-11760E+01	
2	2	2	51522E+00	47620E+00	37574E+00	13000E+01	13000E+01	13000E+01	42607E+00	14000E+01	65022E+00	65022E+00	65022E+00	65022E+00	76675E-02	-38055E+00	
22	22	22	17000E+01	17000E+01	13000E+01	13000E+01	13000E+01	13000E+01	14000E+01	14000E+01	62500E+00	62500E+00	62500E+00	62500E+00	80201E-01	-20133E+01	
228	228	228	54180E-01	11042E+00	91360E-01	91360E-01	91360E-01	91360E-01	75421E-01	75421E-01	43060E+00	43060E+00	43060E+00	43060E+00	41750E-02	10862E-01	
3	3	3	54473E+00	44416E+00	43366E+00	17000E+01	17000E+01	17000E+01	46844E+00	10469E+01	57655E+00	57655E+00	57655E+00	57655E+00	72224E-02	-38253E+00	
21	21	21	10000E+01	10000E+01	17000E+01	17000E+01	17000E+01	17000E+01	10469E+01	10469E+01	40575E+00	40575E+00	40575E+00	40575E+00	75582E-01	-13406E+01	
229	229	229	12576E+00	17015E+00	14939E+00	14939E+00	14939E+00	14939E+00	13913E+00	13913E+00	65065E+00	65065E+00	65065E+00	65065E+00	25416E-02	24720E+00	
3	3	3	47826E+00	43366E+00	37902E+00	13000E+01	13000E+01	13000E+01	41795E+00	17404E+01	54303E+00	54303E+00	54303E+00	54303E+00	65511E-02	-13340E+01	
22	22	22	17000E+01	17000E+01	13000E+01	13000E+01	13000E+01	13000E+01	17404E+01	17404E+01	50778E+00	50778E+00	50778E+00	50778E+00	78098E-01	-58871E+00	
230	230	230	11042E+00	14939E+00	14361E+00	14361E+00	14361E+00	14361E+00	11905E+00	11905E+00	61866E+00	61866E+00	61866E+00	61866E+00	24957E-02	-97539E+00	
4	4	4	49416E+00	42950E+00	37709E+00	17000E+01	17000E+01	17000E+01	43426E+00	10469E+01	46508E+00	46508E+00	46508E+00	46508E+00	79160E-02	-11346E+01	
21	21	21	10000E+01	10000E+01	17000E+01	17000E+01	17000E+01	17000E+01	10469E+01	10469E+01	44208E+00	44208E+00	44208E+00	44208E+00	72501E-01	-51751E+00	
231	231	231	17015E+00	20948E+00	10342E+00	10342E+00	10342E+00	10342E+00	17849E+00	17849E+00	70609E+00	70609E+00	70609E+00	70609E+00	26776E-02	-73560E+00	
4	4	4	43366E+00	37709E+00	32202E+00	17000E+01	17000E+01	17000E+01	37159E+00	17404E+01	44315E+00	44315E+00	44315E+00	44315E+00	71098E-02	-33843E+00	
22	22	22	17000E+01	17000E+01	13000E+01	13000E+01	13000E+01	13000E+01	17404E+01	17404E+01	52221E+00	52221E+00	52221E+00	52221E+00	78824E-01	-14823E+01	
232	232	232	14939E+00	10342E+00	15218E+00	15218E+00	15218E+00	15218E+00	15273E+00	15273E+00	72864E+00	72864E+00	72864E+00	72864E+00	25247E-02	17024E+00	
5	5	5	42950E+00	37407E+00	32912E+00	17000E+01	17000E+01	17000E+01	37609E+00	10469E+01	37649E+00	37649E+00	37649E+00	37649E+00	61743E-02	-36081E+00	
21	21	21	10000E+01	10000E+01	17000E+01	17000E+01	17000E+01	17000E+01	10469E+01	10469E+01	41008E+00	41008E+00	41008E+00	41008E+00	70871E-01	-70600E+00	
233	233	233	20948E+00	23424E+00	20506E+00	20506E+00	20506E+00	20506E+00	20862E+00	20862E+00	83071E+00	83071E+00	83071E+00	83071E+00	11711E-02	24283E+00	
5	5	5	37709E+00	32912E+00	27235E+00	17000E+01	17000E+01	17000E+01	32360E+00	17404E+01	30051E+00	30051E+00	30051E+00	30051E+00	55099E-02	-78164E+00	
22	22	22	17000E+01	17000E+01	13000E+01	13000E+01	13000E+01	13000E+01	17404E+01	17404E+01	48716E+00	48716E+00	48716E+00	48716E+00	72739E-01	-49026E+00	
234	234	234	10342E+00	20506E+00	17017E+00	17017E+00	17017E+00	17017E+00	17051E+00	17051E+00	79543E+00	79543E+00	79543E+00	79543E+00	17257E-02	-56879E+00	
6	6	6	37407E+00	31522E+00	27676E+00	17000E+01	17000E+01	17000E+01	32445E+00	10469E+01	30617E+00	30617E+00	30617E+00	30617E+00	64442E-02	-66154E+00	
21	21	21	10000E+01	10000E+01	17000E+01	17000E+01	17000E+01	17000E+01	10469E+01	10469E+01	30400E+00	30400E+00	30400E+00	30400E+00	70229E-01	-42790E+00	
235	235	235	23424E+00	25526E+00	22441E+00	22441E+00	22441E+00	22441E+00	23014E+00	23014E+00	80807E+00	80807E+00	80807E+00	80807E+00	11355E-02	-37346E+00	
6	6	6	37407E+00	27676E+00	24900E+00	17000E+01	17000E+01	17000E+01	27762E+00	17404E+01	29441E+00	29441E+00	29441E+00	29441E+00	57259E-02	-33243E+00	
22	22	22	17000E+01	17000E+01	13000E+01	13000E+01	13000E+01	13000E+01	17404E+01	17404E+01	40403E+00	40403E+00	40403E+00	40403E+00	71472E-01	-80532E+00	
236	236	236	20506E+00	22441E+00	19544E+00	19544E+00	19544E+00	19544E+00	19093E+00	19093E+00	83546E+00	83546E+00	83546E+00	83546E+00	10943E-02	10190E+00	
7	7	7	31522E+00	25526E+00	20506E+00	17000E+01	17000E+01	17000E+01	27436E+00	10469E+01	24895E+00	24895E+00	24895E+00	24895E+00	49291E-02	-35317E+00	
21	21	21	10000E+01	10000E+01	17000E+01	17000E+01	17000E+01	17000E+01	10469E+01	10469E+01	37360E+00	37360E+00	37360E+00	37360E+00	64672E-01	-57506E+00	
237	237	237	25526E+00	26536E+00	22441E+00	22441E+00	22441E+00	22441E+00	24617E+00	24617E+00	85344E+00	85344E+00	85344E+00	85344E+00	13409E-02	17659E+00	



M	Y1	Y2	X3	X4	XP	XN	A	CZ4
P	Z1	Z2	Z3	Z4	ZF	ZN	FL	CZ5
							CZ1	CZ6
7	.27776E+00	.25558E+00	.17770E+00	.22770E+00	.25770E+00	.24014E+00	.45675E-02	-.59871E+00
22	.17000E+01	.16000E+01	.15000E+01	.14000E+01	.13000E+01	.44721E+00	.62250E-01	-.41811E+00
238	.22411E+00	.23558E+00	.17770E+00	.22770E+00	.25770E+00	.86159E+00	.15816E-02	-.34490E+00
8	.26833E+00	.25558E+00	.17770E+00	.22770E+00	.25770E+00	.20516E+00	.44608E-02	-.52505E+00
21	.16000E+01	.16000E+01	.16000E+01	.16000E+01	.16000E+01	.36432E+00	.62397E-01	-.38590E+00
239	.26833E+00	.27807E+00	.24414E+00	.25558E+00	.25558E+00	.90859E+00	.12747E-02	-.21773E+00
8	.23558E+00	.17770E+00	.15359E+00	.19479E+00	.17654E+00	.15815E+00	.39455E-02	-.33818E+00
22	.17000E+01	.17000E+01	.16000E+01	.16000E+01	.17464E+01	.43054E+00	.63854E-01	-.65205E+00
240	.23558E+00	.24414E+00	.26201E+00	.19479E+00	.21582E+00	.87760E+00	.15097E-02	.14752E+00
9	.22518E+00	.17770E+00	.15740E+00	.19770E+00	.16994E+00	.16465E+00	.47231E-02	-.35256E+00
21	.16000E+01	.16000E+01	.17000E+01	.17000E+01	.16489E+01	.35672E+00	.61939E-01	-.48431E+00
241	.27807E+00	.25558E+00	.22514E+00	.24414E+00	.26537E+00	.91971E+00	.12570E-02	.12615E+00
9	.19770E+00	.12979E+00	.12979E+00	.16359E+00	.16258E+00	.15931E+00	.41723E-02	-.52555E+00
22	.17000E+01	.17000E+01	.16000E+01	.16000E+01	.17484E+01	.42837E+00	.63025E-01	-.38141E+00
242	.24414E+00	.25141E+00	.20803E+00	.25558E+00	.22707E+00	.88945E+00	.12890E-02	-.21533E+00
10	.17843E+00	.14245E+00	.12401E+00	.15740E+00	.15096E+00	.12874E+00	.37251E-02	-.46205E+00
21	.16000E+01	.16000E+01	.17000E+01	.17000E+01	.16489E+01	.35160E+00	.58857E-01	-.36436E+00
243	.26655E+00	.29146E+00	.25558E+00	.25141E+00	.27166E+00	.92725E+00	.11574E-02	-.12422E+00
10	.15710E+00	.12401E+00	.10327E+00	.12979E+00	.12918E+00	.12401E+00	.32875E-02	-.34394E+00
22	.17000E+01	.17000E+01	.16000E+01	.16000E+01	.17484E+01	.42232E+00	.60149E-01	-.54853E+00
244	.25141E+00	.25558E+00	.22174E+00	.20803E+00	.25246E+00	.89784E+00	.12006E-02	.10442E+00
11	.14245E+00	.10160E+00	.69360E-01	.12401E+00	.11470E+00	.96841E-01	.40640E-02	-.35142E+00
21	.16000E+01	.16000E+01	.17000E+01	.17000E+01	.16489E+01	.34800E+00	.56740E-01	-.43715E+00
245	.29146E+00	.29557E+00	.25558E+00	.25558E+00	.27603E+00	.93248E+00	.11627E-02	.77009E-01
11	.12461E+00	.89360E-01	.73960E-01	.10327E+00	.98146E-01	.93924E-01	.35841E-02	-.46862E+00
22	.17000E+01	.17000E+01	.16000E+01	.16000E+01	.17484E+01	.41823E+00	.59739E-01	-.36125E+00
246	.25589E+00	.25757E+00	.21479E+00	.21174E+00	.25620E+00	.90347E+00	.12013E-02	-.12389E+00
12	.16160E+00	.63360E-01	.60410E-01	.89360E-01	.80210E-01	.67214E-01	.33110E-02	-.42908E+00
21	.16000E+01	.16000E+01	.17000E+01	.17000E+01	.16489E+01	.34547E+00	.56721E-01	-.35415E+00
247	.29565E+00	.29802E+00	.26165E+00	.25957E+00	.27912E+00	.93602E+00	.11032E-02	-.62575E-01
12	.89360E-01	.60410E-01	.47960E-01	.73960E-01	.60656E-01	.65114E-01	.24191E-02	-.34780E+00
22	.17000E+01	.17000E+01	.16000E+01	.16000E+01	.17484E+01	.41534E+00	.57893E-01	-.49498E+00
248	.25957E+00	.26165E+00	.22051E+00	.21479E+00	.25664E+00	.90733E+00	.11496E-02	.54249E-01
13	.66860E-01	.31496E-01	.27650E-01	.60410E-01	.47154E-01	.34447E-01	.37352E-02	-.35120E+00
21	.16000E+01	.16000E+01	.17000E+01	.17000E+01	.16489E+01	.34398E+00	.56668E-01	-.42177E+00
249	.29802E+00	.29795E+00	.26305E+00	.26165E+00	.26047E+00	.93814E+00	.11261E-02	.33550E-01
13	.66410E-01	.27650E-01	.46835E-01	.99700E-01	.90550E-01	.30132E-01	.32917E-02	-.47254E+00
22	.17000E+01	.17000E+01	.16000E+01	.16000E+01	.17484E+01	.41370E+00	.57701E-01	-.35218E+00
250	.26165E+00	.26165E+00	.22166E+00	.21666E+00	.24042E+00	.90961E+00	.11673E-02	-.45454E-01

M	X1	X2	X3	X4	X5	X6	X7	X8	X9	A	C74
N	Y1	Y2	Y3	Y4	Y5	Y6	Y7	Y8	Y9	FL	C75
P	Z1	Z2	Z3	Z4	Z5	Z6	Z7	Z8	Z9	CL1	C76
14	.31490E+01	.00000E+00	.00000E+00	.27050E+01	.14000E+01	.14000E+01	.14000E+01	.12220E+01	.12220E+01	.31490E+02	-.42000E+00
21	.10000E+01	.10000E+01	.10000E+01	.10000E+01	.10000E+01	.10000E+01	.10000E+01	.35221E+00	.35221E+00	.50457E+01	-.35100E+00
251	.29000E+00	.29000E+00	.29000E+00	.29000E+00	.29000E+00	.29000E+00	.29000E+00	.73110E+00	.73110E+00	.10070E+02	-.11407E+01
14	.27050E+01	.00000E+00	.00000E+00	.22050E+01	.12000E+01	.12000E+01	.12000E+01	.11377E+01	.11377E+01	.27759E+02	-.35000E+00
22	.17000E+01	.17000E+01	.17000E+01	.17000E+01	.17000E+01	.17000E+01	.17000E+01	.94209E+00	.94209E+00	.50503E+01	-.47203E+00
252	.20330E+00	.20330E+00	.20330E+00	.20330E+00	.20330E+00	.20330E+00	.20330E+00	.91071E+00	.91071E+00	.11339E+02	.69129E+02
1	.43500E+00	.43500E+00	.43500E+00	.43500E+00	.43500E+00	.43500E+00	.43500E+00	.62525E+00	.62525E+00	.61704E+02	-.85442E+00
23	.10000E+01	.10000E+01	.10000E+01	.10000E+01	.10000E+01	.10000E+01	.10000E+01	.76952E+00	.76952E+00	.85008E+01	-.31953E+01
253	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.13000E+00	.13000E+00	.65178E+02	.39207E+00
QUESTIONABLE POINT -PUNK FIT											
WARNING LONG THIN QUAD.											
1	.31225E+00	.30557E+00	.30557E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.38227E+00	.38227E+00	.52762E+02	-.32826E+01
24	.19000E+01	.19000E+01	.19000E+01	.20000E+01	.20000E+01	.20000E+01	.20000E+01	.94956E+00	.94956E+00	.10336E+00	-.85409E+00
254	.00000E+00	.32120E+01	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.91745E+01	.91745E+01	.14002E+01	-.80034E+00
2	.42657E+00	.39574E+00	.39574E+00	.30557E+00	.30557E+00	.30557E+00	.30557E+00	.56745E+00	.56745E+00	.70401E+02	-.27613E+01
23	.10000E+01	.10000E+01	.10000E+01	.19000E+01	.19000E+01	.19000E+01	.19000E+01	.73290E+00	.73290E+00	.67888E+01	-.12693E+01
255	.44830E+01	.91300E+01	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.37576E+00	.37576E+00	.60946E+02	-.18141E+01
QUESTIONABLE POINT -PUNK FIT											
WARNING LONG THIN QUAD.											
2	.30557E+00	.26349E+00	.26349E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.34702E+00	.34702E+00	.58022E+02	-.79607E+00
24	.19000E+01	.19000E+01	.19000E+01	.20000E+01	.20000E+01	.20000E+01	.20000E+01	.89003E+00	.89003E+00	.10106E+00	-.32434E+01
256	.32120E+01	.69450E+01	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.27334E+00	.27334E+00	.10190E+01	-.12984E+00
3	.39574E+00	.35000E+00	.35000E+00	.20349E+00	.20349E+00	.20349E+00	.20349E+00	.46421E+00	.46421E+00	.57441E+02	-.10000E+01
23	.10000E+01	.10000E+01	.10000E+01	.19000E+01	.19000E+01	.19000E+01	.19000E+01	.67912E+00	.67912E+00	.85453E+01	-.22008E+01
257	.91300E+01	.12301E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.59106E+00	.59106E+00	.61126E+02	.94204E+00
QUESTIONABLE POINT -PUNK FIT											
WARNING LONG THIN QUAD.											
3	.28349E+00	.25717E+00	.25717E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.30112E+00	.30112E+00	.44910E+02	-.10725E+01
24	.19000E+01	.19000E+01	.19000E+01	.20000E+01	.20000E+01	.20000E+01	.20000E+01	.86891E+00	.86891E+00	.15335E+00	-.14107E+01
258	.65450E+01	.86550E+01	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.39372E+00	.39372E+00	.10502E+01	-.15017E+01
4	.39000E+00	.31202E+00	.31202E+00	.25717E+00	.25717E+00	.25717E+00	.25717E+00	.40107E+00	.40107E+00	.61282E+02	-.14007E+01
23	.10000E+01	.10000E+01	.10000E+01	.19000E+01	.19000E+01	.19000E+01	.19000E+01	.65397E+00	.65397E+00	.84509E+01	-.10088E+01
259	.12301E+00	.15218E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.60074E+00	.60074E+00	.62780E+02	-.13163E+01
QUESTIONABLE POINT -PUNK FIT											
WARNING LONG THIN QUAD.											
4	.25717E+00	.22320E+00	.22320E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.25090E+00	.25090E+00	.45683E+02	-.84137E+00
24	.19000E+01	.19000E+01	.19000E+01	.20000E+01	.20000E+01	.20000E+01	.20000E+01	.84293E+00	.84293E+00	.14446E+00	-.21728E+01
260	.86550E+01	.19002E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.47595E+00	.47595E+00	.15315E+01	.48785E+00
5	.31202E+00	.27230E+00	.27230E+00	.19000E+00	.19000E+00	.19000E+00	.19000E+00	.33069E+00	.33069E+00	.40933E+02	-.12347E+01
23	.10000E+01	.10000E+01	.10000E+01	.19000E+01	.19000E+01	.19000E+01	.19000E+01	.59700E+00	.59700E+00	.78404E+01	-.13134E+01
261	.15218E+00	.17017E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.00000E+00	.73017E+00	.73017E+00	.59601E+02	.10825E+01





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SOLID ANGLE = 12.500

XYZ POTENTIAL FLOW PROGRAM SOLUTIONS BY VANDERBILT

SAMPLE PROBLEM TRIAXIAL ELLIPSOID

XYZ POTENTIAL FLOW PROGRAM SECTION 99, VERSION 4

SAMPLE PROBLEM TRIAXIAL ELLIPSOID

X VELOCITY=-1.0 Y VELOCITY= .0 Z VELOCITY= .0

ITERATION	SUM OF CHANGES	A	B1	B2
1	.0052E+00			
2	.13219E+00			
3	.07071E+01			
4	.05190E+01			
5	.04774E+02			
6	.05680E+02			
7	.13430E+02			
8	.00600E+03			

X VELOCITY= .0 Y VELOCITY=-1.0 Z VELOCITY= .0

ITERATION	SUM OF CHANGES	A	B1	B2
1	.01142E+01			
2	.02740E+01			
3	.16521E+01			
4	.12150E+01			
5	.09020E+00			
6	.04920E+00			
7	.07463E+00			
8	.04695E+00			
9	.02362E+00			
10	.18540E+00			
11	.13005E+05			

A EXTRAPOLATION

X VELOCITY= .0 Y VELOCITY= .0 Z VELOCITY=-1.0

ITERATION	SUM OF CHANGES	A	B1	B2
1	.15305E+02			
2	.09163E+01			
3	.05055E+00			
4	.16720E+00			
5	.06973E-01			
6	.01712E-02			
7	.13059E-02			

XYZ POTENTIAL FLOW PROGRAM SECTION 99, VERSION 4

SAMPLE PROBLEM TRIAXIAL ELLIPSOID

A Flow

PT.	A	Y	C	X	W	WZ	AUSAY	CP	SOURCE	V NUMMAL
1	90867	04777	60190	-002793	01075	27520	28104	92068	11024	12E-04
2	90671	14777	00127	-000700	00156	27469	26982	94850	11016	12E-04
3	94206	04777	10613	-042322	01309	03955	76018	41217	09881	11E-04
4	94350	14777	12574	-042399	00940	03753	70670	41217	09873	11E-04
5	86521	04777	24044	-079200	00006	69027	105084	-12072	07488	93E-05
6	86304	14777	24502	-079200	00000	69775	105084	-12134	07485	93E-05
7	70974	04777	31617	-1002127	00624	02115	109335	-42806	05888	74E-05
8	76731	14777	31537	-1002127	00624	02115	109335	-42917	05887	74E-05
9	04949	04777	30954	-1001064	00442	02000	109398	-04386	04587	62E-05
10	60821	14777	36001	-1010790	00156	02807	1028213	-04386	04586	62E-05
11	57471	04777	40708	-1024741	00339	45904	1022202	-74933	03697	50E-05
12	57327	14777	40604	-1024741	00339	45904	1022202	-74947	03695	50E-05
13	40538	04777	43505	-1030479	00257	36348	103546	-83400	02954	41E-05
14	40477	14777	43470	-1030479	00257	36348	103546	-83400	02954	41E-05
15	41100	04777	42505	-1033700	00742	30337	1039430	-83430	02954	34E-05
16	40947	14777	42571	-1033700	00742	30337	1039430	-83430	02954	34E-05
17	33655	04777	49391	-1033954	00624	30195	1037073	-87891	02420	34E-05
18	33571	14777	49306	-1033954	00624	30195	1037073	-87891	02420	34E-05
19	26741	04777	46121	-1037040	00126	19148	1039170	-93683	01494	22E-05
20	26674	14777	46000	-1037040	00126	19148	1039170	-93683	01494	22E-05
21	20317	04777	40845	-1036751	00072	14407	1039447	-94545	01127	17E-05
22	20266	14777	40773	-1036751	00072	14407	1039447	-94545	01127	17E-05
23	14208	04777	47442	-1037601	00066	10056	1040021	-92058	00774	12E-05
24	14173	14777	47318	-1037601	00066	10056	1040021	-92058	00774	12E-05
25	08352	04777	47759	-1039905	00132	10033	1040068	-96068	00450	67E-06
26	08332	14777	47645	-1039905	00132	10033	1040068	-96068	00450	67E-06
27	04959	04777	47934	-1040320	00012	01833	1040332	-96930	00141	21E-06
28	04937	14777	47809	-1040320	00012	01833	1040332	-96930	00141	21E-06
29	98124	24777	05101	-1040319	00555	27423	1040331	-90928	00140	22E-06
30	97372	34777	05062	-1040702	00572	27348	1040331	-91358	01160	12E-04
31	93555	24777	15445	-1040622	00572	27348	1040331	-90628	01170	12E-04
32	92859	34777	15377	-1042700	00227	63604	76894	40865	09865	11E-04
33	85809	24777	24428	-107701	04420	63441	77130	40457	09844	10E-04
34	85211	34777	24271	-107701	04420	63441	77130	40457	09844	10E-04
35	76344	24777	31370	-1002190	00262	69519	1066053	-12268	07480	92E-05
36	75759	34777	31133	-1002190	00132	62011	1066053	-12473	07470	92E-05
37	66404	24777	36670	-1002209	00442	01896	1090622	-043045	05878	74E-05
38	65475	34777	36394	-1002209	00242	02843	1090622	-04447	04584	62E-05
39	57038	24777	40434	-1010050	03189	52701	1028249	-04477	04581	62E-05
40	56602	34777	40149	-1010050	01707	43894	1028249	-04477	04581	62E-05
41	46232	24777	43276	-1024709	02411	43648	1032271	-7971	03692	50E-05
42	47803	34777	42945	-1024709	01302	30312	1032271	-7971	03691	50E-05
43	40740	24777	42102	-1030709	01049	30201	1035454	-83428	02950	41E-05
44	40478	34777	44610	-1030709	01049	30105	1035454	-83428	02950	41E-05
45	33401	24777	40021	-1033723	00472	30120	1037073	-87931	02427	34E-05
46	33145	34777	40014	-1033723	00117	24205	1037073	-87931	02427	34E-05
47	26534	24777	47720	-1033970	00151	19129	1038120	-90769	01443	27E-05
48	26335	34777	47352	-1033970	00623	19129	1038120	-90769	01442	27E-05
49	26104	24777	46527	-1037043	00000	14101	1039170	-93609	01494	22E-05
50	26009	34777	46155	-1037043	00000	14101	1039170	-93609	01494	22E-05





A FLUX

PT.	A	F	C	TA	RY	FL	ABS.V	CP	SOURCE	V	MUMMAL
101	.3150	.04900	.04900	-1.30021	.02215	.02190	1.38100	-90802	.01930		.27E-05
102	.3150	.04900	.04900	-1.30032	.02255	.02071	1.38170	-90908	.01934		.27E-05
103	.2275	.04900	.04900	-1.37970	.01902	.17004	1.39201	-93737	.01469		.22E-05
104	.2475	.04900	.04900	-1.37970	.01902	.10946	1.39210	-93811	.01480		.22E-05
105	.19219	.04900	.04900	-1.30270	.02259	.14259	1.39210	-94648	.01121		.16E-05
106	.16834	.04900	.04900	-1.30270	.02259	.14259	1.39220	-94675	.01120		.16E-05
107	.13440	.04900	.04900	-1.37004	.01906	.09973	1.40042	-90118	.00770		.11E-05
108	.13175	.04900	.04900	-1.37004	.01906	.09973	1.40055	-90155	.00777		.11E-05
109	.07401	.04900	.04900	-1.37006	.02259	.09820	1.40100	-90302	.00455		.60E-06
110	.07745	.04900	.04900	-1.37975	.02259	.09819	1.40117	-90327	.00450		.60E-06
111	.02401	.04900	.04900	-1.40330	.01800	.10825	1.40340	-90976	.00141		.21E-06
112	.02432	.04900	.04900	-1.40330	.01800	.10814	1.40359	-90955	.00140		.21E-06
113	.89515	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
114	.87019	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
115	.85347	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
116	.82968	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
117	.85335	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
118	.76151	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
119	.69646	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
120	.67704	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
121	.60651	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
122	.50800	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
123	.52034	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
124	.50303	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
125	.44001	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
126	.42774	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
127	.37211	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
128	.36174	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
129	.30471	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
130	.24621	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
131	.24210	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
132	.23535	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
133	.16345	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
134	.17832	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
135	.12864	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
136	.12305	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
137	.07502	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
138	.07352	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
139	.02374	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
140	.02308	.04900	.04900	-1.02094	.02017	.02017	1.32064	-62304	.11344		.12E-04
141	.84159	1.04970	1.04970	-1.17270	.02270	.02270	1.40120	-96354	.00450		.64E-06
142	.80876	1.14964	1.14964	-1.17270	.02270	.02270	1.40120	-96354	.00450		.64E-06
143	.80241	1.04970	1.04970	-1.17270	.02270	.02270	1.40120	-96354	.00450		.64E-06
144	.77130	1.14964	1.14964	-1.17270	.02270	.02270	1.40120	-96354	.00450		.64E-06
145	.73648	1.04970	1.04970	-1.17270	.02270	.02270	1.40120	-96354	.00450		.64E-06
146	.70793	1.14964	1.14964	-1.17270	.02270	.02270	1.40120	-96354	.00450		.64E-06
147	.65479	1.04970	1.04970	-1.17270	.02270	.02270	1.40120	-96354	.00450		.64E-06
148	.62940	1.14964	1.14964	-1.17270	.02270	.02270	1.40120	-96354	.00450		.64E-06
149	.57022	1.04970	1.04970	-1.17270	.02270	.02270	1.40120	-96354	.00450		.64E-06
150	.54811	1.14964	1.14964	-1.17270	.02270	.02270	1.40120	-96354	.00450		.64E-06

A FLUX

PI.	A	Y	Z	YA	YB	YC	YD	YE	YF	YV	W	CP	SOURCE	V	NUMKAL
151	48921	1.04970	3761	-1.22263	0.0070	4.2623	1.32572	-7.7792	0.3053	0.3053	.47E-05				
152	47024	1.14904	3330	-1.22294	0.0195	4.2243	1.32077	-7.7032	0.3039	0.3039	.40E-05				
153	41303	1.04970	3713	-1.30927	0.0216	3.5501	1.35066	-8.4032	0.2921	0.2921	.34E-05				
154	34705	1.14904	3579	-1.33710	0.0717	3.5004	1.35700	-8.4161	0.2916	0.2916	.36E-05				
155	34905	1.04970	3673	-1.33720	0.0920	2.7401	1.37226	-8.0316	0.2398	0.2398	.32E-05				
156	33623	1.14904	3723	-1.33022	0.0204	2.9176	1.37276	-8.8546	0.2390	0.2390	.32E-05				
157	28647	1.04970	4012	-1.30147	0.0381	2.332	1.38236	-9.1147	0.1920	0.1920	.20E-05				
158	27537	1.14904	3640	-1.32015	0.0306	2.3558	1.38267	-9.1233	0.1917	0.1917	.20E-05				
159	22762	1.04970	4090	-1.34704	0.0409	1.6004	1.39255	-9.3919	0.1475	0.1475	.21E-05				
160	21873	1.14904	3933	-1.33008	0.0337	1.6507	1.39255	-9.4039	0.1472	0.1472	.21E-05				
161	17244	1.04970	4160	-1.30672	0.0211	1.4061	1.39571	-9.4801	0.1115	0.1115	.10E-05				
162	16624	1.14904	4007	-1.33002	0.0205	1.3904	1.39007	-9.4902	0.1113	0.1113	.15E-05				
163	12094	1.04970	4206	-1.33773	0.0322	0.9815	1.40095	-9.6266	0.0772	0.0772	.11E-05				
164	11625	1.14904	4044	-1.33773	0.0177	0.9737	1.40124	-9.6338	0.0768	0.0768	.11E-05				
165	07110	1.04970	4230	-1.40037	0.0087	6.2740	1.40134	-9.6447	0.0452	0.0452	.64E-06				
166	66834	1.14904	4072	-1.40003	0.0102	1.40103	1.40103	-9.6512	0.0452	0.0452	.63E-06				
167	02232	1.04970	4255	-1.40337	0.0276	0.1761	1.40390	-9.7095	0.0138	0.0138	.20E-06				
168	02146	1.14904	4005	-1.40340	0.0310	0.1768	1.40417	-9.7169	0.0139	0.0139	.20E-06				
169	77171	1.24927	0.012	-2.2003	4.261	2.3847	5.8174	-6.0138	1.0830	1.0830	.11E-04				
170	72937	1.34948	0.079	-2.20740	5.874	5.9504	6.3574	-5.9504	1.0613	1.0613	.10E-04				
171	73527	1.24927	12108	-2.3424	3.144	5.6808	6.6346	-2.3338	0.9345	0.9345	.92E-05				
172	69544	1.34948	11516	-2.30337	4.1047	5.4089	8.8723	-2.1203	0.9187	0.9187	.90E-05				
173	67540	1.24927	12230	-2.04040	2.9422	6.3460	1.09305	-1.7906	0.7205	0.7205	.82E-05				
174	63829	1.34948	16101	-2.80266	2.8036	6.2301	1.10300	-2.1673	0.7116	0.7116	.80E-05				
175	60044	1.24927	24601	-1.04002	1.3516	5.7930	1.21234	-4.6908	0.5721	0.5721	.66E-05				
176	56747	1.34948	23325	-1.00024	2.0739	5.6747	1.21712	-4.6138	0.5665	0.5665	.65E-05				
177	52253	1.24927	20347	-1.11024	1.3470	4.9420	1.29016	-6.8437	0.4471	0.4471	.50E-05				
178	49420	1.34948	27262	-1.10004	1.3146	4.5071	1.29236	-6.7077	0.4450	0.4450	.55E-05				
179	44804	1.24927	31823	-1.22036	1.0206	4.1755	1.32742	-7.6338	0.3030	0.3030	.45E-05				
180	42394	1.34948	30075	-1.12070	1.2610	4.1112	1.32734	-7.6768	0.3005	0.3005	.44E-05				
181	37938	1.24927	34039	-1.13104	0.7426	3.4035	1.35006	-8.4431	0.2403	0.2403	.30E-05				
182	32853	1.34948	32159	-1.131220	0.0956	3.4210	1.35401	-8.4691	0.2894	0.2894	.37E-05				
183	32084	1.24927	32522	-1.34144	0.0326	2.6872	1.37337	-8.6613	0.2385	0.2385	.31E-05				
184	30321	1.34948	33570	-1.34245	0.07183	2.6515	1.37426	-8.6863	0.2362	0.2362	.31E-05				
185	26272	1.24927	36674	-1.36205	0.0469	2.3342	1.38334	-9.1377	0.1911	0.1911	.25E-05				
186	24828	1.34948	34078	-1.30032	0.02045	2.3033	1.38379	-9.1487	0.1905	0.1905	.25E-05				
187	20874	1.24927	37504	-1.30001	0.0379	1.8367	1.39332	-9.4135	0.1464	0.1464	.20E-05				
188	14728	1.34948	32500	-1.33110	0.0426	1.6156	1.39366	-9.4228	0.1462	0.1462	.20E-05				
189	12060	1.24927	36169	-1.33725	0.0337	1.3623	1.39643	-9.3000	0.1104	0.1104	.15E-05				
190	11989	1.34948	36072	-1.30761	0.0217	1.3665	1.39670	-9.3078	0.1104	0.1104	.15E-05				
191	11091	1.24927	36545	-1.34000	0.0441	0.9646	1.40152	-9.6425	0.0766	0.0766	.11E-05				
192	10482	1.34948	36475	-1.33763	0.0217	0.9242	1.40177	-9.6497	0.0765	0.0765	.10E-05				
193	06520	1.24927	36621	-1.40003	0.0136	0.9648	1.40204	-9.6570	0.0450	0.0450	.62E-06				
194	06162	1.34948	36717	-1.40115	0.0250	0.9573	1.40232	-9.6650	0.0447	0.0447	.61E-06				
195	02047	1.24927	36760	-1.40434	0.0353	0.1782	1.40443	-9.7250	0.0142	0.0142	.14E-06				
196	01935	1.34948	36036	-1.40400	0.0400	0.1758	1.40471	-9.7322	0.0141	0.0141	.14E-06				
197	60077	1.44936	0.033	-2.34503	5.6531	2.115	6.9737	-5.1367	1.0325	1.0325	.10E-04				
198	62447	1.24927	0.0246	-2.4124	6.1174	2.0200	7.6474	-4.0750	0.9945	0.9945	.97E-05				
199	64303	1.44936	1.0721	-2.0007	4.2132	5.2523	9.1087	-1.3935	0.8984	0.8984	.88E-05				
200	54940	1.54919	0.0901	-2.64921	4.9336	4.9335	9.5386	-0.9015	0.8708	0.8708	.85E-05				

X FLUM

PT.	A	Y	C	VA	VT	VZ	ABS.V	LP	SOURCE	V NORMAL
201	59574	1.44929	10709	-0.90139	22192	60227	1.11475	-24290	07J01	70E-05
202	54047	1.54919	11505	-0.90000	35603	27593	1.13069	-27840	06042	70E-05
203	52706	1.44937	12776	-1.50001	22702	55163	1.22200	-49542	05293	63E-05
204	43066	1.54919	13100	-1.50007	20070	55027	1.23100	-51540	05408	62E-05
205	46126	1.44936	20979	-1.17122	17071	47719	1.29230	-67602	04400	54E-05
206	42311	1.54919	23340	-1.10003	11253	46324	1.29400	-68741	04343	52E-05
207	39572	1.44936	20770	-1.20172	10115	34048	1.33100	-77157	03273	44E-05
208	36300	1.54919	20749	-1.20071	14600	30048	1.33303	-77911	03527	43E-05
209	33403	1.44936	30024	-1.34423	10143	33549	1.36010	-85005	02871	30E-05
210	30596	1.54919	27542	-1.31070	11530	32642	1.36170	-85423	02839	30E-05
211	28000	1.44936	31332	-0.9139	00139	27996	1.37489	-84340	02340	31E-05
212	25959	1.54919	26741	-1.34499	02247	27251	1.37543	-84101	02240	31E-05
213	23173	1.44936	32306	-1.30000	09400	22619	1.38397	-91530	01868	29E-05
214	21257	1.54919	27009	-1.35421	07278	22022	1.38303	-91500	01850	29E-05
215	18412	1.44936	33133	-1.30005	04890	17001	1.39310	-94046	01452	20E-05
216	16009	1.54919	30373	-1.30122	05508	17406	1.39327	-94120	01440	19E-05
217	13969	1.44936	33009	-1.30000	03099	15452	1.39050	-95037	01097	19E-05
218	12452	1.54919	30053	-1.30701	04104	13127	1.39061	-95053	01087	14E-05
219	09703	1.44936	34043	-1.39045	02471	09412	1.40183	-96514	00762	10E-05
220	06974	1.54919	31228	-1.30000	02800	09170	1.40168	-96754	00754	10E-05
221	05751	1.44936	34200	-1.40123	04579	09479	1.40230	-96600	00443	60E-06
222	05275	1.54919	31435	-1.40112	01673	05361	1.40225	-96629	00442	54E-06
223	01806	1.44936	34303	-1.40004	00491	01730	1.40470	-97334	00139	14E-06
224	01656	1.54919	31539	-1.40400	00519	01606	1.40409	-97315	00138	14E-06
225	55815	1.64992	62401	-0.92078	00934	10219	0.85590	-26744	09427	92E-05
226	47761	1.74943	02403	-0.0704	00342	12452	0.60089	07670	08659	82E-05
227	53217	1.64992	60644	-0.71000	05708	45135	1.00241	-00463	08323	81E-05
228	45537	1.74943	67542	-0.81076	57323	39132	1.06742	-13938	07730	76E-05
229	46844	1.64992	61913	-0.94104	43408	53435	1.15272	-32877	06619	73E-05
230	41745	1.74943	11100	-0.79223	43408	47519	1.18520	-40470	06252	64E-05
231	43426	1.64992	17099	-1.01945	29203	49954	1.24262	-54409	05342	60E-05
232	37159	1.74943	12273	-1.13024	32768	42257	1.26081	-56903	05094	57E-05
233	37818	1.64992	20002	-1.00000	21014	44041	1.30539	-70405	04244	51E-05
234	32360	1.74943	17051	-1.22723	24709	40441	1.31570	-73108	04070	49E-05
235	32445	1.64992	25014	-1.27320	16928	37203	1.33750	-78809	03459	42E-05
236	27762	1.74943	17633	-1.20431	14401	34504	1.34450	-80769	03330	40E-05
237	27436	1.64992	24617	-1.30007	13174	31292	1.36381	-85998	02772	38E-05
238	25477	1.74943	21004	-1.32721	11197	29104	1.36722	-86929	02642	36E-05
239	25262	1.64992	25639	-1.34704	10000	26177	1.37632	-89425	02260	30E-05
240	19854	1.74943	21902	-1.35071	12297	24377	1.37003	-85896	02180	30E-05
241	18999	1.64992	26537	-1.30000	00308	21201	1.38441	-91600	01821	24E-05
242	16258	1.74943	22707	-1.30726	07420	19841	1.38225	-91892	01768	24E-05
243	15096	1.64992	27106	-1.30000	09426	16747	1.39329	-94120	01417	19E-05
244	12918	1.74943	23245	-1.30000	07477	13677	1.39378	-94261	01375	19E-05
245	11470	1.64992	27003	-1.30000	07806	12600	1.39039	-94992	01072	14E-05
246	09815	1.74943	23040	-1.30045	03006	11800	1.39062	-95055	01043	14E-05
247	06021	1.64992	27912	-1.30000	03248	08622	1.40125	-96351	00741	10E-05
248	06864	1.74943	23054	-1.30000	03099	08273	1.40125	-96350	00714	98E-06
249	04715	1.64992	20097	-1.40000	04959	05167	1.40195	-96546	00439	50E-06
250	04035	1.74943	24042	-1.40002	02252	04847	1.40164	-96401	00423	57E-06



Y FLUID

PT.	A	T	L	VX	VY	VZ	ABSXAV	CP	SOURCE	V NUKMAL
1	9.0019	0.0000	0.0000	0.0000	-1.12617	0.0200	1.12600	-2.1297	00118	-1.14E-08
2	9.0021	0.0000	0.0000	0.0000	-1.12617	0.0041	1.12750	-2.1719	00354	-1.56E-08
3	9.0026	0.0000	0.0000	0.0000	-1.12605	0.0008	1.12692	-2.1732	00105	-1.6E-08
4	9.0030	0.0000	0.0000	0.0000	-1.12725	0.0074	1.12700	-2.2707	00314	-1.48E-08
5	9.0031	0.0000	0.0000	0.0000	-1.12703	0.0084	1.12788	-2.2721	00087	-1.4E-08
6	9.0034	0.0000	0.0000	0.0000	-1.12703	0.0042	1.12747	-2.2719	00261	-1.43E-08
7	9.0024	0.0000	0.0000	0.0000	-1.12601	0.0062	1.12705	-2.2719	00078	-1.2E-08
8	9.0031	0.0000	0.0000	0.0000	-1.12696	0.0047	1.12732	-2.2705	00231	-1.56E-08
9	9.0029	0.0000	0.0000	0.0000	-1.12722	0.0078	1.12732	-2.2708	00064	-1.1E-08
10	9.0021	0.0000	0.0000	0.0000	-1.12673	0.0035	1.12703	-2.2702	00206	-1.31E-08
11	9.0031	0.0000	0.0000	0.0000	-1.12710	0.0078	1.12713	-2.2702	00066	-1.93E-04
12	9.0037	0.0000	0.0000	0.0000	-1.12601	0.0034	1.12687	-2.2694	00194	-1.28E-08
13	9.0039	0.0000	0.0000	0.0000	-1.12691	0.0057	1.12693	-2.2694	00062	-1.85E-09
14	9.0047	0.0000	0.0000	0.0000	-1.12695	0.0025	1.12669	-2.2694	00184	-1.25E-08
15	9.0039	0.0000	0.0000	0.0000	-1.12677	0.0039	1.12674	-2.2696	00060	-1.80E-04
16	9.0037	0.0000	0.0000	0.0000	-1.12656	0.0025	1.12659	-2.2692	00174	-1.24E-08
17	9.0035	0.0000	0.0000	0.0000	-1.12606	0.0029	1.12668	-2.2691	00054	-1.75E-04
18	9.0037	0.0000	0.0000	0.0000	-1.12627	0.0016	1.12659	-2.2684	00175	-1.24E-08
19	9.0041	0.0000	0.0000	0.0000	-1.12634	0.0015	1.12657	-2.2691	00056	-1.75E-04
20	9.0037	0.0000	0.0000	0.0000	-1.12618	0.0010	1.12634	-2.2687	00170	-1.24E-08
21	9.0037	0.0000	0.0000	0.0000	-1.12647	0.0070	1.12644	-2.2689	00056	-1.71E-04
22	9.0037	0.0000	0.0000	0.0000	-1.12644	0.0042	1.12631	-2.2688	00169	-1.21E-08
23	9.0039	0.0000	0.0000	0.0000	-1.12644	0.0079	1.12646	-2.2684	00056	-1.70E-04
24	9.0032	0.0000	0.0000	0.0000	-1.12608	0.0029	1.12628	-2.2682	00167	-1.21E-08
25	9.0032	0.0000	0.0000	0.0000	-1.12642	0.0024	1.12644	-2.2687	00056	-1.69E-04
26	9.0032	0.0000	0.0000	0.0000	-1.12600	0.0024	1.12626	-2.2686	00167	-1.20E-08
27	9.0022	0.0000	0.0000	0.0000	-1.12640	0.0075	1.12642	-2.2683	00055	-1.69E-04
28	9.0016	0.0000	0.0000	0.0000	-1.12605	0.0025	1.12625	-2.2683	00166	-1.20E-08
29	9.0024	0.0000	0.0000	0.0000	-1.12635	0.0107	1.12615	-2.2687	00593	-1.92E-08
30	9.0032	0.0000	0.0000	0.0000	-1.12605	0.0179	1.12396	-2.2632	00834	-1.13E-07
31	9.0056	0.0000	0.0000	0.0000	-1.12643	0.0343	1.12680	-2.2641	00534	-1.80E-08
32	9.0039	0.0000	0.0000	0.0000	-1.12645	0.0487	1.12493	-2.2656	00751	-1.11E-07
33	9.0068	0.0000	0.0000	0.0000	-1.12633	0.0400	1.12662	-2.2656	00437	-1.71E-08
34	9.0021	0.0000	0.0000	0.0000	-1.12675	0.0768	1.12532	-2.2627	00615	-1.97E-08
35	9.0034	0.0000	0.0000	0.0000	-1.12605	0.0418	1.12607	-2.2638	00384	-1.56E-08
36	9.0059	0.0000	0.0000	0.0000	-1.12357	0.0579	1.12560	-2.2669	00548	-1.80E-08
37	9.0084	0.0000	0.0000	0.0000	-1.12609	0.0401	1.12654	-2.2690	00346	-1.52E-08
38	9.0075	0.0000	0.0000	0.0000	-1.12392	0.0564	1.12561	-2.2670	00491	-1.71E-08
39	9.0038	0.0000	0.0000	0.0000	-1.12607	0.0800	1.12642	-2.2682	00326	-1.45E-08
40	9.0062	0.0000	0.0000	0.0000	-1.12614	0.0547	1.12563	-2.2670	00461	-1.62E-08
41	9.0023	0.0000	0.0000	0.0000	-1.12600	0.0764	1.12626	-2.2685	00309	-1.42E-08
42	9.0063	0.0000	0.0000	0.0000	-1.12642	0.0326	1.12556	-2.2668	00435	-1.57E-08
43	9.0070	0.0000	0.0000	0.0000	-1.12652	0.0315	1.12616	-2.2665	00300	-1.39E-08
44	9.0073	0.0000	0.0000	0.0000	-1.12624	0.0250	1.12551	-2.2678	00423	-1.53E-08
45	9.0041	0.0000	0.0000	0.0000	-1.12640	0.0600	1.12609	-2.2680	00294	-1.37E-08
46	9.0059	0.0000	0.0000	0.0000	-1.12625	0.0214	1.12547	-2.2668	00414	-1.50E-08
47	9.0059	0.0000	0.0000	0.0000	-1.12646	0.0618	1.12599	-2.2676	00265	-1.36E-08
48	9.0035	0.0000	0.0000	0.0000	-1.12623	0.0506	1.12540	-2.2652	00401	-1.50E-08
49	9.0064	0.0000	0.0000	0.0000	-1.12635	0.0564	1.12593	-2.2672	00282	-1.35E-08
50	9.0009	0.0000	0.0000	0.0000	-1.12642	0.0025	1.12535	-2.2664	00596	-1.46E-08

Y FLUX

PT.	A	Y	Z	Y	VZ	AB3-V	UP	SOURCE	V NORMAL
51	.14101	.24975	.62029	.00293	-1.12533	1.12589	-.20764	.00274	-.34E-08
52	.13993	.24974	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.47E-08
53	.00293	.24975	.62029	.00144	-1.12531	1.12587	-.20759	.00274	-.34E-08
54	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.40E-08
55	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.34E-08
56	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.40E-08
57	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.10E-07
58	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.10E-07
59	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.14E-07
60	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.10E-07
61	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.12E-07
62	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.14E-07
63	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.10E-07
64	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.12E-07
65	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.80E-08
66	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.10E-07
67	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.77E-08
68	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.91E-08
69	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.71E-08
70	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.86E-08
71	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.66E-08
72	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.77E-08
73	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.62E-08
74	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.75E-08
75	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.62E-08
76	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.72E-08
77	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.59E-08
78	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.56E-08
79	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.68E-08
80	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.57E-08
81	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.60E-08
82	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.57E-08
83	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.60E-08
84	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.57E-08
85	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.66E-08
86	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.21E-07
87	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.23E-07
88	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.18E-07
89	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.20E-07
90	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.16E-07
91	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.17E-07
92	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.14E-07
93	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.12E-07
94	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.13E-07
95	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.10E-07
96	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.11E-07
97	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.95E-08
98	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.10E-07
99	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.80E-08
100	.00293	.24975	.62029	.00293	-1.12421	1.12533	-.20638	.00393	-.95E-08

Y FLUX

PT.	A	Y	Z	YA	YZ	ABS.Y	LP	SOURCE	Y NORMAL
101	.31459	.00000	.00000	-.01007	-.111724	1.12200	-.25900	.00799	-.81E-08
102	.31625	.74902	.43200	.01721	-1.11494	1.12000	-.25448	.00436	-.87E-08
103	.25295	.64785	.49220	.01202	-1.11772	1.12200	-.25900	.00773	-.80E-08
104	.24793	.74762	.44620	.01475	-1.11434	1.12000	-.25520	.00908	-.82E-08
105	.19219	.54735	.40233	.00920	-1.11700	1.12212	-.25914	.00767	-.70E-08
106	.10833	.74762	.45330	.01113	-1.11421	1.12000	-.25527	.00902	-.62E-08
107	.13470	.64705	.46770	.00920	-1.11794	1.12212	-.25915	.00750	-.75E-08
108	.13175	.74762	.42845	.00700	-1.11407	1.12000	-.25541	.00893	-.80E-08
109	.07701	.64705	.47000	.00367	-1.11000	1.12214	-.25920	.00757	-.73E-08
110	.07745	.74762	.46147	.00421	-1.11472	1.12000	-.25541	.00891	-.78E-08
111	.02481	.64705	.47235	.00115	-1.11800	1.12212	-.25916	.00754	-.73E-08
112	.02432	.74762	.46302	.00124	-1.11474	1.12000	-.25548	.00885	-.70E-08
113	.89215	.54778	.00000	.24335	-1.00709	1.00935	-.20638	.02160	-.24E-07
114	.87019	.94974	.00224	.27215	-1.02236	1.00821	-.18630	.02468	-.24E-07
115	.85347	.84978	.14130	.13725	-1.06104	1.10424	-.21934	.01955	-.21E-07
116	.82968	.74774	.15742	.14200	-1.00027	1.09071	-.20277	.02231	-.21E-07
117	.78335	.84978	.22312	.12002	-1.07277	1.11009	-.23230	.01002	-.18E-07
118	.70151	.94974	.21670	.14440	-1.00191	1.0901	-.21944	.01840	-.16E-07
119	.64646	.84978	.26025	.00744	-1.09770	1.11347	-.23909	.01430	-.15E-07
120	.67704	.94974	.27027	.10199	-1.07026	1.10901	-.22989	.01039	-.15E-07
121	.66021	.84978	.23428	.00377	-1.10409	1.11250	-.24448	.01280	-.13E-07
122	.59360	.94974	.32525	.07202	-1.07077	1.11181	-.23612	.01467	-.13E-07
123	.52034	.84978	.30710	.04840	-1.10052	1.11007	-.24695	.01207	-.11E-07
124	.50583	.94974	.32001	.05529	-1.10004	1.11337	-.23958	.01384	-.11E-07
125	.44001	.84978	.39400	.03671	-1.10014	1.13806	-.24800	.01144	-.10E-07
126	.42774	.94974	.30379	.04200	-1.10222	1.11434	-.24166	.01311	-.10E-07
127	.37211	.84978	.41221	.02721	-1.10900	1.11174	-.24935	.01110	-.90E-08
128	.30174	.94974	.40021	.03337	-1.10343	1.11490	-.24301	.01273	-.95E-08
129	.30471	.84978	.42529	.02249	-1.10971	1.11005	-.25003	.01086	-.84E-08
130	.24621	.94974	.41372	.02575	-1.10431	1.11200	-.24304	.01247	-.88E-08
131	.24210	.84978	.42208	.01709	-1.10100	1.11017	-.25030	.01052	-.88E-08
132	.23535	.94974	.42323	.01944	-1.10470	1.11250	-.24434	.01208	-.86E-08
133	.16395	.84978	.44239	.01206	-1.11039	1.11027	-.25053	.01045	-.84E-08
134	.17802	.94974	.43035	.01469	-1.10220	1.11200	-.24400	.01199	-.82E-08
135	.12804	.84978	.44704	.00000	-1.11027	1.11008	-.25063	.01032	-.82E-08
136	.12505	.94974	.43520	.01000	-1.11009	1.11273	-.24405	.01180	-.80E-08
137	.07502	.84978	.45061	.00325	-1.11009	1.11036	-.25073	.01032	-.80E-08
138	.04352	.94974	.43805	.00344	-1.10209	1.11277	-.24495	.01163	-.78E-08
139	.02374	.84978	.45210	.00125	-1.11071	1.11035	-.25072	.01026	-.80E-08
140	.02308	.94974	.43949	.00173	-1.10578	1.11200	-.24504	.01178	-.70E-08
141	.84159	1.04970	.04375	.30011	-1.10000	1.07780	-.16105	.02792	-.24E-07
142	.80356	1.14564	.04205	.34213	-1.00501	1.06640	-.13209	.03134	-.24E-07
143	.80341	1.04970	.13770	.23900	-1.04808	1.08743	-.16250	.02520	-.21E-07
144	.77130	1.14564	.12775	.20007	-1.02729	1.07004	-.15706	.02844	-.19E-07
145	.73543	1.04970	.20977	.10300	-1.00700	1.09759	-.20470	.02090	-.16E-07
146	.70743	1.14564	.20139	.10207	-1.02191	1.08091	-.16572	.02369	-.17E-07
147	.65479	1.04970	.23912	.11290	-1.00009	1.10364	-.21802	.01870	-.15E-07
148	.62429	1.14564	.22007	.12475	-1.00005	1.09050	-.20245	.02110	-.13E-07
149	.57022	1.04970	.24509	.00249	-1.00700	1.10732	-.22615	.01694	-.13E-07
150	.54811	1.14564	.30220	.07273	-1.07029	1.10140	-.21321	.01899	-.11E-07

Y FLUX

PT.	A	T	Z	FA	TF	YZ	ABS.Y	CP	SOURCE	Y NORMAL
151	40921	1.04970	34701	0.0071	-1.07419	1.0449	1.10734	-23075	01360	-1.1E-07
152	47024	1.14909	35296	0.0069	-1.06196	0.0025	1.10408	-21900	01193	-0.9E-06
153	41303	1.24970	35118	0.0075	-1.07409	0.0020	1.11065	-22323	01494	-0.97E-08
154	34705	1.14964	35567	0.0072	-1.06902	0.0027	1.10506	-22293	01700	-0.85E-08
155	34905	1.04970	35073	0.0077	-1.07690	1.7717	1.11134	-23508	01453	-0.89E-08
156	33679	1.14964	37233	0.0060	-1.06765	0.0031	1.10677	-22443	01650	-0.77E-08
157	26647	1.04970	40012	0.0071	-1.07107	1.1476	1.11180	-23627	01420	-0.82E-08
158	27537	1.14964	36901	0.0060	-1.06923	1.1760	1.10749	-22624	01616	-0.71E-08
159	27662	1.04970	40901	0.0069	-1.07474	1.1206	1.11217	-23693	01372	-0.80E-08
160	21879	1.14964	34373	0.0067	-1.07027	1.1920	1.10794	-22723	01564	-0.60E-08
161	17294	1.04970	41020	0.0066	-1.07497	1.1145	1.11234	-23740	01300	-0.70E-08
162	16624	1.14964	40007	0.0070	-1.07923	1.1939	1.10620	-22811	01554	-0.64E-08
163	11044	1.04970	42030	0.0149	-1.07427	1.1702	1.11241	-23754	01350	-0.74E-08
164	11625	1.14964	40934	0.0071	-1.07154	1.1296	1.10634	-22823	01537	-0.63E-08
165	07119	1.04970	46309	0.0076	-1.07990	1.0999	1.11254	-23765	01344	-0.72E-08
166	06834	1.14964	40723	0.0062	-1.07105	1.1253	1.10044	-22875	01530	-0.61E-08
167	02232	1.04970	42929	0.0190	-1.07926	1.0909	1.11256	-23764	01343	-0.72E-08
168	02146	1.14964	40827	0.0214	-1.07101	1.1498	1.10056	-22890	01528	-0.60E-08
169	77179	1.24957	07012	3.7725	-0.97322	0.7727	1.04076	-09571	03919	-0.20E-07
170	72954	1.34948	03741	4.1390	-0.93392	0.6538	1.02511	-05065	03937	-0.16E-07
171	73567	1.24957	12108	2.9590	-1.00021	1.1956	1.06170	-12722	03210	-0.17E-07
172	69544	1.34948	12910	3.2005	-0.96606	2.1640	1.04341	-08870	03000	-0.13E-07
173	67540	1.24957	14236	2.0420	-1.03115	2.3867	1.07799	-16206	02672	-0.15E-07
174	63829	1.34948	10101	2.2757	-1.00474	2.6610	1.06396	-13201	03017	-0.11E-07
175	60049	1.24957	24001	1.4921	-1.02000	2.4422	1.06780	-16331	02401	-0.11E-07
176	56744	1.34948	23325	1.6267	-1.02812	2.7339	1.07621	-15823	02117	-0.83E-08
177	52243	1.24957	22647	1.0448	-1.06225	2.4012	1.09405	-19694	02149	-0.95E-08
178	49420	1.34948	27262	1.1711	-1.04362	2.0474	1.08426	-17562	02444	-0.65E-08
179	44804	1.24957	3323	0.7746	-1.06924	2.3449	1.09753	-20426	02037	-0.77E-08
180	42347	1.34948	30075	0.8912	-1.07230	2.6363	1.08052	-18468	02313	-0.50E-08
181	37938	1.24957	34039	0.6020	-1.07376	2.4956	1.09960	-20929	01928	-0.67E-08
182	35853	1.34948	32169	0.6700	-1.02814	2.5860	1.09143	-19121	02190	-0.40E-08
183	32034	1.24957	35522	0.4769	-1.07637	2.2567	1.10086	-21169	01875	-0.59E-08
184	30321	1.34948	33570	0.2373	-1.00136	2.5495	1.09476	-19417	02118	-0.33E-08
185	26272	1.24957	30074	0.3699	-1.07837	2.2208	1.10179	-21394	01834	-0.53E-08
186	24828	1.34948	34678	0.4175	-1.06308	2.5131	1.09390	-19674	02091	-0.27E-08
187	20374	1.24957	37504	0.2758	-1.07475	2.2026	1.10240	-21529	01777	-0.50E-08
188	14728	1.34948	35500	0.3101	-1.00554	2.4866	1.09472	-19842	02024	-0.24E-08
189	12800	1.24957	30169	0.2111	-1.06050	2.1066	1.10271	-21546	01760	-0.46E-06
190	14589	1.34948	36072	0.2377	-1.00806	2.4706	1.09536	-19960	02012	-0.21E-08
191	11091	1.24957	36595	0.1427	-1.00119	2.1770	1.10259	-21628	01746	-0.45E-08
192	10402	1.34948	36475	0.1640	-1.06724	2.4570	1.09557	-20028	01984	-0.19E-06
193	06520	1.24957	36821	0.0024	-1.00154	2.1706	1.10314	-21693	01746	-0.43E-08
194	06162	1.34948	36717	0.0924	-1.06808	2.4441	1.09564	-20068	01980	-0.16E-08
195	02047	1.24957	36930	0.0026	-1.00173	2.1670	1.10323	-21711	01737	-0.43E-08
196	01935	1.34948	36335	0.0201	-1.00820	2.4457	1.09544	-20109	01977	-0.18E-08
197	60077	1.44926	03359	4.5106	-0.80430	0.9273	0.97701	00546	04407	-0.10E-07
198	62447	1.54919	03246	4.6767	-0.81925	1.0034	0.92040	06135	04440	-0.20E-08
199	64908	1.44926	10721	3.9718	-0.92363	2.3802	1.01951	-03940	04052	-0.84E-08
200	54540	1.54919	07801	3.9240	-0.86610	2.6026	0.96291	02794	04578	-0.14E-08



Y FLUM

PT.	A	Y	Z	X	Y	VZ	ADJUST	CP	SOURCE	V NUMAL
201	59574	1.44936	1.1005	22201	-90967	27538	1.09467	-09133	03414	-03E-00
202	59597	1.44939	1.1006	22209	-92220	32743	1.01790	-03611	03690	33E-04
203	52905	1.44936	2.1770	1.1336	-97603	30242	1.00021	-1.4000	03085	-04E-08
204	48506	1.44939	1.1925	2.0325	-97036	34125	1.03770	-07648	03524	20E-00
205	46126	1.44936	2.2445	1.1662	-1.04331	30204	1.07752	-1.4608	02782	-023E-08
206	42311	1.44939	2.3346	1.1072	-90331	34055	1.05137	-1.0534	03197	34E-00
207	39572	1.44936	2.6070	1.0066	-1.02905	29683	1.07630	-1.1542	02630	-011E-00
208	36300	1.44939	2.9749	1.1361	-99771	33533	1.05067	-1.1208	03034	43E-08
209	33403	1.44936	3.0024	0.7637	-1.09714	29124	1.08004	-1.0644	02506	-019E-04
210	30596	1.44939	2.7542	0.6050	-1.00757	35017	1.00380	-1.1367	02887	50E-08
211	26300	1.44936	3.1332	0.6074	-1.04141	26738	1.08204	-1.1061	02414	39E-04
212	25959	1.44939	3.2741	0.0910	-1.01330	32575	1.06662	-1.1367	02763	50E-08
213	23173	1.44936	3.2309	0.4704	-1.04406	26301	1.08357	-1.1743	02350	89E-04
214	21257	1.44939	2.9009	0.3505	-1.01708	32205	1.06684	-1.1421	02714	59E-08
215	16412	1.44936	3.3133	0.0559	-1.04709	26111	1.08474	-1.17676	02312	12E-08
216	16289	1.44939	3.0393	0.4204	-1.02092	37433	1.07054	-1.14607	02665	62E-08
217	13989	1.44936	3.6607	0.2750	-1.04800	27925	1.08544	-1.17824	02294	14E-08
218	12332	1.44939	3.0803	0.3104	-1.02319	31747	1.07174	-1.14873	02640	63E-06
219	09783	1.44936	3.4043	0.1006	-1.04904	27800	1.08619	-1.17960	02274	16E-08
220	08474	1.44939	3.2228	0.2260	-1.02462	31622	1.07273	-1.15075	02620	65E-08
221	05751	1.44936	3.4201	0.1091	-1.05400	27719	1.08041	-1.16029	02271	17E-08
222	05275	1.44939	3.1435	0.1296	-1.02575	31543	1.07324	-1.15164	02616	65E-08
223	01806	1.44936	3.4303	0.0333	-1.05097	27602	1.08062	-1.16118	02254	16E-08
224	01650	1.44939	3.1594	0.0391	-1.02630	31504	1.07357	-1.15256	02604	66E-08
225	58815	1.64842	0.2901	5.4912	-73361	10720	90509	10062	05574	87E-08
226	47761	1.74843	0.2493	5.3984	-61534	11209	82622	31736	06353	23E-07
227	53217	1.64842	0.0814	4.4445	-70797	26200	93660	11904	05201	82E-06
228	45537	1.74843	0.7542	4.0056	-67591	30161	86057	24905	05990	21E-07
229	46844	1.64842	1.3913	3.0910	-65525	36104	97893	04171	04475	94E-08
230	41745	1.74843	1.1905	3.3757	-75500	39600	91721	15874	05240	22E-07
231	43426	1.64842	1.7044	2.2742	-90063	36207	1.00441	-00804	04083	10E-07
232	37159	1.74843	1.5273	2.2247	-81013	42706	94496	09757	04825	22E-07
233	37813	1.64842	2.0002	1.0740	-93245	36444	1.02239	-04528	03714	11E-07
234	32360	1.74843	1.7851	1.0608	-85026	43549	97304	05203	04414	23E-07
235	32445	1.64842	2.0114	1.2008	-92027	38024	1.03186	-06473	03533	12E-07
236	27762	1.74843	1.7673	1.4542	-86527	43469	98612	02361	04207	22E-07
237	27436	1.64842	2.4617	0.9050	-90340	37509	1.03075	-07900	03341	12E-07
238	23477	1.74842	2.1064	1.1364	-67276	43109	99016	00367	03934	23E-07
239	23202	1.64842	2.5689	0.9011	-97116	37141	1.04280	-08756	03225	13E-07
240	19854	1.74843	2.1962	0.7273	-90311	42824	1.00381	-00764	03840	23E-07
241	16999	1.64842	2.0537	0.0307	-97708	36767	1.04390	-09467	03164	13E-07
242	16258	1.74843	2.2707	0.7334	-91142	42573	1.00065	-01736	03793	23E-07
243	15090	1.64842	2.7100	0.4072	-90136	36536	1.04031	-04846	03107	13E-07
244	12418	1.74843	2.3249	0.2601	-97130	42324	1.01197	-02404	03722	23E-07
245	11470	1.64842	2.7603	0.3002	-90421	36351	1.04404	-1.0216	03084	13E-07
246	09815	1.74842	2.3005	0.4202	-92131	42205	1.01426	-02876	03690	22E-07
247	06021	1.64842	2.7942	0.2420	-96608	36213	1.05377	-1.0442	03057	13E-07
248	06804	1.74843	2.6094	0.2420	-92404	42091	1.01566	-03146	03660	23E-07
249	04715	1.64842	2.0172	0.2476	-90729	36137	1.05142	-1.0548	03054	13E-07
250	04635	1.74842	2.4642	0.1726	-92571	42050	1.01600	-03309	03650	22E-07

Y FLUX

PL	A	f	g	RA	RY	VZ	ASBY	CP	SOUKLE	V NORMAL
251	.01401	1.84725	.60170	.00920	-.70767	.36094	1.001175	-.10618	.03040	.13E-07
252	.01207	1.74745	.59421	.00921	-.70094	.41999	1.011730	-.00490	.03043	.22E-07
253	.03304	1.84725	.01911	.00620	-.40027	.11025	.09080	.51108	.07249	.44E-07
254	.20594	1.84725	.00071	.00707	-.15106	.60350	.40940	.83252	.08450	.71E-07
255	.30006	1.84725	.00997	.40304	.00000	.30695	.74756	.42115	.08928	.42E-07
256	.19635	1.93333	.00202	.00137	-.00031	.24212	.45755	.79005	.08467	.65E-07
257	.31603	1.84725	.07300	.00572	-.00947	.42004	.81130	.06195	.06195	.42E-07
258	.10022	1.93333	.00103	.00204	-.00009	.34707	.51602	.73372	.08234	.62E-07
259	.29095	1.84725	.11942	.07943	-.00360	.40920	.00512	.26362	.05744	.40E-07
260	.10023	1.93333	.00506	.03467	-.00275	.41238	.50292	.68312	.08238	.55E-07
261	.25302	1.84724	.10077	.07071	-.00101	.49136	.09423	.20036	.05205	.42E-07
262	.13994	1.93333	.00997	.10500	-.00009	.42142	.59035	.64436	.08086	.53E-07
263	.21708	1.84725	.10998	.10927	-.074994	.49600	.91524	.16233	.04911	.42E-07
264	.11971	1.93333	.00491	.10236	-.00047	.47500	.61091	.16946	.08068	.50E-07
265	.16326	1.84725	.10470	.10079	-.077447	.49866	.93107	.13310	.04716	.41E-07
266	.10123	1.93333	.09063	.10397	-.037826	.49221	.03547	.59618	.08065	.47E-07
267	.15524	1.84724	.17108	.11145	-.070997	.49807	.94093	.11466	.04013	.40E-07
268	.00501	1.93333	.07476	.11248	-.00999	.50244	.64585	.50208	.08013	.40E-07
269	.12712	1.84725	.17754	.00900	-.00228	.49903	.94901	.09938	.04547	.39E-07
270	.07010	1.93333	.10791	.09125	-.039910	.49903	.05918	.57205	.08051	.49E-07
271	.10100	1.84725	.10176	.00800	-.01138	.49900	.95502	.08743	.04460	.30E-07
272	.05570	1.93333	.10023	.07141	-.00596	.51619	.00554	.56369	.08052	.44E-07
273	.07674	1.84725	.10400	.00128	-.01746	.49801	.95900	.08032	.04442	.37E-07
274	.04232	1.93333	.10109	.00374	-.01074	.52024	.00501	.0032	.08036	.43E-07
275	.05367	1.84724	.10075	.00400	-.02100	.49801	.96180	.07463	.04384	.30E-07
276	.02900	1.93333	.10299	.00303	-.01309	.52279	.66789	.55392	.08031	.43E-07
277	.03125	1.84725	.10798	.00000	-.00439	.49841	.96350	.07156	.04362	.37E-07
278	.01740	1.93333	.10307	.00213	-.01594	.52459	.66969	.55151	.08048	.42E-07
279	.00991	1.84725	.10800	.00623	-.00253	.49827	.96435	.07003	.04340	.30E-07
280	.00546	1.93333	.10401	.00090	-.01693	.52522	.67062	.55026	.08044	.42E-07

Z FLOW

PT.	X	Y	Z	VX	VY	VZ	405.8 V	CP	SOURCE	V NORMAL
1	93409	04999	05140	04999	00624	-2.00249	2.43333	-4.92137	04424	-30E-05
2	90621	14997	05127	04972	04672	-2.00204	2.43355	-4.92200	04425	-30E-05
3	94206	04999	05013	04917	01917	-1.71191	2.05343	-3.21677	11993	-81E-05
4	94030	14997	05074	04930	04239	-1.71259	2.05369	-3.21832	11990	-81E-05
5	80521	04999	04944	04765	01765	-1.60919	1.64494	-1.70563	15702	-90E-05
6	86304	14997	04982	04903	04403	-1.60654	1.64570	-1.70832	15690	-90E-05
7	76924	04999	04917	04826	01826	-1.60957	1.28024	-1.70832	17793	-10E-04
8	76731	14997	04957	04921	04921	-1.60711	1.28045	-1.70832	17780	-10E-04
9	62989	04999	04954	04979	01979	-1.42369	1.02540	-0.65209	18051	-98E-05
10	60821	14997	04961	04936	04936	-1.42433	1.02650	-0.65433	18045	-98E-05
11	57471	04999	04700	04712	01712	-2.27322	0.2203	3.2028	19325	-10E-04
12	57327	14997	04004	04055	04055	-2.27443	0.22409	3.2008	19323	-10E-04
13	48548	04999	04305	04300	01300	-1.7759	0.6150	5.6242	19382	-10E-04
14	46477	14997	04345	04376	04376	-1.7853	0.6127	5.6007	19378	-10E-04
15	41160	04999	04555	04516	01516	-1.1940	0.54174	7.0652	19796	-10E-04
16	40977	14997	04371	04380	04380	-1.1928	0.54378	7.0430	19792	-10E-04
17	35055	04999	04700	04603	01603	-0.67531	0.42702	8.1765	19997	-10E-04
18	35571	14997	04638	04606	04606	-0.67614	0.42942	8.1557	19994	-10E-04
19	26741	04999	04512	04490	01490	-0.04504	0.33428	8.8025	19017	-95E-05
20	26674	14997	04500	04473	04473	-0.04647	0.33439	8.8818	19012	-95E-05
21	20317	04999	04695	04603	01603	-0.62597	0.25088	9.3706	19980	-97E-05
22	20266	14997	04773	04609	04609	-0.62677	0.25486	9.3505	19977	-97E-05
23	14208	04999	04942	04910	01910	-0.01270	0.17030	9.8092	19965	-98E-05
24	14173	14997	04970	04923	04923	-0.01345	0.18138	9.8710	19962	-98E-05
25	06352	04999	04970	04924	01924	-0.00444	0.10524	9.8907	19066	-97E-05
26	08332	14997	04645	04636	04636	-0.00525	0.11384	9.8704	19061	-97E-05
27	02616	04999	04934	04904	01904	-0.00021	0.03423	9.9803	19033	-98E-05
28	02616	14997	04909	04904	04904	-0.00130	0.05025	9.9604	19030	-98E-05
29	90124	04999	05101	04935	01935	-2.30299	2.43377	-4.92325	04421	-30E-05
30	97372	14997	05002	04910	04910	-2.30325	2.43305	-4.92361	04412	-30E-05
31	93550	04999	05495	04993	01993	-1.71416	2.05482	-3.22230	11974	-80E-05
32	92859	14997	05377	04992	04992	-1.71635	2.05004	-3.22729	11959	-80E-05
33	85868	04999	04458	04455	01455	-1.00836	1.64758	-1.71386	15684	-90E-05
34	85211	14997	04271	04457	04457	-1.00915	1.64463	-1.72195	15665	-90E-05
35	76344	04999	04376	04329	01329	-0.67264	1.29111	-0.66698	17778	-10E-04
36	75759	14997	04138	04366	04366	-0.67612	1.29474	-0.67654	17759	-10E-04
37	66484	04999	04676	04644	01644	-0.42636	1.02416	-0.05918	18630	-98E-05
38	65975	14997	04394	04625	04625	-0.42938	1.03323	-0.06750	18622	-98E-05
39	57033	04999	04059	04406	01406	-0.27631	0.62097	3.1612	19314	-10E-04
40	56602	14997	04149	04401	04401	-0.27933	0.63561	3.0851	19302	-10E-04
41	48232	04999	04276	04372	01372	-1.6025	0.66036	5.5594	19572	-10E-04
42	47803	14997	04245	04367	04367	-1.60328	0.67252	5.4712	19558	-10E-04
43	46740	04999	04102	04270	01270	-1.2241	0.54013	6.9955	19784	-10E-04
44	46478	14997	04010	04262	04262	-1.22472	0.54401	6.9307	19775	-10E-04
45	35461	04999	04001	04199	01199	-0.07763	0.43441	8.1129	19987	-10E-04
46	33145	14997	04274	04194	04194	-0.00444	0.44180	8.0476	19970	-10E-04
47	26539	04999	04770	04607	01607	-0.00019	0.34109	8.8309	19607	-95E-05
48	26355	14997	04632	04607	04607	-0.0001	0.34444	8.7709	19797	-95E-05
49	20104	04999	04562	04503	01503	-0.02043	0.26241	9.3008	19970	-97E-05
50	20007	14997	04155	04500	04500	-0.03043	0.27479	9.2449	19959	-97E-05

4 FLOW

PT.	A	Y	Z	XA	YA	ZA	VB	YB	ZB	VC	YC	ZC	AB, B, Y	CP	SOURCE	V NORMAL
51	14161	24995	42009	17402	07907	-01920	19279	90263	19955	-90E-05						
52	13943	34912	40074	17413	11213	-01734	20780	95600	19444	-90E-05						
53	02293	24995	42009	13024	07909	-00606	13042	90000	20057	-97E-05						
54	02225	34913	42010	13021	11105	-00932	13041	97677	20046	-97E-05						
55	02603	44995	43526	03024	07925	-00290	08489	94279	20024	-90E-05						
56	02503	34912	44170	03044	11108	-00537	11288	90657	20014	-90E-05						
57	95302	44995	05009	40921	07719	-238445	243474	-492822	04415	-31E-05						
58	95004	54908	04948	40905	07015	-238536	243333	-493091	04397	-31E-05						
59	91876	44990	15217	11227	13092	-171974	2405623	-323031	11926	-80E-05						
60	90057	54908	15010	11201	17025	-172374	2400000	-324807	11097	-80E-05						
61	83226	44940	24049	122070	10459	-109602	105328	-173332	15641	-89E-05						
62	83203	54908	23700	122102	20314	-110160	105747	-174721	15609	-89E-05						
63	74973	44973	30615	109901	10595	-06034	1029661	-00638	17739	-10E-04						
64	73973	54908	30476	109012	20332	-06634	103045	-70158	17707	-10E-04						
65	62290	44950	35017	92924	10215	-43380	103825	-07796	18005	-98E-05						
66	64424	54908	35539	92090	20094	-4390	104941	-04288	18577	-98E-05						
67	56014	44930	37733	77237	13763	-28339	83769	29828	19284	-10E-04						
68	55271	54908	37000	77028	19509	-28869	84569	28481	19261	-10E-04						
69	47307	44990	42930	63913	13559	-18604	67809	53938	19545	-10E-04						
70	46738	54907	41936	63357	14047	-19212	68093	52537	19223	-10E-04						
71	40057	44990	44391	52652	15145	-12852	56274	60333	19759	-10E-04						
72	39526	54908	43703	52532	10701	-13338	57335	67127	19741	-10E-04						
73	32601	44990	45014	42021	14008	-08401	45201	74508	19964	-10E-04						
74	32366	54908	45206	41732	10391	-08809	46470	78398	19940	-10E-04						
75	26062	44990	40990	32705	14734	-05429	36279	86838	19783	-95E-05						
76	2716	54908	40278	32540	10202	-05874	37752	85748	19764	-95E-05						
77	19802	44990	47655	24898	14598	-03449	29067	91551	19946	-97E-05						
78	19537	54908	47023	24799	16079	-03877	30330	90430	19929	-97E-05						
79	13848	44990	40130	17450	14506	-02107	22041	94783	19931	-90E-05						
80	13664	54908	47548	17309	17990	-02550	25130	93602	19910	-90E-05						
81	06141	44990	40507	13313	14503	-01274	17841	90817	20033	-97E-05						
82	00032	54908	47504	10204	17944	-01722	20754	95653	20016	-97E-05						
83	02556	44990	40608	03010	14404	-00877	14621	97803	20001	-90E-05						
84	02522	54908	40022	03032	17907	-01322	18210	96604	19984	-90E-05						
85	93226	64905	04052	40109	00596	-238743	243703	-493910	04379	-30E-05						
86	91676	74902	04706	40070	00770	-238808	243778	-494276	04357	-30E-05						
87	84172	64905	14770	10047	20369	-172947	240629	-326131	11893	-81E-05						
88	87403	74902	14477	100673	23817	-173002	240826	-327771	11845	-81E-05						
89	81945	64905	23312	121500	24243	-110851	106251	-176393	15570	-89E-05						
90	80226	74902	22851	120700	20392	-111746	106819	-178618	15516	-89E-05						
91	72707	64905	29908	100374	24514	-069305	131135	-171963	17671	-99E-05						
92	71327	74902	29310	107990	20704	-70294	132025	-174366	17624	-99E-05						
93	63303	64905	34957	92300	24976	-44649	107376	-10999	18549	-98E-05						
94	62115	74902	34205	91998	20154	-45545	106445	-13305	18507	-98E-05						
95	54306	64905	30504	70805	20359	-29530	85538	-26833	19232	-10E-04						
96	53290	74902	37801	70550	27305	-30374	86783	-24687	19197	-10E-04						
97	49973	64905	41249	63118	22745	-19626	69459	-51058	19500	-10E-04						
98	45003	74902	40433	63023	20877	-20057	71494	-40867	19465	-10E-04						
99	30879	64905	43090	22490	22400	-13959	58714	-65226	19716	-10E-04						
100	30110	74902	42174	22226	20251	-14708	60274	-63670	19680	-10E-04						

Z FLOW

PT.	A	Y	Z	FX	FY	VZ	ABSJAV	CP	SOURCE	V NORMAL
101	.31456	.74905	.47900	.74049	.66001	-.00905	.46040	.70916	.19423	-.10E-04
102	.31200	.74902	.47900	.74049	.62922	-.10212	.47906	.75034	.19695	-.10E-04
103	.29705	.64705	.47900	.50000	.24627	-.00473	.59731	.04214	.19741	-.45E-03
104	.24750	.74905	.47900	.74049	.62922	-.07102	.41007	.06472	.19712	-.47E-03
105	.14713	.64905	.47900	.24700	.24030	-.04473	.52004	.08976	.19906	-.47E-03
106	.10853	.74902	.47900	.74049	.62922	-.05190	.55004	.87100	.19676	-.47E-03
107	.13440	.64905	.47900	.1367	.24574	-.03125	.27680	.92223	.19693	-.96E-03
108	.15175	.74902	.47900	.17203	.25242	-.03031	.50671	.90470	.19867	-.47E-03
109	.07901	.64905	.47900	.10276	.24306	-.02204	.23926	.94275	.19994	-.47E-03
110	.07745	.74902	.47900	.10237	.25221	-.02493	.27303	.92502	.19966	-.47E-03
111	.02431	.64905	.47900	.03031	.24473	-.01008	.21771	.95260	.19364	-.46E-03
112	.02432	.74902	.47900	.03000	.24150	-.02504	.25460	.93518	.19436	-.46E-03
113	.84515	.64905	.47900	.47003	.11220	-.239079	.24329	-.44325	.04325	-.30E-03
114	.7019	.74904	.47900	.40316	.12072	-.2399308	.244078	-.495740	.04301	-.30E-03
115	.85347	.84906	.47900	.10335	.27320	-.174440	.207366	-.330006	.11777	-.80E-03
116	.84908	.74904	.47900	.107342	.31003	-.175403	.208016	-.332706	.11702	-.80E-03
117	.70335	.64905	.47900	.11900	.32615	-.112879	.207777	-.1.81492	.15425	-.80E-03
118	.76151	.64904	.47900	.11000	.37092	-.114245	.108006	-.1.64925	.15425	-.84E-03
119	.64646	.64906	.47900	.107300	.33000	-.711422	.133082	-.77107	.17567	-.94E-03
120	.67704	.64904	.47900	.100438	.37702	-.72800	.134351	-.80503	.17496	-.96E-03
121	.60651	.64906	.47900	.34911	.32403	-.40614	.107690	-.15971	.18457	-.97E-03
122	.56900	.64904	.47900	.34911	.30900	-.47448	.109233	-.19318	.18398	-.97E-03
123	.52034	.64906	.47900	.70146	.31572	-.31384	.08204	.22200	.19151	-.10E-04
124	.50583	.64904	.47900	.75750	.30035	-.32659	.90074	.16949	.19098	-.10E-04
125	.4001	.64906	.47900	.62037	.30052	-.21559	.73074	.40595	.19424	-.10E-04
126	.42774	.64904	.47900	.62426	.33174	-.22032	.75207	.43439	.19374	-.94E-03
127	.37211	.64906	.47900	.5001	.30333	-.15673	.62224	.61226	.19650	-.10E-04
128	.36174	.64904	.47900	.34787	.34630	-.16830	.64332	.50336	.19605	-.10E-04
129	.30471	.64906	.47900	.41291	.27601	-.11120	.52162	.72791	.19858	-.10E-04
130	.24621	.64905	.47900	.41159	.24130	-.12267	.54054	.69905	.19814	-.10E-04
131	.24210	.64908	.47900	.52300	.27503	-.08103	.44001	.80107	.19676	-.95E-03
132	.23335	.64904	.47900	.32122	.33795	-.09197	.47517	.77422	.19632	-.95E-03
133	.16345	.64906	.47900	.24020	.27341	-.06061	.38801	.84945	.19643	-.97E-03
134	.17302	.64904	.47900	.24936	.33328	-.07160	.42167	.82203	.19800	-.97E-03
135	.12404	.64906	.47900	.17274	.27232	-.04709	.34282	.89247	.19633	-.96E-03
136	.07505	.64905	.47900	.17154	.33412	-.05792	.37993	.85565	.19790	-.96E-03
137	.07502	.64906	.47900	.10230	.24129	-.03864	.51114	.69019	.19933	-.97E-03
138	.07352	.64904	.47900	.10131	.25322	-.04941	.55177	.87626	.19890	-.97E-03
139	.02374	.64908	.47900	.03011	.24132	-.03464	.29491	.91303	.19903	-.96E-03
140	.02305	.64904	.47900	.02772	.25237	-.04928	.33676	.86659	.19863	-.96E-03
141	.84159	.10490	.04375	.45317	.14220	-.239079	.244208	-.496705	.04258	-.30E-03
142	.80896	.114904	.04205	.43204	.19705	-.2399308	.244215	-.497874	.04201	-.29E-03
143	.80241	.10490	.10276	.103023	.34707	-.176730	.208026	-.330004	.11594	-.74E-03
144	.77130	.114904	.12775	.103970	.30903	-.1870302	.209041	-.340333	.11480	-.78E-03
145	.73608	.10490	.10237	.11234	.49006	-.115993	.207016	-.1.85556	.15332	-.84E-03
146	.70793	.114904	.10194	.113494	.40908	-.116163	.171742	-.1.94952	.15263	-.84E-03
147	.65474	.10490	.10212	.105102	.40944	-.174523	.135990	-.84950	.17471	-.10E-04
148	.60940	.10490	.10200	.104101	.49701	-.176704	.137961	-.90333	.17364	-.94E-03
149	.57622	.114904	.10132	.10132	.47603	-.2399308	.111083	-.23344	.18324	-.97E-03
150	.54511	.114904	.10200	.10200	.47104	-.239079	.113401	-.28597	.18280	-.97E-03

Z FLOW

PI	A	I	L	RA	RT	YZ	ABJ.V	CP	SOURCE	V NORMAL
151	46021	1.04970	0.0701	75016	40703	-0.34244	0.2120	0.15128	19077	-1.10E-04
152	47024	1.14964	0.0000	74015	40125	-0.00203	0.4614	0.10103	18999	-1.10E-04
153	47300	1.04970	0.1110	74016	39000	-0.24307	0.7573	0.34825	19316	-0.95E-05
154	34765	1.14964	0.0000	62479	40000	-0.20219	0.0024	0.34889	19280	-1.10E-04
155	34915	1.04970	0.0701	54411	39276	-0.10229	0.6724	0.54809	19582	-1.10E-04
156	33629	1.14964	0.1633	51000	44457	-0.01007	0.7023	0.50124	19512	-1.10E-04
157	26647	1.04970	0.0000	40017	40702	-0.13022	0.7081	0.60498	19759	-1.10E-04
158	27537	1.14964	0.0001	40000	40771	-0.19443	0.6104	0.19288	19728	-1.10E-04
159	22702	1.04970	0.0001	34901	40307	-0.10577	0.51007	0.19579	19579	-0.94E-05
160	21879	1.14964	0.1413	31711	43326	-0.12312	0.5085	0.04656	19540	-0.95E-05
161	17244	1.04970	0.4250	24400	36024	-0.00924	0.5977	0.78861	19740	-0.97E-05
162	16624	1.14964	0.0000	24204	43014	-0.10242	0.5040	0.74540	19710	-0.97E-05
163	12004	1.04970	0.4000	17110	37923	-0.07143	0.4213	0.82181	19738	-0.96E-05
164	11625	1.14964	0.0004	10734	42024	-0.00826	0.4094	0.78010	19894	-0.96E-05
165	07110	1.04970	0.2305	10121	37708	-0.00278	0.39621	0.84302	19838	-0.97E-05
166	06834	1.14964	0.0723	10034	42708	-0.07454	0.44587	0.80120	19804	-0.98E-05
167	02232	1.04970	0.4000	02976	37172	-0.00870	0.38541	0.85244	19813	-0.96E-05
168	02146	1.14964	0.0000	02900	42003	-0.07537	0.4394	0.81170	19774	-0.97E-05
169	77179	1.24957	0.0012	42902	17447	-2.40515	2.44942	-4.99965	0.4133	-0.29E-05
170	72937	1.34948	0.0791	41200	14125	-2.41031	2.45286	-5.01651	0.4045	-0.28E-05
171	73587	1.24957	0.1210	1.00902	43204	-1.00198	2.41033	-3.45344	0.11367	-0.70E-05
172	69544	1.34948	0.1210	47600	47844	-1.02603	2.41240	-3.51756	0.11175	-0.77E-05
173	67540	1.24957	0.1230	1.13327	32595	-1.20007	1.87372	-2.02035	0.15120	-0.89E-05
174	63829	1.34948	0.1010	1.10490	0.6004	-1.24145	1.76223	-2.10544	0.14740	-0.88E-05
175	60049	1.24957	0.2400	1.02013	0.5810	-0.79471	1.44050	-0.97411	0.17283	-1.00E-04
176	50740	1.34948	0.2325	1.00544	0.0320	-0.8314	1.43600	-1.06362	0.17118	-0.99E-05
177	52293	1.24957	0.2587	0.0190	0.5053	-0.54321	1.10574	-0.34228	0.18165	-0.97E-05
178	44420	1.34948	0.2720	0.0000	0.5627	-0.57705	1.119461	-0.43906	0.18066	-0.98E-05
179	44804	1.24957	0.3123	73600	0.7035	-0.38679	0.98000	-0.05949	0.19186	-1.00E-04
180	42390	1.34948	0.0070	72404	0.5010	-0.41922	1.02144	-0.04335	0.18606	-1.00E-04
181	37938	1.24957	0.4034	60717	0.0899	-0.20975	0.84232	-0.24050	0.19186	-1.00E-04
182	35853	1.34948	0.3210	54751	0.7371	-0.31009	0.8889	-0.21342	0.19107	-1.00E-04
183	32084	1.24957	0.3522	0.0530	0.0201	-0.22449	0.74682	-0.42226	0.19461	-1.00E-04
184	26272	1.34948	0.0070	47776	0.6700	-0.25507	0.79093	-0.30491	0.19207	-0.97E-05
185	24828	1.24957	0.0070	40220	0.47507	-0.17739	0.66206	-0.50167	0.19643	-1.00E-04
186	20874	1.34948	0.3676	37920	0.5052	-0.20702	0.71084	-0.48327	0.19575	-1.00E-04
187	14728	1.24957	0.37504	0.1434	0.0906	-0.14593	0.60000	-0.63993	0.19487	-0.95E-05
188	15820	1.34948	0.3500	0.1243	0.5540	-0.17541	0.66095	-0.56315	0.19381	-0.95E-05
189	14989	1.24957	0.36169	2.3717	0.0678	-0.12492	0.5582	-0.60945	0.19662	-0.96E-05
190	11091	1.34948	0.3672	2.3824	0.0187	-0.15344	0.1087	-0.17000	0.19554	-0.98E-05
191	10402	1.24957	0.38595	1.0890	0.4030	-0.11009	0.52466	-0.72494	0.19659	-0.96E-05
192	00520	1.34948	0.30475	1.0742	0.4602	-0.13842	0.59018	-0.65169	0.19550	-0.96E-05
193	00162	1.24957	0.30817	0.7074	0.40200	-0.10122	0.50316	-0.74603	0.19758	-0.98E-05
194	02000	1.34948	0.30717	0.0802	0.46031	-0.12962	0.57011	-0.67498	0.19660	-0.98E-05
195	01935	1.24957	0.30900	0.2979	0.40177	-0.09693	0.49234	-0.75760	0.19723	-0.97E-05
196	06077	1.34948	0.30830	0.2400	0.4520	-0.12523	0.2615	-0.68623	0.19626	-0.97E-05
197	02447	1.24957	0.3359	0.37144	0.20844	-2.41708	2.445747	-0.03918	0.03442	-0.27E-05
198	02447	1.34948	0.3240	0.4010	0.22007	-2.42204	2.446242	-5.06523	0.03801	-0.26E-05
199	04903	1.24957	0.1071	0.7006	0.2200	-1.85508	2.414365	-3.55523	0.0927	-0.75E-05
200	04919	1.34948	0.0701	0.07471	0.57670	-1.65570	2.416000	-3.70023	0.10596	-0.73E-05

Z FLUX

PT.	Z	Y	X	VX	VY	VZ	ABS.V	CP	SOURCE	V NORMAL
201	55874	1.44936	1.50000	0.0218	-1.22050	-2.22080	1.79400	-2.22080	.14090	-.87E-05
202	54017	1.29417	1.01000	1.7303	-1.35394	-2.30720	1.83499	-2.30720	.14360	-.82E-05
203	52006	1.44936	1.70000	0.7990	-0.76446	-1.18107	1.47700	-1.18107	.16897	-.96E-05
204	40006	1.54919	1.90000	0.7990	-0.93009	-1.34005	1.52999	-1.34005	.16289	-.97E-05
205	40129	1.44936	1.80000	0.7990	-0.82207	-1.25544	1.24717	-1.25544	.17874	-.97E-05
206	42311	1.29419	1.80000	0.7990	-0.82207	-1.11140	1.30020	-1.11140	.17601	-.90E-05
207	39572	1.44936	1.70000	0.7990	-0.82207	-1.07326	1.07326	-1.07326	.18625	-.10E-04
208	36300	1.29419	1.80000	0.7990	-0.82207	-1.30109	1.14022	-1.30109	.18374	-.10E-04
209	33403	1.44936	1.90000	0.7990	-0.82207	-1.1054	1.1054	-1.1054	.18944	-.10E-04
210	30690	1.54919	1.90000	0.7990	-0.82207	-1.01060	1.01060	-1.01060	.18709	-.99E-05
211	26300	1.44936	1.80000	0.7990	-0.82207	-1.26639	0.85051	-1.26639	.19050	-.97E-05
212	25059	1.54919	1.80000	0.7990	-0.82207	-1.2202	0.85051	-1.2202	.18069	-.97E-05
213	23173	1.44936	1.90000	0.7990	-0.82207	-1.30109	0.85051	-1.30109	.19181	-.97E-05
214	21257	1.54919	1.90000	0.7990	-0.82207	-1.23903	0.85051	-1.23903	.18964	-.96E-05
215	16412	1.44936	1.80000	0.7990	-0.82207	-1.21500	0.85051	-1.21500	.19241	-.95E-05
216	10809	1.54919	1.80000	0.7990	-0.82207	-1.21080	0.85051	-1.21080	.19033	-.95E-05
217	13089	1.44936	1.90000	0.7990	-0.82207	-1.31805	0.85051	-1.31805	.19422	-.96E-05
218	12832	1.54919	1.90000	0.7990	-0.82207	-1.24652	0.85051	-1.24652	.19219	-.97E-05
219	09763	1.44936	1.80000	0.7990	-0.82207	-1.16000	0.85051	-1.16000	.19415	-.97E-05
220	08974	1.54919	1.80000	0.7990	-0.82207	-1.20000	0.85051	-1.20000	.19215	-.96E-05
221	05751	1.44936	1.90000	0.7990	-0.82207	-1.16774	0.85051	-1.16774	.19522	-.98E-05
222	05275	1.54919	1.90000	0.7990	-0.82207	-1.22100	0.85051	-1.22100	.19327	-.98E-05
223	01806	1.44936	1.80000	0.7990	-0.82207	-1.26000	0.85051	-1.26000	.19493	-.97E-05
224	01606	1.54919	1.80000	0.7990	-0.82207	-1.25719	0.85051	-1.25719	.19300	-.97E-05
225	55815	1.54932	0.2901	0.2914	-2.43578	-5.09897	2.46901	-5.09897	.03590	-.25E-05
226	47701	1.74843	0.2403	0.2476	-2.42104	-2.47409	2.47409	-2.47409	.03299	-.22E-05
227	53217	1.54932	0.0814	0.0814	-1.89485	-1.89485	2.46901	-1.89485	.0131	-.70E-05
228	45537	1.74843	0.7542	0.7542	-2.40247	-2.43304	2.43304	-2.43304	.09416	-.65E-05
229	48844	1.54932	1.3913	1.3913	-1.89485	-1.89485	1.89485	-1.89485	.13886	-.83E-05
230	41745	1.74843	1.1905	1.1905	-1.89485	-1.89485	1.89485	-1.89485	.13122	-.80E-05
231	43426	1.54932	1.7049	1.7049	-1.89485	-1.89485	1.89485	-1.89485	.16137	-.95E-05
232	37159	1.74843	1.5273	1.5273	-1.89485	-1.89485	1.89485	-1.89485	.15402	-.95E-05
233	37818	1.54932	2.0002	2.0002	-1.89485	-1.89485	1.89485	-1.89485	.17200	-.95E-05
234	32300	1.74843	1.7051	1.7051	-1.89485	-1.89485	1.89485	-1.89485	.16235	-.95E-05
235	32445	1.54932	2.3014	2.3014	-1.89485	-1.89485	1.89485	-1.89485	.17996	-.97E-05
236	27762	1.74843	1.9693	1.9693	-1.89485	-1.89485	1.89485	-1.89485	.17361	-.95E-05
237	27436	1.54932	2.4617	2.4617	-1.89485	-1.89485	1.89485	-1.89485	.18212	-.95E-05
238	23477	1.74843	2.1004	2.1004	-1.89485	-1.89485	1.89485	-1.89485	.17390	-.87E-05
239	23502	1.54932	2.5009	2.5009	-1.89485	-1.89485	1.89485	-1.89485	.18330	-.92E-05
240	19854	1.74843	2.1902	2.1902	-1.89485	-1.89485	1.89485	-1.89485	.17749	-.90E-05
241	18999	1.54932	2.6537	2.6537	-1.89485	-1.89485	1.89485	-1.89485	.18632	-.95E-05
242	16258	1.74843	2.2707	2.2707	-1.89485	-1.89485	1.89485	-1.89485	.18055	-.93E-05
243	15076	1.54932	2.7106	2.7106	-1.89485	-1.89485	1.89485	-1.89485	.18709	-.94E-05
244	12918	1.74843	2.5296	2.5296	-1.89485	-1.89485	1.89485	-1.89485	.18154	-.92E-05
245	11470	1.54932	2.7603	2.7603	-1.89485	-1.89485	1.89485	-1.89485	.18897	-.96E-05
246	09815	1.74843	2.5025	2.5025	-1.89485	-1.89485	1.89485	-1.89485	.18348	-.94E-05
247	08021	1.54932	2.7912	2.7912	-1.89485	-1.89485	1.89485	-1.89485	.18902	-.95E-05
248	06504	1.74843	2.5004	2.5004	-1.89485	-1.89485	1.89485	-1.89485	.18364	-.93E-05
249	04715	1.54932	2.8077	2.8077	-1.89485	-1.89485	1.89485	-1.89485	.19013	-.97E-05
250	03055	1.74843	2.6042	2.6042	-1.89485	-1.89485	1.89485	-1.89485	.18476	-.95E-05

Z FLOW

PT.	X	Y	Z	VA	VY	VZ	ABS.V	UP	SOURCE	V NORMAL
201	.01401	1.84725	1.84725	.02500	.01090	-.01960	.00347	.02937	.18988	-.90E-05
202	.01267	1.84725	1.84725	.02500	.01137	-.01850	1.03026	-.07308	.18453	-.94E-05
203	.01344	1.84725	1.84725	.02500	.01022	-.01792	4.00042	-.01212	.02041	-.20E-05
204	.00544	1.84725	1.84725	.01012	.02225	-.00227	4.00400	-.01097	.02130	-.13E-05
205	.00906	1.84725	1.84725	.02074	.00370	-.01441	4.01307	-.01307	.08234	-.57E-05
206	.01425	1.84725	1.84725	.02705	.01577	-.01430	4.03353	-.01428	.06701	-.01E-05
207	.02680	1.84725	1.84725	.04306	.03401	-.01740	4.09240	-.03302	.11775	-.70E-05
208	.04022	1.84725	1.84725	.05948	.01006	-.01901	4.09150	-.04719	.09187	-.69E-05
209	.04905	1.84725	1.84725	.06375	1.04119	-.01349	1.06470	-.04771	.14056	-.85E-05
200	.06023	1.84725	1.84725	.09001	1.00704	-.01906	4.24154	-.04045	.11135	-.80E-05
201	.05302	1.84725	1.84725	.07540	1.00036	-.01392	1.09003	-.01805	.15055	-.81E-05
202	.03994	1.84725	1.84725	.05697	1.07120	-.01853	4.13345	-.01802	.11908	-.80E-05
203	.02108	1.84725	1.84725	.02309	1.01420	-.01853	1.57005	-.01402	.15779	-.82E-05
204	.01191	1.84725	1.84725	.01041	1.01044	-.01001	4.00400	-.01211	.12350	-.82E-05
205	.00396	1.84725	1.84725	.00470	1.11123	-.01705	1.49398	-.01211	.16275	-.82E-05
206	.00123	1.84725	1.84725	.00771	1.10609	-.01298	1.99339	-.01939	.12948	-.84E-05
207	.00524	1.84725	1.84725	.02491	1.12908	-.01298	1.43063	-.01034	.16057	-.84E-05
208	.00501	1.84725	1.84725	.02406	1.13710	-.01455	1.95406	-.01834	.13131	-.84E-05
209	.02712	1.84725	1.84725	.07229	1.13347	-.01705	1.98074	-.01000	.16980	-.87E-05
270	.07010	1.84725	1.84725	.17606	1.14471	-.01542	1.92251	-.01034	.13394	-.93E-05
271	.10100	1.84725	1.84725	.24520	1.11200	-.01714	1.95309	-.01034	.17101	-.86E-05
272	.05570	1.84725	1.84725	.14264	1.14001	-.01504	1.89028	-.01034	.13526	-.95E-05
273	.07674	1.84725	1.84725	.10413	1.11237	-.01651	1.92987	-.01034	.17294	-.88E-05
274	.04232	1.84725	1.84725	.10185	1.15117	-.01403	1.80090	-.01034	.13594	-.93E-05
275	.05307	1.84725	1.84725	.10075	1.11113	-.01006	1.91232	-.01034	.17264	-.86E-05
276	.02960	1.84725	1.84725	.10274	1.13117	-.01400	1.86894	-.01034	.13644	-.93E-05
277	.03195	1.84725	1.84725	.10744	1.11037	-.01507	1.90187	-.01034	.17306	-.80E-05
278	.01740	1.84725	1.84725	.04548	1.14406	-.01454	1.86209	-.01034	.13725	-.94E-05
279	.00991	1.84725	1.84725	.02202	1.11045	-.01705	1.87142	-.01034	.17304	-.85E-05
280	.00540	1.84725	1.84725	.01422	1.13510	-.01455	1.85023	-.01034	.13736	-.94E-05

XYZ POTENTIAL FLOW PROGRAM SECTION 3, VERSION 4

SAMPLE PROBLEM TRIAXIAL ELLIPSOID

NORP = 3  
 ILCIT5= 0  
 IWEAD = 0

UFF BODY PLINKS

PT.	X	Y	Z
1	.00000	.00000	.00000
2	.00000	.00000	1.00000
3	.00000	1.00000	.00000



AD-A168 167

FORMULATION OF NUMERICAL METHODS USED IN THE XYZ  
THREE-DIMENSIONAL POTENT. (U) TEXAS A AND M UNIV  
COLLEGE STATION COLL OF ENGINEERING W J BERRY MAY 86  
N00228-85-G-3303

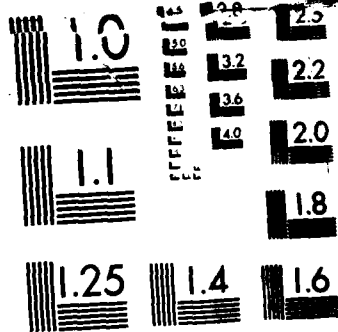
3/3

UNCLASSIFIED

F/G 20/4

NL





MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS-1963-A

SAMPLE PROBLEM TRIAXIAL ELLIPSOID

PT.	X	Y	Z	VX	VY	VZ	CP
1	.00000	.00000	.00000	-.99173	.00000	.00000	.12608
2	.00000	.00000	.00000	-1.00000	.00000	.00000	-.16671
3	.00000	3.00000	.00000	-1.02501	.00000	.00000	-.06051

SAMPLE PROBLEM TRIAXIAL ELLIPSOID

PT.	X	Y	Z	VX	VY	VZ	CP
1	2.00000	.00000	.00000	.00000	-1.03010	.00000	-.00112
2	.00000	.00000	.00000	.00000	-1.004127	.00000	-.00425
3	.00000	3.00000	.00000	.00000	-.95129	.00000	.09512

SAMPLE PROBLEM TRIAXIAL ELLIPSOID

PT.	X	Y	Z	VX	VY	VZ	CP
1	2.00000	.00000	.00000	.00000	.00000	-1.10658	-.22452
2	.00000	.00000	.00000	.00000	.00000	-.70719	.41142
3	.00000	3.00000	.00000	.00000	.00000	-1.02703	-.11731

XYZ POTENTIAL FLW PROGRAM SECTION 03 VERSION 4

SAMPLE PROBLEM TRIAXIAL ELLIPSOID

2 STARTING POINTS TO BE COMPUTED AT 20 STEPS UP -1.0000 T FOR AN UNSET VELOCITY OF .1000 .0000 .0000

PT	A	Y	Z	VA	VY	VZ	CP
1	1.00000	1.00000	.00000	.00000	.00000	.00000	.60518
2	1.50000	.00000	.00000	.07772	.00000	.00000	.39600
STEP 0							
LINE	X	Y	Z	VA <td>VY <td>VZ <td>CP</td> </td></td>	VY <td>VZ <td>CP</td> </td>	VZ <td>CP</td>	CP
1	1.00000	1.00000	.00000	.04334	-.02240	.00000	.74008
2	1.50000	.00000	.00000	.07394	.00000	.00000	.45332
STEP 1							
LINE	X	Y	Z	VA <td>VY <td>VZ <td>CP</td> </td></td>	VY <td>VZ <td>CP</td> </td>	VZ <td>CP</td>	CP
1	.95270	1.02485	.00000	.04305	-.02605	.00000	.77660
2	1.42411	.00000	.00000	.07394	.00000	.00000	.51910
STEP 2							
LINE	X	Y	Z	VA <td>VY <td>VZ <td>CP</td> </td></td>	VY <td>VZ <td>CP</td> </td>	VZ <td>CP</td>	CP
1	.91327	1.05377	.00000	.04305	-.03100	.00000	.77660
2	1.35240	.00000	.00000	.06935	.00000	.00000	.51910
STEP 3							
LINE	X	Y	Z	VA <td>VY <td>VZ <td>CP</td> </td></td>	VY <td>VZ <td>CP</td> </td>	VZ <td>CP</td>	CP
1	.88110	1.08683	.00000	.04305	-.03506	.00000	.79351
2	1.28576	.00000	.00000	.06378	.00000	.00000	.54326
STEP 4							
LINE	X	Y	Z	VA <td>VY <td>VZ <td>CP</td> </td></td>	VY <td>VZ <td>CP</td> </td>	VZ <td>CP</td>	CP
1	.85496	1.12394	.00000	.04306	-.03914	.00000	.78987
2	1.22519	.00000	.00000	.05724	.00000	.00000	.67230
STEP 5							
LINE	X	Y	Z	VA <td>VY <td>VZ <td>CP</td> </td></td>	VY <td>VZ <td>CP</td> </td>	VZ <td>CP</td>	CP
1	.83254	1.16476	.00000	.04107	-.04231	.00000	.77664
2	1.17163	.00000	.00000	.04969	.00000	.00000	.75305
STEP 6							
LINE	X	Y	Z	VA <td>VY <td>VZ <td>CP</td> </td></td>	VY <td>VZ <td>CP</td> </td>	VZ <td>CP</td>	CP
1	.81250	1.20800	.00000	.01957	-.04567	.00000	.75312
2	1.12610	.00000	.00000	.04126	.00000	.00000	.82956
STEP 7							
LINE	X	Y	Z	VA <td>VY <td>VZ <td>CP</td> </td></td>	VY <td>VZ <td>CP</td> </td>	VZ <td>CP</td>	CP
1	.79209	1.25254	.00000	.02102	-.04747	.00000	.72574
2	1.08423	.00000	.00000	.03242	.00000	.00000	.89491

STEP 8									
LINE	X	Y	Z	VA	VY	VZ	CP		
1	.77077	1.39475	.00000	.02174	-.05033	.00000	.67439		
2	1.06120	.00000	.00000	.02377	.00000	.00000	.94352		
STEP 9									
LINE	A	Y	Z	VA	VY	VZ	CP		
1	.74004	1.35000	.00000	.02520	-.05274	.00000	.05504		
2	1.04141	.00000	.00000	.02000	.00000	.00000	.97430		
STEP 10									
LINE	A	Y	Z	VA	VY	VZ	CP		
1	.72070	1.41410	.00000	.02702	-.05004	.00000	.00403		
2	1.02000	.00000	.00000	.03775	.00000	.00000	.95000		
STEP 11									
LINE	X	Y	Z	VA	VY	VZ	CP		
1	.69033	1.46037	.00000	.03107	-.05000	.00000	.50129		
2	1.02435	.00000	.00000	.00520	.00000	.00000	.94724		
STEP 12									
LINE	A	Y	Z	VA	VY	VZ	CP		
1	.65062	1.35000	.00000	.03703	-.05170	.00000	.47045		
2	1.01701	.00000	.00000	.00204	.00000	.00000	.94936		
STEP 13									
LINE	A	Y	Z	VA	VY	VZ	CP		
1	.61701	1.35000	.00000	.04124	-.05400	.00000	.41909		
2	1.01580	.00000	.00000	.00113	.00000	.00000	.94987		
STEP 14									
LINE	A	Y	Z	VA	VY	VZ	CP		
1	.57000	1.05000	.00000	.07000	-.06010	.00000	.31229		
2	1.01512	.00000	.00000	.00000	.00000	.00000	.94998		

STEP 15

LINE	X	Y	Z	VA	VY	VZ	CP
1	.51644	1.72440	.00000	.00108	-.06935	.00000	.14599
2	1.01480	.00000	.00000	.00020	.00000	.00000	1.00000

STEP 16

LINE	X	Y	Z	VA	VY	VZ	CP
1	.45208	1.79351	.00000	.00804	-.07144	.00000	.01838
2	1.01467	.00000	.00000	.00008	.00000	.00000	1.00000

STEP 17

LINE	X	Y	Z	VA	VY	VZ	CP
1	.37174	1.85222	.00000	.00620	-.06674	.00000	-.18642
2	1.01461	.00000	.00000	.00003	.00000	.00000	1.00000

STEP 18

LINE	X	Y	Z	VA	VY	VZ	CP
1	.27359	1.92497	.00000	.11722	-.07705	.00000	-.59555
2	1.01454	.00000	.00000	.00001	.00000	.00000	1.00000

STEP 19

LINE	X	Y	Z	VA	VY	VZ	CP
1	.15407	1.97074	.00000	.12240	-.05020	.00000	-.62923
2	1.01450	.00000	.00000	.00001	.00000	.00000	1.00000

STEP 20

LINE	X	Y	Z	VA	VY	VZ	CP
1	.02453	1.99005	.00000	.14659	.00103	.00000	-1.10076
2	1.01458	.00000	.00000	.00000	.00000	.00000	1.00000

XYZ POTENTIAL FLUX PROGRAM SECTION 7, VERSION 4

SAMPLE PROBLEM TRIAXIAL ELLIPSOID

ON BODY STREAMLINES - INPUT DATA

VX1 = -1.00000  
 VY1 = .00000  
 VZ1 = .00000  
 NLIN = 1  
 JMAX = 0  
 IWRITE = 1  
 MACH NO = .00000

STREAMLINE STARTING POINTS

LINE	X	Y	Z	NBP
1	1.00000	.05000	.00000	1

SAMPLE PROBLEM TRIAXIAL ELLIPSOID

UNSET FLOW, VX1=1.000 VY1= .000 VZ1= .000

LINE NU. 1 PASSING THROUGH QUADRILATERAL 1 WITH STARTING POINT, X= 1.00000 Y= .05000 Z= .00000

I	X	Y	Z	VX	VY	VZ	CP	K1	K2	M2	SL	V	P
1	.99940	.04944	-.00012	-.01221	.01724	.05748	.94614	-10.55732	-16.67703	1.00000	.00000	.00172	0.00000
2	.99403	.10000	.01980	-.02490	.03434	.14164	.97705	-3.72831	-5.86713	1.42924	.05404	.14883	0.00000
3	.99403	.10000	.01980	-.02490	.03434	.14164	.97705	-3.72831	-5.86713	1.42924	.05404	.14883	0.00000

13.27.20.UCLP, AA, NOTTY549  
 6.942KLN3.  
 \*\* LNU OF LISTING \*\*

ENTR  
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